

Honda Aircraft Company

AIRPLANE FLIGHT MANUAL



HondaJet

Model HA-420

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FAA Approved in the Normal Category based on 14 CFR Part 23. This document must be carried in the airplane at all times and kept within reach of the pilot during all flight operations. This manual includes the material required to be furnished to the pilot by 14 CFR Part 23.

FAA Approved by:

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LIST OF EFFECTIVE PAGES

This list contains all current pages with effective revision date.

Use this list to maintain the most current version of the manual:
Insert the latest revised pages. Then destroy superseded or deleted pages.

Note: A vertical revision bar in the left margin of the page indicates pages that have been added, revised or deleted.

MODEL HA-420 AFM

Title Page	March 3, 2017
Copyright Page	March 3, 2017
List of Effective Pages	March 3, 2017
Record of Revisions	March 3, 2017
Record of Temporary Revisions	March 3, 2017
List of Service Bulletins	March 3, 2017
Documentation Group	March 3, 2017

SECTION 1 - INTRODUCTION

Pages 1 – 20	October 30, 2016
--------------------	------------------

SECTION 2 – LIMITATIONS

Pages 1 – 2	March 3, 2017
Pages 3 – 16	October 30, 2016
Pages 17 – 18	March 3, 2017
Pages 19 – 24	October 30, 2016
Pages 25 – 26	March 3, 2017
Pages 27 – 34	October 30, 2016
Pages 35 – 38	March 3, 2017
Pages 39 – 44	October 30, 2016

Honda Aircraft Company

SECTION 3 – EMERGENCY PROCEDURES

Pages 1 – 38	October 30, 2016
Pages 39 – 40	March 3, 2017
Pages 41 – 62	October 30, 2016
Pages 63 – 66	March 3, 2017

SECTION 3A – ABNORMAL PROCEDURES

Pages 1 – 132	October 30, 2016
Pages 133 – 134	March 3, 2017
Pages 135 – 154	October 30, 2016
Pages 155 – 156	March 3, 2017
Pages 157 – 188	October 30, 2016

SECTION 3B – ADVISORY PROCEDURES

Pages 1 – 26	October 30, 2016
--------------------	------------------

SECTION 4 – NORMAL PROCEDURES

Pages 1 – 4	October 30, 2016
Pages 5 – 6	March 3, 2017
Pages 7 – 60	October 30, 2016

SECTION 5 – PERFORMANCE

Pages 1 – 168	October 30, 2016
Pages 169 – 170	March 3, 2017
Pages 171 – 176	October 30, 2016
Pages 177 – 178	March 3, 2017
Pages 179 – 182	October 30, 2016
Pages 183 – 184	March 3, 2017
Pages 185 – 262	October 30, 2016

Honda Aircraft Company

SECTION 6 – WEIGHT AND BALANCE

Pages 1 – 32 October 30, 2016

SECTION 7 – SUPPLEMENTS

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RECORD OF REVISIONS

REV. NO.	ISSUE DATE	REVISION DESCRIPTION
A-1	12-18-2015	Placard information added in Section 2
A-2	3-31-2016	ICE DETECTED CAS maintenance instructions added in Section 3B. EASA requirements were added in Sections 2 and 4 and minor format/font updates were made throughout document.
A3	7-1-2016	Reduced Vertical Separation Minimum (RVSM) Airspace related changes
B	10-30-2016	Flight into Known Icing (FIKI) information added. Also added CPDLC, SurfaceWatch, Windshear and general formatting updates.
B1	12-23-2016	Removed required fuel additive section. Added icing inhibitor to approved fuel additives section.
B2	3-3-2017	Added content for additional approved fuels, RVSM, and grammatical and formatting enhancements.

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RECORD OF TEMPORARY REVISIONS

Approved temporary changes must be included in this manual for all flight operations.

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LIST OF SERVICE BULLETINS

The following is a list of active Service Bulletins applicable to the operation of this airplane and incorporated into this manual.

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DOCUMENTATION GROUP

The following is a list that defines the group of documents that provides pilots with the information required for the safe and efficient operation of the airplane.

This documentation group is effective for serial numbers SN-49 and after, and prior airplanes SN-11 thru SN-48 with HAC Service Bulletin HA-420-42-001 installed.

Serial number SN-11 thru SN-48 without HAC Service Bulletin HA-420-42-001 installed continue using HJ1-29000-003-001 Rev A3 until compliance with HAC Service Bulletin HA-420-42-001 is achieved.

MANUAL PART NO.	REV. NO.	MANUAL	RELEASE DATE
HJ1-29000-003-001	B2	Airplane Flight Manual	3-3-2017
HJ1-29000-035-001	B2	Electronic Checklist	3-3-2017
HJ1-29000-005-001	B2	Pilot's Operating Manual	3-3-2017
HJ1-29000-007-001	B2	Quick Reference Handbook, Vol I and II	3-3-2017

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HA-420 AFM

INTRODUCTION

INTRODUCTION

This Airplane Flight Manual (AFM) contains the material required to be furnished to the pilot by Federal Aviation Regulations.

This AFM must be carried in the airplane at all times and it is the owner/operator's responsibility to ensure that all issued, applicable revisions and supplements are incorporated.

This AFM has been prepared to provide pilots with the information required for the safe and efficient operation of the airplane.

Honda Aircraft Company supports authorized facilities worldwide for ease of maintenance and service on our aircraft. For information on how to obtain revisions for this manual or other Honda Aircraft Company service publications, visit the HondaJet Connect website: <https://hondajetconnect.com/>.

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HA-420 AFM

INTRODUCTION

SECTION CONTENTS

Section 2 – Limitations Contains FAA approved operating limitations, which must be observed during aircraft operations except where a deviation is specifically authorized.

Section 3 – Emergency Procedures Contains FAA approved procedures to protect the occupants and airplane from serious harm during critical conditions requiring immediate response. Pilots should memorize procedural steps highlighted with bold type in a red box, so they can execute without the checklist.

Section 3A – Abnormal Procedures Contains FAA approved procedures to maintain an acceptable level of airworthiness or reduce operational risk resulting from a failure.

Section 3B – Advisory Procedures Contains FAA approved procedures to maintain an acceptable level of airworthiness or reduce operational risk resulting from a failure.

Section 4 – Normal Procedures Contains FAA approved routine operating procedures for the aircraft and its systems.

Section 5 – Performance Contains FAA approved data defining the aircraft performance under standard performance conditions.

Section 6 – Weight and Balance Contains FAA approved data intended to assist the operator in ensuring the aircraft is properly loaded.

Section 7 – Supplements Contains FAA approved supplements. The supplements include operating limitations, procedures, performance data, and other necessary data for the aircraft conducting special operations and/or equipped with specific equipment not covered by the basic AFM.

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HA-420 AFM

INTRODUCTION

REVISION INSTRUCTIONS

Honda Aircraft Company Technical Publications will update this manual and distribute revisions to this manual as required.

The List of Effective Pages shows the revision status of each page by notating the date of the revision. The Record of Revisions shows each revision and its date of release.

MANUAL CONVENTIONS

TABLE OF CONTENTS

A table of contents precedes each section of the manual, except Section 3B. In the Emergency and Abnormal Procedures sections, the table of contents is grouped by system. Within each system, procedures without an associated crew alerting system (CAS) message are listed first, followed by procedures with an associated CAS message. Within each group, the table lists procedures alphabetically.

OPTIONAL EQUIPMENT

This manual contains limitations and procedures for using standard and optional equipment. Omit procedural steps referring to any optional equipment that is not installed.

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USE OF L(R) AND L-R

L(R) or L-R may precede a CAS message text. L(R) applies if the message annunciates as either L or R, depending on the affected side. L-R applies to a condition affecting both sides simultaneously.

USE OF L(R) VERSUS 1(2)

L(R) terminology is used for systems that are located on the left or right side of the aircraft. An example is the engine anti-ice system, shown with either an L or R, preceding the associated CAS text. Other systems that are redundant but not associated with the left or right side use 1 or 2. An example is AHRS 1 which is a redundant system that can couple to the left or right side flight instruments.

AURAL ALERTS

Aural alerts accompany specific conditions. AFM procedure titles include the aural alert where applicable. Aural alerts may be voice alerts or tones. A description of the aural alert is adjacent to the aural symbol as shown below.



The Aural symbol indicates a tone or voice alert will sound. Next to the symbol is the applicable tone or voice alert enclosed in quotation marks.

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HA-420 AFM

INTRODUCTION

NOTES, CAUTIONS, AND WARNINGS

This AFM uses the following notations to categorize procedures addressing safety or airplane operation.

WARNING

Operating procedures or techniques which may result in personal injury or loss of life if not carefully followed.

CAUTION

Operating procedures or techniques which may result in damage to equipment if not carefully followed.

NOTE

Additional, significant operating information requiring emphasis.

CHECKLIST EXECUTION

A Checklist is complete when the --- END OF PROCEDURE --- statement is reached. In cases where the checklist continues for more than one page, the text **┌ Procedure Continued ┐** appears at the bottom of the page.

Some more complex emergency procedures have significant impact to the actions required to be followed during approach and landing. In these cases the emergency procedure has been written to incorporate all applicable normal procedure steps, and allows the crew to use the single checklist all the way through landing. These checklists include subsections titled the same as the associated normal procedures checklist. The number for each subsection starts at “1” to indicate the subsection represents the complete set of normal / abnormal steps.

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October 30, 2016

HJ1-29000-003-001

Page 1-5

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CHECKLIST EXECUTION (continued)

In some cases a checklist may be partially completed enroute, but have additional steps or considerations during the descent, approach, landing, or following landing. For these cases, the procedure contains subsection headers indicating in which phase of flight the step(s) should be executed. These checklists have continuous numbering.

NOTE

Checklists may contain a step directing "Land as soon as possible" or "Land at nearest suitable airfield", but these steps are not necessarily the last step in the checklist. Continue through the remaining checklist steps while diverting to the selected airport. Checklist is not complete until --- END OF PROCEDURE --- is reached.

SYMBOLS AND "IF" STATEMENTS

Certain procedures are dependent on prescribed conditions (see the example below). These conditions are presented in bold italics as independent "If" statements. If additional levels of "If" statements are required, they will be preceded by a symbol (e.g., • or ○).

Procedures completed by following the actions in the "If" statement will be indicated by an --- END OF PROCEDURE --- statement following the last step. Otherwise, the additional steps following the other possible "If" statements should be followed.

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HA-420 AFM

INTRODUCTION

SYMBOLS AND “IF” STATEMENTS (continued)

WING BLEED LEAK

A leak has been detected in wing anti-ice system

1. WING ANTI-ICE.....OFF

2. Icing conditionsExit

If WING BLEED LEAK message remains

3. L ENGINE BLEED OFF

4. Altitude..... Descend to FL 250 or below

5. Land at nearest suitable airport

If WING BLEED LEAK message remains

6. L ENGINE BLEED NORM

7. R ENGINE BLEED OFF

○ *If no icing conditions are encountered prior to landing or SAT above 5 °C*

8. Land using normal procedures

Figure 1. Example Procedure

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HA-420 AFM

INTRODUCTION

CAS MESSAGE INHIBITING

The following describes features that inhibit certain CAS messages during certain conditions or phases of flight.

TAKEOFF AND LANDING INHIBIT

CAS messages deemed non-critical during the takeoff or landing are inhibited until the takeoff or landing phase is complete.

OTHER PHASE INHIBITS

During other phases of normal operations such as battery only operation on the ground or engine starting, some CAS messages which reflect unpowered equipment or automatic operations are inhibited. This simplifies the CAS list display and allows the pilot to more easily recognize a true failure condition.

CASCADING FAILURES

Failure of some systems cause other associated failures. An example is an electrical bus failure which would subsequently fail equipment powered solely by that bus. In most cases, the design mitigates flight crew distraction by inhibiting cascading CAS messages related to the associated failure and allows the crew to more easily understand the primary failure. In these cases, the procedures for the posted CAS message include all the required steps for the un-annunciated failure, and a table is included at the end of the procedure listing inoperative equipment.

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HA-420 AFM

INTRODUCTION

CIRCUIT BREAKERS AND SOLID STATE RELAYS (SSR)

With the exception of the ALTERNATE GEAR RELEASE Extension procedure, these checklists do not require circuit breaker manipulation. However, if a circuit breaker is found to be open, it may be reset once. If the circuit breaker opens again, do not attempt another reset.

Solid State Relays are associated with specific electrical buses and equipment. In cases where an SSR trip may cause a failure, resetting the SSR is referenced in the appropriate checklist. As with circuit breakers, SSRs may be reset once. If the SSR trips again, do not attempt another reset.

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HA-420 AFM

INTRODUCTION

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HJ1-29000-003-001

Page 1-10

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY

Additional performance related terminology is included in Section 5 – Performance. Additional weight and balance terminology can be found in Section 6 – Weight and Balance.

TERM	DEFINITION
%MAC	Percent of Mean Aerodynamic Chord is the CG location expressed as a percentage of the average chord length of the wing.
ADS-B	Automatic Dependent Surveillance-Broadcast
AMMD	Airport Moving Map Display
Arm	The horizontal distance from the reference datum to the center of gravity (CG) of an item. The airplane's arm is obtained by adding the airplane's individual moments and dividing the sum by the total weight.
ASI	Aircraft System Indication refers to the system information normally presented on the left side of the center display below the engine indications.
BEW	The Basic Empty Weight is the actual weight of the airplane including all operating equipment that has a fixed location in the airplane. The basic empty weight includes the weight of unusable fuel and full oil.

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
CG	Center of Gravity is the point at which an airplane would balance if suspended. Its distance from the reference datum is found by dividing the total moment by the total weight of the airplane.
EDM	Emergency Descent Mode
EIS	Engine Indication System refers to the engine information display which is normally presented on the top left of the center display.
FRT	Fixed Radius Transitions
GNSS	Global Navigation Satellite System
GP	Glidepath

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
IDLE	<p>The engine has four idle modes: flight idle, ground idle, ground anti-ice idle, and ground APU mode. Automatic selection between modes is accomplished by the FADEC based on inputs from the aircraft.</p> <p><i>Flight idle</i> mode is activated when the aircraft weight is off wheels and the thrust lever is in the IDLE detent. Flight idle provides minimum engine bleed pressure sufficient for ECS and anti-ice systems.</p> <p><i>Ground idle</i> mode is activated when the thrust lever is in the IDLE detent with aircraft weight on wheels. Ground idle provides a stable, minimum engine thrust level for ground operations.</p> <p><i>Ground Anti-Ice idle</i> mode is activated so a higher N_1 is commanded by the FADEC when either the wing anti-ice or engine anti-ice systems are activated.</p> <p><i>Ground APU idle</i> mode is activated when the thrust lever is in the IDLE detent with aircraft weight on wheels and only one engine is running. Ground APU idle provides a stable, minimum engine thrust level for single engine ground operations.</p>
Ind.	Indication

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
ISA	International Standard Atmosphere (standard day) is an atmosphere where: <ul style="list-style-type: none">(1) the air is a dry perfect gas(2) the temperature at sea level is 15 °C,(3) the pressure at sea level is 29.92 in. Hg (1013.2 hPA), and(4) the temperature decreases by 2 °C for every 1000 ft up to 36,089 ft, and is a constant -56.5 °C above 36,089 ft.
KCAS	Knots Calibrated Airspeed is the indicated airspeed corrected for position and instrument error.
KGS	Knots Ground Speed
KIAS	Knots Indicated Airspeed is the speed shown on the airspeed indicator when corrected for instrument error. The IAS values published in this manual assume no instrument error.
M	Mach Number is the ratio of true airspeed to the speed of sound.
M _I	Indicated Mach Number is the Mach number shown on the airspeed indicator. Zero instrument error is assumed.

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
M _{SB}	Speed Brake Operating Mach Number is the maximum Mach number that the speed brake may be operated.
MCT	Maximum Continuous Thrust is the maximum thrust rating available for continuous use.
MLW	Maximum Landing Weight is the maximum weight of the airplane allowed by structural design which should normally not be exceeded at touchdown.
Moment	The product of the weight of an item multiplied by its arm.
MRW	Maximum Ramp Weight is the highest weight of the airplane allowed by structural design.
MTOW	Maximum Takeoff Weight is the highest permissible weight of the airplane at brake release for takeoff.
MZFW	Maximum Zero Fuel Weight is the maximum permissible weight of the loaded airplane prior to usable fuel being added.
NWS	Nosewheel Steering
OSP	Overspeed Protection

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
PFD	The Primary Flight Displays are the left and right displays which present primary flight information as well as a multi-function tile.
Pressure Altitude	Pressure Altitude is the altitude read from the altimeter when the altimeter's barometric adjustment has been set to 29.92 in. Hg (1013 hPA).
Reference Datum	An imaginary vertical plane from which all horizontal distances are measured for balance purposes.
RF	Radius to a Fix
RNAV	Area Navigation
RNP	Required Navigation Performance
SAT	Static Air temperature is the ambient free air static temperature obtained from either 1) ground meteorological sources or 2) from the total air temperature obtained from onboard temperature measurement adjusted for compressibility effects.
SBAS	Satellite Based Augmentation System
SSR	Solid State Relay

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
Station	A location along the airplane fuselage measured in inches from the reference datum and expressed as a number. For example, a point 123 inches aft of the reference datum is Fuselage Station 123.0 (FS 123).
Tare	The weight of all items used to hold or position the airplane on the scales for weighing. Tare includes blocks, shims, and chocks. Tare weight must be subtracted from the associated scale reading.
TAT	Total Air Temperature is the onboard measurement of temperature not corrected for compressibility effects on the temperature probe. TAT will always be greater than SAT in proportion to Mach number.
TLD	Time Limited Dispatch is the period of time the operator may be allowed to defer short-time and long-time faults up to their specified time limits, at which time repair of the faults is required. Deferral and subsequent dispatch is only authorized in accordance with an approved MEL.
TO	Takeoff Thrust is the highest thrust rating available.
TOAC	Time of Arrival Control

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
V_A	Maneuvering Speed is the maximum speed at which application of full control movement will not overstress the airplane.
V_{AC}	Approach Climb speed which is equal to $V_{REF} + 5$ knots.
V_{ENR}	Enroute (single-engine) Climb Speed with gear and flaps up.
V_{FE}	Maximum Flap Extended Speed is the highest speed permissible with wing flaps in a prescribed extended position.
V_{LE}	Maximum Landing Gear Extended Speed is the maximum speed at which an aircraft can be safely flown with the landing gear extended.
V_{LO}	Maximum Landing Gear Operating Speed is the maximum speed at which the landing gear can be safely extended or retracted.
V_{MCA}	Minimum Control Speed Air is the minimum flight speed out of ground effect in the takeoff configuration at which the airplane is controllable with up to 5° of bank when one engine suddenly becomes inoperative and the remaining engine is operating at takeoff thrust.

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HA-420 AFM

INTRODUCTION

SYMBOLS, ABBREVIATIONS, AND TERMINOLOGY (continued)

TERM	DEFINITION
V_{MCL}	Minimum Control Speed Landing is the minimum flight speed out of ground effect in the landing configuration at which the airplane is controllable with up to 5° of bank when one engine suddenly becomes inoperative and the remaining engine is operating at takeoff thrust.
V_{MO}/M_{MO}	Maximum Operating Limit Speed is the speed limit that may not be deliberately exceeded in normal flight operations. V is expressed in knots, and M in Mach Number.
VNAV	Vertical Navigation
V_{REF}	Reference Landing Speed is the landing approach speed with the airplane configured for landing.
V_{SB}	Maximum Speedbrake Speed is the maximum speed where the speedbrake can be safely extended.
V_{SR}	Stall speed is the lowest airspeed at which level flight can be sustained by the airplane's wings.
V_{TIRE}	Maximum Tire Speed is the maximum ground speed permissible during ground operations.

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HA-420 AFM

INTRODUCTION

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October 30, 2016

HJ1-29000-003-001

Page 1-20

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HA-420 AFM

LIMITATIONS

SECTION 2 LIMITATIONS

TABLE OF CONTENTS

INTRODUCTION	2-3
AIRSPEED LIMITS	2-3
ENGINE LIMITS.....	2-6
WEIGHT LIMITS.....	2-12
CENTER OF GRAVITY LIMITS.....	2-14
MANEUVER LIMITS	2-15
MINIMUM FLIGHT CREW/OCCUPANCY LIMITS.....	2-15
KINDS OF OPERATION.....	2-16
OPERATIONAL LIMITS.....	2-17
AVIONICS / AFCS LIMITS	2-23
ELECTRICAL SYSTEM LIMITS	2-34
ENVIRONMENTAL / PRESSURIZATION LIMITS.....	2-35
FUEL SYSTEM LIMITS.....	2-36
HYDRAULIC SYSTEM LIMITS.....	2-38
MISCELLANEOUS SYSTEM LIMITS	2-38
PLACARDS	2-38

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

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FAA APPROVED
March 3, 2017

HJ1-29000-003-001

Page 2-2

FOR TRAINING PURPOSES ONLY

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HA-420 AFM

LIMITATIONS

LIMITATIONS

INTRODUCTION

The limitations included in this section are required for safe operation of the airplane. Compliance with these limitations when operating the airplane is required by Federal Aviation Regulations.

AIRSPEED LIMITS

The listed speeds are as displayed on the Primary Flight Display (PFD) unless otherwise specified.

MAXIMUM OPERATING SPEED V_{MO}/M_{MO}

V_{MO} 270 KIAS

M_{MO} 0.72 M_I

MANEUVERING SPEED V_A

Full application of pitch, roll, or yaw controls should be confined to speeds below the maneuvering speed. Avoid rapid and large alternating control inputs, especially in combination with large changes in pitch, roll, or yaw (e.g., large sideslip angles) as they may result in structural failures at any speed, including below V_A .

V_A 200 KIAS

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HA-420 AFM

LIMITATIONS

MAXIMUM LANDING GEAR SPEEDS

V_{LO} 200 KIAS

V_{LE} 200 KIAS

MAXIMUM FLAP EXTENDED SPEED V_{FE}

Flaps TO/APPR 200 KIAS

Flaps LDG..... 160 KIAS

MAXIMUM SPEEDBRAKE SPEED V_{SB}/M_{SB}

V_{SB}/M_{SB} No limit

MINIMUM CONTROL SPEED AIR V_{MCA}

The speed shown is the minimum demonstrated speed with rudder bias operational.

Flaps UP 105 KIAS

Flaps TO/APPR 100 KIAS

MINIMUM CONTROL SPEED LANDING V_{MCL}

The speed shown is the minimum demonstrated speed with rudder bias operational.

Flaps LDG..... 95 KIAS

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HA-420 AFM

LIMITATIONS

MAXIMUM TIRE SPEED V_{TIRE}

V_{TIRE} is the maximum ground speed for both nose and main wheel tires.

V_{TIRE} 165 KGS

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HA-420 AFM

LIMITATIONS

ENGINE LIMITS

ENGINE OPERATING LIMITS

OPERATING CONDITIONS		OPERATING LIMIT				
Thrust Setting	Time Limit (Minutes)	N ₁ Fan % RPM	ITT °C	N ₂ Turbine % RPM	Oil Press (1) PSIG	Oil Temp °C
Takeoff (TO)	5 10 OEI	100.0	860	100.9	179 max 55 min	165 max 10 min
	2	100.0	885	100.9	179 max 55 min	165 max 10 min
Maximum Continuous (MCT)	None	100.0	860	100.9	179 max 55 min	143 max 10 min
	15	100.0	860	100.9	179 max 55 min	165 max 10 min
Idle	None	(2)	860	(2)	47 max 15 min	143 max -40 min
	15	(2)	860	(2)	47 max 15 min	165 max -40 min
Start	None	-----	556 (3)	-----	--- (4)	-40 min

NOTES:

- (1) Minimum and maximum oil pressure varies based on N₂ speed. Values in the table are based on maximum N₂ for TO and MCT, and typical N₂ for Idle. Higher oil pressure indications are acceptable when oil temperature is below 60 °C. Oil pressure outside the normal range is indicated by red digits and a cautionary range is provided for low oil pressure and is indicated by amber digits. Operation above the maximum limit (with oil temperature above 60 °C) or in the cautionary range is indicative of an oil system malfunction that must be corrected before next engine operation.
- (2) Idle N₁ and N₂ speeds vary with ambient conditions.
- (3) ITT Start limits only apply when the FADEC START mode ICON is displayed on the Engine Indicating System.
- (4) Oil pressure indications are inhibited when oil temperatures are below -40 °C.

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 2-6

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Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ENGINE INDICATOR MARKINGS

Instrument	Lower Limit (red)	Normal Operating Range (green)	Transient / Caution Range (amber)	Upper Limit (red)
N ₁	-----	0 to 100.0 %	-----	100.0%
ITT		200 to 860 °C (1) 200 to 556 °C (2)	860 to 885 °C (3)	860 °C (1) 885 °C (3) 556 °C (2)
N ₂	-----	0 to 100.9 %	-----	100.9 %
Oil Pressure	(4)	(4)	(4)	(4)
Oil Temperature	-40 °C	10 to 143 °C (5)	143 to 165 °C (6) -40 to 10 °C (7)	165 °C (8) 143 °C (9)
Fuel Flow	-----	0 to 1300 pph	-----	-----

NOTES:

- (1) Applies for continuous operations
- (2) Applies for engine starting
- (3) Applies for 2 minutes with TO thrust
- (4) Oil pressure operating limits are calculated by the FADEC based on engine operating conditions. For more specific information see the Pilot's Operating Manual, HJ1-29000-005-001.
- (5) Oil temperature between 10 °C and 165 °C is normal for 5 minutes of operation at TO thrust.
- (6) Applies from 5 to 15 minutes of operation above 143 °C
- (7) Applicable to engine warm-up near IDLE only. Oil temperature should be in the normal range before selecting high power
- (8) Applies for up to 15 minutes of operation
- (9) Applies following 15 minutes of operation

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 2-7

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HA-420 AFM

LIMITATIONS

WIND LIMITS FOR ENGINE START

Tailwind..... 15 knots maximum

NOTE

It is recommended the aircraft be aligned with the prevailing wind when the velocity is greater than 20 knots to minimize the possibility of an aborted start.

AMBIENT TEMPERATURE LIMITS FOR ENGINE START

Engine starts are permitted following a ground cold soak at ambient temperatures to -40 °C, as defined in the Cold Weather Operations (Refer to Section 4 – Normal Procedures).

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HA-420 AFM

LIMITATIONS

APPROVED ENGINE GROUND OPERATIONS ENVELOPE

The following envelope has been approved for engine ground operations.

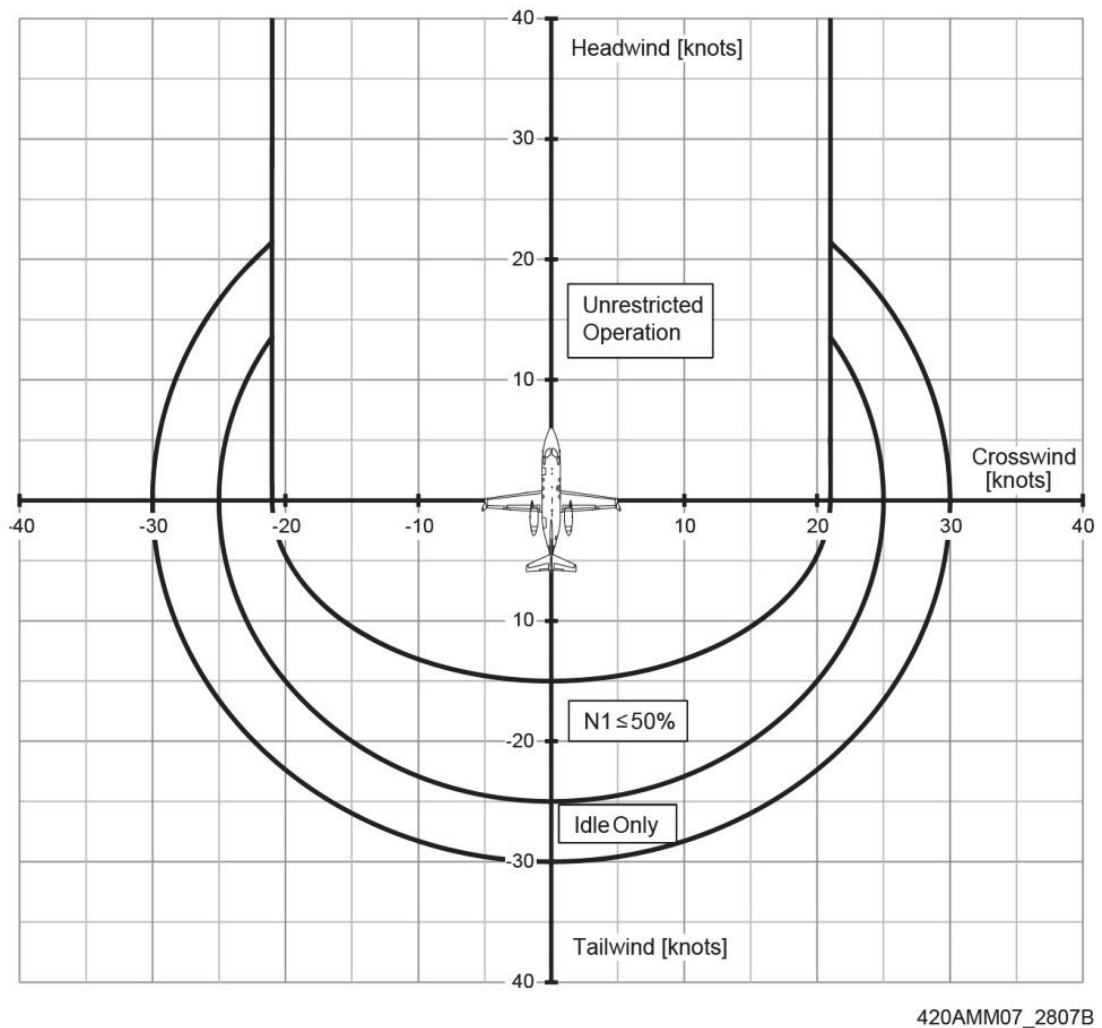


Figure 2. Approved Engine Ground Operations Envelope

NOTES:

- Headwind is not limited.
- Unrestricted operations with crosswind up to 21 knots and tailwind up to 15 knots.
- N_1 must be 50 percent or less for crosswind from 22 to 25 knots, and tailwind from 15 to 25 knots.
- IDLE only operations for crosswinds or tailwinds from 26 to 30 knots.
- For gusty conditions, an additional 5 knots are allowed.

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 2-9

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

AMBIENT TEMPERATURE OPERATING LIMITS

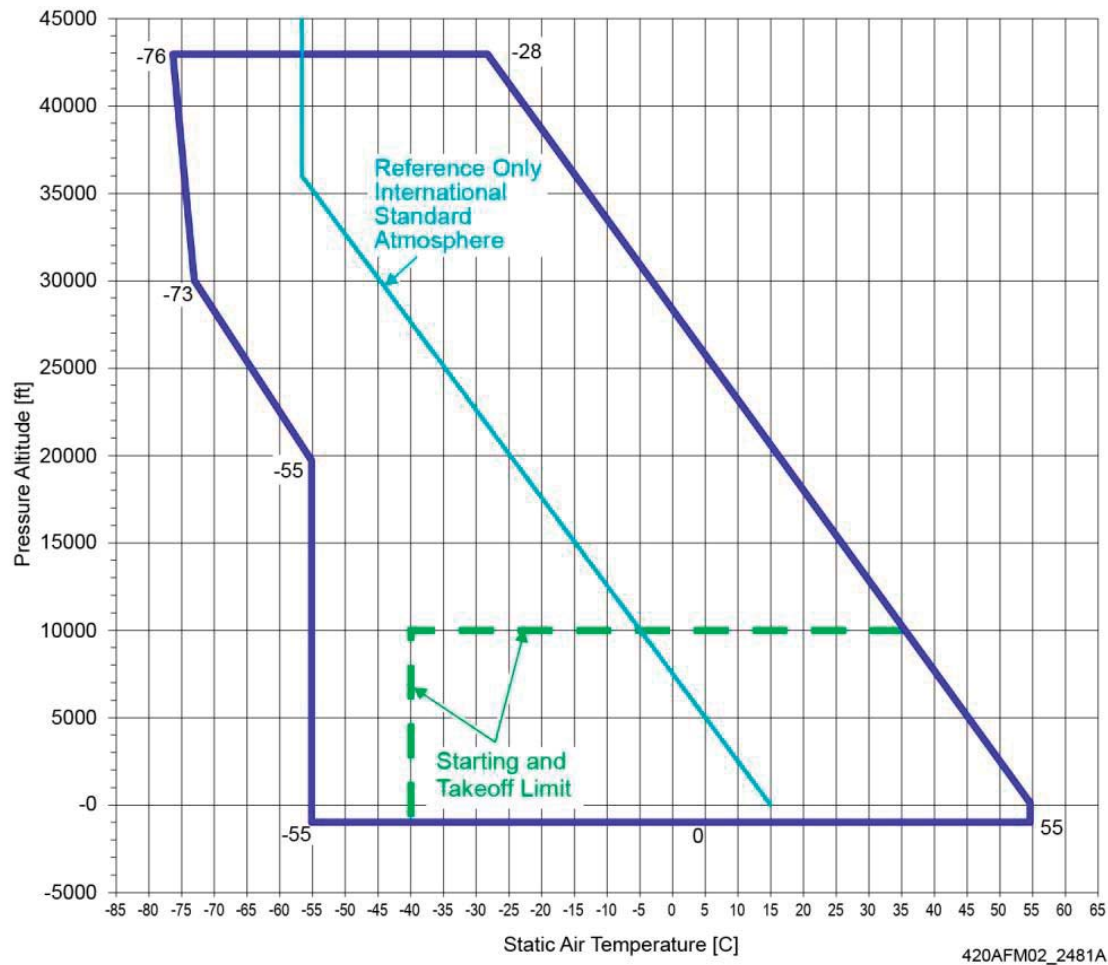


Figure 3. Ambient Temperature Operating Limits

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HA-420 AFM

LIMITATIONS

STARTER LIMITS (GROUND STARTS)

Starter duty cycle is as follows:

- 30 seconds on, 30 seconds off
- 30 seconds on, 30 seconds off
- 30 seconds on, 30 minutes off

APPROVED OIL

The following Type II oils are approved. Additional approved oils may be found in the HF120 Service Bulletin 79-0001.

Oil brands have no incompatibilities, but it is recommended to avoid indiscriminate mixing of different brands of oil. Refer to the most recent version of HF120 Service Bulletin 79-0001 for details on changing from one brand of oil to another.

- AeroShell Ascender
- BP Turbo Oil 2380 / Exxon Turbo Oil 2380
- BP Turbo Oil 2197 / Exxon Turbo Oil 2197
- Mobil Jet Oil II
- Mobil Jet Oil 254
- Mobil Jet Oil 387

APPROVED OIL CONSUMPTION

The maximum allowable oil consumption rate is 48 cc (0.05 quarts) per hour during normal operation.

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October 30, 2016

HJ1-29000-003-001

Page 2-11

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HA-420 AFM

LIMITATIONS

WEIGHT LIMITS

Maximum Ramp Weight	10,680 lbs (4,844 kg)
Maximum Takeoff Weight	10,600 lbs (4,808 kg)
Maximum Landing Weight	9,860 lbs (4,472 kg)
Maximum Zero Fuel Weight	8,800 lbs (3,992 kg)
Maximum Fwd Baggage Compartment Loading	100 lbs (45 kg)
Maximum Aft Baggage Compartment Loading	400 lbs (181 kg)

TAKEOFF WEIGHT LIMIT

Takeoff weight is limited by the most restrictive of the following:

- Maximum Takeoff Weight
- Max Weight to Achieve Takeoff Climb
Requirements..... Refer to Section 5, Performance
- Max Weight to Meet Brake Energy
Requirements..... Refer to Section 5, Performance
- Takeoff Field Length Refer to Section 5, Performance

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

LANDING WEIGHT LIMIT

Landing weight is limited by the most restrictive of the following:

- Maximum Landing Weight

NOTE Perform Hard/Overweight Landing Inspection if the Maximum Certified Landing Weight is exceeded.

- Maximum Weight to Achieve Approach Climb
Requirements..... Refer to Section 5, Performance
- Maximum Weight to Meet Brake Energy
Requirements..... Refer to Section 5, Performance
- Landing Field Length..... Refer to Section 5, Performance

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

CENTER OF GRAVITY LIMITS

The center-of-gravity must be maintained within the Weight and CG Envelope during all flight conditions.

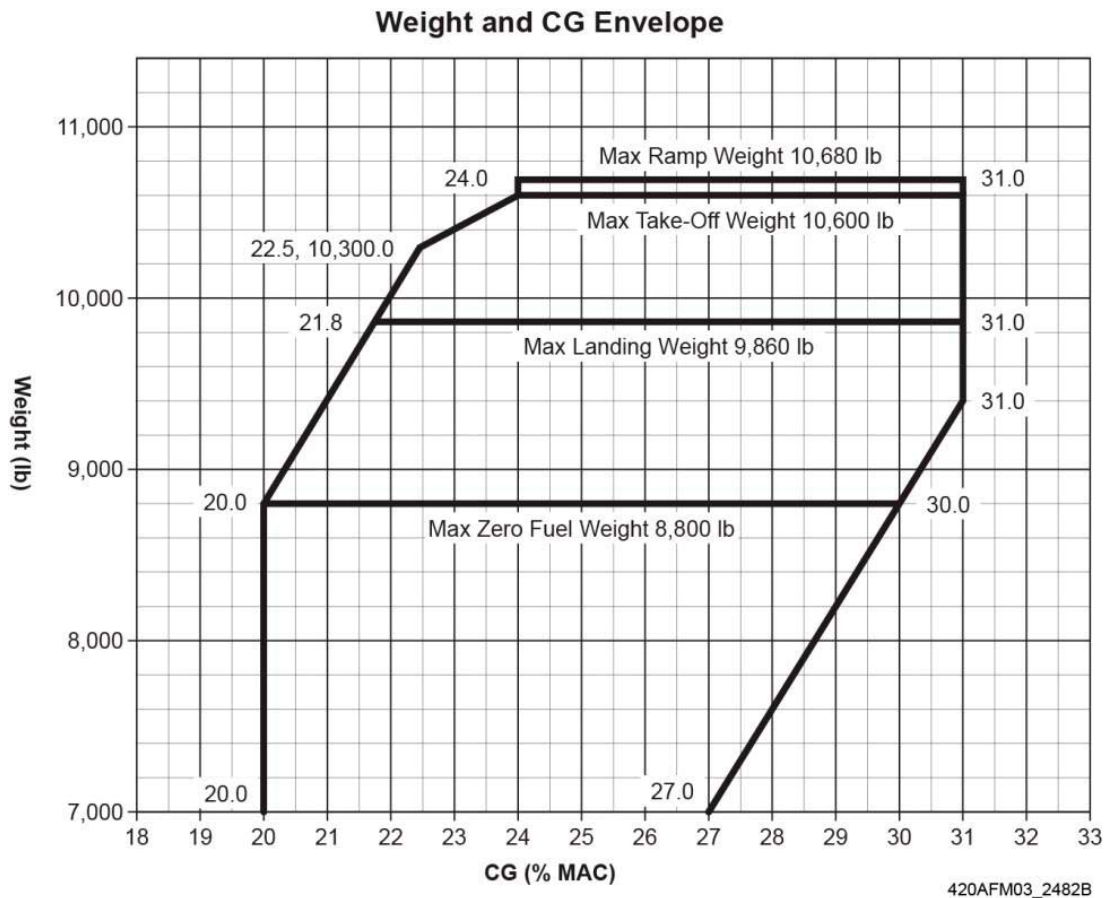


Figure 4. Weight and CG Envelope

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LIMITATIONS

MANEUVER LIMITS

This airplane is approved in the Normal Category.

Maneuvers are limited to any maneuver incident to normal flying, stalls (except whip stalls), and steep turns in which the angle of bank is not more than 60 degrees.

Acrobatic maneuvers, including spins, are prohibited.

MANEUVERING LOAD FACTOR LIMITS

Flaps UP+3.27g, -1.31g

Flaps TO/APPR and LDG+2.0g, 0.0g

MINIMUM FLIGHT CREW/OCCUPANCY LIMITS

MINIMUM FLIGHT CREW

The minimum flight crew is one pilot in the left seat. For single pilot operation, the following equipment must be operative and available:

- Autopilot
- Quick Reference Handbook – Normal Procedures
- Quick Reference Handbook – Emergency/Abnormal Procedures

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

MAXIMUM OCCUPANCY

- Single Pilot, 1 Pilot and 1 Passenger (cockpit), 5 Passengers (cabin when equipped with optional side-facing seat)
- Two Pilots, 1 Pilot and 1 Copilot, 5 Passengers (cabin when equipped with optional side-facing seat)

NOTE The lavatory seat is not certified for use during takeoff and landing.

KINDS OF OPERATION

This airplane is certificated in the normal category and is eligible for the following kinds of operations when the appropriate instruments and equipment required by the airworthiness and the Minimum Equipment List are installed and operational:

- VFR Day
- VFR Night
- IFR
- RVSM
- Flight into Known Icing

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

OPERATIONAL LIMITS

The following equipment must be checked and operable in accordance with the procedures in Section 4 – Normal Procedures prior to flight:

- Pitch Trim Systems (normal and standby)
- Stall Warning System
- Fire Detection and Suppression Systems
- Switchlights
- Cockpit Speakers

OPERATION IN RVSM DESIGNATED AIRSPACE

This airplane has been demonstrated to meet the requirements of 14 CFR 91 Appendix G and FAA Advisory Circular 91-85, dated 21 August 2009, “Authorization of Aircraft for Flight in Reduced Vertical Separation Minimum Airspace” and is qualified for flight operations in RVSM airspace. This does not constitute operational approval.

In addition, this airplane has been demonstrated to meet the requirements of CS ACNS.E.RVSM (Reduced Vertical Separation Minimum) and is qualified for flight operations in European RVSM airspace. This does not constitute operational approval.

The following equipment must be operable prior to entering RVSM airspace:

- Air Data Computer 1 and 2
- Pilot and Copilot Primary Flight Displays
- Altitude Alerter
- Transponder 1 or 2
- Autopilot

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March 3, 2017

HJ1-29000-003-001

Page 2-17

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HA-420 AFM

LIMITATIONS

TAKEOFF AND LANDING

Takeoff Flap SettingsUP, TO/APPR
Landing Flap SettingLDG
Landing Flap Setting (with ice accumulated on airframe)TO/APPR
Runway Surface..... Smooth, Hard, and Paved Only
Runway Slope..... +/- 2%
Airfield Pressure Altitude..... -1,000 to 10,000 ft.
Ambient Temperature.....-40 °C to ISA+40 °C
Tailwind.....10 knots
Crosswind.....20 knots
Turnaround Time.....Refer to Section 5, Performance

ENROUTE

Maximum Pressure Altitude.....43,000 ft
Maximum Altitude for landing gear extension.....18,000 ft
Maximum Altitude for flap extension
• TO/APPR.....18,000 ft
• LDG18,000 ft

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HA-420 AFM

LIMITATIONS

ALL OPERATIONS

ICING LIMITATIONS

The aircraft has been approved for operation within the icing envelope defined by 14 CFR Part 25, Appendix C. Conditions outside of the approved envelope may be identified by:

- Unusually extensive ice accreted on the airframe in areas not normally observed to collect ice
- Inability of the protected areas (wing leading edge or engine inlet) to prevent ice buildup (following normal warm-up time) or unusually extensive ice accretes on the wing or engine nacelle aft of the protected area
- Erratic air data and angle of attack indications
- Inability of the windshield heat system to keep the inboard zones free of ice
- Performance losses larger than normally encountered in icing conditions
- Frequent autopilot retrimming or mistrim alerts during straight and level flight
- Unusual control force or control deflection or uncommanded control movement or unusual lateral trim requirements
- Moderate to severe aircraft buffet

Supercooled cloud environment and freezing rain, freezing drizzle, or mixed conditions have not been tested. Operation in these conditions is prohibited. These icing environmental conditions are outside the icing envelope of 14 CFR Part 25, Appendix C, and may exceed the capabilities of the ice protection system, and may result in a serious degradation of performance or handling characteristics.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ICING LIMITATIONS (continued)

For operations in known or forecast icing, the airplane must be operated with its ice protection systems used as described in OPERATING IN ICING CONDITIONS (Section 4 – Normal Procedures). Where specific operational speeds and performance information have been established for such conditions, this information must be used.

Ground Operations

Engine anti-ice must be ON in ground icing conditions.

A visual and tactile check of the wing leading edge and wing upper surfaces is required prior to takeoff in ground icing conditions.

Ground icing conditions are defined as outside air temperature of 5 °C (41 °F) or less, or either L or R wing fuel temperature cannot be determined to be above 0 °C (32 °F) and:

- Visible moisture (rain, drizzle, sleet, snow, fog, etc.) is present; or
- The airplane was exposed to visible moisture (rain, drizzle, sleet, snow, fog, etc.) since the previous landing; or
- The difference between the dew point and the outside air temperature is 3 °C (5 °F) or less; or
- The airplane was exposed to atmospheric conditions conducive to formation of frost; or
- Water is present on the wing

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ICING LIMITATIONS (continued)

Takeoff is prohibited with any ice, snow, slush, or frost (including polished frost) adhering on aircraft critical areas. Aircraft critical areas are defined as:

- Wing and winglet surfaces
- Horizontal and vertical stabilizer
- Flaps and control surfaces
- Engine inlets and pylons
- Air data probes
- Windshield
- Fuel vents
- Landing Gear

Wing anti-ice must be ON for takeoff in ground icing conditions. Thrust settings above that used for normal taxi are prohibited for ground operations with the wing anti-ice on.

Ground deicing/anti-icing operations are limited to Type I, II de-icing fluids and Type IV anti-icing fluids. Refer to TAKEOFF – ANTI-ICE FLUID PERFORMANCE ADDITIVES (Section 5 – Performance) for increased takeoff speeds and distances, if Type II or IV fluids are used.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ICING LIMITATIONS (continued)

Flight Operations

Engine anti-ice must be ON for flight in visible moisture when SAT is 5 °C (41 °F) to -40 °C (-40 °F).

Wing anti-ice must be NORM for all operations, except wing anti-ice must be ON for takeoff in icing conditions and flight above FL 340 in icing conditions. Inflight operation of wing anti-ice should not be necessary at temperatures below -40°C, but the wing anti-ice switch must remain in NORM.

Use of autopilot is prohibited in conditions exceeding the approved icing envelope defined by 14 CFR Part 25, Appendix C.

Minimum thrust setting when wing anti-ice is operating is 62% N_1 . Thrust settings as low as 50% N_1 are allowed if required for descent or deceleration, but these thrust reductions must be limited to less than 5 minutes.

Minimum speed in icing conditions is 180 KIAS except as required for takeoff, approach and landing.

Flaps and landing gear must remain retracted in icing conditions except as required for takeoff, approach and landing.

Retraction of flaps to UP is prohibited following an icing encounter with flaps extended until the flaps can be confirmed free of ice, except as required for normal takeoff.

Extension of flaps to LDG is prohibited following an icing encounter unless the airframe can be confirmed free of ice.

The wing ice inspection lights must be operable prior to flight into known or forecast icing at night.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

AVIONICS / AFCS LIMITS

GENERAL

The installed Garmin G3000™ system complies with AC 20-138D for IFR navigation using GPS for enroute, terminal area, and approach operations under the conditions described in the Navigation Operational Capabilities section. The Global Navigational Satellite System (GNSS) has been approved per TSO-C115c and TSO-146c.

NAVIGATION OPERATIONAL CAPABILITIES

The *G3000* system must use at least software version Honda HA-420 System 1792.09 as the minimum version. The aircraft must have other operational, approved navigation equipment installed appropriate to the operation.

The *G3000* Integrated Avionics System Cockpit Reference Guide, P/N 190-01490-01 Rev B, dated September 2016 (or later appropriate revision) must be immediately available to the flight crew whenever navigation is predicated on the use of the system.

When using the *G3000* VNAV system, the barometric altimeter must be used as the primary altitude reference for all operations; including instrument approach procedure step-down fixes.

The navigation equipment as installed has been found to comply with the requirements established for the following navigation specifications. This does not constitute an operational approval.

GNSS FDE availability, where required per the following table, must be verified using the Garmin WFDE Prediction Program, part number 006-A0154-01 (010-G1000-00) or later.

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HA-420 AFM

LIMITATIONS

NAVIGATION OPERATIONAL CAPABILITIES (continued)

Navigation Specification	Operational Requirements/ Authorizations	Required Equipment	Reference Guidance
Oceanic and Remote Areas of Operation (Class II Navigation)	GNSS FDE availability must be verified prior to flight.	Two Avionics Computer systems, operating and receiving usable navigation information from each of the dual GNSS sensors (or one navigation system and one GNSS sensor for those routes requiring only one long range navigation sensor).	Meets the applicable requirements of AC20-138D, AC90-100A, AC91-70A, FAA Order 8400.33.
North Atlantic Tracks (NAT) Minimum Navigation Performance Specifications (MNPS)	GNSS FDE availability must be verified prior to flight.	Two Avionics Computer systems, operating and receiving usable navigation information from each of the dual GNSS sensors (or one FMS and one GNSS sensor for those routes requiring only one long range navigation sensor).	Meets the applicable requirements of AC20-138D, AC91-70A.

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HA-420 AFM

LIMITATIONS

NAVIGATION OPERATIONAL CAPABILITIES (continued)

Navigation Specification	Operational Requirements/ Authorizations	Required Equipment	Reference Guidance
RNAV-10 RNP-10	GNSS FDE availability must be verified prior to flight.	Two Avionics Computer systems, operating and receiving usable navigation information from each of the dual GNSS sensors.	Meets the applicable requirements of AC20-138D FAA Order 8400.12C.
B-RNAV/ RNAV-5 RNP-5	EPU/ANP does not exceed RNP.	At least one Avionics Computer system is receiving usable navigation information from the GNSS sensor.	Meets the applicable requirements of AC20-138D, AC90-96A, AC90-100A, EASA AMC 20-4.
RNP-4 Oceanic and Remote Area Operations	GNSS FDE availability must be verified prior to flight. EPU/ANP does not exceed RNP.	Two Avionics Computer navigation systems, operating and receiving usable navigation information from each of the dual GNSS sensors.	Meets the applicable requirements of AC20-138D, FAA Order 8400.33.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

NAVIGATION OPERATIONAL CAPABILITIES (continued)

Navigation Specification	Operational Requirements/ Authorizations	Required Equipment	Reference Guidance
RNAV-2 RNAV-1 P-RNAV RNAV Routes (DPs, STARS, Q and T Routes) RNP-2 RNP-1	GNSS is required for takeoff in P-RNAV airspace. GNSS FDE availability must be verified prior to flight for DPs that require GNSS. EPU/ANP does not exceed RNP.	At least one Avionics Computer system is receiving usable navigation information from one or more GNSS sensor (required for takeoff in P-RNAV airspace and some DPs).	Meets the applicable requirements of AC20-138D, AC90-105, AC90-96A, AC90-100A, JAA TGL 10.
RNP-APCH [titled RNAV (GPS) or RNAV (GNSS)] -including RNP procedures to a minimum value of RNP-0.3 (LNAV or LPV minimums). RNP AR-APCH procedures, and approach procedures with RF legs are NOT authorized.	All instrument approach procedures that are retrieved from the navigation system database are authorized. GNSS is required to initiate RNAV (GPS) approach procedures. For RNAV (GPS) approach procedures, a missed approach is required if both GNSS sensors become unavailable.	At least one Avionics Computer system is receiving usable navigation information from a GNSS sensor (required for RNAV (GPS) approach procedures).	Meets the applicable requirements of AC20-138D, AC90-105, EASA AMC 20-27.

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HA-420 AFM

LIMITATIONS

NAVIGATION OPERATIONAL CAPABILITIES (continued)

Navigation Specification	Operational Requirements/ Authorizations	Required Equipment	Reference Guidance
	<p>EPU/ANP does not exceed RNP (except during a missed approach procedure following loss of GNSS navigation).</p> <p>Maximum predicted RAIM outage is 5 minutes.</p> <p>For ILS, LOC, LOC-BC, LDA, and SDF approach procedures, the active navigation source must be LOC or BC (green needles) prior to crossing the final approach fix.</p>		
Enroute, Terminal and Approach Vertical Navigation (VNAV)	Use of Vertical Glidepath (GP) guidance to a published DA is approved.	The selected navigation system is receiving usable information for a baro-VNAV (or SBAS, if applicable) solution.	Meets the applicable requirements of AC20-138D, AC90-100A, AC90-105, EASA AMC 20-27.

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 2-27

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Honda Aircraft Company

HA-420 AFM

LIMITATIONS

NAVIGATION OPERATIONAL CAPABILITIES (continued)

The following Advanced RNP functions are supported:

- Parallel Offsets
- RNAV Holding
- Fixed Radius Transitions (FRT)

The following Advanced RNP functions are not supported:

- Radius to a Fix (RF Legs)
- Scalable RNP
- Time of Arrival Control (TOAC)

VOR NAVIGATION

VHF NAV must not be used as the primary means of navigation beyond 100 nm from the VOR station.

SAFETAXI

The SafeTaxi airport moving map display (AMMD) functionality with own-ship position symbol is designed to assist flight crews in orienting themselves on the airport surface to improve pilot positional awareness during taxi and flight operations. Do not use the AMMD function as the basis for ground maneuvering. SafeTaxi is limited to ground operations only.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

WEATHER RADAR

Do not operate the weather radar on the ground in close proximity to ground crew. All personnel must remain beyond the safe operating zone which is 11 feet (3.4 meters) from the antenna.

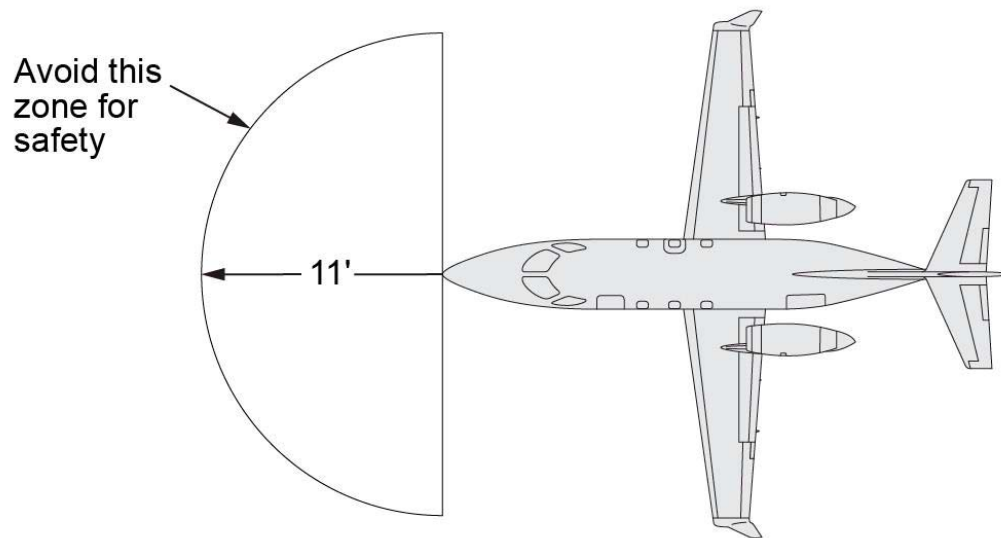


Figure 5. Safe Zone for Weather Radar Operation

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LIMITATIONS

ATTITUDE AND HEADING REFERENCE SYSTEM (AHRS)

Operation in the following regions is not authorized due to unsuitability of the magnetic fields near the Earth's poles:

- 1) North of 72° North latitude at all longitudes
- 2) South of 70° South latitude at all longitudes
- 3) North of 65° North latitude between longitude 75° W and 120° W. (Northern Canada)
- 4) North of 70° North latitude between longitude 70° W and 128° W. (Northern Canada)
- 5) North of 70° North latitude between longitude 85° E and 114° E. (Northern Russia)
- 6) South of 55° South latitude between longitude 120° E and 165° E. (Region south of Australia and New Zealand)

SYNTHETIC VISION SYSTEM

Navigation must not be predicated upon the use of the Synthetic Vision System display.

NOTE

The synthetic vision system may not provide either the accuracy or fidelity, or both, on which to solely base decisions and plan maneuvers to avoid terrain or obstacles.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

TRAFFIC COLLISION AVOIDANCE SYSTEM (TCAS)

Pilots are authorized to deviate from their current Air Traffic Control (ATC) clearance to comply with a TCAS II resolution advisory (RA).

Maneuvers based solely on a traffic advisory (TA) or on information displayed on the traffic display are not authorized.

TERRAIN AWARENESS AND WARNING SYSTEM (TAWS)

Navigation or terrain clearance must not be predicated upon the use of the TAWS.

NOTE

The terrain display is intended to serve as a situational awareness tool only. It may not provide either the accuracy or fidelity, or both, on which to solely base decisions and plan maneuvers to avoid terrain or obstacles.

VERTICAL SITUATION DISPLAY

Navigation or terrain clearance must not be predicated upon the use of the Vertical Situation Display.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ELECTRONIC CHECKLIST

The Electronic Checklist must not be edited without the appropriate certification and/or operational approval from the applicable competent authority.

CONTROLLER-PILOT DATA LINK COMMUNICATIONS (CPDLC)

The installed CPDLC system is fully compliant with the requirements of CS ACNS.B.DLS (Data Link Services). Compliance with the above does not constitute an operational approval.

AUTOMATIC DEPENDENT SURVEILLANCE-BROADCAST (ADS-B)

The installed ADS-B Out system has been demonstrated to meet the equipment requirements of 14 CFR 91.225 and 91.227. The installed ADS-B Out system is fully compliant with the requirements of CS ACNS.D.ADSB (1090 MHz Extended Squitter ADS-B Out). Compliance with the above does not constitute an operational approval.

TRANSPONDER

The installed transponder system is able to respond to interrogations in Modes A, C and S and is fully compliant with the requirements of CS ACNS.D.ELS/EHS (Mode S Elementary/Enhanced Surveillance).

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

AUTOPILOT

In order to prevent a mistrim condition, do not manually override the autopilot.

Minimum Use Heights:

- Climb, Enroute, or Descent: 500 ft AGL
- Approach:
 - Visual and Non-precision: 200 ft AGL
 - Precision: 100 ft AGL

YAW DAMPER

The Yaw Damper must be disengaged for taxi, takeoff, and landing.

The Yaw Damper must be engaged in flight except as required for takeoff and landing.

Minimum Use Height

- 50 ft AGL

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ELECTRICAL SYSTEM LIMITS

GENERATOR LIMITS

All Operations 325 amps (continuous) each

NOTE

The GCU will automatically trip offline due to an overcurrent. The trip point varies based on the amount of overcurrent and duration. A GCU trip is imminent anytime **GENERATOR OVERLOAD** is posted.

EXTERNAL POWER LIMITS

External power cart minimum voltage is 27.0 Volts DC (VDC). The power cart must be capable of generating a minimum of 1,000 amps momentarily and 300 amps continuously.

BATTERY LIMITS FOR GROUND STARTS

Minimum battery voltage for start 23.5 volts

Minimum battery cold-soak temperature -20 °C

NOTE

If the batteries have been cold-soaked as defined in Cold Weather Operations (Section 4 – Normal Procedures), engine start is permitted to 22.5 volts.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

ENVIRONMENTAL / PRESSURIZATION LIMITS

ALTITUDE LIMITS FOR DEGRADED OPERATIONS

Maximum Pressure Altitude

during single engine bleed operations25,000 ft

Maximum Pressure Altitude

during single cabin inflow operations25,000 ft

CABIN PRESSURIZATION LIMITS

Maximum Differential Pressure9.0 psid

Maximum Differential Pressure for Landing0.3 psid

GROUND COOLING MODE

Operation with Ground Cooling Mode enabled is prohibited during taxi or flight.

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

FUEL SYSTEM LIMITS

APPROVED FUELS

Description	Specification
Jet A	ASTM-D 1655
Jet A-1 (AVTUR)	ASTM-D1655 (DEF STAN 91-91)
JP-8	MIL-DTL-83133
PRC No. 3 Jet Fuel	GB6537-2006 (1)

NOTES:

- (1) See the Prohibited Fuels Additives section

MINIMUM/MAXIMUM FUEL TEMPERATURE

-40 °C / 60 °C

MAXIMUM FUEL IMBALANCE

All Operations100 lbs

FAA APPROVED
March 3, 2017

HJ1-29000-003-001

Page 2-36

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

APPROVED FUEL ADDITIVES

Fuel biocide SOHIO Biobor JF additive is approved at a concentration not to exceed 20 ppm (270 ppm total additive) of elemental boron.

- Shock Treatment (Max) (270 PPM) = 2.6 Fl Oz/100 Gal of Jet Fuel
- Maintenance Level (135 PPM) = 1.3 Fl Oz/100 Gal of Jet Fuel

Fuel Static Dissipator Dupont Stadis 450 anti-static additive or equivalent is approved to bring fuel up to 300 conductive units, but must not exceed 1 part per million (ppm).

- 0.5-3.0 mg/l or 0.189-1.135g/100 Gal of Jet Fuel

Fuel icing inhibitors (FSII) meeting specification MIL-DTL-27686 (EGME) or MIL-DTL-85470 (DiEGME) are approved in amounts not more than 0.15% by volume.

- 0.15 Gal/100 Gal or 0.6 Qts/100 Gal or 19.2 Oz/100 Gal

NOTE If used, the icing inhibitor must be well blended into the fuel and not poured or splash-blended into the fuel tank. The icing inhibitor must be added either by an injector system on the fuel unit (truck or stand) or by injection into the fuel stream at the re-fuel port using aerosol cans.

PROHIBITED FUEL ADDITIVES

The following additives defined in GB6537-2006 are not approved for use:

- T1502 Antistatic Agent
- T1601 Anti-Wear Agent (Corrosion Inhibitor/Lubricity Improver)
- T1602 Anti-Wear Agent (Corrosion Inhibitor/Lubricity Improver)

FAA APPROVED
March 3, 2017

HJ1-29000-003-001

Page 2-37

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

HYDRAULIC SYSTEM LIMITS

APPROVED HYDRAULIC FLUID

Approved Hydraulic Fluid..... MIL-PRF-87257

MISCELLANEOUS SYSTEM LIMITS

Oxygen supply must be adequate for the intended flight.

Each pilot's onside audio speaker must be on if headsets are not worn.

The cockpit and cabin must have been warmed to at least -15 °C prior to takeoff if the aircraft has been cold-soaked, as defined in Cold Weather Operations (Section 4 – Normal Procedures).

Crew and passenger oxygen masks are not approved for use above 40,000 ft cabin altitude. Prolonged operation of passenger masks above 25,000 ft cabin altitude is not approved.

Only the cabin wireless system may be used to provide airborne access services (Internet/World Wide Web) to the aircraft's cabin passengers with access via portable electronic devices (PEDs). Use of cell phones for this purpose (other than through 802.11 b/g WiFi) is prohibited.

Maintenance personnel must use the Cabin Telecommunications Router Configuration Page during ground operations only. Use of this page at other times is prohibited.

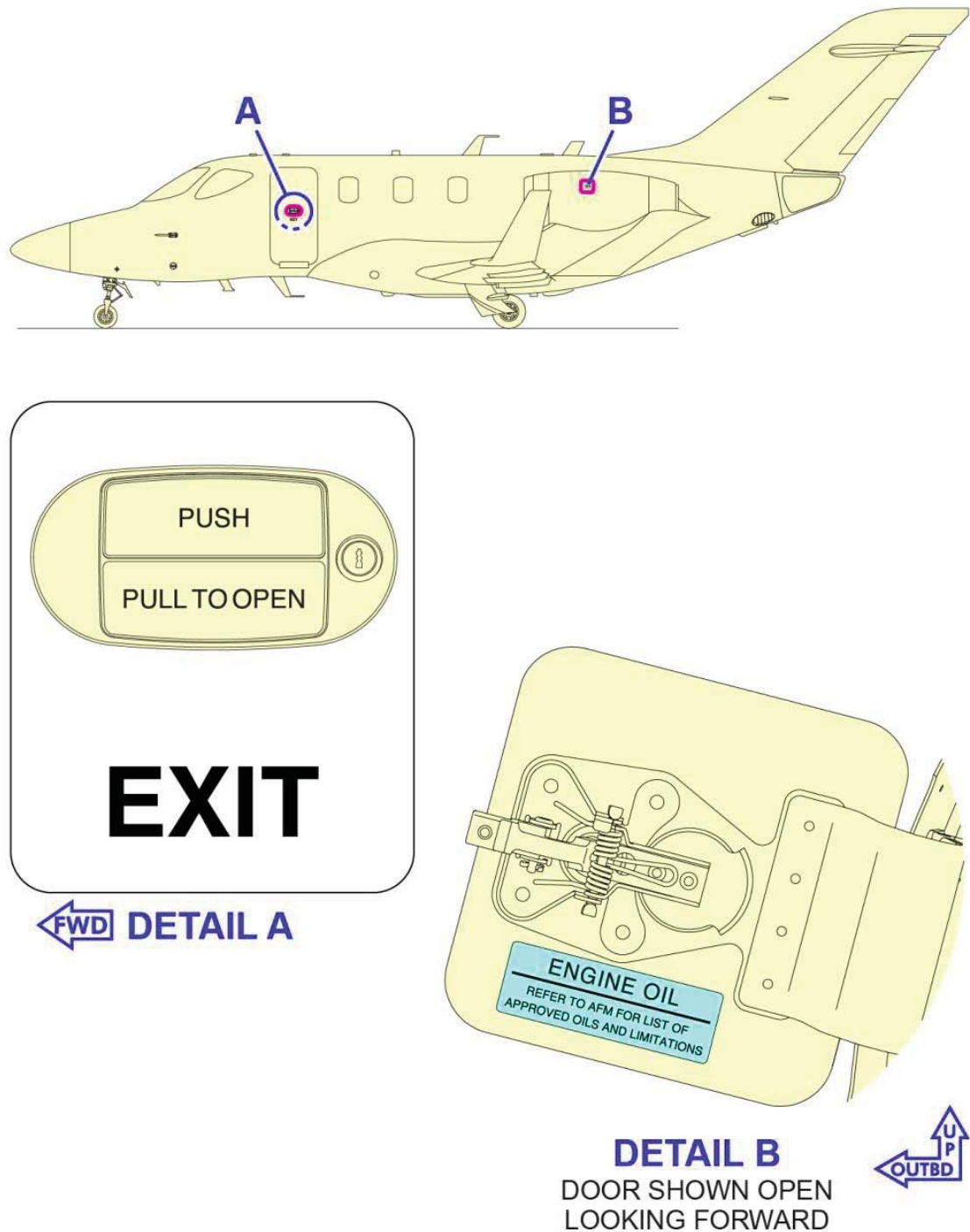
PLACARDS

Placards remind the flight crew and passengers of certain types of equipment and safety devices.

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HA-420 AFM

LIMITATIONS



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Figure 6. Exterior Placards

HJ1-29000-003-001

FAA APPROVED
October 30, 2016

Page 2-39

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HA-420 AFM

LIMITATIONS

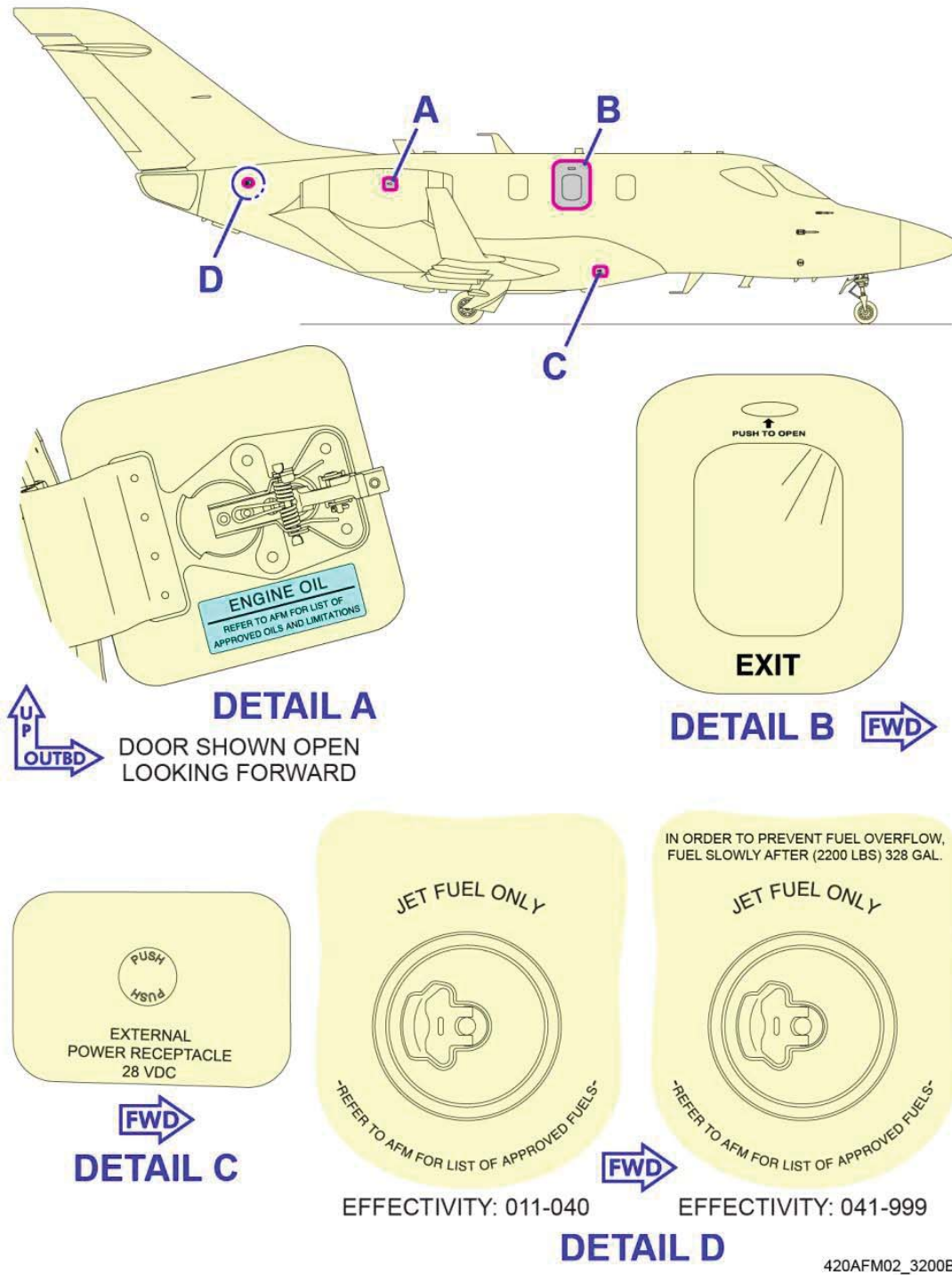


Figure 7. Exterior Placards

HJ1-29000-003-001

FAA APPROVED
October 30, 2016

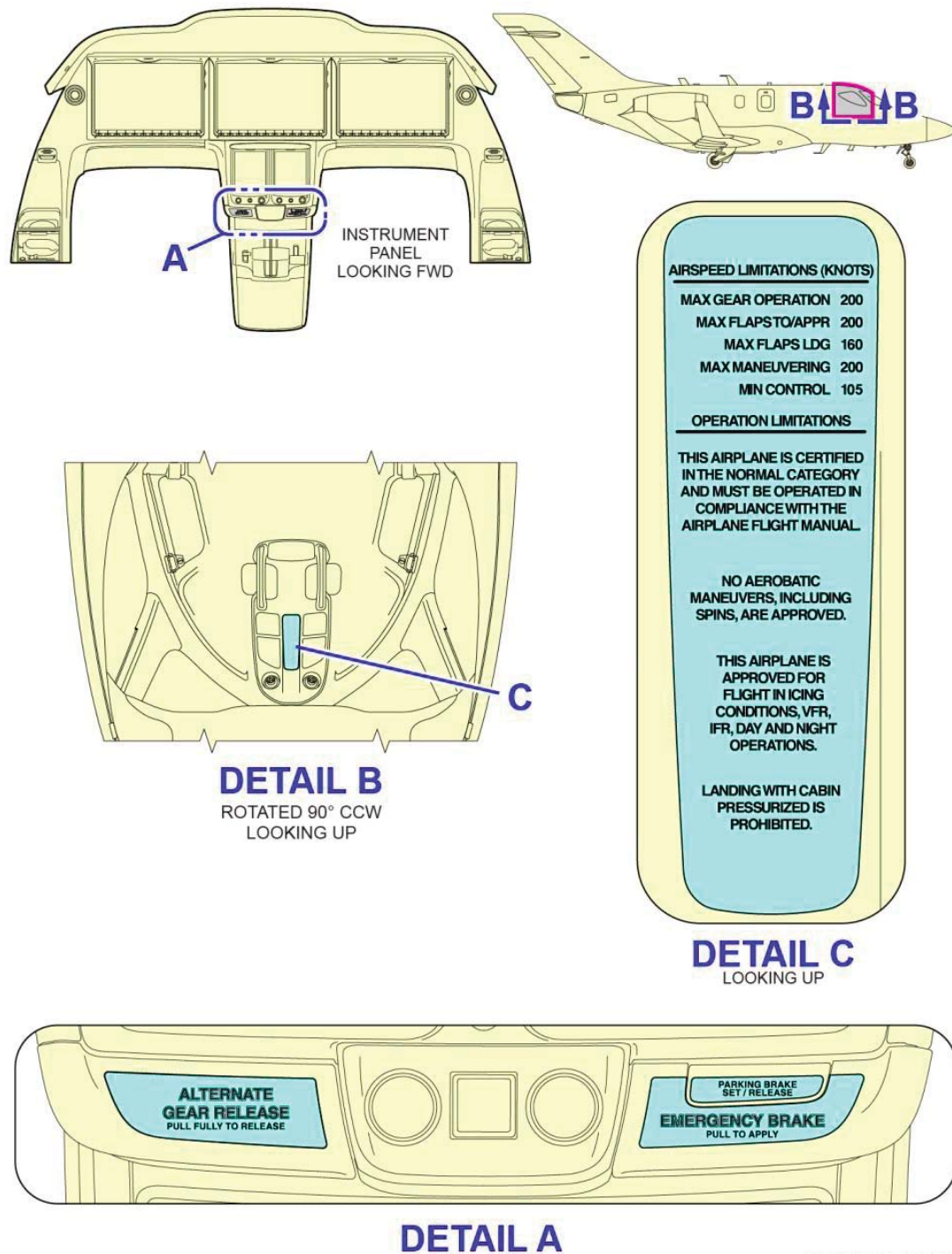
Page 2-40

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Honda Aircraft Company

HA-420 AFM

LIMITATIONS



420AFM02_3203B

Figure 8. Interior Placards

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 2-41

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

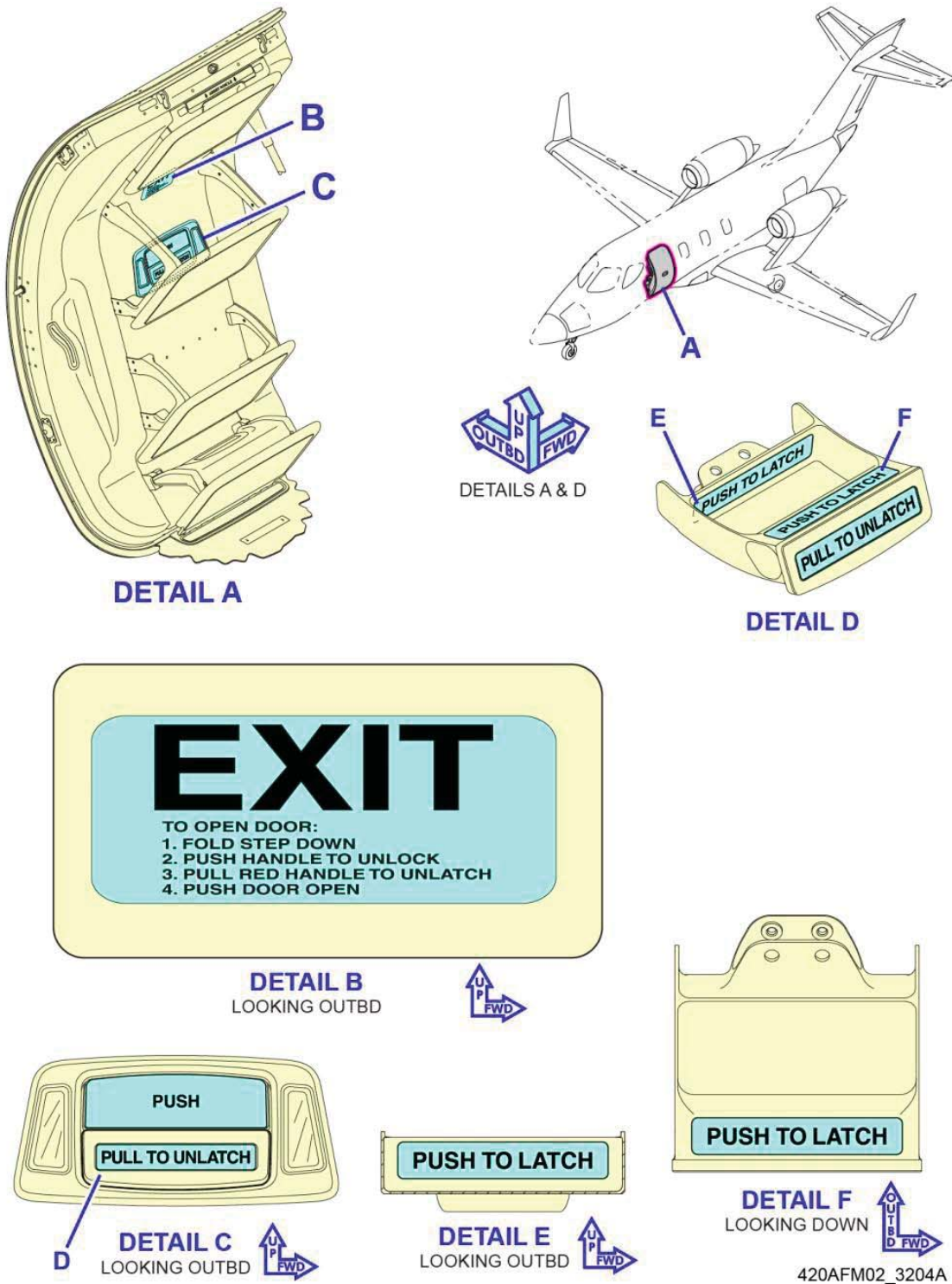


Figure 9. Interior Placards

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 2-42

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

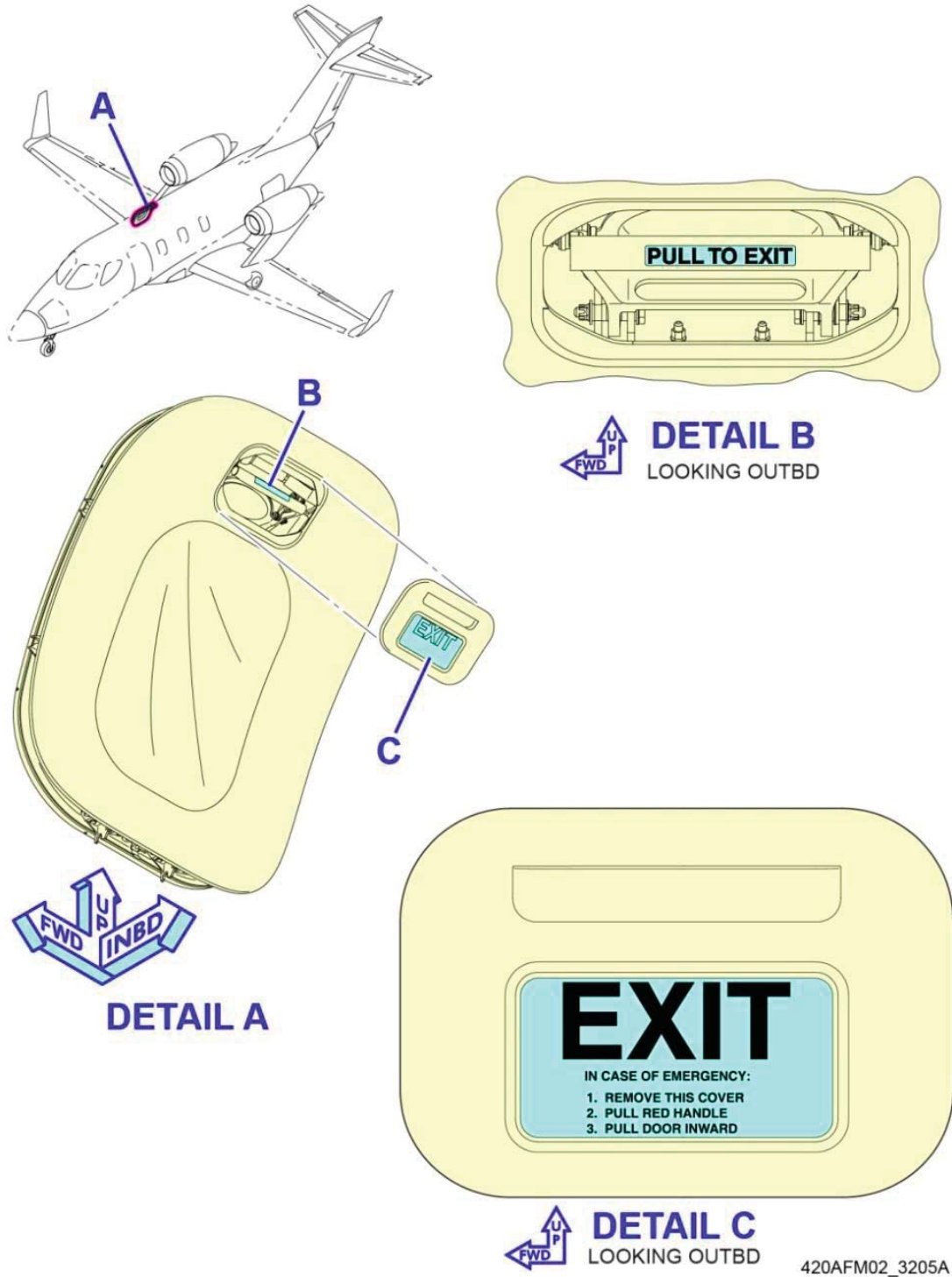


Figure 10. Interior Placards
HJ1-29000-003-001

FAA APPROVED
October 30, 2016

Page 2-43

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

LIMITATIONS

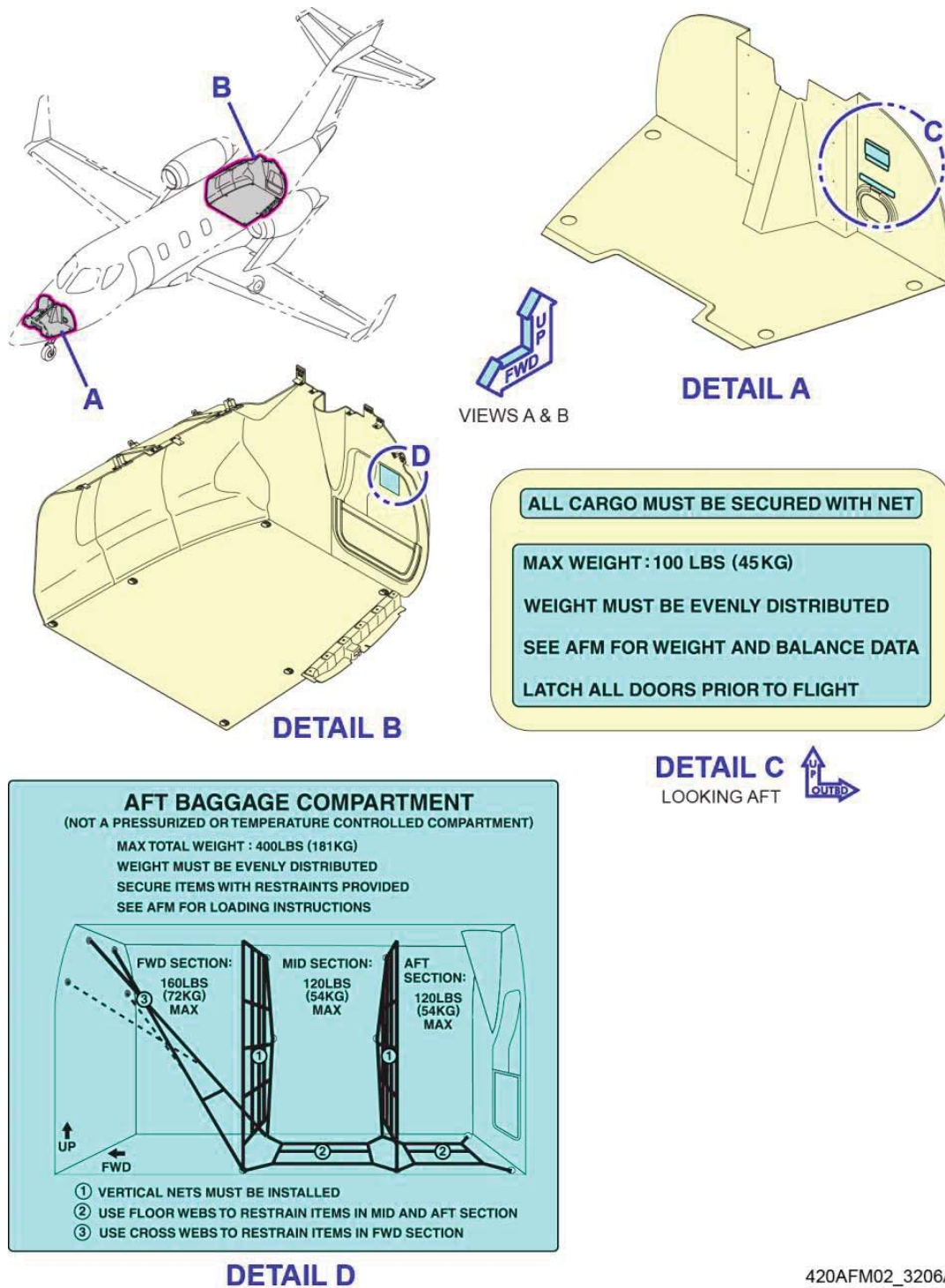


Figure 11. Interior Placards
HJ1-29000-003-001

FAA APPROVED
October 30, 2016

Page 2-44

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

EMERGENCY

SECTION 3 EMERGENCY PROCEDURES

CAS INDEX

CAS Message	Page
CABIN ΔP HIGH	3-17
CABIN ALT HIGH	3-19
L(R) CABIN BLEED LEAK	3-20
ELEC EMER POWER	3-7
L(R) ENG BLEED LEAK	3-21
L(R) ENG EXCEEDANCE	3-9
L(R) ENGINE FIRE	3-35
L(R) ENGINE FLAMEOUT	3-10
L-R ENGINE FLAMEOUT	3-13
L(R) ENG OIL PRESS LOW	3-15
L(R) ENG VIBRATION HIGH	3-16
LDG GEAR UNSAFE	3-27
NORMAL BRAKES FAIL	3-28
TAKEOFF CONFIG	3-64
WING BLEED LEAK	3-23

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EMERGENCY

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HJ1-29000-003-001

Page 3-2

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EMERGENCY

SECTION 3 EMERGENCY PROCEDURES

TABLE OF CONTENTS

ELECTRICAL SYSTEM

ELEC EMER POWER 3-7

ENGINES

L(R) ENG EXCEEDANCE 3-9
L(R) ENGINE FLAMEOUT 3-10
L-R ENGINE FLAMEOUT 3-13
L(R) ENG OIL PRESS LOW 3-15
L(R) ENG VIBRATION HIGH 3-16

ENVIRONMENTAL / PRESSURIZATION SYSTEMS

CABIN ΔP HIGH 3-17
CABIN ALT HIGH 3-19
L(R) CABIN BLEED LEAK 3-20
L(R) ENG BLEED LEAK 3-21
WING BLEED LEAK 3-23

FLIGHT CONTROLS

Flight Control Malfunction 3-25

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HA-420 AFM

EMERGENCY

LANDING GEAR AND BRAKES

LDG GEAR UNSAFE 3-27

NORMAL BRAKES FAIL 3-28

SMOKE OR FIRE

Smoke or Fire..... 3-29

Electrical Fire or Smoke 3-30

Environmental System Smoke or Odor 3-33

L(R) ENGINE FIRE 3-35

OPERATIONAL

Ditching 3-37

Emergency Descent 3-40

Emergency Evacuation 3-41

Inadvertent Overspeed..... 3-42

Operation with Emergency Power Only 3-43

Operation With #1 Buses Only 3-50

Operation With #2 Buses Only..... 3-55

Rejected Takeoff..... 3-63

Stall Recovery 3-63

Windshear or Terrain Avoidance 3-64

TAKEOFF CONFIG..... 3-64

PFD / EIS / ASI Flags..... 3-65

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HA-420 AFM

EMERGENCY

EMERGENCY PROCEDURES

GENERAL

The procedures contained in this section have been developed by the manufacturer and approved by the certifying agency to provide a means to protect the occupants and the aircraft from harm during a critical situation requiring immediate response. Immediate actions which should be accomplished without reference to this manual are delineated by bold text within a red box.

Emergency procedures are grouped by functional system. Within each functional system area, the procedures are sorted by non-CAS procedures, CAS based, and other visual indications such as PFD flags. Each subgroup is sorted by alphabetical order.

Certain failures are capable of compromising multiple airplane systems. The pilot must respond directly to each annunciated or otherwise identified system failure and consult the AFM for each specific Abnormal or Emergency procedure. For cases where multiple procedures specify different airplane configurations for continued safe flight and landing, the most restrictive must be used. If multiple procedures specify additional landing distance factors, each factor is additive and applied to the normal landing distance.

Some more complex emergency procedures have significant impact to the actions required to be followed during approach and landing. In these cases the emergency procedure has been written to incorporate all applicable normal procedure steps and allow the crew to use the single checklist all the way through landing.

Procedures are for use in flight unless other conditions are specifically called out. If not specifically addressed, while on the ground conditions addressed in this section must be corrected prior to flight, unless dispatch

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EMERGENCY

is allowed by an approved MEL or otherwise authorized by the appropriate aviation authority.

TERMINOLOGY

Sound judgment, a thorough knowledge of the aircraft, its characteristics, and the flight manual procedures are essential for handling any emergency situation. Some emergencies may compromise airworthiness or functionality, and therefore, the terms “land at nearest suitable airport” or “land as soon as possible” may be used which are defined as:

Land at nearest suitable airport – the mission should be terminated, and the aircraft landed at a suitable airfield. The airfield and duration of the flight is left to the aircrew’s discretion based on their specific circumstances. Considerations should include (but are not limited to):

- Severity of the emergency
- Aircraft performance
- Field facilities
- Weather
- Ambient conditions such as lighting
- Degraded aircraft functionality

Land as soon as possible – the mission should be terminated, and the aircraft landed as soon as possible. The same considerations apply when selecting a suitable airfield, but this term is used when prolonged flight is not recommended.

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ELECTRICAL SYSTEM

ELEC EMER POWER

Both generators are off-line in the air, or on the ground with both engines running

1. GENERATORS (one at a time)..... OFF, then NORM

NOTE

During ground operations, N₂ may need to be increased above 55% on the associated engine prior to cycling the generator switch to reset the generator.

If only one generator comes on

2. GENERATOR (inoperative side)..... OFF
3. Accomplish L(R) GENERATOR FAIL
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

If neither generator comes on

2. GENERATOR..... Both OFF

┌ Procedure Continued ┐

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EMERGENCY

ELEC EMER POWER (continued)

NOTE

The Cockpit Flood light may be used to provide cockpit illumination at night when operating on Emergency Power.

3. Accomplish OPERATION WITH EMERGENCY POWER ONLY (Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

ENGINES

L(R) ENG EXCEEDANCE

An exceedance of N_1 , ITT, N_2 , oil pressure, or oil temperature has been detected

1. Thrust Lever (affected engine) Retard

2. Engine Indications Monitor within limits

If engine parameters cannot be maintained within limits

3. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

L(R) ENGINE FLAMEOUT

An un-commanded N_2 deceleration of the associated engine has been detected

DURING TAKEOFF (BELOW V_1 – TAKEOFF ABORTED)

1. Accomplish REJECTED TAKEOFF (Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

DURING TAKEOFF (ABOVE V_1 – TAKEOFF CONTINUED)

1. Pitch Attitude..... Flaps TO/APPR - 10° Nose Up
Flaps UP - 11° Nose Up

NOTE Flight Director TO mode pitch targets are automatically adjusted for flap position and engine failure.

2. Landing Gear (positive rate of climb) UP
3. Airspeed..... Maintain V_2
4. Trim.....As required
5. Airspeed (1500 ft AGL or clear of obstacles) Accelerate to $V_2 + 10$

┐ Procedure Continued ┐

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EMERGENCY

L(R) ENGINE FLAMEOUT (continued)

6. FLAPS UP
7. Airspeed.....Accelerate to 140 KIAS
8. Continue planned takeoff flight path
9. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

DURING FLIGHT

1. Thrust Lever (affected engine) IDLE

NOTE Successful engine relight is indicated by steady increase of ITT and N₂ within 10 seconds.

If engine relights within 10 seconds

2. Thrust Lever (affected engine) As required following
Stabilized idle

--- END OF PROCEDURE ---

┌ Procedure Continued ┐

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EMERGENCY

L(R) ENGINE FLAMEOUT (continued)

If engine does not relight within 10 seconds

2. Thrust Lever (affected engine) CUT OFF

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

3. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

L-R ENGINE FLAMEOUT

An un-commanded N_2 deceleration of both engines has been detected

1. Thrust Levers..... IDLE

NOTE Successful engine relight is indicated by steady increase of ITT and N_2 within 10 seconds.

If one or both engines automatically relights within 10 seconds

2. Thrust Lever (affected engine) As required following
Stabilized idle
3. Accomplish L(R) ENGINE FLAMEOUT
(Section 3 – Emergency Procedures) if required
--- END OF PROCEDURE ---

If neither engine automatically relights within 10 seconds

2. Thrust Levers..... CUT OFF
3. Airspeed..... 140 KIAS Minimum

┌ Procedure Continued ┐

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EMERGENCY

L-R ENGINE FLAMEOUT (continued)

CAUTION

To minimize the risk of rotor lock:

- *Maintain airspeed above 140 KIAS to ensure core rotation.*
- *Following commanded or uncommanded inflight shutdown, maintain positive core rotation throughout the engine-out scenario. If core rotation has stopped, take action to achieve core rotation as soon as practical. Failure to maintain positive core rotation may preclude a successful start.*

4. Altitude Descend below 30,000 ft
5. Accomplish ENGINE AIRSTART (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

L(R) ENG OIL PRESS LOW

Engine oil pressure is below the minimum limit

1. Oil Pressure Indication (affected engine) Monitor

NOTE

Aircraft negative-g maneuvers can cause temporary oil supply interruption resulting in low oil pressure indications (below redline) and an **ENG OIL PRESS LOW** message. Engine operations below the oil pressure limits are permitted for a maximum of 15 seconds before engine shutdown is required.

If ENG OIL PRESS LOW message remains

2. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

L(R) ENG VIBRATION HIGH

Excessive engine vibration has been detected

1. Thrust Lever (affected engine) Slowly reduce thrust

NOTE If icing conditions were encountered, cycling Thrust Lever between IDLE and MCT may reduce engine vibration due to ice accumulation.

If vibration condition remains

2. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)
--- END OF PROCEDURE ---

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EMERGENCY

ENVIRONMENTAL / PRESSURIZATION SYSTEMS

CABIN ΔP HIGH

Cabin differential pressure has exceeded 9.2 psid

1. Oxygen MasksDON
2. Crew CommunicationsEstablish

NOTE

Crew communications can be established by use of headsets, selecting cockpit speakers ON, or selecting the OXY MASK AUDIO switch to EMER.

3. CABIN INFLOWL or R OFF
4. Altitude.....Descend to FL 250 or below
5. Land at nearest suitable airport

If cabin differential pressure remains high

6. Cycle remaining CABIN INFLOW between OFF and NORM as required to maintain differential pressure below 9.0 psid

┌ Procedure Continued ┐

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EMERGENCY

CABIN ΔP HIGH (continued)

If cabin differential pressure remains high

7. CABIN DUMPDUMP
8. CABIN OXYGEN..... DROP MASK
9. Descend to 10,000 ft MSL or Minimum Safe Altitude,
whichever is higher

If cabin differential pressure remains high

10. CABIN INFLOW Both OFF

--- END OF PROCEDURE ---

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EMERGENCY

CABIN ALT HIGH



**“CABIN
ALTITUDE”**

Cabin altitude has exceeded 10,000 ft (15,000 ft in High Field Mode)

1. Oxygen Masks..... **DON**
 2. OXY MASK AUDIO..... **EMER**

NOTE

“Cabin Altitude” aural alert will repeat every 30 seconds after acknowledgement via the Master Warning Switch.

NOTE

If the airplane is above 25,000 ft MSL and the autopilot is engaged, it will automatically enter Emergency Descent Mode (EDM) when the **CABIN ALT HIGH** warning posts.

If cabin altitude is greater than 15,000 ft

3. Accomplish EMERGENCY DESCENT
(Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

L(R) CABIN BLEED LEAK

A leak has been detected in cabin bleed air system

1. CABIN INFLOW (affected side)..... OFF

2. AltitudeDescend to FL 250 or below

3. Land at nearest suitable airport

If CABIN BLEED LEAK message remains

4. ENGINE BLEED (affected side) OFF

5. WING FLOW FROM R(L) (operable side)

If CABIN BLEED LEAK message remains

6. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN
FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

L(R) ENG BLEED LEAK

A leak has been detected in engine bleed air system or the engine anti-ice system

1. **ENGINE BLEED (affected side)..... OFF**

CAUTION When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

2. Altitude.....Descend to FL 250 or below
3. WING FLOW FROM R(L) (operable side)
4. Land at nearest suitable airport

If ENG BLEED LEAK message remains

5. Icing ConditionsExit
6. ENGINE ANTI-ICE (affected engine)..... OFF

┐ **Procedure Continued** ┐

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EMERGENCY

L(R) ENG BLEED LEAK (continued)

If ENG BLEED LEAK message remains

7. WING ANTI-ICE OFF

If ENG BLEED LEAK message remains

8. Thrust Lever (affected engine) IDLE

If ENG BLEED LEAK message remains

9. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN
FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

WING BLEED LEAK

A leak has been detected in wing anti-ice system

1. WING ANTI-ICE OFF

2. Icing conditionsExit

If WING BLEED LEAK message remains

3. L ENGINE BLEED OFF

4. Altitude.....Descend to FL 250 or below

5. Land at nearest suitable airport

If WING BLEED LEAK message remains

6. L ENGINE BLEED NORM

7. R ENGINE BLEED OFF

○ *If no icing conditions are encountered prior to landing*

8. Land using normal procedures

--- END OF PROCEDURE ---

┌ Procedure Continued ┐

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EMERGENCY

WING BLEED LEAK (continued)

- *If wing protected surfaces cannot be confirmed to be free of ice, or icing conditions are expected during approach or landing*
- 8. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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EMERGENCY

FLIGHT CONTROLS

FLIGHT CONTROL MALFUNCTION

An unexpected aircraft or flight control motion has occurred whose cause is unknown

1. AFCS / TRIM MASTER..... Press and Hold

2. Airplane attitude Recover

If pitch anomaly

3. PITCH SERVO POWER OFF

4. PITCH TRIM MODE..... STBY

5. AFCS / TRIM MASTER..... Release

6. Adjust trim by use of the STANDBY PITCH switch

7. Yaw Damper..... Engage

8. RVSM airspace..... Exit

9. Land at nearest suitable airport

--- END OF PROCEDURE ---

If roll anomaly

3. ROLL SERVO POWER..... OFF

┌ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

FLIGHT CONTROL MALFUNCTION (continued)

4. ROLL TRIM POWER..... OFF
5. AFCS / TRIM MASTER..... Release
6. Yaw Damper..... Engage
7. RVSM airspace..... Exit
8. Land at nearest suitable airport

--- END OF PROCEDURE ---

If yaw anomaly

3. YAW SERVO POWER OFF

NOTE

The rudder bias system is inoperative. Rudder forces with one engine inoperative will increase.

4. YAW TRIM POWER..... OFF
5. AFCS / TRIM MASTER..... Release
6. Autopilot..... As required

○ *If the autopilot is not available*

7. Altitude..... FL 240 or lower
8. Land at nearest suitable airport

--- END OF PROCEDURE ---

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HA-420 AFM

EMERGENCY

LANDING GEAR AND BRAKES

LDG GEAR UNSAFE



**“LANDING
GEAR”**

One or more landing gear is not safe for landing

NOTE “Landing Gear” aural alert will repeat every 10 seconds after acknowledgement via the Master Warning Switch.

1. LANDING GEARDN
2. Landing Gear Indications Verify gear down and locked
(three green)

If unsafe landing gear indications persist and both main gear are down and locked

3. Accomplish ALTERNATE GEAR RELEASE EXTENSION
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

If unsafe landing gear indications persist and only one main gear is down and locked

3. LANDING GEAR UP
4. Accomplish ALTERNATE GEAR RELEASE EXTENSION
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3-27

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EMERGENCY

NORMAL BRAKES FAIL

Normal wheel brakes have failed

1. EMERGENCY BRAKE Apply gradually

CAUTION

Anti-skid will not operate. Avoid cycling the brake handle. Approximately ten applications are available with a fully charged system.

NOTE

Gradually pull emergency brake handle until desired braking action is achieved.

NOTE

Landing distance will increase by 50% on a dry runway and 100% on a wet runway.

--- END OF PROCEDURE ---

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EMERGENCY

SMOKE OR FIRE

SMOKE OR FIRE

Smoke or fire has been detected in the cockpit or cabin area

1. Oxygen Masks.....DON and 100% OXYGEN
2. Smoke Goggles..... DON
3. OXY MASK AUDIO..... EMER
4. CABIN DUMP DUMP

NOTE Attempt to identify the source of the smoke or fumes. Electrical smoke is generally gray or tan in color and irritating to the nose and eyes. Environmental system smoke is usually white in color and less irritating.

NOTE The oxygen mask microphone will be turned off if the OXY MASK AUDIO switch is returned from EMER to NORM. The oxygen mask microphone can be re-enabled through the AUDIO AND RADIOS page on the CDU.

5. CABIN OXYGEN.....As required
6. Accomplish ELECTRICAL FIRE OR SMOKE or ENVIRONMENTAL SYSTEM SMOKE OR ODOR as appropriate (Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

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HA-420 AFM

EMERGENCY

ELECTRICAL FIRE OR SMOKE

Fire or smoke has been determined to be caused electrically

If source is known:

1. Faulty EquipmentIsolate

--- END OF PROCEDURE ---

If source is not known:

1. Generators Both OFF

NOTE

The Cockpit Flood light may be used to provide cockpit illumination at night when operating on Emergency Power, or on #1 Buses Only.

- *If smoke or fire decreases or ceases*

2. Accomplish OPERATION WITH EMERGENCY POWER ONLY (Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

- *If smoke or fire persists*

2. L GENERATOR NORM
3. BUS TIE OPEN
4. BATTERY OFF

┌ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

ELECTRICAL FIRE OR SMOKE (continued)

5. Left DISPLAY REVERSION REV
6. Left PFD Sensors..... AHRS 1, ADC 1

- *If smoke or fire decreases or ceases*

7. Accomplish OPERATION WITH #1 BUSES ONLY
(Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

- *If smoke or fire persists*

7. R GENERATOR NORM
8. Left PFD Sensors..... AHRS 2, ADC 2
9. L GENERATOR OFF

NOTE

The Cockpit Map lights may be used to provide cockpit illumination at night when operating on #2 Buses Only.

┌ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

ELECTRICAL FIRE OR SMOKE (continued)

➤ *If smoke or fire decreases or ceases*

10. Accomplish OPERATION WITH #2 BUSES ONLY
(Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---

➤ *If smoke or fire persists*

10. L GENERATOR NORM
11. BUS TIE Normal
12. BATTERY NORM
13. Left DISPLAY REVERSION Normal
14. Left PFD Sensors AHRS 1, ADC 1

NOTE

Repowering left side buses will result in the
L ENG VIBRATION HIGH and
R ENG VIBRATION HIGH messages posting
briefly following touchdown.

15. Land as soon as possible

--- END OF PROCEDURE ---

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HA-420 AFM

EMERGENCY

ENVIRONMENTAL SYSTEM SMOKE OR ODOR

Smoke or odor has been determined to originate from the environmental system

1. CABIN INFLOWL OFF

If smoke or fire decreases or ceases

2. Altitude.....Descend to FL 250 or below
3. Land at nearest suitable airport

--- END OF PROCEDURE ---

If smoke persists

2. CABIN INFLOWL NORM, R OFF

○ *If smoke or fire decreases or ceases*

3. Altitude.....Descend to FL 250 or below
4. Land at nearest suitable airport

--- END OF PROCEDURE ---

○ *If smoke persists*

3. CABIN INFLOW Both OFF
4. CABIN OXYGEN..... DROP MASK

┐ Procedure Continued ┘

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HA-420 AFM

EMERGENCY

ENVIRONMENTAL SYSTEM SMOKE OR ODOR (continued)

5. Land as soon as possible
 6. Accomplish EMERGENCY DESCENT
(Section 3 – Emergency Procedures)
- END OF PROCEDURE ---

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HA-420 AFM

EMERGENCY

L(R) ENGINE FIRE



**“LEFT (RIGHT)
ENGINE FIRE”**

A fire has been detected in the indicated engine

1. Thrust Lever (affected engine).....IDLE

If ENGINE FIRE indications still present after 15 seconds

2. Thrust Lever (affected engine).....CUT OFF
3. ENGINE FIRE PUSH Switch (affected engine)Lift
cover
and push
4. FIRE EXT PUSH Switch (affected engine).....Push

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE

The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

5. FUEL CROSSFEEDAs required
6. WING FLOW FROM L(R) (operable side)

┌ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

L(R) ENGINE FIRE (continued)

7. TCAS (TCAS II only)TA ONLY
8. ENGINE ANTI-ICE (affected engine).....OFF
9. AltitudeDescend to FL 250 or below
10. Land at nearest suitable airport
11. Accomplish SINGLE-ENGINE APPROACH AND LANDING
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

If ENGINE FIRE indications are not still present after 15 seconds

2. Thrust Lever (affected engine)Maintain at IDLE
3. FUEL CROSSFEED As Required
4. WING FLOW FROM L(R) (operable side)

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

5. TCAS (TCAS II only)TA ONLY
6. Land at nearest suitable airport
7. Accomplish SINGLE-ENGINE APPROACH AND LANDING
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

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HA-420 AFM

EMERGENCY

OPERATIONAL

DITCHING

The aircraft is required to be force landed in water

NOTE

Ditching has not been demonstrated. However, the following procedures are recommended.

DESCENT

1. Airspeed..... 140 KIAS
2. Transponder..... 7700
3. ELT.....ON
4. Brief and prepare passengers for ditching
5. CABIN SIGNSAs required
6. Seat belts and harness..... Fastened
7. Cabin seats.....Positioned for landing
8. TAWS Warnings Disable

┌ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

DITCHING (continued)

APPROACH

1. FLAPS TO/APPR (if operable)
2. Landing Gear Do not extend
3. CABIN SIGNS As required

NOTE

Ditch parallel to and near the crest of the swell unless there is a crosswind of 20 knots or more. In strong wind, heading should be more into the wind and slightly across the swell, planning to touch down on the upslope of the swell near the top.

Wave motion is indicative of wind direction, but the swell does not necessarily move with the wind. Water surface conditions are indicative of wind speed, as related below.

Surface Condition	Wind Speed (kts)
Few white crests	10 – 15
Many white crests	15 – 25
Streaks of foam from crests	25 – 35
Spray blown from tops of waves	35 – 45

┐ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

DITCHING (continued)

LANDING

1. FLAPS LDG (if operable)
2. CABIN DUMP DUMP
3. ENGINE BLEED (Both) OFF
4. Sink rate 200 – 300 ft / minute
5. GENERATOR Both OFF

AFTER LANDING

1. Touchdown Slightly nose high
2. Thrust Levers CUT OFF
3. Emergency Exit Door Open, evacuate airplane
4. Main Entrance Door Open, evacuate airplane

CAUTION

The lower edge of the Main Entrance Door should be above the waterline during calm seas. During rough seas only the Emergency Exit Door should be used.

--- END OF PROCEDURE ---

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HA-420 AFM

EMERGENCY

EMERGENCY DESCENT

EDM

A maximum rate descent is required due to an emergency condition

1. Thrust Levers..... IDLE
2. SPEEDBRAKE (if installed)EXT
3. Airspeed.....Maintain V_{MO} / M_{MO}

CAUTION

If structural damage is suspected, limit airspeed to a speed lower than V_{MO} / M_{MO} and limit maneuvering loads until damage assessment can be made.

CAUTION

Emergency Descent Mode (EDM) will automatically command a descent at V_{MO}/M_{MO} . If a lower speed is desired, the autopilot must be disengaged.

NOTE

EDM cannot guarantee a maximum rate of descent. The pilot must ensure the thrust levers are commanded to IDLE.

NOTE

If the airplane is above 25,000 ft MSL, and the autopilot is engaged, it will automatically enter EDM when the **CABIN ALT HIGH** warning message posts.

┐ Procedure Continued ┐

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HA-420 AFM

EMERGENCY

EMERGENCY DESCENT (continued)

4. Transponder..... 7700
5. CABIN OXYGEN..... DROP MASK (as required)
6. CABIN SIGNS ON
7. ATC Notify and obtain local altimeter setting
8. Descend to 10,000 ft MSL or Minimum Safe Altitude,
whichever is higher
9. Land at nearest suitable airport

NOTE EDM will automatically set the altitude pre-selector to 15,000 ft.

--- END OF PROCEDURE ---

EMERGENCY EVACUATION

The aircraft is required to be evacuated in an emergency situation

1. PARKING BRAKE..... Set
2. Thrust Levers..... CUT OFF
3. ENGINE FIRE PUSH Switches..... LIFT
COVER
and PUSH
4. FIRE EXT PUSH Switch Push (if required)
5. BATTERY..... OFF
6. Evacuate the aircraft using the cabin door and/or the emergency
exit

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3-41

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HA-420 AFM

EMERGENCY

INADVERTENT OVERSPEED



Warble Tone

The aircraft has inadvertently exceeded V_{MO} or M_{MO}

1. Thrust Levers..... IDLE
2. SPEEDBRAKE (if installed)..... EXT
3. Perform wings level pull-up

NOTE

AFCS Overspeed Protection (OSP) mode will engage when speed exceeds 276 KIAS or M 0.73 if the Flight Director is active and the vertical mode is not ALT. OSP will command a 1.5g pull until speed decreases to below 267 KIAS / M 0.715.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH EMERGENCY POWER ONLY

The aircraft is operating on battery power only

NOTE When selected on, the Cockpit Flood light provides illumination during all electrical system configurations and may be used when executing this procedure at night.

1. Icing ConditionsExit

CAUTION Windshield heat, wing anti-ice, and tail de-ice systems are not available when operating with emergency power only.

2. L PFD Verify displaying AHRS 1 and STBY ADC

3. Radio COM 1

NOTE COM 1, NAV 1 and 2, and GPS are the only communication and navigation sources available on emergency power.

4. NavigationNAV 1 or 2, or GPS based only

5. Transponder..... Verify XPDR 1

6. SPEEDBRAKE (if installed).....RET

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH EMERGENCY POWER ONLY

(continued)

7. PRESSURIZATION CONTROL MODE..... HOLD
8. RVSM airspace.....Exit
9. Land as soon as possible

NOTE

A minimum of 60 minutes of operation on battery power has been demonstrated.

DESCENT

1. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. Avionics..... Set
4. Landing Data Set and confirmed
 - a. Radios and Navigation Set

NOTE

COM 1, NAV 1 and 2, and GPS are the only communication and navigation sources available on emergency power.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH EMERGENCY POWER ONLY

(continued)

- b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
- c. Landing Distance.....Confirm

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

NOTE If icing conditions cannot be avoided, reference LANDING WITH ICE ACCUMULATION ON WINGS and SIDE WINDOW LANDING (Section 3A – Abnormal Procedures).

- 5. CAS Messages..... Check
- 6. Approach Briefing.....Complete
- 7. LANDING LightON
- 8. FLAPS TO/APPR

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH EMERGENCY POWER ONLY

(continued)

BEFORE LANDING

1. Landing Gear Alternate extension
 - a. LDG GEAR CTRL circuit breaker.....PULL (B7)
 - b. LANDING GEAR..... DN
 - c. Airspeed..... 150 KIAS minimum
 - d. ALTERNATE GEAR RELEASE handle..... Pull fully
 - e. Yaw airplane, if necessary, to obtain gear locked down
 - f. LANDING GEAR indicator.... Verify three green DN

NOTE The gear DOOR icon will remain posted but a normal landing is possible without gear door-ground contact.

- g. ALTERNATE GEAR RELEASE handle..... Stow

CAUTION *The ALTERNATE GEAR RELEASE handle could interfere with thrust lever operation if not stowed following use.*

NOTE Normal landing gear operations cannot be restored in flight after activation of the alternate gear release.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH EMERGENCY POWER ONLY

(continued)

2. FLAPS LDG
3. Airspeed $V_{REF} + 5$
4. CABIN DUMP DUMP

WARNING

The airplane must be unpressurized prior to landing.

LANDING

1. Thrust Levers IDLE
2. Brakes Apply (after touchdown)
3. SPEEDBRAKE (if installed) EXT

NOTE

The speedbrake may not be operational, or the deflection may be reduced depending on the system and normal accumulator pressure at the time of landing.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH EMERGENCY POWER ONLY

(continued)

FOLLOWING LANDING ROLLOUT

1. Do not taxi
2. Wheel chocks Install

CAUTION

Following landing rollout, the emergency and brake accumulators may drop below the level required for nosewheel steering and braking.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

The following equipment is <u>OPERATIVE</u> for a limited time	
Communication	
L and R Intercom	COM 1
Navigation / AFCS	
L PFD (using Standby Air Data)	MFD
NAV 1	L CDU
Transponder 1	NAV 2
AHRS 1	Standby Instrument
Standby AHRS and ADC	GPS
Low Speed Awareness (Degraded Mode)	Rudder Bias
Engine	
Engines*	FADEC Control*
Engine Instruments	Ignition
Fire Detection	Fire Extinguishers
Vibration Detection	
Flight Control	
Normal Flight Controls*	All Trim
Flaps	
Landing Gear and Brakes	
Landing Gear (alternate extension)*	Normal Brakes (accumulator)*
Hydraulic Pump (ground only)	Anti-Skid (if normal brakes available)
Parking Brake (accumulator)	Emergency Brake (accumulator)*
Nosewheel Steering (accumulator)*	
Fuel	
Fuel Pumps	Fuel Shutoff Valves
Fuel Quantity Indications	Fuel Low Level Sensors
Fuel Crossfeed	
Environmental	
Engine Bleed	Pressurization (HOLD Mode)
Dump Mode	Oxygen*
Bleed Leak Detection	
Ice Protection	
Ice Detection	Standby Probe Heat
Engine Anti-ice (failed on)	
Lighting	
Cockpit Flood Light	L Ice Inspection Light
L Landing Light	Footwell Lights

NOTE: * These items do not require aircraft electrical power to operate and are not time limited.

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3-49

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #1 BUSES ONLY

Operation of the aircraft with #1 buses only is required

1. Thrust Levers.....Avoid rapid movements

CAUTION When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

2. Icing ConditionsExit
3. Radio COM 1

NOTE Copilot will lose all communication capability.

4. NavigationNAV 1 or GPS

NOTE COM 1, NAV 1, and GPS are the only communication and navigation sources available during #1 bus only operation.

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #1 BUSES ONLY (continued)

5. Transponder..... Verify XPDR 1
6. R WINDSHIELD HEAT..... OFF
7. R ENGINE BLEED..... OFF
8. WING FLOWFROM L
9. PRESSURIZATION CONTROL MODE..... HOLD
10. Altitude.....Descend to FL 250 or below
11. Land at nearest suitable airport

DESCENT

1. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. Avionics..... Set
4. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance..... Confirm

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #1 BUSES ONLY (continued)

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

NOTE If icing conditions cannot be avoided, reference LANDING WITH ICE ACCUMULATION ON WINGS and SIDE WINDOW LANDING (Section 3A – Abnormal Procedures).

- 5. CAS Messages..... Check
- 6. Approach Briefing..... Complete
- 7. FLAPS TO/APPR

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #1 BUSES ONLY (continued)

BEFORE LANDING

1. LANDING GEARDN

NOTE Hydraulic pressure indications will be unavailable during #1 bus operation, but the hydraulic pump will be operable.

2. Cabin Differential Pressure Verify less than 0.3 psi
3. CABIN DUMP DUMP (if required)

WARNING The airplane must be unpressurized prior to landing.

4. FLAPS LDG
5. Airspeed..... $V_{REF} + 5$
6. Autopilot/Yaw Damper Disengage

LANDING

1. Thrust Levers..... IDLE
2. BrakesApply (after touchdown)

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

The following equipment will be <u>INOPERATIVE</u> during #1 bus only operations	
Communication	
COM 2	R Audio
Navigation / AFCS	
R PFD	MFD
R CDU	TCAS II
AHRS 2 and ADC 2	NAV 2
ADF	CVFDR
Transponder 2	DME
Radio Altimeter	Weather Radar
SATCOM / Connex TM Weather	Stall Pusher
Engine	
Right Oil Temperature	R Fire Detection
R Fire Extinguisher	Vibration Detection
Flight Control	
Standby Pitch Trim	Speedbrake
Landing Gear and Brakes	
Hydraulic (indication)	
Fuel	
R Fuel Low Level Sensor	Fuel Crossfeed
R Fuel Quantity	R Fuel Shutoff Valve
R Fuel Pump	
Environmental	
Air Conditioner	R Bleed Leak Detection
R Engine Bleed	Cabin Oxygen Automatic Deployment
Pressure Control (normal)	Oxygen Quantity (Ind)
	Cabin Fan
Ice Protection	
R Wing Anti-ice Valve	R Engine Anti-ice (failed on)
R Windshield Zone Heat	Right Probe Heat
Ice Detector 2	
Lighting	
R Landing Light	Strobe Lights
Logo Lights	Taxi Lights
RECOG Lights	NAV Lights
Cockpit Map Lights	Cabin Overhead Lights
Copilot Footwell Light	Cabin Signs
Ice Inspection Lights	

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3-54

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY

Operation of the aircraft with #2 buses only is required

1. PITCH TRIM MODE.....STBY
2. Thrust Levers.....Avoid rapid movements

CAUTION When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

3. Icing ConditionsExit
4. Radio COM 2

CAUTION Pilot will not hear intercom or alerts through headset, but alerts will be automatically routed to cross-side speaker. Speaker volume may need to be adjusted to ensure alerts can be heard while wearing headset.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY (continued)

NOTE The pilot's boom mic, or oxygen mask mic is automatically routed to COM 2 for communication with ATC.

NOTE Marker beacon functionality (aural and visual) will be inoperative on the pilot's side.

5. NavigationNAV 2 or GPS

NOTE COM 2, NAV 2, and GPS are the only communication and navigation sources available for #2 bus only operation.

6. Transponder..... XPDR 2 (if applicable) or notify ATC

7. L WINDSHIELD HEATOFF

8. L ENGINE BLEEDOFF

9. WING FLOW FROM R

10. AltitudeDescend to FL 250 or below

11. NOSE WHEEL STEERINGOFF

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY (continued)

12. Select a long, wide, and dry runway with minimal crosswind

WARNING

Anti-skid is not available. Any braking above light in wet runway conditions could result in blown tires and loss of directional control.

CAUTION

Nosewheel steering is inoperative. Crosswinds should be minimized to ensure adequate directional control during the low-speed portion of the rollout using differential braking.

NOTE

Landing distance will double on a wet runway using light braking.

13. Land at nearest suitable airport

NOTE

Stall Pusher is inoperative. Low Speed Awareness remains operative in a degraded mode based on remaining ADC AOA data.

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY (continued)

DESCENT

1. Landing Field Elevation Verify set
2. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. Avionics..... Set
4. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance..... Confirm

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY (continued)

Flap Setting	Min Approach Speed	Landing Distance
UP	$V_{REF} + 20$	Add 105%
TO/APPR	$V_{REF} + 10$	Add 75%
LDG	$V_{REF} + 5$	Add 60%

NOTE Flaps are inoperative during #2 bus only operation. The flap setting at the time #2 bus only operation was initiated will be the final flap setting.

NOTE Due to anti-skid and the stall pusher being inoperative, the minimum approach speed and landing distance factors are defined in the table.

NOTE Landing distance will double on a wet runway using light braking.

NOTE If icing conditions cannot be avoided, reference LANDING WITH ICE ACCUMULATION ON WINGS and SIDE WINDOW LANDING (Section 3A – Abnormal Procedures).

- 5. CAS Messages..... Check
- 6. Approach Briefing Complete

┌ Procedure Continued ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3-59

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY (continued)

BEFORE LANDING

- | | |
|---|-----------------------|
| 1. Landing Gear | Alternate Extension |
| a. LDG GEAR CTRL circuit breaker..... | PULL (B7) |
| b. LANDING GEAR | DN |
| c. Airspeed..... | 150 KIAS minimum |
| d. ALTERNATE GEAR RELEASE handle..... | Pull fully |
| e. Yaw airplane, if necessary, to obtain gear locked down | |
| f. LANDING GEAR indicator.... | Verify three green DN |

NOTE The gear DOOR icon will remain posted but a normal landing is possible without gear door - ground contact.

- g. ALTERNATE GEAR RELEASE handle..... Stow

CAUTION *The ALTERNATE GEAR RELEASE handle could interfere with thrust lever operation if not stowed following use.*

NOTE Normal landing gear operations cannot be restored in flight after activation of the alternate gear release.

┐ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

OPERATION WITH #2 BUSES ONLY (continued)

2. Airspeed..... (Flaps UP) - $V_{REF} + 20$
(Flaps TO/APPR) - $V_{REF} + 10$
(Flaps LDG) - $V_{REF} + 5$
3. Yaw Damper..... Disengage

LANDING

1. Thrust Levers..... IDLE
2. Brakes..... Apply Gradually
3. Maintain directional control with rudder and differential braking

CAUTION *Brakes must be applied gradually. Light to moderate braking can be applied without skidding tires on a dry surface, however, the pilot should consider runway surface conditions when applying brakes.*

FOLLOWING LANDING ROLLOUT

1. R ENGINE BLEED..... OFF
2. Do not taxi
3. Wheel chocks Install

CAUTION *Following landing rollout, the emergency and brake accumulators may drop below the level required for braking.*

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

EMERGENCY

The following equipment will be <u>INOPERATIVE</u> during #2 bus only operations	
Communication	
COM 1	L Audio
Navigation / AFCS	
Autopilot	L PFD
AHRS 1 and ADC 1	XM Weather
Standby Instrument	NAV 1
Transponder 1	Control Yoke System Control
Standby AHRS and ADC	L CDU
Weather Radar	TCAS
Stall Pusher	SATCOM / Connex TM Weather
Low Speed Awareness (degraded mode)	
Engine	
L Engine Instruments	L Fire Detection
L Fire Extinguisher	
Flight Control	
Pitch (primary), Roll and Yaw Trim	Speedbrake
Flaps	
Landing Gear and Brakes	
Landing Gear (normal extension)	Anti-Skid
Hydraulic Pump	Nosewheel Steering
Fuel	
L Fuel Pump	L Fuel Shutoff Valve
Left Fuel Quantity	Left Fuel Low Level Sensor
Center Fuel Quantity	
Environmental	
Left Engine Bleed	L Bleed Leak Detection
Cockpit Fan	Ground Cooling Fan
Air Conditioner	
Ice Protection	
Ice Detector 1	L Engine Anti-ice (failed on)
Standby Probe Heat	L Wing Anti-ice Valve
L Windshield Zone Heat	L Probe Heat
Lighting	
Beacon	L Landing Light
Cabin Overhead Lights	Cabin Signs
Cockpit Flood Light	NAV Lights
Footwell Lights	Right Taxi Light
Ice Inspection Lights	

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3-62

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Honda Aircraft Company

HA-420 AFM

EMERGENCY

REJECTED TAKEOFF

A condition requiring the takeoff to be rejected has been encountered

1. Thrust Levers.....IDLE
 2. Brakes..... Apply maximum braking
 3. SPEEDBRAKE (if installed) EXT

NOTE

Consult the Aircraft Maintenance Manual (AMM) for required maintenance actions if a rejected takeoff is conducted above 100 knots ground speed.

--- END OF PROCEDURE ---

STALL RECOVERY

STALL

The aircraft has stalled or is approaching stall

1. Pitch Attitude.....Lower to reduce angle of attack
2. Thrust Levers..... TO
3. Level the wings
4. Airspeed..... Accelerate above stall condition

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

EMERGENCY

WINDSHEAR OR TERRAIN AVOIDANCE

PULL UP

WINDSHEAR

A condition has been encountered requiring a vertical escape maneuver

1. Pitch Attitude..... 15° initially
2. Thrust Levers..... TO
3. Do not change gear or flap configuration
4. Pitch Attitude..... Adjust to maintain close to shaker speed

--- END OF PROCEDURE ---

TAKEOFF CONFIG



“TAKEOFF
CONFIGURATION”

NO TAKEOFF

One or more configuration items are not properly set for takeoff

1. Accomplish REJECTED TAKEOFF
(Section 3 – Emergency Procedures)

--- END OF PROCEDURE ---





Honda Aircraft Company

HA-420 AFM

EMERGENCY

PFD / EIS / ASI FLAGS


The following malfunctions are indicated by flags

Display Flag / Indication	Description
AFCS	
	The autopilot has disconnected abnormally. This alert is accompanied by continuous disconnect aural tones. These alerts will repeat continuously until acknowledged. Refer to the Cockpit Reference Guide for details.
SurfaceWatch	
	The aircraft is aligned with a runway during takeoff or landing that is too short, as determined by the SurfaceWatch. This is accompanied by the “Runway Too Short” aural alert. Refer to the Cockpit Reference Guide for details.
	The aircraft is aligned with a taxiway during the landing phase, as detected by the SurfaceWatch. This is accompanied by the “Taxiway” aural alert. Refer to the Cockpit Reference Guide for details.
	The aircraft is aligned with a taxiway and thrust levers are set to TO, as detected by the SurfaceWatch. This is accompanied by the “Taxiway” aural alert. Refer to the Cockpit Reference Guide for details.

Honda Aircraft Company

HA-420 AFM

EMERGENCY

Display Flag / Indication	Description
Traffic Collision and Avoidance System (TCAS)	
	Resolution Advisory (RA) is active. A potentially hazardous intruding aircraft has been detected and is closing to within 15-35 seconds of a potential collision area. An appropriate aural alert is also issued, providing the flight crew with maneuver instructions to avoid the hazardous traffic. This alert is only applicable to TCAS II installations. Refer to the Cockpit Reference Guide for details.

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SECTION 3A ABNORMAL PROCEDURES

CAS INDEX

CAS Message		Page
AFCS MISTRIM		3A-16
AFT BAG DOOR UNSAFE		3A-49
1(2)	AHRS FAIL	3A-16
1(2)	AIR DATA FAIL	3A-17
ANTI-SKID FAIL		3A-170
L(R)	AUDIO FAIL	3A-18
AVIONICS COMPUTER 1 FAIL		3A-19
AVIONICS COMPUTER 2 FAIL		3A-21
BATTERY BUS 1 FAIL		3A-55
BATTERY BUS 1A FAIL		3A-64
BATTERY BUS 1B FAIL		3A-70
BATTERY BUS 1C FAIL		3A-75
BATTERY BUS 2 FAIL		3A-77
BATTERY BUS 2A FAIL		3A-82
BATTERY BUS 2B FAIL		3A-86
BATTERY BUS 2C FAIL		3A-91

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-1

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

	CAS Message	Page
	BUS TIE FAIL	3A-92
	CABIN ALT CTRL FAIL	3A-115
	CABIN ALT CTRL FAULT	3A-116
	CABIN ALT HIGH FIELD	3A-116
L(R)	CABIN BLEED FAIL	3A-117
L(R)	CABIN BLEED TEMP HIGH	3A-118
	CABIN DOOR UNSAFE	3A-50
	CABIN OXYGEN OFF	3A-117
	CPDLC FAIL	3A-23
	DATA ACQUISITION 1 FAIL	3A-23
	DATA ACQUISITION 2 FAIL	3A-26
	DATA ACQUISITION 3 FAIL	3A-29
	DATA CONCENTRATOR 1 FAIL	3A-31
	DATA CONCENTRATOR 2 FAIL	3A-33
	ECS AIR COND FAIL	3A-119
	ECS GND COOLING FAN FAIL	3A-119
	EMER EXIT DOOR UNSAFE	3A-51
L(R)	ENG ANTI-ICE FAIL	3A-143
L(R)	ENG BLEED FAIL	3A-120
L(R)	ENG CONTROL FAIL	3A-109
L(R)	ENG CONTROL FAULT	3A-110

Honda Aircraft Company

HA-420 AFM

ABNORMAL

CAS Message		Page
L-R	ENG FUEL BYPASS	3A-111
L(R)	ENG OIL TEMP HIGH	3A-112
L(R)	ENG OVERSPD PROT FAIL	3A-113
L(R)	ENG START FAIL	3A-106
	EXT PWR DOOR UNSAFE	3A-52
L(R)	FIRE DETECTOR FAIL	3A-114
	FLAP FAIL	3A-127
	FLAP FAULT	3A-128
	FLT DIR MODE CHANGE	3A-35
	FUEL CROSSFEED	3A-131
	FUEL CROSSFEED FAIL	3A-131
	FUEL IMBALANCE	3A-132
	FUEL LEVEL CTRL FAULT	3A-133
L-R	FUEL PRESSURE LOW	3A-134
L(R)	FUEL PRESSURE LOW	3A-134
L(R)	FUEL PUMP FAIL	3A-135
	FUEL QTY FAULT	3A-136
L-R	FUEL QTY LOW	3A-138
L(R)	FUEL QTY LOW	3A-138
	FUEL TEMP HIGH	3A-139
	FUEL TEMP LOW	3A-140

Honda Aircraft Company

HA-420 AFM

ABNORMAL

	CAS Message	Page
	FWD BAG DOOR UNSAFE	3A-53
L(R)	GENERATOR FAIL	3A-92
L(R)	GENERATOR OVERLOAD	3A-94
	HYD PUMP FAIL	3A-141
	HYD PRESSURE LOW	3A-141
	ICE DETECT FAIL	3A-144
	ICE PROT NOT ACTIVE	3A-148
	LDG GEAR FAIL	3A-171
	MAIN BUS 1 FAIL	3A-94
	MAIN BUS 2 FAIL	3A-96
	NO LDG FIELD ELEV	3A-122
	NOSEWHEEL STEER FAIL	3A-172
	OXYGEN LOW	3A-123
	OXYGEN QTY FAIL	3A-124
	OXYGEN UNAVAILABLE	3A-125
	PARK BRAKE FAIL	3A-174
	PARK BRAKE ON	3A-174
	PITCH TRIM FAIL	3A-129
L(R)	PROBE HEAT FAIL	3A-150
	RUDDER BIAS FAIL	3A-35
	SPEEDBRAKE EXTENDED	3A-130

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-4

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

CAS Message		Page
STALL PUSHER FAIL		3A-36
STALL WARN MISCOMP		3A-37
STBY PROBE HEAT FAIL		3A-151
TAIL DE-ICE FAIL		3A-151
TRANSPONDER MODE		3A-37
L-R	WING A/I TEMP LOW	3A-155
WING ANTI-ICE FAIL		3A-152
WING ANTI-ICE FAIL ON		3A-153
L(R)	WING ANTI-ICE OVERHEAT	3A-154
L(R)	WING ANTI-ICE TEMP LOW	3A-158
WING ANTI-ICE UNAVAIL		3A-161
L(R)	WSHD ZONE FAIL	3A-163
L(R)	WSHD ZONE OVERHEAT	3A-164
YAW DAMPER FAIL		3A-38
YAW DAMPER OFF		3A-39

Honda Aircraft Company

HA-420 AFM

ABNORMAL

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-6

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SECTION 3A ABNORMAL PROCEDURES

TABLE OF CONTENTS

AVIONICS / AFCS

Primary Flight or Multi-Function Display Failure.....	3A-15
AFCS MISTRIM.....	3A-16
AHRS 1(2) FAIL	3A-16
AIR DATA 1(2) FAIL.....	3A-17
L(R) AUDIO FAIL	3A-18
AVIONICS COMPUTER 1 FAIL.....	3A-19
AVIONICS COMPUTER 2 FAIL.....	3A-21
CPDLC FAIL.....	3A-23
DATA ACQUISITION 1 FAIL	3A-23
DATA ACQUISITION 2 FAIL	3A-26
DATA ACQUISITION 3 FAIL	3A-29
DATA CONCENTRATOR 1 FAIL	3A-31
DATA CONCENTRATOR 2 FAIL	3A-33
FLT DIR MODE CHANGE.....	3A-35
RUDDER BIAS FAIL.....	3A-35
STALL PUSHER FAIL.....	3A-36

Honda Aircraft Company

HA-420 AFM

ABNORMAL

STALL WARN MISCOMP	3A-37
TRANSPONDER MODE.....	3A-37
YAW DAMPER FAIL.....	3A-38
YAW DAMPER OFF	3A-39
PFD / EIS / ASI FLAGS	3A-41

DOORS

AFT BAG DOOR UNSAFE.....	3A-49
CABIN DOOR UNSAFE.....	3A-50
EMER EXIT DOOR UNSAFE.....	3A-51
EXT PWR DOOR UNSAFE.....	3A-52
FWD BAG DOOR UNSAFE.....	3A-53

ELECTRICAL SYSTEMS

BATTERY BUS 1 FAIL.....	3A-55
BATTERY BUS 1A FAIL	3A-64
BATTERY BUS 1B FAIL	3A-70
BATTERY BUS 1C FAIL	3A-75
BATTERY BUS 2 FAIL.....	3A-77
BATTERY BUS 2A FAIL	3A-82
BATTERY BUS 2B FAIL	3A-86
BATTERY BUS 2C FAIL	3A-91
BUS TIE FAIL	3A-92
L(R) GENERATOR FAIL	3A-92

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) GENERATOR OVERLOAD.....	3A-94
MAIN BUS 1 FAIL	3A-94
MAIN BUS 2 FAIL	3A-96

ENGINES

Engine Airstart.....	3A-99
Engine Precautionary Shutdown in Flight	3A-103
L(R) ENG START FAIL	3A-106
Engine Start Abort.....	3A-106
Engine Tailpipe Fire After Shutdown.....	3A-108
L(R) ENG CONTROL FAIL	3A-109
L(R) ENG CONTROL FAULT	3A-110
L-R ENG FUEL BYPASS	3A-111
L(R) ENG OIL TEMP HIGH.....	3A-112
L(R) ENG OVERSPD PROT FAIL.....	3A-113
L(R) FIRE DETECTOR FAIL.....	3A-114

ENVIRONMENTAL / PRESSURIZATION SYSTEMS

Cracked Windshield or Window	3A-115
CABIN ALT CTRL FAIL.....	3A-115
CABIN ALT CTRL FAULT.....	3A-116
CABIN ALT HIGH FIELD	3A-116
L(R) CABIN BLEED FAIL	3A-117
CABIN OXYGEN OFF	3A-117

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) CABIN BLEED TEMP HIGH.....	3A-118
ECS AIR COND FAIL	3A-119
ECS GND COOLING FAN FAIL	3A-119
L(R) ENG BLEED FAIL.....	3A-120
NO LDG FIELD ELEV	3A-122
OXYGEN LOW	3A-123
OXYGEN QTY FAIL.....	3A-124
OXYGEN UNAVAILABLE.....	3A-125

FLIGHT CONTROLS

FLAP FAIL.....	3A-127
FLAP FAULT.....	3A-128
PITCH TRIM FAIL	3A-129
SPEEDBRAKE EXTENDED	3A-130

FUEL SYSTEMS

FUEL CROSSFEED	3A-131
FUEL CROSSFEED FAIL	3A-131
FUEL IMBALANCE.....	3A-132
FUEL LEVEL CTRL FAULT.....	3A-133
L-R FUEL PRESSURE LOW	3A-134
L(R) FUEL PRESSURE LOW	3A-134
L(R) FUEL PUMP FAIL.....	3A-135
FUEL QTY FAULT.....	3A-136

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L-R FUEL QTY LOW.....	3A-138
L(R) FUEL QTY LOW	3A-138
FUEL TEMP HIGH	3A-139
FUEL TEMP LOW.....	3A-140

HYDRAULIC SYSTEMS

HYD PUMP FAIL.....	3A-141
HYD PRESSURE LOW	3A-141

ICE PROTECTION SYSTEMS

L(R) ENG ANTI-ICE FAIL.....	3A-143
ICE DETECT FAIL	3A-144
ICE PROT NOT ACTIVE	3A-148
L(R) PROBE HEAT FAIL.....	3A-150
STBY PROBE HEAT FAIL	3A-151
TAIL DE-ICE FAIL	3A-151
WING ANTI-ICE FAIL	3A-152
WING ANTI-ICE FAIL ON	3A-153
L(R) WING ANTI-ICE OVERHEAT.....	3A-154
L-R WING A/I TEMP LOW	3A-155
L(R) WING ANTI-ICE TEMP LOW	3A-158
WING ANTI-ICE UNAVAIL	3A-161
L(R) WSHD ZONE FAIL	3A-163
L(R) WSHD ZONE OVERHEAT	3A-164

Honda Aircraft Company

HA-420 AFM

ABNORMAL

LANDING GEAR AND BRAKES

ALTERNATE GEAR RELEASE Extension.....	3A-165
ANTI-SKID FAIL	3A-170
LDG GEAR FAIL	3A-171
NOSEWHEEL STEER FAIL	3A-172
PARK BRAKE FAIL	3A-174
PARK BRAKE ON	3A-174

OPERATIONAL

Cabin System Shedding	3A-175
Landing With Ice Accumulation on Wings	3A-176
Reduced Flap Landing	3A-180
Side Window Landing	3A-183
Single-Engine Approach and Landing	3A-184
Single-Engine Missed Approach.....	3A-187

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ABNORMAL PROCEDURES

GENERAL

The procedures contained in this section have been developed to provide a means to maintain an acceptable level of airworthiness or reduce operational risk resulting from a failure condition.

Abnormal procedures are grouped by functional system. Within each functional system area, the procedures are sorted by non-CAS procedures, CAS based, and other visual indications such as PFD flags. Each sub-group is sorted by alphabetical order.

Certain failures are capable of compromising multiple airplane systems. The pilot must respond directly to each annunciated or otherwise identified, system failure and consult the AFM for each specific Abnormal or Emergency procedure. For cases where multiple procedures specify different airplane configurations for continued safe flight and landing, the most restrictive must be used. If multiple procedures specify additional landing distance factors, each factor is additive and applied to the normal landing distance.

Procedures are for use in flight unless other conditions are specifically called out. If not specifically addressed while on the ground, conditions addressed in this section must be corrected prior to flight, unless dispatch is allowed by an approved MEL or otherwise authorized by the appropriate aviation authority.

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HA-420 AFM

ABNORMAL

TERMINOLOGY

Sound judgment, a thorough knowledge of the aircraft, its characteristics, and the flight manual procedures are essential for handling any abnormal situation. Some failures may compromise airworthiness or functionality, and therefore, the term “land at nearest suitable airport” may be used which is defined as:

Land at nearest suitable airport – the mission should be terminated, and the aircraft landed at a suitable airfield. The airfield and duration of the flight is left to the aircrew’s discretion based on their specific circumstances. Considerations should include (but are not limited to):

- Severity of the emergency
- Aircraft performance
- Field facilities
- Weather
- Ambient conditions such as lighting
- Degraded aircraft functionality

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HA-420 AFM

ABNORMAL

AVIONICS / AFCS

PRIMARY FLIGHT or MULTI-FUNCTION DISPLAY FAILURE

A Primary Flight Display or Multi-Function Display has failed.

LEFT PFD FAILED

1. DISPLAY REVERSION (pilot side) REV
--- END OF PROCEDURE ---

RIGHT PFD FAILED

1. Transfer controls to left seat pilot
--- END OF PROCEDURE ---

MFD FAILED

1. DISPLAY REVERSION (pilot flying) REV
--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

AFCS MISTRIM

Autopilot is holding constant force.

1. Flight Controls..... Hold firmly

NOTE

Expect high flight control forces when autopilot is disengaged. Axis and direction of mistrim is indicated by flag in upper left hand corner of the PFD.

2. Trim aircraft as necessary
3. Yaw Damper.....Engage as required
4. Autopilot.....Engage as required

--- END OF PROCEDURE ---

AHRS 1(2) FAIL

The associated AHRS unit has failed.

If system does not automatically revert to operable AHRS

1. PFD (affected side)..... Select operable AHRS

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

AIR DATA 1(2) FAIL

The associated air data computer is not providing proper data.

1. PFD (affected side)..... Ensure operable ADC selected
2. RVSM airspace.....Exit
3. Do not perform intentional stalls

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

PRIOR TO LANDING

4. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

L(R) AUDIO FAIL

The intercom and associated speaker, handmic, and oxygen mask mic have failed.

1. Headset (affected side) Don
2. Speaker Volume (opposite side).....Adjust

CAUTION *Pilot with failed audio will not hear intercom or alerts through headset, but alerts will be automatically routed to cross-side speaker. Speaker volume may need to be adjusted to ensure alerts can be heard while wearing headset.*

NOTE The handheld microphone, speaker and oxygen mask mic on the failed side are inoperative.

NOTE The audio system on the failed side will default to use of the onside radio only at a fixed volume.

NOTE Marker beacon functionality (aural and visual) will be inoperative on the failed side.

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

AVIONICS COMPUTER 1 FAIL

Avionics computer 1 has failed.

1. Radio COM 2
2. Navigation Select NAV 2 or GPS
3. Transponder Ensure operating transponder selected
4. Do not perform intentional stalls

NOTE

Stall Pusher is inoperative. Stall Warning remains operative based on available valid ADC AOA data.

NOTE

Failure of Avionics Computer 1 will cause associated failures which will result in the following CAS messages: **FLAP FAULT**, **NOSEWHEEL STEER FAULT**, **TAIL DEICE FAULT**. The procedures for those messages should be followed in addition to this procedure.

┌Procedure Continued ┐

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HA-420 AFM

ABNORMAL

AVIONICS COMPUTER 1 FAIL (continued)

PRIOR TO LANDING

5. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

NOTE

Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

The following equipment will be <u>INOPERATIVE</u> with Avionics Computer 1 failure	
Communication	
COM 1 tuning (defaults to 121.5) GPS 1	NAV 1 Transponder 1 (only for dual installations)
Navigation / AFCS	
AHRS 1 and 2 (degraded) TAWS (degraded) Stall Pusher	CSC Radio Altimeter TCAS
Engine	
L Fire Suppression CAS	
Flight Control	
Flaps (Degraded)	

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3A-20

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HA-420 AFM

ABNORMAL

AVIONICS COMPUTER 2 FAIL

Avionics computer 2 has failed.

1. Radio COM 1
2. Navigation NAV 1 or GPS
3. Transponder Verify XPDR 1
4. Do not perform intentional stalls

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on available valid ADC AOA data.

NOTE Failure of Avionics Computer 2 will cause associated failures which will result in the following CAS messages: **FLAP FAULT**, **NOSEWHEEL STEER FAULT**. The procedures for those messages should be followed in addition to this procedure.

┌ **Procedure Continued** ┐

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HA-420 AFM

ABNORMAL

AVIONICS COMPUTER 2 FAIL (continued)

PRIOR TO LANDING

5. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

NOTE

Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

The following equipment will be <u>INOPERATIVE</u> with an Avionics Computer 2 failure	
Communication	
COM 2 tuning (defaults to 121.5) GPS 2	NAV 2 Transponder 2
Navigation / AFCS	
AHRS 1 and 2 (degraded) ILS/BC on Standby Instrument	Stall Pusher CSC
Engine	
R Fire Suppression CAS	
Flight Control	
Flaps (Degraded)	Speedbrake Auto-retract

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3A-22

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HA-420 AFM

ABNORMAL

CPDLC FAIL

The CPDLC has failed.

1. Select other means of communication

--- END OF PROCEDURE ---

DATA ACQUISITION 1 FAIL

Data Acquisition Unit 1 has failed.

1. L FUEL PUMP.....ON
2. Icing ConditionsExit
3. ENGINE ANTI-ICE.....Both OFF, if practical
4. WING ANTI-ICE.....OFF
5. ITT..... Monitor

CAUTION

Engines will not sense operation of wing anti-ice or left engine anti-ice. This could result in an engine ITT exceedance if TO or MCT power is selected and the wing or engine anti-ice is on.

6. Land at nearest suitable airport

┌Procedure Continued┐

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 1 FAIL (continued)

BEFORE SHUTDOWN

7. Wheel Chocks..... Install

NOTE

Failure of Data Acquisition Unit 1 will cause associated failures which will result in the following CAS messages: **L ENG CONTROL FAULT**,

R ENG CONTROL FAULT,
WING ANTI-ICE FAULT,
L ENG ANTI-ICE FAULT,
L ENG BLEED FAULT,
PITCH TRIM FAULT,
L ENG FAULT,
NOSEWHEEL STEER FAULT.

The procedures for those messages should be followed in addition to this procedure.

┌Procedure Continued ┐

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 1 FAIL (continued)

The following equipment will be <u>INOPERATIVE</u> with Data Acquisition Unit 1 failure	
Engine	
L Fuel Filter Pressure Sensing L Chip Detector Sensing L Vibration Detection L and R Wing Anti-ice Sensing	L Oil Filter Pressure Sensing L Oil Level L Oil Temperature
Hydraulic	
Volume and System Pressure (Ind.) Anti-Skid Fail monitor	Emergency/Parking Brake Pressure (Ind.)
Fuel	
L Fuel Pump Automatic Activation L and R SOV Position (Ind.) L Fuel Temperature	L Fuel Pump Fail Sensing Crossfeed Valve (Ind.) L Fuel Pump (Ind.)
Environmental	
L Manifold Temp and Press (Ind.) L ENG Bleed (Ind.)	L and R CABIN INFLOW (Ind.) L Eng Bleed Leak Detection
Ice Protection	
L Eng Anti-ice Temp and Press (Ind.) L WINDSHIELD Zone (Ind.) Wing Bleed Leak Detection	L Wing Anti-ice Valve (Ind.) Wing Anti-ice High Flow

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3A-25

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 2 FAIL

Data Acquisition Unit 2 has failed.

1. R FUEL PUMP ON
2. Icing ConditionsExit
3. ENGINE ANTI-ICEBoth OFF, if practical
4. WING ANTI-ICEOFF
5. ITT Monitor

CAUTION

Engines will not sense operation of wing anti-ice or right engine anti-ice. This could result in an engine ITT exceedance if TO or MCT power is selected and the wing or engine anti-ice is on.

6. Land at nearest suitable airport

BEFORE SHUTDOWN

7. Wheel Chocks..... Install

┌ Procedure Continued ┐

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 2 FAIL (continued)

NOTE

Failure of Data Acquisition Unit 2 will cause associated failures which will result in the following CAS messages: **R ENG CONTROL FAULT**,

WING ANTI-ICE FAULT,

R ENG ANTI-ICE FAULT,

R ENG BLEED FAULT,

L CABIN BLEED FAULT,

PITCH TRIM FAULT,

R ENG FAULT, **NOSEWHEEL STEER FAULT**.

The procedures for those messages should be followed in addition to this procedure.

┌ **Procedure Continued** ┐

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 2 FAIL (continued)

The following equipment will be <u>INOPERATIVE</u> with Data Acquisition Unit 2 failure	
Engine	
R Fuel Filter Pressure Sensing R Chip Detector Sensing R Vibration Detection L and R Wing Anti-ice Sensing	R Oil Filter Pressure Sensing R Oil Level R Oil Temperature
Electrical	
R Generator Voltage or Current (Ind.)	
Hydraulic	
Anti-Skid Fail Monitor Hydraulic Pump (Ind.)	Parking/Emer Brake Position Sensing Main Accumulator Pressure (Ind.)
Fuel	
R Fuel Pump Automatic Activation L and R SOV Closed Position (Ind.)	R Fuel Pump Fail Sensing Crossfeed Valve Closed (Ind.)
Environmental	
R Manifold Temp and Press (Ind.) R Eng Bleed Leak Detection	L and R ENG Bleed (Ind.) L and R HPRSOV (Ind.)
Ice Protection	
R Eng Anti-ice Temp and Press (Ind.) R WINDSHIELD Zone (Ind.) Wing Crossflow (Ind.) Wing Anti-ice High Flow	Ice Detector 2 R Zone Temp (Ind.) R Eng Anti-ice (Ind.)

--- END OF PROCEDURE ---

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October 30, 2016

HJ1-29000-003-001

Page 3A-28

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 3 FAIL

Data Acquisition Unit 3 has failed.

1. Icing ConditionsExit
2. WING ANTI-ICE OFF
3. ITT Monitor

CAUTION

Engines will not sense operation of wing anti-ice. This could result in an engine ITT exceedance if TO or MCT power is selected and the wing or engine anti-ice is on.

4. Land at nearest suitable airport

NOTE

Failure of Data Acquisition Unit 3 will cause associated failures which will result in the following CAS messages: **L CABIN BLEED FAIL**, **R ENG CONTROL FAULT**, **WING ANTI-ICE FAULT**, **R CABIN BLEED FAULT**, **R ENG FAULT**, **NOSEWHEEL STEER FAULT**. The procedures for those messages should be followed in addition to this procedure.

┌ Procedure Continued ┐

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HA-420 AFM

ABNORMAL

DATA ACQUISITION 3 FAIL (continued)

The following equipment will be <u>INOPERATIVE</u> with Data Acquisition Unit 3 failure	
Engine	
L and R Fire Extinguisher Status	
Hydraulic	
Normal Brake Pressure (Ind.)	Brake Accumulator Pressure (Ind.)
Fuel	
L and R Fuel Pressure (Ind.)	R Fuel Temp Sensing
Fuel ISO Valve (Ind.)	L and R Fuel Pump Control (Latched ON)
Environmental	
R Cabin Bleed Temp (Ind.)	Air Conditioner Status (Ind.)
L and R Eng Bleed Leak Detection	Cockpit Duct Temp
Doors	
External Power Door (Ind.)	Electrical Service Doors (Ind.)
Ice Protection	
L and R Wing Anti-ice Valve (Ind.)	L or R Wing Anti-ice Sensing and Indication
Wing Bleed Leak Detection	

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

DATA CONCENTRATOR 1 FAIL

Data Concentrator 1 has failed.

1. Land at nearest suitable airport

CAUTION *Loss of data will suppress some subsequent failure indications including CAS alerts.*

CAUTION *Left engine fire warning and suppression will be unavailable except the light in the ENGINE FIRE PUSH Switch.*

NOTE Failure of Data Concentrator 1 will cause associated failures which will result in the following CAS messages: **FUEL QTY FAULT**, **L ENG CONTROL FAULT**, **STALL PUSHER FAIL**, **L GENERATOR FAULT**, **ENG SYNC FAIL**, **R ENG FAULT**, **L ENG FAULT**, **AIR DATA 1 FAULT**, **AFCS FAULT**, **NOSEWHEEL STEER FAULT**. The procedures for those messages should be followed in addition to this procedure.

NOTE If use of the oxygen mask is required, the OXYGEN MASK AUDIO switch must be selected to EMER to activate the oxygen mask microphone.

┌ Procedure Continued ┐

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HA-420 AFM

ABNORMAL

DATA CONCENTRATOR 1 FAIL (continued)

The following equipment will be <u>INOPERATIVE</u> with Data Concentrator Unit 1 failure	
Communication	
Pilot Scroll Wheel	FDR Data to CVFDR
Pilot O2 Mask Mic Door Switch	
Navigation / AFCS	
Stall Pusher	ADC 1 Temperature Indication
ADF	XM Weather and Music
AHRS 1 (degraded)	INMARSAT
Weather Radar	CSC
Engine	
L Engine Fire Warning Aural	L Eng Fire Warning CAS and EI flag
L and R FADEC Data Redundancy	L Engine Fire Pushbutton
L Fire Extinguisher Control	Engine Sync
Electrical	
No indication of #1 side information	
Fuel	
L and CTR Fuel Quantity	Fuel Iso Valve Position (Ind.)
L and R SOV Position (Ind.)	
Environmental	
L ENG Bleed Valve (Ind.)	
Lighting	
Lighting Control 1	Ice Inspection Light(s)
NAV Lights	L Landing Light
Beacon	R Taxi Light

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

DATA CONCENTRATOR 2 FAIL

Data Concentrator 2 has failed.

1. Land at nearest suitable airport

CAUTION *Loss of data will suppress some subsequent failure indications including CAS alerts.*

CAUTION *Right engine fire warning and suppression will be unavailable except the light in the ENGINE FIRE PUSH Switch.*

NOTE Failure of Data Concentrator 2 will cause associated failures which will result in the following CAS messages: **FUEL QTY FAULT**, **R ENG CONTROL FAULT**, **STALL PUSHER FAIL**, **R GENERATOR FAULT**, **ENG SYNC FAIL**, **R ENG FAULT**, **AIR DATA 2 FAULT**, **AFCFS FAULT**, **NOSEWHEEL STEER FAULT**. The procedures for those messages should be followed in addition to this procedure.

NOTE If use of the oxygen mask is required, the OXYGEN MASK AUDIO switch must be selected to EMER to activate the oxygen mask microphone.

┌ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

DATA CONCENTRATOR 2 FAIL (continued)

The following equipment will be <u>INOPERATIVE</u> with Data Concentrator Unit 2 failure	
Communication	
Copilot Scroll Wheel Copilot O2 Mask Mic Door Switch	SATCOM / Connex TM Weather
Navigation / AFCS	
Stall Pusher Weather Radar ADC 2 Temperature Indication	AHRS 2 (degraded) DME CSC
Engine	
R Engine Fire Warning Aural L and R FADEC Data Redundancy R Engine Fire Extinguisher Control	R Engine Fire CAS and EI flag R Engine Fire Pushbutton Engine Sync
Electrical	
No indication of #2 side information	
Fuel	
R Fuel Quantity L and R SOV Position (Ind.)	Fuel Iso Valve Position (Ind.) Crossfeed Valve Position (Ind.)
Environmental	
Cabin Fan Speed (Ind.)	
Lighting	
Lighting Control 2 L Taxi Light R Landing Light	Strobe Lights RECOG Lights Logo Lights

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

FLT DIR MODE CHANGE

The flight director function has automatically reverted to a non-armed mode.

1. Mode Control Panel..... Re-engage desired modes

NOTE During approach, loss or unavailability of the navigation source signal could cause a flight director mode change. In this case, the desired mode may be unavailable until the navigation signal is available.

--- END OF PROCEDURE ---

RUDDER BIAS FAIL

The rudder bias system has failed.

1. Maintain directional control with rudder inputs

NOTE Rudder forces with one engine inoperative will increase with Rudder Bias inoperative. Use of additional bank angle into the operable engine may be used to reduce residual forces.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

STALL PUSHER FAIL

The stall pusher has failed.

1. Do not perform intentional stalls.

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on available valid ADC AOA data.

PRIOR TO LANDING

2. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

STALL WARN MISCOMP

AOA 1 and AOA 2 do not agree within allowable limits.

1. Compare Left and Right PFD air data information
2. PFD (affected side)..... Select operable ADC
3. RVSM airspace.....Exit

--END OF PROCEDURE ---

TRANSPONDER MODE

The transponder is not in ALT mode while airborne.

1. Select appropriate transponder mode

If TRANSPONDER MODE message remains posted

2. Transponder.....Select other transponder (if installed)

If TRANSPONDER MODE message remains posted

3. RVSM airspace.....Exit

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

YAW DAMPER FAIL

The Yaw Damper system has failed.

If the autopilot is available

1. Autopilot..... Engage

NOTE

Dutch Roll damping improves at all altitudes with the autopilot engaged

If the autopilot is not available

1. Altitude..... FL 240 or lower

NOTE

Dutch Roll damping improves at lower altitudes

2. Land at nearest suitable airport

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

YAW DAMPER OFF

The Yaw Damper is operational but not selected during enroute flight operations.

1. Yaw Damper..... Engage

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

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October 30, 2016

HJ1-29000-003-001

Page 3A-40

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HA-420 AFM

ABNORMAL

PFD / EIS / ASI FLAGS

The following malfunctions are indicated by flags

Display Flag / Indication	Description
Air Data	
	An Indicated Airspeed miscompare of greater than 6 knots has been detected between the two primary air data probes.
	An Altitude miscompare of greater than 200 ft. has been detected between the two primary air data probes. The aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited.
	Low Speed Awareness is failed.
	The pilot and copilot PFDs are displaying air data from the #1 air data probe. The aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited.
	The pilot and copilot PFDs are displaying air data from the #2 air data probe. The aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited.

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HA-420 AFM

ABNORMAL

Display Flag / Indication	Description
Air Data (continued)	
BOTH ON ADC STBY	<p>The pilot and copilot PFDs are displaying air data from the standby air data probe.</p> <p>The aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited.</p>
USING ADC 1	<p>The copilot PFD is displaying air data from the #1 air data probe.</p>
USING ADC 2	<p>The pilot PFD is displaying air data from the #2 air data probe.</p>
USING ADC STBY	<p>The pilot or copilot PFD is displaying air data from the standby air data probe.</p> <p>The aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited.</p>
IAS	<p>Indicates the avionics system is not able to perform the airspeed comparator function. Airspeed data is missing or invalid from one or both primary air data probes.</p>
ALT	<p>Indicates the avionics system is not able to perform the altitude comparator function. Altitude data is missing or invalid from one or both primary air data probes.</p> <p>The aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited.</p>

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HA-420 AFM







ABNORMAL

Display Flag / Indication	Description
Attitude Heading Reference System (AHRS)	
BOTH ON AHRS 1	The pilot and copilot PFDs are displaying attitude and heading data from the #1 AHRS.
BOTH ON AHRS 2	The pilot and copilot PFDs are displaying attitude and heading data from the #2 AHRS.
BOTH ON AHRS STBY	The pilot and copilot PFDs are displaying attitude and heading data from the standby AHRS.
USING AHRS 1	The copilot PFD is displaying attitude and heading data from the #1 AHRS.
USING AHRS 2	The pilot PFD is displaying attitude and heading data from the #2 AHRS.
USING AHRS STBY	The pilot or copilot PFD is displaying attitude and heading data from the standby AHRS.
PIT	A Pitch miscompare of greater than 5° has been detected between the two primary AHRS.
ROL	A Roll miscompare of greater than 6° has been detected between the two primary AHRS.

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HA-420 AFM






ABNORMAL

Display Flag / Indication	Description
Attitude Heading Reference System (AHRS) (continued)	
	A Heading miscompare of greater than 6° has been detected between the two primary AHRS. Note: This flag may be observed on the ground in the vicinity of equipment or buildings.
	Indicates the avionics system is not able to perform the pitch comparator function. Pitch data is missing or invalid from one or both primary AHRS.
	Indicates the avionics system is not able to perform the roll comparator function. Roll data is missing or invalid from one or both primary AHRS.
	Indicates the avionics system is not able to perform the heading comparator function. Heading data is missing or invalid from one or both primary AHRS.
Global Positioning System (GPS)	
	GPS 2 is failed. The system is using GPS 1 for GPS position, vector, and time information.
	GPS 1 is failed. The system is using GPS 2 for GPS position, vector, and time information.

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HA-420 AFM

ABNORMAL

Display Flag / Indication	Description
Global Positioning System (GPS) (continued)	
	Dead Reckoning mode is active. The system is using non-navigation source information (heading, airspeed, etc.) to estimate aircraft position from last known navigation source position.
	GPS Loss of Integrity (LOI). This display flag indicates loss of GPS integrity monitoring functions or the integrity is insufficient for the current phase of flight. This flag is also displayed when both GPS units are failed. During this condition FMS course deviation data is removed from the PFD.
	The system is unable to achieve the required accuracy.
Radio Altimeter	
	The Radio Altimeter has failed.
SurfaceWatch	
	The aircraft is aligned with a runway that is not entered in the avionics, as detected by the SurfaceWatch. This is accompanied by the “Check Runway” aural alert. Refer to the Cockpit Reference Guide for details.

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-45

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

Display Flag / Indication	Description
Terrain Awareness and Warning System	
TERRAIN	Terrain Caution, accompanied by the appropriate aural and CDU pop-up alert, as determined by the TAWS. Refer to the Cockpit Reference Guide for details.
TAWS FAIL	The TAWS function has failed. All or portions of the TAWS function are not available. Refer to the Cockpit Reference Guide for details. This alert is accompanied by a “TAWS System Failure” aural alert.
GPWS FAIL	The GPWS function has failed. All or portions of the GPWS function are not available. Refer to the Cockpit Reference Guide for details. This alert is accompanied by a “GPWS System Failure” aural alert. This alert is only displayed in TAWS-A configurations.
TAWS N/A	Portions of the TAWS function are not available, depending on the failure cause. Refer to the Cockpit Reference Guide for details.
GLIDESLOPE	An excessive glideslope deviation has been detected by the TAWS. This alert is only displayed in TAWS-A configurations.
GLIDEPATH	An excessive glide path deviation has been detected by the TAWS. This alert is only displayed in TAWS-A configurations.

FAA APPROVED
October 30, 2016

HJ1-29000-003-001






Page 3A-46

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM


ABNORMAL

Display Flag / Indication	Description
Traffic Collision and Avoidance System (TCAS) (continued)	
	Traffic Advisory (TA) is active. A potentially hazardous intruding aircraft has been detected and is closing to within 20-48 seconds of a potential collision area. In TCAS I installations, this visual alert is accompanied by an aural alert indicating the presence of traffic and relative bearing/altitude (“Traffic! 11 o’clock, high, six miles!”). In TCAS II installations, this visual alert is accompanied by a “Traffic! Traffic!” aural alert.
	The TCAS has failed.
	The TCAS is in Standby mode during flight.
	TA mode has been selected by the flight crew or automatically engaged during takeoff or landing. During this mode, Resolution Advisories are inhibited. This selection is only available in TCAS II installations.
	Standby mode has been selected by the flight crew while the aircraft is on ground. In flight the annunciation is amber. During this mode, all TCAS alerting functions are unavailable.

Honda Aircraft Company

HA-420 AFM

ABNORMAL

Display Flag / Indication	Description
Windshear	
	Windshear Caution indicating increasing performance windshear. This alert is accompanied by a repeating “Caution Windshear” aural alert.

Honda Aircraft Company

HA-420 AFM

ABNORMAL

DOORS

AFT BAG DOOR UNSAFE

The aft baggage door is not closed and latched with an engine running and the parking brake not set, on ground or in flight.

ON GROUND

1. Aft Baggage Door Close

--- END OF PROCEDURE ---

DURING FLIGHT

1. Airspeed..... Reduce
2. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

CABIN DOOR UNSAFE

The main cabin door is not closed and latched with an engine running and the parking brake not set, on ground or in flight.

ON GROUND

1. Main Cabin Door Close

--- END OF PROCEDURE ---

DURING FLIGHT

1. Seatbelts Verify fastened
2. CABIN SIGNS
 - a. SEATBELTS ON
3. Airspeed Reduce
4. Descend to 10,000 ft MSL or Minimum Safe Altitude, whichever is higher
5. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

EMER EXIT DOOR UNSAFE

The Emergency Exit door is not closed and latched, on ground or in flight.

ON GROUND

1. Emergency Exit Door Verify properly installed
--- END OF PROCEDURE ---

DURING FLIGHT

1. Emergency Exit Door Handle Verify closed

If EMER EXIT DOOR UNSAFE message remains

2. Continue planned flight

NOTE

The emergency exit door is a plug type door.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

EXT PWR DOOR UNSAFE

The external power door is not closed with an engine running and the parking brake not set.

ON GROUND

1. Do not taxi
2. External power cord Verify removed
3. External power door Close

--- END OF PROCEDURE ---

DURING FLIGHT

1. Continue planned flight

NOTE

The external power door will remain closed due to airflow during flight.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FWD BAG DOOR UNSAFE

The forward baggage door is not closed and latched with an engine running and the parking brake not set, on ground or in flight.

ON GROUND

1. Fwd Baggage Door..... Close

--- END OF PROCEDURE ---

DURING FLIGHT

1. Airspeed..... Reduce
2. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-54

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ELECTRICAL SYSTEMS

BATTERY BUS 1 FAIL

Battery bus 1 is isolated from the aircraft electrical system

1. PITCH TRIM MODE..... STBY
2. Thrust Levers.....Avoid rapid movements
3. Icing ConditionsExit
4. Left DISPLAY REVERSION REV
5. Radio COM 2

CAUTION

Pilot will not hear intercom or alerts through headset, but alerts will be automatically routed to cross-side speaker. Speaker volume may need to be adjusted to ensure alerts can be heard while wearing headset.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

NOTE

The pilot's handheld microphone and speaker are inoperative.

NOTE

The pilot's boom mic, or oxygen mask mic is automatically routed to COM 2 for communication with ATC.

NOTE

Marker beacon functionality (aural and visual) will be inoperative on the failed side.

6. Navigation NAV 2 or GPS

NOTE

COM 2, NAV 2, and GPS are the only communication and navigation sources available during a Battery Bus 1 failure.

7. Transponder XPDR 2 (if applicable) or notify ATC

8. L ENGINE BLEED OFF

9. WING FLOW FROM R

10. L WINDSHIELD HEAT OFF

11. Altitude Descend to FL 250 or below

12. NOSE WHEEL STEERING OFF

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

13. Select a long, wide, and dry runway with minimal crosswind

WARNING Anti-skid is not functional. Any braking above light in wet runway conditions could result in blown tires and loss of directional control.

NOTE Landing distance will double on a wet runway using light braking.

CAUTION *Nosewheel steering is inoperative. Crosswinds should be minimized to ensure adequate directional control during the low-speed portion of the rollout using differential braking.*

14. Land at nearest suitable airport

CAUTION *Loss of data will suppress some subsequent failure indications including CAS alerts.*

NOTE Stall Pusher is inoperative. Low Speed Awareness remains operative in a degraded mode based on available valid ADC AOA data.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

DESCENT

1. Landing Field Elevation Verify set
2. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. Avionics..... Set

┌Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

4. Landing Data Set and confirmed
- a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance.....Confirm

Flap Setting	Min Approach Speed	Landing Distance
UP	$V_{REF} + 20$	Add 105%
TO/APPR	$V_{REF} + 10$	Add 75%
LDG	$V_{REF} + 5$	Add 60%

NOTE Flaps are inoperative during a Battery Bus 1 failure. The flap setting at the time the failure occurred will be the final flap setting.

NOTE Due to anti-skid and the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

NOTE Landing distance will double on a wet runway using light braking.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

- 5. CAS Messages..... Check
- 6. Approach Briefing..... Complete

BEFORE LANDING

- 1. Landing Gear Alternate Extension
 - a. LDG GEAR CTRL circuit breaker..... PULL (B7)
 - b. LANDING GEAR..... DN
 - c. Airspeed..... 150 KIAS minimum
 - d. ALTERNATE GEAR RELEASE handle..... Pull fully
 - e. Yaw airplane, if necessary, to obtain gear locked down
 - f. LANDING GEAR indicator.... Verify three green DN

NOTE

The gear DOOR icon will remain posted but a normal landing is possible without gear door - ground contact.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

g. ALTERNATE GEAR RELEASE handle Stow

CAUTION *The ALTERNATE GEAR RELEASE handle could interfere with thrust lever operation if not stowed following use.*

NOTE Normal landing gear operations cannot be restored in flight after activation of the alternate gear release.

2. Airspeed.....(Flaps UP) - $V_{REF} + 20$
(Flaps TO/APPR) - $V_{REF} + 10$
(Flaps LDG) - $V_{REF} + 5$
3. Yaw Damper..... Disengage

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1 FAIL (continued)

LANDING

1. Thrust Levers..... IDLE
2. Brakes..... Apply Gradually
3. Maintain directional control with rudder and differential braking

CAUTION

Brakes must be applied gradually. Light to moderate braking can be applied without skidding tires on a dry surface, however, the pilot should consider runway surface conditions when applying brakes.

FOLLOWING LANDING ROLLOUT

1. Do not taxi
2. Wheel chocks Install

CAUTION

Following landing rollout, the emergency and brake accumulators may drop below the level required for braking.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

The following equipment will be <u>INOOPERATIVE</u> with a Battery Bus 1 failure	
Communication	
COM 1 Transponder 1	L Audio
Navigation / AFCS	
Autopilot AHRS 1 and ADC 1 Standby Instrument Standby AHRS and ADC ADF TCAS Radio Altimeter	L PFD SiriusXM [®] Weather Datalink NAV 1 L CDU Weather Radar SATCOM / Connex [™] Weather GPWS
Engine	
L Engine Instruments L Fire Extinguisher	L Fire Detection
Flight Control	
Normal Pitch Trim Roll and Yaw Trim Stall Pusher	Stall Warning Flaps Speedbrake
Landing Gear and Brakes	
Landing Gear Control Hydraulic Pump	Anti-Skid Hydraulic Pump
Fuel	
L Fuel Pump Left Fuel Quantity Center Fuel Quantity	L Fuel Pump Left Fuel Quantity Center Fuel Quantity
Environmental	
L ENG Bleed (Ind.) Cockpit Fan	L Eng Bleed Leak Detection
Ice Protection	
Ice Detector 1 Standby Probe Heat L Probe Heat	L Engine Anti-ice (failed on) L Wing Anti-ice L Windshield Zone Heat
Lighting	
Cockpit Flood Light L Landing Light Ice Inspection Light(s) Strobe Light Footwell Lights	Cabin Signs Beacon Position Lights R Taxi Light

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-63

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1A FAIL

Battery bus 1A has failed or is isolated from the aircraft electrical system.

1. Solid State Relay (SSR)..... Reset if necessary

If BATTERY BUS 1A FAIL message remains

2. Thrust Levers.....Avoid rapid movements

CAUTION When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

3. Icing ConditionsExit
4. Left DISPLAY REVERSION REV
5. Radio COM 2

CAUTION Pilot will not hear intercom or alerts through headset, but alerts will be automatically routed to cross-side speaker. Speaker volume may need to be adjusted to ensure alerts can be heard while wearing headset.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1A FAIL (continued)

- NOTE** The pilot's handheld microphone and speaker are inoperative.
- NOTE** The pilot's boom mic, or oxygen mask mic is automatically routed to COM 2 for communication with ATC.
- NOTE** Marker beacon functionality (aural and visual) will be inoperative on the failed side.

6. NavigationNAV 2 or GPS

- NOTE** COM 2, NAV 2, and GPS are the only communication and navigation sources available during a Battery Bus 1A failure.

7. Transponder..... XPDR 2 (if applicable) or notify ATC
8. L ENGINE BLEED.....OFF
9. WING FLOW.....FROM R
10. Altitude.....Descend to FL 250 or below
11. Land at nearest suitable airport

- CAUTION** *Loss of data will suppress some subsequent failure indications including CAS alerts.*

┌ **Procedure Continued** ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-65

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1A FAIL (continued)

DESCENT

1. Landing Field Elevation Verify set
2. ENGINE ANTI-ICE As required
3. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts Adjusted and secure
2. Passenger Briefing Complete
3. CABIN SIGNS As required
4. Avionics Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance Set,
programmed,
and modes selected
 - c. Landing Distance Confirm

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1A FAIL (continued)

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

6. CAS Messages..... Check
7. Approach Briefing Complete
8. FLAPS TO/APPR

BEFORE LANDING

1. Landing Gear Alternate Extension
 - a. LDG GEAR CTRL circuit breaker..... PULL (B7)
 - b. LANDING GEAR DN
 - c. Airspeed 150 KIAS minimum
 - d. ALTERNATE GEAR RELEASE handle Pull fully
 - e. Yaw airplane, if necessary, to obtain gear locked down
 - f. LANDING GEAR indicator.... Verify three green DN

NOTE The gear DOOR icon will remain posted but a normal landing is possible without gear door - ground contact.

┌ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1A FAIL (continued)

- g. ALTERNATE GEAR RELEASE handle..... Stow

CAUTION

The ALTERNATE GEAR RELEASE handle could interfere with thrust lever operation if not stowed following use.

NOTE

Normal landing gear operations cannot be restored in flight after activation of the alternate gear release.

2. FLAPS LDG
3. Airspeed..... $V_{REF} + 5$
4. Yaw Damper..... Disengage

LANDING

1. Thrust Levers..... IDLE
2. Brakes..... Apply
3. SPEEDBRAKE (if installed)..... EXT

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1A FAIL (continued)

<p>The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 1A failure</p>	
Communication	
COM 1 L Audio	Transponder 1
Navigation / AFCS	
L PFD Stall Pusher Standby Instrument Standby AHRS and ADC SiriusXM® Weather Datalink TCAS Radio Altimeter	L CDU ADF NAV 1 Control Yoke System Control and Scroll Wheel Weather Radar GPWS
Engine	
L Fire Detection	
Landing Gear and Brakes	
Landing Gear Control	
Fuel	
Left Fuel Quantity R Fuel Pump Control (latches ON)	L Fuel Shutoff Valve
Environmental	
L Bleed Leak Detection	L Engine Bleed
Ice Protection	
L Wing Anti-ice	
Lighting	
Cockpit Flood Light L Landing Light NAV Lights Right Taxi Light	Lighting Control 1 Ice Inspection Light(s) Ground Beacon

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-69

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1B FAIL

Battery bus 1B has failed or is isolated from the aircraft electrical system.

1. Solid State Relay (SSR)..... Reset if necessary

If BATTERY BUS 1B FAIL message remains

2. PITCH TRIM MODE..... STBY
3. Thrust Levers..... Avoid rapid movements

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE

The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

4. WING FLOW FROM R
5. Icing Conditions Exit
6. Left PFD Verify AHRS 2 selected
7. RVSM airspace..... Exit
8. NOSE WHEEL STEERING OFF

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1B FAIL (continued)

9. Select a long, wide, and dry runway with minimal crosswind

WARNING Anti-skid function is not available. Any braking above light in wet runway conditions could result in blown tires and loss of directional control.

CAUTION *Nosewheel steering is inoperative. Crosswinds should be minimized to ensure adequate directional control during the low-speed portion of the rollout using differential braking.*

NOTE Landing distance will double on a wet runway using light braking.

10. Land at nearest suitable airport

NOTE Stall Pusher is inoperative. Low Speed Awareness remains operative in a degraded mode based on available valid ADC AOA data.

DESCENT

1. Landing Field Elevation Verify set
2. Altimeters (transition altitude) Set

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1B FAIL (continued)

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. CABIN SIGNS As required
4. Avionics..... Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance..... Confirm

Flap Setting	Min Approach Speed	Landing Distance
UP	$V_{REF} + 20$	Add 105%
TO/APPR	$V_{REF} + 10$	Add 75%
LDG	$V_{REF} + 5$	Add 60%

NOTE

Flaps are inoperative during a Battery Bus 1B failure. The flap setting at the time the failure occurred will be the final flap setting.

Procedure Continued

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-72

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1B FAIL (continued)

NOTE Due to anti-skid and the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

NOTE Landing distance will double on a wet runway using light braking.

6. CAS Messages..... Check
7. Approach Briefing..... Complete

BEFORE LANDING

1. Landing Gear.....DN
2. Airspeed.....(Flaps UP) - $V_{REF} + 20$
(Flaps TO/APPR) - $V_{REF} + 10$
(Flaps LDG) - $V_{REF} + 5$
3. Yaw Damper..... Disengage

LANDING

1. Thrust Levers..... IDLE
2. Brakes..... Apply Gradually

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1B FAIL (continued)

3. Use rudder and differential braking to maintain direction control

CAUTION *Brakes must be applied gradually. Light to moderate braking can be applied without skidding tires on a dry surface, however, the pilot should consider runway surface conditions when applying brakes.*

4. SPEEDBRAKE (if installed)..... EXT

FOLLOWING LANDING ROLLOUT

1. Taxi to parking using differential braking

The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 1B failure	
Navigation / AFCS	
AHRS 1	AFCS Panel
Stall Pusher	Yaw Damper (once disengaged)
TCAS RA Mode	Flight Director
Engine	
L Fire Extinguisher	
Flight Control	
Pitch (normal), Roll and Yaw Trim	Flaps
Landing Gear and Brakes	
Anti-Skid	Nosewheel Steering
Ice Protection	
Ice Detector 1	L Engine Anti-ice (failed on)
Standby Probe Heat	
Lighting	
L Landing Light	

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-74

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1C FAIL

Battery bus 1C has failed or is isolated from the aircraft electrical system.

1. Solid State Relay (SSR) Reset if necessary

If BATTERY BUS 1C FAIL message remains

2. RVSM airspace.....Exit
3. Do not perform intentional stalls

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

4. L PFDVerify ADC 2 selected

PRIOR TO LANDING

5. Final approach speed..... $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 1C FAIL (continued)

NOTE

Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 1C failure	
Navigation / AFCS	
ADC 1	SiriusXM [®] Weather Datalink
Stall Pusher	
Environmental	
Cockpit Fan	
Lighting	
Beacon	

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2 FAIL

Battery bus 2 is isolated from the aircraft electrical system.

1. Thrust Levers.....Avoid rapid movements

CAUTION When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

2. Icing ConditionsExit
3. Radio COM 1

NOTE Copilot will lose all communication capability.

4. NavigationNAV 1 or GPS

NOTE COM 1, NAV 1, and GPS are the only communication and navigation sources available following a Battery Bus 2 failure.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2 FAIL (continued)

5. PRESSURIZATION CONTROL MODE HOLD
6. R ENGINE BLEED OFF
7. WING FLOW FROM L
8. Altitude Descend to FL 250 or below
9. Land at nearest suitable airport

CAUTION *Loss of data will suppress some subsequent failure indications including CAS alerts.*

DESCENT

1. Landing Field Elevation Verify set
2. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts Adjusted and secure
2. Passenger Briefing Complete
3. CABIN SIGNS As required
4. Avionics Set

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2 FAIL (continued)

5. Landing Data Set and confirmed
- a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance.....Confirm

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

6. CAS Messages..... Check
7. Approach Briefing Complete
8. FLAPS TO/APPR

┐ Procedure Continued ┘

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2 FAIL (continued)

BEFORE LANDING

1. LANDING GEAR..... DN

NOTE Hydraulic pressure indications will be unavailable with a loss of Battery Bus 2, but the hydraulic pump will be operable.

2. Cabin Differential Pressure Verify less than 0.3 psi
3. CABIN DUMP DUMP (if required)

WARNING The airplane must be unpressurized prior to landing.

4. FLAPS LDG
5. Airspeed..... $V_{REF} + 5$
6. Yaw Damper..... Disengage

LANDING

1. Thrust Levers..... IDLE
2. Brakes.....Apply (after touchdown)

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2 FAIL (continued)

The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 2 failure	
Communication	
CPDLC R Audio	COM 2
Navigation / AFCS	
MFD NAV 2 DME	Stall Pusher Weather Radar
Engine	
Right Engine Instruments R Fire Extinguisher	R Fire Detection Vibration Detection
Flight Control	
Standby Pitch Trim	Speedbrake
Landing Gear and Brakes	
Hydraulic (indication)	
Fuel	
R Fuel Low R Fuel Quantity R Fuel Pump	Fuel Crossfeed R Fuel Shutoff Valve
Environmental	
Air Conditioner R Engine Bleed Pressure Control (normal)	R Bleed Leak Cabin Oxygen Automatic Deployment Cabin Fan
Ice Protection	
R Wing Anti-ice	R Engine Anti-ice (failed on)
Lighting	
Ice Inspection Light Lighting Control 2 RECOG Lights Logo Lights	Strobe Lights L Taxi Light Right Landing Lights

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-81

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2A FAIL

Battery bus 2A has failed or is isolated from the aircraft electrical system.

1. Solid State Relay (SSR)..... Reset if necessary

If BATTERY BUS 2A FAIL message remains

2. Thrust Levers.....Avoid rapid movements

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE

The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

3. Icing ConditionsExit
4. Radio COM 1
5. NavigationNAV 1 or GPS

NOTE

COM 1, NAV 1, and GPS are the only communication and navigation sources available following a Battery Bus 2A failure.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2A FAIL (continued)

6. Transponder..... Verify XPDR 1
7. R ENGINE BLEED..... OFF
8. WING FLOWFROM L
9. Altitude.....Descend to FL 250 or below
10. Land at nearest suitable airport

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

DESCENT

1. Landing Field Elevation Verify set
2. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. CABIN SIGNS As required
4. Avionics..... Set

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2A FAIL (continued)

5. Landing Data Set and confirmed
- a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance.....Confirm

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

6. CAS Messages..... Check
7. Approach Briefing.....Complete
8. FLAPSTO/APPR

BEFORE LANDING

- 1. LANDING GEAR..... DN
- 2. FLAPS LDG
- 3. Airspeed..... $V_{REF} + 5$
- 4. Yaw Damper..... Disengage

┌ Procedure Continued ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-84

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HA-420 AFM

ABNORMAL

BATTERY BUS 2A FAIL (continued)

LANDING

1. Thrust Levers..... IDLE
2. Brakes.....Apply (after touchdown)
3. SPEEDBRAKE (if installed).....EXT

The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 2A failure	
Navigation / AFCS	
Stall Pusher NAV 2 Transponder 2	COM 2 GPS 2
Engine	
Right Engine Instruments	R Fire Detection
Fuel	
R Fuel Low R Fuel Quantity	Fuel Crossfeed
Environmental	
R Engine Bleed	R Bleed Leak
Ice Protection	
R Wing Anti-ice	

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2B FAIL

Battery bus 2B has failed or is isolated from the aircraft electrical system.

1. Solid State Relay (SSR)..... Reset if necessary

If BATTERY BUS 2B FAIL message remains

2. Thrust Levers.....Avoid rapid movements

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE

The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

3. Icing ConditionsExit
4. WING FLOW FROM L
5. Left DISPLAY REVERSION REV
6. Headset (right side)..... Don

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2B FAIL (continued)

7. Speaker Volume (left side).....Adjust

CAUTION *Copilot will not hear intercom or alerts through headset, but alerts will be automatically routed to cross-side speaker. Speaker volume may need to be adjusted to ensure alerts can be heard while wearing headset.*

NOTE The handheld microphone, speaker and oxygen mask mic on the right side are inoperative.

NOTE The audio system on the right side will default to use of the onside radio only at a fixed volume

8. Transponder..... Verify XPDR 1

9. Land at nearest suitable airport

CAUTION *Loss of data will suppress some subsequent failure indications including CAS alerts.*

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

┌ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2B FAIL (continued)

DESCENT

1. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. CABIN SIGNS As required
4. Avionics..... Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance..... Confirm

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

NOTE

Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

Procedure Continued

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-88

FOR TRAINING PURPOSES ONLY

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HA-420 AFM

ABNORMAL

BATTERY BUS 2B FAIL (continued)

6. CAS Messages..... Check
7. Approach Briefing Complete
8. FLAPS TO/APPR

BEFORE LANDING

1. LANDING GEAR DN

NOTE Hydraulic pressure indications will be unavailable with a failure of Battery Bus 2B, but the hydraulic pump will be operable.

2. FLAPS LDG or TO/APPR (ice)

CAUTION *Do not extend the flaps to LDG unless the airframe can be confirmed free of ice and icing conditions are not expected during approach and landing.*

NOTE If TAWS-A is installed, the TAWS Warnings will annunciate when landing with Flaps at TO/APPR, unless Flap Override is selected.

3. Airspeed..... $V_{REF} + 5$
4. Yaw Damper..... Disengage

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

BATTERY BUS 2B FAIL (continued)

LANDING

1. Thrust Levers..... IDLE
2. Brakes.....Apply (after touchdown)

The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 2B failure	
Communication	
CPDLC	R Audio Functions
Navigation / AFCS	
MFD Weather Radar Stall Pusher Transponder 2	DME TCAS RA Mode GPS 2
Engine	
R Fire Extinguisher	
Flight Control	
Speedbrake	STBY Pitch Trim
Landing Gear and Brakes	
Hydraulic Press (indication)	
Environmental	
Cabin Oxygen Automatic Deployment	
Ice Protection	
R Engine Anti-ice (failed on)	
Lighting	
Ice Inspection Light Lighting Control 2 RECOG Light Logo Lights	Strobe Lights L Taxi Light R Landing Light

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-90

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HA-420 AFM

ABNORMAL

BATTERY BUS 2C FAIL

Battery bus 2C has failed or is isolated from the aircraft electrical system.

1. Solid State Relay (SSR) Reset if necessary

If BATTERY BUS 2C FAIL message remains

2. PRESSURIZATION CONTROL MODE HOLD

PRIOR TO LANDING

3. Cabin Differential Pressure Verify less than 0.3 psi
4. CABIN DUMP DUMP (if required)

WARNING The airplane must be unpressurized prior to landing.

The following equipment will be <u>INOPERATIVE</u> with a Battery Bus 2C failure	
Communication	
SATCOM / Connex TM Weather	TCAS RA Mode
Environmental	
Pressure Control (normal)	Cabin Fan

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-91

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HA-420 AFM

ABNORMAL

BUS TIE FAIL

Bus Tie has failed to achieve required state

ON THE GROUND

1. Do not dispatch

--- END OF PROCEDURE ---

L(R) GENERATOR FAIL

The generator has failed

1. GENERATOR (affected side)..... OFF, then NORM

NOTE

During ground operations, N_2 may need to be increased above 55% on the associated engine prior to cycling the generator switch to reset the generator.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) GENERATOR FAIL (continued)

If GENERATOR FAIL message remains posted

2. BUS TIE Verify Normal
3. GENERATOR (affected side) OFF
4. Icing Conditions Exit

NOTE The left zone heating of both windshields is inoperative during single generator operations.

NOTE The air conditioning compressor is inoperative during single generator operations in the air.

5. Generator amps Monitor

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) GENERATOR OVERLOAD

Generator load is above allowable limits

1. Generator load Reduce

NOTE

The GCU will automatically trip offline due to an overcurrent. The trip point varies based on the amount of overcurrent and duration. A GCU trip is imminent anytime the **GENERATOR OVERLOAD** is posted.

--- END OF PROCEDURE ---

MAIN BUS 1 FAIL

Main Bus 1 has failed or is isolated from the airplane electrical system

1. Icing ConditionsExit
2. L WINDSHIELD HEATOFF
3. Do not perform intentional stalls

NOTE

Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

MAIN BUS 1 FAIL (continued)

4. L PFD Ensure ADC 2 selected
5. RVSM airspace.....Exit
6. Land at nearest suitable airport

PRIOR TO LANDING

7. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

The following equipment will be <u>INOPERATIVE</u> with a Main Bus 1 failure	
Navigation / AFCS	
Stall Pusher TCAS	Weather Radar Autopilot
Ice Protection	
L Windshield Heat Zone	L Probe Heat

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-95

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HA-420 AFM

ABNORMAL

MAIN BUS 2 FAIL

Main Bus 2 has failed or is isolated from the airplane electrical system

1. Icing ConditionsExit
2. Radio COM 1
3. R WINDSHIELD HEAT.....OFF
4. Transponder..... Verify XPDR 1
5. Do not perform intentional stalls

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

6. RVSM airspace.....Exit
7. Land at nearest suitable airport

PRIOR TO LANDING

8. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

MAIN BUS 2 FAIL (continued)

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

The following equipment will be <u>INOPERATIVE</u> with a Main Bus 2 failure	
Communication	
COM 2	
Navigation / AFCS	
R PFD Stall Pusher CVFDR DME TCAS II	R CDU AHRS 2 and ADC 2 ADF Transponder 2 Radio Altimeter GPWS
Ice Protection	
R Windshield Heat Zone Ice Detector 2	Right Probe Heat
Lighting	
Strobe Lights Cockpit Map Lights Copilot Footwell Light Lighting Control 1	Taxi Lights R Landing Light RECOG Lights NAV Lights

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

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October 30, 2016

HJ1-29000-003-001

Page 3A-98

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HA-420 AFM

ABNORMAL

ENGINES

ENGINE AIRSTART

An airborne restart of a shutdown engine is required

CAUTION

Do not attempt to restart an engine that has been shut down due to an engine fire or indicates obvious mechanical failures.

CAUTION

Ensure positive core rotation before attempting a relight. Positive N_2 or oil pressure >1 psi can be used to determine core rotation. If there is no indication of positive N_2 or oil pressure, the starter may be engaged as long as N_2 rotation is verified prior to moving the Thrust Lever out of CUT OFF.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENGINE AIRSTART (continued)

PRE-START

1. Thrust Lever (affected engine) CUT OFF
2. ENGINE ANTI-ICE (affected engine) OFF
3. Airspeed and Altitude Check within limits

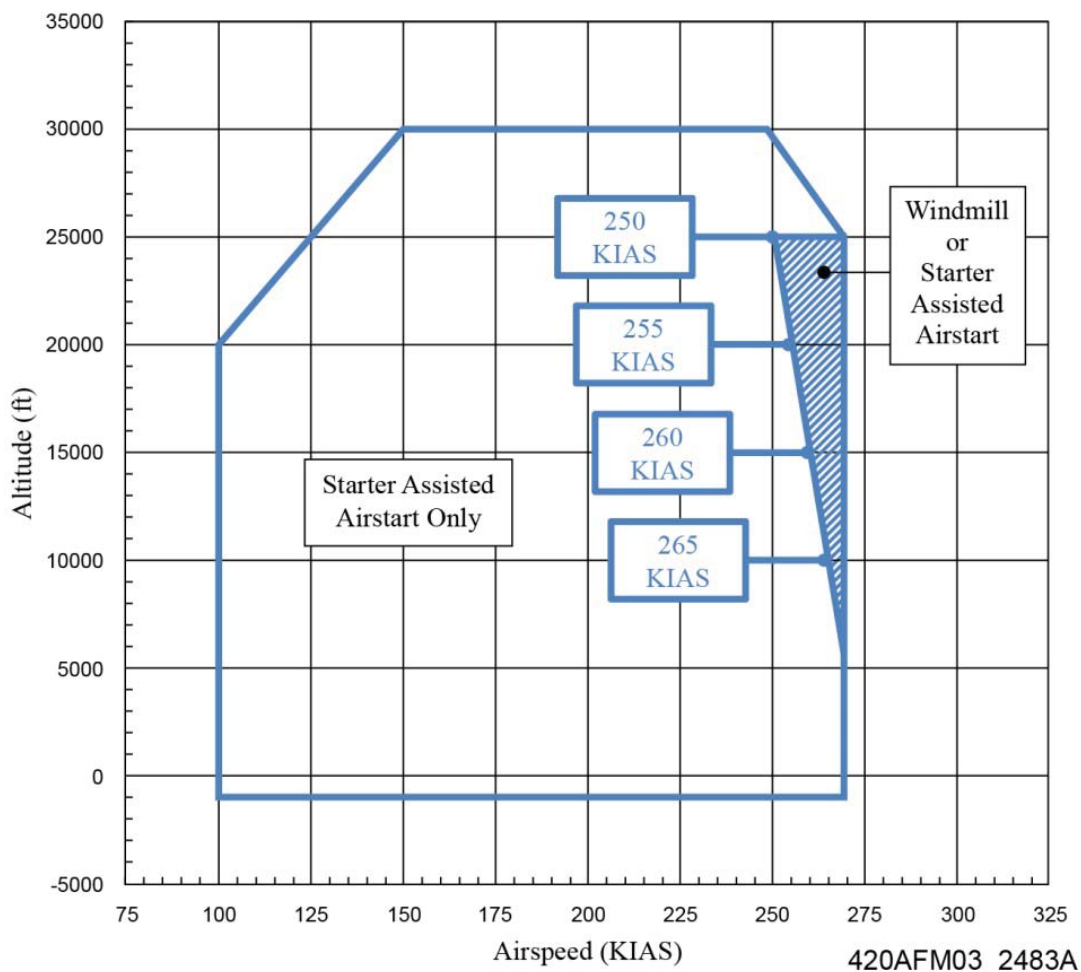


Figure 12. Airstart Envelope

┌ Procedure Continued ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-100

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HA-420 AFM

ABNORMAL

ENGINE AIRSTART (continued)

STARTER ASSISTED AIRSTART

1. Engine START Switch (affected engine).....Press
2. Thrust Lever (affected engine) IDLE
3. Engine Instruments..... Monitor

NOTE

The FADEC will not automatically abort an in-flight engine start. The pilot should abort the start if any of the following indications are observed:

- No ITT rise in 30 seconds
- No positive oil pressure indication within 10 seconds after N₂ increase
- Failure of N₁ or N₂ to accelerate to idle
- ITT approaching limit

Reference ENGINE START ABORT
(Section 3A – Abnormal Procedures) as appropriate.

WINDMILLING AIRSTART

1. FUEL PUMP (affected engine).....ON
2. Thrust Lever (affected engine)IDLE at 8% N₂ minimum
3. Engine Instruments..... Monitor

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENGINE AIRSTART (continued)

NOTE The FADEC will not automatically abort an in-flight engine start. The pilot should abort the start if any of the following indications are observed:

- No ITT rise in 30 seconds
- No positive oil pressure indication within 10 seconds after N_2 increase
- Failure of N_1 or N_2 to accelerate to idle
- ITT approaching limit

Reference ENGINE START ABORT (Section 3A – Abnormal Procedures) as appropriate.

POST START

1. Thrust Lever (affected engine)As required

NOTE If conditions permit, allow the engine to warm up at IDLE until the oil temperature is above 10 °C prior to selecting a higher thrust setting.

2. FUEL PUMP (affected engine) NORM
3. ENGINE ANTI-ICE (affected engine)As required
4. WING FLOW NORM
5. TCASAUTO (TCAS II only)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT

An inflight shutdown of an engine is required

CAUTION

Severe damage to the engine, and possibly the aircraft, may occur if engine operation is continued with a serious engine malfunction. A serious malfunction can be recognized by one or more of the following symptoms:

- *Increase in engine vibration accompanied by higher than normal ITT or fuel flow*
- *Repeated or uncontrollable engine stalls*
- *Loss of thrust*
- *A shift in engine-to-engine parameters, or in the relationship of one parameter to another during steady-state operation*
- *Oil pressure increase or decrease of ± 8 psid or more from the normal steady-state operating pressure, and/or an increase in oil temperature, or indications of oil filter bypass.*

NOTE

If conditions permit with no abnormal engine indications, stabilize the engine at idle thrust for 2 minutes prior to shut down.

┌ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT

(continued)

1. Thrust Lever (affected engine) CUT OFF

CAUTION

To minimize the risk of rotor lock:

- *Maintain airspeed above 140 KIAS to ensure core rotation.*
- *Following commanded or uncommanded inflight shutdown, maintain positive core rotation throughout the engine-out scenario. If core rotation has stopped, take action to achieve core rotation as soon as practical. Failure to maintain positive core rotation may preclude a successful start.*

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE

The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

2. Fuel CROSSFEEDAs required
3. WING FLOW FROM L(R) (operable side)
4. TCAS (TCAS II only) TA ONLY
5. ENGINE ANTI-ICE (affected engine).....OFF

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT

(continued)

6. Altitude.....FL 250 or below

NOTE The left zone heating of both windshields is inoperative during single generator operations.

7. Accomplish ENGINE AIRSTART (Section 3A – Abnormal Procedures) or SINGLE-ENGINE APPROACH AND LANDING (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG START FAIL

ENGINE START ABORT

The FADEC has detected a starting abnormality or has automatically aborted a ground engine start, or the pilot has detected a reason to abort the start attempt.

GROUND OR STARTER ASSISTED AIRSTART

1. Thrust Lever (affected engine) CUT OFF

NOTE

For a pilot initiated start abort on the ground, the START icon and start ITT limits will remain displayed for 60 seconds after the thrust lever is selected to CUT OFF.

If ITT rise observed or ITT is above 120 °C

2. Engine START Switch (affected engine) Press

CAUTION

If the ENGINE START ABORT is performed after the starter is disengaged, allow N_2 to decrease below 45% before energizing the starter for motoring or a subsequent start attempt.

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG START FAIL (continued)

ENGINE START ABORT (continued)

3. Engine Indications..... Verify ITT decrease
4. Engine PUSH TO DISCPress
5. Accomplish SHUTDOWN (Section 4 – Normal Procedures)
or ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT
(Section 3A – Abnormal Procedures) as appropriate

--- END OF PROCEDURE ---

WINDMILLING AIRSTART

1. Thrust Lever (affected engine) CUT OFF
2. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN
FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENGINE TAILPIPE FIRE AFTER SHUTDOWN

Engine ITT does not decrease after shutdown indicating a tailpipe fire

1. Thrust Lever (affected engine) Verify CUT OFF

If ITT rise observed

2. Engine START Switch (affected engine) Press
3. Engine Indications Verify ITT decrease
4. Engine PUSH TO DISC Press (after 30 seconds)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG CONTROL FAIL

The FADEC has detected a loss of thrust control

ON GROUND

1. Thrust Lever (affected engine) CUT OFF
2. ENGINE FIRE PUSH Switch (affected engine) LIFT
COVER
AND PUSH

--- END OF PROCEDURE ---

IN FLIGHT

1. Thrust Lever (responsive engine) As required
2. Fuel CROSSFEED As required

WHEN AIRCRAFT IS IN A POSITION TO LAND

3. Thrust Lever (affected engine) CUT OFF
4. ENGINE FIRE PUSH Switch (affected engine) LIFT
COVER
AND PUSH
5. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN
FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG CONTROL FAULT

The FADEC has detected a fault that may result in degraded engine response.

ON GROUND

1. Do not dispatch

--- END OF PROCEDURE ---

IN FLIGHT

1. WING FLOW FROM L(R) (operable side)

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE

The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active on a single bleed source until WING FLOW is selected from the operable side.

If ENG CONTROL FAULT is still illuminated

2. Thrust Lever (affected engine) Avoid rapid movements
3. Engine Indications (affected engine) Monitor
4. Land at nearest suitable airport

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-110

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HA-420 AFM

ABNORMAL

L-R ENG FUEL BYPASS

An impending bypass of the fuel filters on both engines has been detected.

1. Thrust Levers.....Avoid rapid movements
2. Land at nearest suitable airport

NOTE

The **ENG FUEL BYPASS** message will post for an impending bypass condition. The filter has been designed to allow for continued engine operation during the diversion to a suitable landing airport even with a maximum expected level of fuel contamination.

NOTE

High differential pressure across the fuel filter is most likely to occur at high power settings such as takeoff, and the message may self clear after power is reduced. If the message posts and then clears after takeoff, notify maintenance after landing.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG OIL TEMP HIGH

Engine oil temperature is above normal limits.

If high oil temperature occurs after a thrust reduction

1. Thrust Lever (affected engine)Advance, if possible

If high oil temperature occurs with engine at high thrust

1. Thrust Lever (affected engine)Retard

If ENG OIL TEMP HIGH message remains or other abnormal engine indications are present

2. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG OVERSPD PROT FAIL

The FADEC has detected a failure of the overspeed protection system.

1. Thrust Lever (affected engine) IDLE
2. Engine indications (affected engine) Monitor

If engine indications remain within limits

3. Thrust Lever (affected engine) Maintain at IDLE
4. FUEL CROSSFEED As Required
5. WING FLOW FROM L(R) (operable side)

CAUTION

When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

6. TCAS (TCAS II only) TA ONLY
7. Land at nearest suitable airport
8. Accomplish SINGLE-ENGINE APPROACH AND LANDING
(Section 3A – Abnormal Procedures)

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG OVERSPD PROT FAIL (continued)

If un-commanded acceleration of affected engine occurs

9. Thrust lever (affected engine) CUT OFF
10. Land at nearest suitable airport
11. Accomplish ENGINE PRECAUTIONARY SHUTDOWN IN FLIGHT (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

L(R) FIRE DETECTOR FAIL

The associated fire detection system has failed.

If ENGINE FIRE indications were present prior to the failure

1. Consider fire still burning
2. Land at nearest suitable airport

--- END OF PROCEDURE ---

If ENGINE FIRE indications were not present prior to the failure

1. Engine Indications (affected side) Monitor
2. Continue planned flight

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ENVIRONMENTAL / PRESSURIZATION SYSTEMS

CRACKED WINDSHIELD OR WINDOW

A cracked or shattered windshield or window has been detected.

1. Airspeed.....200 KIAS maximum
2. Descend to 15,000 ft MSL or Minimum Safe Altitude,
whichever is higher
3. Land at nearest suitable airport

--- END OF PROCEDURE ---

CABIN ALT CTRL FAIL

The Cabin Pressurization system automatic operation has failed.

1. PRESSURIZATION CONTROL MODE..... HOLD

PRIOR TO LANDING

2. CABIN DUMPDUMP

WARNING

The airplane must be unpressurized prior to landing.

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-115

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HA-420 AFM

ABNORMAL

CABIN ALT CTRL FAULT

A fault has been detected which affects the cabin dump function.

FOLLOWING LANDING

1. CABIN DUMPDUMP

WARNING

The pressurization controller may not be able to ensure all residual cabin pressure is dumped following landing.

NOTE

The first motion of the main entry door handle operates a small pressure relief valve which will release any residual pressure.

--- END OF PROCEDURE ---

CABIN ALT HIGH FIELD

The Cabin Pressurization system has been operating in the High Field Mode for more than 30 minutes.

1. Oxygen MaskDon as required

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

L(R) CABIN BLEED FAIL

There has been a failure of the cabin bleed system.

1. CABIN INFLOW (affected side)OFF
 2. Altitude.....Descend to FL 250 or below
- END OF PROCEDURE ---

CABIN OXYGEN OFF

Cabin oxygen is off and cabin altitude is greater than 15,500 ft.

If passengers occupy the cabin

1. CABIN OXYGEN.....DROP MASK
- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) CABIN BLEED TEMP HIGH

The cabin bleed temperature is above normal operating limits when operating in Manual Mode.

If L CABIN BLEED TEMP HIGH message is posted

1. CABIN TEMP Press COLD and release
(3-5 times)

NOTE

The COLD button must be released for approximately one second prior to each subsequent re-selection.

If R CABIN BLEED TEMP HIGH message is posted

1. COCKPIT TEMP Press COLD and release
(3-5 times)

If L(R) CABIN BLEED TEMP HIGH message remains posted

2. CABIN INFLOW (affected side) OFF
3. Altitude Descend to FL 250 or below

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ECS AIR COND FAIL

The Temperature Control system has detected a failure of the automatic system.

1. ECS MODE.....MANUAL
2. COCKPIT TEMPAs required
3. CABIN TEMPAs required

--- END OF PROCEDURE ---

ECS GND COOLING FAN FAIL

Ground cooling fan has failed to operate when commanded.

ON GROUND

1. CABIN INFLOW Both OFF
2. Dispatch per MEL

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG BLEED FAIL

There has been a failure of the bleed system, or the aircraft was powered up with a high bleed manifold temperature.

ON GROUND

1. Associated bleed manifold temperature Check

NOTE

The **L(R) ENG BLEED FAIL** may post on the ground following power-up if the associated bleed manifold temperature is 75 °C or greater. In this case, the message may clear by de-powering the aircraft and allowing the manifold temperature to cool prior to re-powering.

--- END OF PROCEDURE ---

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) ENG BLEED FAIL (continued)

DURING FLIGHT

1. ENGINE BLEED (affected side) OFF

CAUTION When operating on a single engine bleed with wing anti-ice ON above FL 340, large or sudden thrust changes may result in an ITT exceedance.

NOTE The **L(R) ENG CONTROL FAULT** message may post if wing anti-ice is active until WING FLOW is selected from the operable side.

2. WING FLOW FROM R(L) (operable side)
3. Altitude Descend to FL 250 or below

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

NO LDG FIELD ELEV

The Cabin Pressurization system does not have landing field elevation information.

1. Enter flight plan destination or manually enter landing field elevation

If NO LDG FIELD ELEV message remains

2. PRESSURIZATION CONTROL MODE..... HOLD

PRIOR TO LANDING

3. CABIN DUMPDUMP

WARNING

The airplane must be unpressurized prior to landing.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

OXYGEN LOW

Oxygen quantity is less than 465 liters.

1. Descend to 10,000 ft MSL or Minimum Safe Altitude, whichever is higher
2. Oxygen quantity Monitor
3. CABIN OXYGEN..... OFF

NOTE

The CABIN OXYGEN knob is mechanical and may be difficult to move to the OFF position.

4. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

OXYGEN QTY FAIL

The oxygen pressure transducer is indicating out of range.

1. Descend to 10,000 ft MSL or Minimum Safe Altitude, whichever is higher
2. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

OXYGEN UNAVAILABLE

Oxygen quantity is depleted.

1. OXYGEN SUPPLY Verify ON

If OXYGEN UNAVAILABLE message remains

2. Descend to 10,000 ft MSL or Minimum Safe Altitude, whichever is higher
3. Land at nearest suitable airport

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-126

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HA-420 AFM

ABNORMAL

FLIGHT CONTROLS

FLAP FAIL

The flap system has reported a failure.

1. Flaps are inoperative

CAUTION

If the flap position is unknown, limit inflight speed to 160 KIAS and assume the flaps are UP for subsequent landing.

CAUTION

*If the avionics system is unable to determine flap position, the **STALL PUSHER FAIL** message will post in combination with this message. Adjustments to the landing speed and distance must be applied for both failures.*

2. Accomplish REDUCED FLAP LANDING
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FLAP FAULT

The flap system has reported a fault or has been powered for three days without performing the Built-In Test.

ON GROUND

1. In some circumstances, the fault may be cleared by cycling aircraft power.

--- END OF PROCEDURE ---

DURING FLIGHT

1. FLAPS Monitor operation

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

PITCH TRIM FAIL

The primary pitch trim has failed, or a stuck switch has been detected.

NOTE

In case of a stuck switch, autopilot may continue to function, and pitch trim functionality may be available from the other switch.

1. AUTOPILOT..... Disengage
2. PITCH TRIM MODE..... Cycle STBY then NORM

If pitch trim remains failed

3. PITCH TRIM MODE..... STBY
4. Adjust trim by use of the STANDBY PITCH switch

NOTE

The autopilot will not be available with a primary pitch trim failure.

5. RVSM airspace.....Exit

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SPEEDBRAKE EXTENDED

Speedbrake is not fully closed when commanded to close.

1. SPEEDBRAKE LeverRET

If SPEEDBRAKE EXTENDED is still posted

2. Land at nearest suitable airport

CAUTION

*Range will be reduced with the speedbrake extended.
FMS fuel planning information is based on current
flight conditions and may be beneficial to determine
available range.*

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL SYSTEMS

FUEL CROSSFEED

Fuel Crossfeed has been selected for more than 5 minutes with both engines running, or for 15 minutes with only one engine running.

1. Fuel QuantityMonitor for excessive decrease

If fuel leak is suspected

2. FUEL CROSSFEED NORM
3. Land at nearest suitable airport

--- END OF PROCEDURE ---

FUEL CROSSFEED FAIL

Fuel crossfeed has been selected but is not functioning.

1. FUEL PUMP Verify both NORM
2. FUEL CROSSFEED Select NORM for 2 seconds,
then reselect desired direction

If FUEL CROSSFEED FAIL message remains

3. FUEL PUMP To side OFF, From side ON

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL CROSSFEED FAIL (continued)

If FUEL CROSSFEED FAIL message remains

4. FUEL PUMPBoth NORM
5. FUEL CROSSFEED NORM
6. Fuel Quantity Monitor
7. Thrust Levers..... Use asymmetric thrust to
maintain fuel balance

--- END OF PROCEDURE ---

FUEL IMBALANCE

A 100 lb fuel imbalance between the left and right wing tanks has been detected

1. FUEL CROSSFEEDSelect to side with lower fuel quantity

When FUEL IMBALANCE indication clears

2. FUEL CROSSFEED NORM

NOTE

An emergency fuel imbalance of 600 pounds has been demonstrated for safe return and landing.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL IMBALANCE (continued)

NOTE

If a crosswind landing is required with a fuel imbalance greater than 100 lbs, select a runway with the crosswind from the heavy wing side.

--- END OF PROCEDURE ---

FUEL LEVEL CTRL FAULT

A fault has been detected in the fuel level control system.

1. Avoid rapid rolling and abrupt pitching maneuvers
2. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L-R FUEL PRESSURE LOW

Fuel pressure is below minimum allowable limits for both engines.

1. FUEL PUMP (both) ON

If FUEL PRESSURE LOW message remains

2. THRUST LEVER (affected engine)..... Minimum
practical setting
3. Altitude..... Maintain 20,000 ft MSL or below
4. Land at nearest suitable airport

--- END OF PROCEDURE ---

L(R) FUEL PRESSURE LOW

Fuel pressure is below minimum allowable limits.

1. FUEL PUMP (affected side) ON

If FUEL PRESSURE LOW message remains

2. THRUST LEVER (affected engine)..... Minimum
practical setting
3. Altitude..... Maintain 20,000 ft MSL or below
4. Land at nearest suitable airport

--- END OF PROCEDURE --

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) FUEL PUMP FAIL

Fuel pump has failed to function as commanded.

1. Solid State Relay (SSR) Reset if necessary

*If FUEL PUMP FAIL message remains and is accompanied by
FUEL QUANTITY LOW indication*

2. FUEL PUMP (affected side)ON

NOTE

If the **FUEL PUMP FAIL** indication remains,
indicated fuel quantity below 20 lbs is unusable on the
affected side.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL QTY FAULT

A failure of fuel quantity system has been detected which affects the left, right, or center fuel quantity indication.

1. Indicated Fuel Quantity Check to determine failed system

CAUTION

A **FUEL QTY FAULT** which affects the left or right wing tanks will cause the associated **FUEL QTY LOW** message to delay posting until the wing tank fuel quantity is as low as 110 lbs.

NOTE

Faults affecting the fuel quantity indication may also affect the Fuel Used information on the Fuel Synoptic page, and Fuel Remaining indications of the Weight and Balance Information on the CDU.

┐ Procedure Continued ┘

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL QTY FAULT (continued)

If all fuel tank quantities are still displayed, or the center tank fuel quantity is invalid (“----“)

2. Avoid rapid rolling and abrupt pitch maneuvers
3. Land at nearest suitable airport

CAUTION

If all fuel tank quantities are still displayed, faults may be affecting one or more fuel tanks. All fuel quantity information should be considered inaccurate.

If L or R fuel tank quantities are displayed as invalid (“----“)

2. Replan flight as necessary

NOTE

Displayed fuel quantity information is valid and can be assumed to be available for inflight planning.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L-R FUEL QTY LOW

L(R) FUEL QTY LOW

Approximately 220 lbs or less of fuel remains in one or both wings based on either the optical low level sensor, or the fuel quantity indicator.

CAUTION

A **FUEL QTY FAULT** which affects the left or right wing tanks will cause the associated **FUEL QTY LOW** message to delay posting until the wing tank fuel quantity is as low as 110 lbs.

1. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL TEMP HIGH

The left and/or right wing fuel temperature is above 60 °C.

ON THE GROUND

1. Do not dispatch

--- END OF PROCEDURE ---

IN FLIGHT

1. FUEL PUMP (affected side)ON

If FUEL TEMP HIGH message remains

2. Land at nearest suitable airport

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

FUEL TEMP LOW

The left and/or right wing fuel temperature is below -40 °C.

1. FUEL PUMP (affected side) ON
 2. Aircraft altitude Descend to warmer air
- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

HYDRAULIC SYSTEMS

HYD PUMP FAIL

The Hydraulic Pump has failed.

1. Hydraulic pressure..... Monitor

--- END OF PROCEDURE ---

HYD PRESSURE LOW

Main system hydraulic pressure is below allowable limits.

1. Hydraulic pressure..... Monitor
2. Speedbrake (if installed).....Do not extend
3. Land at nearest suitable airport

PRIOR TO LANDING

4. Accomplish ALTERNATE GEAR RELEASE EXTENSION
(Section 3A – Abnormal Procedures)

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

HYD PRESSURE LOW (continued)

AFTER TOUCHDOWN

5. Normal braking and steering are operable using accumulator pressure

FOLLOWING LANDING ROLLOUT

6. Do not taxi

CAUTION

Following landing rollout, the emergency and brake accumulators may drop below the level required for nosewheel steering and braking.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ICE PROTECTION SYSTEMS

L(R) ENG ANTI-ICE FAIL

The engine anti-ice system has failed.

ON GROUND

1. ENGINE ANTI-ICE (affected engine)..... OFF

If ground icing conditions are present

2. Thrust lever (affected engine) CUT OFF
3. Do not dispatch

--- END OF PROCEDURE ---

DURING FLIGHT

1. ENGINE ANTI-ICE (affected engine)..... OFF
2. Icing ConditionsExit

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ICE DETECT FAIL

Both ice detectors have failed.

1. Icing ConditionsExit

CAUTION

With a failure of the ice detectors, windshield high heat mode is not available, and the windshield heat may not provide sufficient anti-icing of the windshield.

NOTE

When the ice detectors are failed, icing conditions are defined as a SAT of 5 °C to -40 °C in visible moisture.

2. Select a landing airport free of icing conditions for approach and landing

If icing conditions are inadvertently encountered

1. ICE PROTECTION (ON)
 - a. WING ANTI-ICE..... ON

NOTE

Monitor ITT during operation of wing anti-ice.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ICE DETECT FAIL (continued)

- b. ENG ANTI-ICEON
- c. TAIL DE-ICE.....ON

NOTE Ice protection must remain ON for a minimum of 3 minutes after exiting icing conditions.

NOTE When the ice detectors are failed, icing conditions are defined as a SAT of 5 °C to -40 °C in visible moisture.

If an approach and landing must be made through icing conditions

- 1. Select longest, widest runway available
- 2. Adjust pilot seat in order to look through the lower forward portion of the side window

CAUTION *With a failure of the ice detectors, windshield high heat mode is not available and the windshield heat may not provide sufficient anti-icing of the windshield.*

- 3. Use copilot, if available, to adjust power during approach
- 4. Avoid left crosswind if possible for single pilot operations

┌ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ICE DETECT FAIL (continued)

DESCENT

1. Landing Field Elevation Verify set
2. ICE PROTECTION (ON)
 - a. WING ANTI-ICE ON

NOTE Monitor ITT during operation of wing anti-ice.

- b. ENG ANTI-ICE ON
 - c. TAIL DE-ICE ON
3. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts Adjusted and secure
2. Passenger Briefing Complete
3. CABIN SIGNS As required
4. Avionics Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance Set,
programmed,
and modes selected
 - c. Landing Distance Confirm

┌ Procedure Continued ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-146

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Honda Aircraft Company

HA-420 AFM

ABNORMAL

ICE DETECT FAIL (continued)

NOTE

With the wing anti-ice ON, the **STALL WARN ICE ADVANCE** message will post. Do not fly slower than the approach reference speed (green circle). Refer to Uncorrected Landing Field Length, Flaps TO/APPR – Icing (Section 5 – Performance).

- 6. CAS Messages..... Check
- 7. Approach Briefing..... Complete
- 8. FLAPS..... TO/APPR
- 9. TAWS Warnings..... Flap Override

BEFORE LANDING

- 1. LANDING GEAR.....DN
- 2. SPEEDBRAKE (if installed).....RET
- 3. Airspeed..... V_{REF}
- 4. Autopilot/Yaw Damper..... Disengage

LANDING

- 1. At approximately 300 ft AGL, transition to visual reference looking out side window
- 2. Thrust Levers..... IDLE
- 3. Land with minimum flare

┌ Procedure Continued ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-147

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HA-420 AFM

ABNORMAL

ICE DETECT FAIL (continued)

4. BrakesApply (after touchdown)
 5. SPEEDBRAKE (if installed)..... EXT
- END OF PROCEDURE ---

ICE PROT NOT ACTIVE

Ice detected and one or more protection systems are not active.

ON GROUND

1. ICE PROTECTION (ON and NORM)
 - a. WING ANTI-ICE..... ON

NOTE Monitor ITT during operation of wing anti-ice.

- b. ENGINE ANTI-ICE..... ON
- c. TAIL DE-ICE..... ON
- d. WINDSHIELD HEAT NORM

┐ **Procedure Continued** ┘

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ICE PROT NOT ACTIVE (continued)

IN FLIGHT

1. ICE PROTECTION (ON and NORM)

- a. WING ANTI-ICE.....ON

NOTE

Wing anti-ice will not activate automatically above FL 340 when the wing anti-ice switch is in the NORM position, and it must be manually selected on for flight in icing conditions above FL 340.

NOTE

Monitor ITT during operation of wing anti-ice.

- b. ENGINE ANTI-ICE.....ON
- c. TAIL DE-ICE..... NORM or ON
- d. WINDSHIELD HEAT NORM

If message does not clear

- 2. Icing ConditionsExit

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) PROBE HEAT FAIL

The associated air data probe heater has failed or is not powered.

1. PFD (affected side)..... Select operable ADC
2. RVSM airspace.....Exit
3. Do not perform intentional stalls

NOTE Stall Pusher is inoperative. Stall Warning remains operative based on remaining ADC AOA data.

PRIOR TO LANDING

4. Final approach speed $V_{REF} + 5$

Flap Setting	Min Approach Speed	Landing Distance
LDG	$V_{REF} + 5$	Add 10%
TO/APPR (ice)	$V_{REF} + 5$	Add 10%

NOTE Due to the stall pusher being inoperative, the minimum approach speed and landing distance factor are defined in the table.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

STBY PROBE HEAT FAIL

The standby air data probe heater has failed or is not powered.

1. Disregard standby instrument air data

--- END OF PROCEDURE ---

TAIL DE-ICE FAIL

The tail de-ice system has failed.

1. Solid State Relay (SSR) Reset if necessary

If TAIL DE-ICE FAIL message remains

2. Icing ConditionsExit

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

WING ANTI-ICE FAIL

Wing anti-ice system pre-flight test has failed, wing anti-ice has failed, or wing bleed air leak detection has failed.

ON GROUND

1. WING ANTI-ICE.....OFF

NOTE

The wing anti-ice system performs a self-test after the second engine is started on the ground. If an engine is subsequently shut down or manifold pressure did not remain above 10 psi during the self-test, the **WING ANTI-ICE-FAIL** message will post. In this case, the message can be cleared by shutting down the engines and cycling aircraft power off and back on again.

--- END OF PROCEDURE ---

IN THE AIR

1. WING ANTI-ICE.....OFF
2. Icing ConditionsExit

If wing protected surfaces cannot be confirmed to be free of ice, or icing conditions are expected during approach or landing

3. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-152

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

ABNORMAL

WING ANTI-ICE FAIL ON

Wing anti-ice system has failed on.

1. Icing ConditionsExit
2. WING ANTI-ICE OFF

If WING ANTI-ICE FAIL ON message remains posted

3. Continue with planned flight

FOLLOWING LANDING

4. ENGINE BLEED Both OFF

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) WING ANTI-ICE OVERHEAT

An overheat has been detected in the associated wing anti-ice system.

1. WING FLOW FROM R(L) (opposite engine)

If L(R)WING ANTI-ICE OVERHEAT message remains

2. Icing ConditionsExit
3. WING ANTI-ICEOFF

If wing protected surfaces cannot be confirmed to be free of ice, or icing conditions are expected during approach or landing

4. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L-R WING A/I TEMP LOW

The wing anti-ice system is operating at a lower than expected temperature.

NOTE Operation of wing anti-ice is not required when SAT is below -40 °C. The **L-R WING A/I TEMP LOW** message may post if the system is operated in these conditions.

NOTE Monitor ITT during operation of wing anti-ice.

NOTE Stall warning ice advance will go to the failure schedule after one minute in icing conditions with the **L-R WING A/I TEMP LOW** message posted.

1. Thrust Levers..... Increase (Minimum 62% N_1)

NOTE If thrust was set below 62% N_1 , it may take up to three minutes for the system to warm up sufficiently to clear the message.

NOTE Posting of the **L-R WING A/I TEMP LOW** message may occur if operating in icing conditions outside the approved envelope. However, if the message does not clear within approximately one minute after exiting icing conditions, a system failure is likely and further icing encounters must be avoided.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L-R WING A/I TEMP LOW (continued)

If L-R WING A/I TEMP LOW message remains posted

2. Icing ConditionsExit
3. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

If L-R A/I TEMP LOW message clears

2. Continue flight in icing conditions as required
- *When clear of icing conditions*

CAUTION

Do not attempt more than one reset of the **STALL WARN ICE ADVANCE**. Multiple occurrences of the **L-R WING A/I TEMP LOW** message likely indicates a system failure.

3. Stall Warn Ice Advance Reset Select
4. WING ANTI-ICE ON

NOTE

Operate wing anti-ice until the **STALL WARN ICE ADVANCE** message posts (approximately one minute).

┐ **Procedure Continued** ┘

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L-R WING A/I TEMP LOW (continued)

5. WING ANTI-ICE NORM

--- END OF PROCEDURE ---

○ *If icing conditions expected through approach and landing*

3. Accomplish LANDING WITH ICE ACCUMULATION ON
WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) WING ANTI-ICE TEMP LOW

The affected wing anti-ice system is operating at a lower than expected temperature.

NOTE Operation of wing anti-ice is not required when SAT is below -40 °C. If the system is operated in these conditions, the **L(R) WING ANTI-ICE TEMP LOW** message may post.

NOTE Monitor ITT during operation of wing anti-ice.

NOTE Stall warning ice advance will go to the failure schedule after one minute with the **L(R) WING ANTI-ICE TEMP LOW** message posted.

1. Thrust Levers..... Increase (Minimum 62% N_1)

NOTE If thrust was set below 62% N_1 , it may take up to three minutes for the system to warm up sufficiently to clear the message.

NOTE Posting of the **L(R) WING ANTI-ICE TEMP LOW** message may occur if operating in icing conditions outside the approved envelope. However, if the message does not clear within approximately one minute after exiting icing conditions, a system failure is likely and further icing encounters must be avoided.

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) WING ANTI-ICE TEMP LOW (continued)

2. WING FLOWFROM R(L) (opposite side)

If L(R) WING ANTI-ICE TEMP LOW message remains posted

3. Icing ConditionsExit
4. WING ANTI-ICE..... OFF
5. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

If L(R) WING ANTI-ICE TEMP LOW message clears

3. Continue flight in icing conditions as required

- *When clear of icing conditions*

CAUTION

Do not attempt more than one reset of the **STALL WARN ICE ADVANCE**. Multiple occurrences of the **L(R) WING ANTI-ICE TEMP LOW** message likely indicates a system failure.

4. Stall Warn Ice Advance Reset Select
5. WING ANTI-ICE ON

NOTE

Operate wing anti-ice until the **STALL WARN ICE ADVANCE** message posts (approximately one minute).

┐ **Procedure Continued** ┐

HJ1-29000-003-001

FAA APPROVED
October 30, 2016

Page 3A-159

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HA-420 AFM

ABNORMAL

L(R) WING ANTI-ICE TEMP LOW (continued)

6. WING ANTI-ICE..... NORM

--- END OF PROCEDURE ---

○ *If icing conditions expected through approach and landing*

4. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

WING ANTI-ICE UNAVAIL

Wing anti-ice has been selected ON but is unavailable during single bleed operations.

NOTE

When operating in the Single Bleed Inoperative Keep Out Zone, the engine is incapable of supporting wing anti-ice using a single bleed source. If wing anti-ice is manually selected ON, the **WING ANTI-ICE UNAVAIL** message will post alerting the pilot that wing anti-ice is unavailable.

┐ Procedure Continued ┘

Honda Aircraft Company

HA-420 AFM

ABNORMAL

WING ANTI-ICE UNAVAIL (continued)

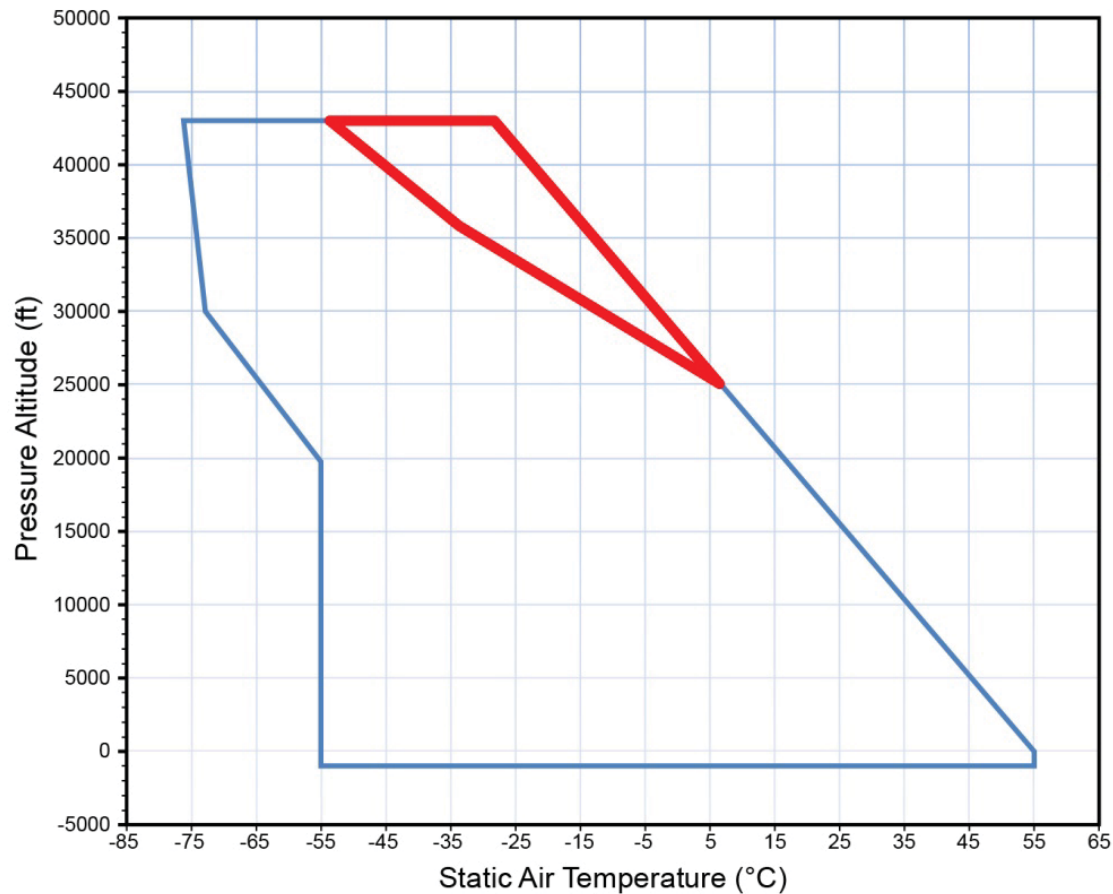


Figure 13. Single Bleed Inoperative Keep Out Zone

1. Remain clear of icing conditions
2. WING ANTI-ICE..... NORM

NOTE Monitor ITT during operation of wing anti-ice.

┌ Procedure Continued ┐

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-162

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Honda Aircraft Company

HA-420 AFM

ABNORMAL

WING ANTI-ICE UNAVAIL (continued)

If icing conditions are encountered

3. Altitude..... Expedite descent to FL 250 or below

If wing protected surfaces cannot be confirmed to be free of ice

4. Accomplish LANDING WITH ICE ACCUMULATION ON WINGS (Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

L(R) WSHD ZONE FAIL

There is a failure of the windshield zone heat.

1. Solid State Relay (SSR) Reset if necessary

If WSHD ZONE FAIL message remains

2. Icing ConditionsExit
3. WINDSHIELD HEAT (affected zone) OFF

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

L(R) WSHD ZONE OVERHEAT

There is an overheat of the windshield zone.

1. Icing ConditionsExit
 2. WINDSHIELD HEAT (affected zone)OFF
- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

LANDING GEAR AND BRAKES

ALTERNATE GEAR RELEASE EXTENSION

Extension of the landing gear using the alternate system is required

1. LDG GEAR CTRL circuit breaker..... PULL (B7)
2. LANDING GEARDN
3. Airspeed..... 150 KIAS minimum
4. ALTERNATE GEAR RELEASE handle..... Pull fully
5. Yaw airplane, if necessary, to obtain gear locked down
6. LANDING GEAR indicator..... Verify three green DN

NOTE The gear DOOR icon will remain posted but a normal landing is possible without gear door - ground contact.

7. ALTERNATE GEAR RELEASE handle..... Stow

CAUTION *The ALTERNATE GEAR RELEASE handle could interfere with thrust lever operation if not stowed following use.*

NOTE Normal landing gear operations cannot be restored in flight after activation of the alternate gear release.

┐ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ALTERNATE GEAR RELEASE EXTENSION (continued)

If all landing gear indicates down and locked

8. Land using normal procedures

--- END OF PROCEDURE ---

If all landing gear doesn't indicate down and locked

DESCENT

1. Select a long, wide, and dry runway with minimal crosswind
2. Landing Field Elevation Verify set
3. ENGINE ANTI-ICE As required
4. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts Adjusted and secure
2. Passenger Briefing Complete
3. CABIN SIGNS As required
4. Avionics Set
5. Landing Data Set and confirmed
6. Radios and Navigation Set

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ALTERNATE GEAR RELEASE EXTENSION (continued)

7. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
8. CAS Messages..... Check
9. Approach Briefing Complete
10. FLAPS TO/APPR

BEFORE LANDING

1. SPEEDBRAKE (if installed).....RET
2. FLAPS LDG or TO/APPR (ice)

CAUTION

Do not extend the flaps to LDG unless the airframe can be confirmed free of ice and icing conditions are not expected during approach and landing.

NOTE

If TAWS-A is installed, the TAWS Warnings will annunciate when landing with Flaps at TO/APPR, unless Flap Override is selected.

3. Airspeed.....V_{REF}
4. Autopilot/Yaw Damper Disengage

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ALTERNATE GEAR RELEASE EXTENSION (continued)

LANDING

1. Thrust Levers..... IDLE
- *If the nose wheel is not down and locked*
2. After touchdown..... Hold nose off ground after landing for as long as possible. Lower prior to loss of elevator control
3. Maintain directional control with rudder and differential braking

NOTE

When the anti-skid system is active during braking, the amount of available differential braking is reduced.

- *If one main wheel is not down and locked*
2. After touchdown..... Hold unsupported wing off ground after landing for as long as possible. Lower wing prior to loss of lateral control
3. Maintain directional control with rudder and nosewheel steering
4. EMERGENCY BRAKE Apply gradually

CAUTION

Use of the emergency brake is required to achieve satisfactory braking as anti-skid would cause brakes to release due to one main gear not being down if normal brakes were used.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ALTERNATE GEAR RELEASE EXTENSION (continued)

NOTE Gradually pull emergency brake handle until desired braking action is achieved.

NOTE Landing distance will increase by 50%.

5. SPEEDBRAKE (if installed).....EXT

AFTER AIRCRAFT STOPS

1. PARKING BRAKE..... Set
2. Thrust Levers..... CUT OFF
3. ENGINE FIRE PUSH Switches.....LIFT
COVER
and PUSH
4. FIRE EXT PUSH Switch Push (if required)
5. BATTERY..... OFF
6. Evacuate the aircraft using the cabin door and/or the emergency exit

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

ANTI-SKID FAIL

The antiskid braking system has failed.

PRIOR TO LANDING

1. Select a dry runway

Flap Setting	Min Approach Speed	Landing Distance
LDG	V_{REF}	Add 50%
TO/APPR (ice)	V_{REF}	Add 50%

WARNING

Any braking above light in wet runway conditions could result in blown tires and loss of directional control.

NOTE

Landing distance will increase by 50% on a dry runway and 100% on a wet runway using light braking.

AFTER TOUCHDOWN

2. Brakes..... Apply gradually

CAUTION

Brakes must be applied gradually. Light to moderate braking can be applied without skidding tires on a dry surface, however, the pilot should consider runway surface conditions when applying brakes.

--- END OF PROCEDURE ---

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3A-170

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HA-420 AFM

ABNORMAL

LDG GEAR FAIL

The landing gear has not reached the handle commanded position after 20 seconds.

If LANDING GEAR handle is UP

1. Airspeed.....200 KIAS maximum

--- END OF PROCEDURE ---

If LANDING GEAR handle is DN and both main gear are indicating down

1. Accomplish ALTERNATE GEAR RELEASE EXTENSION
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

If LANDING GEAR handle is DN and both main gear are not indicating down

1. LANDING GEAR..... UP
2. Accomplish ALTERNATE GEAR RELEASE EXTENSION
(Section 3A – Abnormal Procedures)

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

NOSEWHEEL STEER FAIL

The nosewheel steering system has failed, or is selected OFF during taxi operations.

ON GROUND

1. NOSE WHEEL STEERING NORM

If NOSEWHEEL STEER FAIL message remains

2. Do not taxi

--- END OF PROCEDURE ---

IN THE AIR

1. NOSE WHEEL STEERING OFF

PRIOR TO LANDING

2. Select runway aligned with wind

CAUTION

Nosewheel steering is inoperative. Crosswinds should be minimized to ensure adequate directional control during the low-speed portion of the rollout using differential braking.

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

NOSEWHEEL STEER FAIL (continued)

Flap Setting	Min Approach Speed	Landing Distance
LDG	V_{REF}	Add 10%
TO/APPR (ice)	V_{REF}	Add 10%

NOTE Nosewheel Steering has failed. Due to use of differential braking, landing distance will increase by 10%.

AFTER TOUCHDOWN

3. Maintain directional control with rudder and differential braking

NOTE When the anti-skid system is active during braking, the amount of available differential braking is reduced.

FOLLOWING LANDING ROLLOUT

4. Taxi to parking using differential braking

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

PARK BRAKE FAIL

Brake pressure is less than 1450 psi with the PARK BRAKE set.

1. Wheel chocks Install

--- END OF PROCEDURE ---

PARK BRAKE ON

The parking brake is pressurized during flight.

1. EMERGENCY/PARKING BRAKE handle Verify
fully released

If PARKING BRAKE ON message remains posted

2. Brake pressure Monitor

If brake pressure is greater than 175 psi

3. Select long wide runway

CAUTION *Main landing gear tires may fail at touchdown.*

If one or both tires fail

4. EMERGENCY BRAKE Apply gradually

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

OPERATIONAL

CABIN SYSTEM SHEDDING

An anomaly has been detected with a cabin system (including lighting, inverters, or wireless transmitter) requiring the cabin systems to be depowered

1. CABIN POWER.....OFF

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

LANDING WITH ICE ACCUMULATION ON WINGS

A landing is required with ice buildup on the wing protected surfaces, asymmetric ice buildup or larger than usual buildup aft of the protected surfaces.

1. Select a long, wide, and dry runway with minimal crosswind

CAUTION

Aircraft response is degraded with ice accumulation on the wings. Crosswinds should be minimized to ensure satisfactory control during approach and landing rollout.

DESCENT

1. Airspeed.....200 KIAS or greater
2. Landing Field Elevation Verify set
3. ENGINE ANTI-ICE.....As required
4. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts.....Adjusted and secure
2. Passenger Briefing.....Complete
3. CABIN SIGNSAs required
4. Avionics..... Set

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

LANDING WITH ICE ACCUMULATION ON WINGS

(continued)

5. Landing Data Set and confirmed
- a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance.....Confirm

Flap Setting	Min Approach Speed	Landing Distance
TO/APPR (ice)	$V_{REF} + 35$	Add 75%

NOTE If **STALL WARN ICE ADVANCE** message is posted with Wing Anti-ice OFF or failed, the minimum speeds, as indicated by the low-speed awareness cues, will be increased. Do not fly slower than the approach reference speed (green circle).

NOTE The minimum approach speed and landing distance factor are defined in the table. Refer to Uncorrected Landing Field Length, Flaps TO/APPR – Icing (Section 5 – Performance).

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

LANDING WITH ICE ACCUMULATION ON WINGS

(continued)

6. CAS Messages..... Check
7. Approach Briefing..... Complete
8. Airspeed..... $V_{REF} + 40$ minimum
9. TAWS Warnings Flap Override

BEFORE LANDING

1. FLAPS TO/APPR
2. LANDING GEAR..... DN
3. SPEEDBRAKE (if installed)..... RET
4. Airspeed..... $V_{REF} + 35$
5. Autopilot/Yaw Damper Disengage

LANDING

1. Thrust Levers..... IDLE
2. Brakes..... Apply (after touchdown)
3. SPEEDBRAKE (if installed)..... EXT

┌ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

LANDING WITH ICE ACCUMULATION ON WINGS

(continued)

BALKED LANDING

1. TO/GA.....Press
2. Thrust Levers.....TO
3. Airspeed.....Maintain $V_{REF} + 35$

When clear of obstacles

CAUTION

Retracting the flaps to UP following an icing encounter with the flaps extended may result in damage to the flaps or airframe.

NOTE

See performance tables in the Airplane Flight Manual (Section 5 – Performance) and either the Quick Reference Handbook (Volume 1, Performance Section) or the Pilot's Operating Manual (Section 2 – Flight Planning).

4. LANDING GEAR.....UP (positive rate of climb)
5. Yaw Damper..... Engaged
6. Thrust Levers.....MCT
7. Flight GuidanceAs required

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

REDUCED FLAP LANDING

A landing with flaps TO/APPR or UP is required.

DESCENT

1. Landing Field Elevation Verify set
2. ENGINE ANTI-ICE As required
3. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts Adjusted and secure
2. Passenger Briefing Complete
3. CABIN SIGNS As required
4. Avionics Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance Set,
programmed,
and modes selected
 - c. Landing Distance Confirm

┐ Procedure Continued ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

REDUCED FLAP LANDING (continued)

Flap Setting	Min Approach Speed	Landing Distance
UP	$V_{REF} + 15$	Add 30%
TO/APPR	$V_{REF} + 5$	Add 10%
UP (ice)	$V_{REF} + 10$	Add 30%

NOTE The minimum approach speed and landing distance factor are defined in the table.

NOTE For landing in icing conditions, refer to Uncorrected Landing Field Length, Flaps TO/APPR – Icing (Section 5 – Performance).

- 6. CAS Messages..... Check
- 7. Approach Briefing..... Complete
- 8. TAWS Warnings Flap Override

┐ **Procedure Continued** ┐

Honda Aircraft Company

HA-420 AFM

ABNORMAL

REDUCED FLAP LANDING (continued)

BEFORE LANDING

1. LANDING GEAR..... DN
2. SPEEDBRAKE (if installed).....RET
3. Airspeed.....(Flaps UP) - $V_{REF} + 15$
(Flaps TO/APPR) - $V_{REF} + 5$
(Flaps UP [ice]) - $V_{REF} + 10$
4. Autopilot/Yaw Damper Disengage

LANDING

1. Thrust Levers..... IDLE
2. BrakesApply (after touchdown)
3. SPEEDBRAKE (if installed)..... EXT

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SIDE WINDOW LANDING

A landing is to be conducted using the side windows only due to obstruction of the forward windows.

1. Select longest, widest runway available.
2. Adjust pilot seat in order to look through the lower forward portion of the side window.
3. Use copilot, if available, to adjust power during approach.
4. Avoid left crosswind if possible for single pilot operations.

If executing an ILS or LPV APPROACH

5. At approximately 300 ft AGL, transition to visual reference looking out side window.
6. Land with minimum flare.

--- END OF PROCEDURE ---

If executing a VISUAL APPROACH

5. Conduct approach using a left base to short final.
6. Establish airplane on runway centerline at 300 ft AGL and transition to visual reference looking out the side window.
7. Land with minimum flare.

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SINGLE-ENGINE APPROACH AND LANDING

A landing with one engine shut down is required.

DESCENT

1. Landing Field Elevation Verify set
2. ENGINE ANTI-ICE As required
3. Altimeters (transition altitude) Set

APPROACH

1. Seats and Seat Belts Adjusted and secure
2. Passenger Briefing Complete
3. CABIN SIGNS As required
4. Avionics Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance Set,
programmed,
and modes selected
 - c. Landing Distance Confirm

┐ Procedure Continued ┘

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SINGLE-ENGINE APPROACH AND LANDING (continued)

Flap Setting	Min Approach Speed	Landing Distance
TO/APPR	$V_{REF} + 5$	Add 10%
LDG	V_{REF}	Normal
TO/APPR (ice)	V_{REF}	Normal

6. CAS Messages..... Check
7. Approach Briefing Complete
8. TAWS Warnings Flap Override
9. FLAPS TO/APPR

BEFORE LANDING

1. LANDING GEAR DN
2. SPEEDBRAKE (if installed) RET
3. FLAPS TO/APPR
4. Airspeed $V_{REF} + 5$
5. AFCS/TRIM MASTER Push and release

NOTE

Yaw damper status is not annunciated while the rudder bias function is active. Momentary selection of the AFCS/TRIM MASTER will ensure the yaw damper is disengaged for landing.

┌ Procedure Continued ┐

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October 30, 2016

HJ1-29000-003-001

Page 3A-185

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HA-420 AFM

ABNORMAL

SINGLE-ENGINE APPROACH AND LANDING (continued)

NOTE

Rudder bias function will be momentarily disabled when the AFCS/TRIM MASTER button is pressed. The pilot should be prepared to apply additional rudder pedal force to compensate.

LANDING ASSURED

1. FLAPS LDG or TO/APPR (ice)
2. Airspeed..... V_{REF}

LANDING

1. Thrust Lever (operable engine) IDLE
2. Brakes Apply (after touchdown)
3. SPEEDBRAKE (if installed)..... EXT

--- END OF PROCEDURE ---

Honda Aircraft Company

HA-420 AFM

ABNORMAL

SINGLE-ENGINE MISSED APPROACH

A missed approach with one engine shut down is required.

1. TO/GA.....Press
2. Thrust Lever (operable engine) TO
3. FLAPS TO/APPR
4. Climb airspeed..... V_{AC}
5. LANDING GEAR UP (Positive rate of climb)
6. Yaw Damper.....As required

When clear of obstacles

7. FLAPSUP (130 KIAS minimum) or TO/APPR (ice)

CAUTION

Retracting the flaps to UP following an icing encounter with the flaps extended may result in damage to the flaps or airframe.

--- END OF PROCEDURE ---

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HA-420 AFM

ABNORMAL

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October 30, 2016

HJ1-29000-003-001

Page 3A-188

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HA-420 AFM

ADVISORY

SECTION 3B ADVISORY PROCEDURES

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS

ADF FAIL	The Automatic Direction Finding system has failed	Use other means of navigation. Note: GNSS may be substituted for ADF per AC 90-108	MEL
ADS B OUT 1(2) FAIL	The ADS-B Out function has failed	Select other transponder (if installed) Consider operational impact	MEL
AFCS FAULT	A fault has been detected by the AFCS system	Crew Awareness	No Dispatch

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

AFCS PITCH SERVO OFF	Pitch Servo is selected OFF	Crew Awareness Exit RVSM airspace Autopilot and Stall Pusher functions are unavailable	No Dispatch
AFCS ROLL SERVO OFF	Roll Servo is selected OFF	Crew Awareness Exit RVSM airspace Autopilot function is unavailable	No Dispatch
AFCS YAW SERVO OFF	Yaw Servo is selected OFF	Crew Awareness Yaw Damper and Rudder Bias functions are unavailable	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

AIR DATA 1(2) FAULT	The affected air data probe is operating in a degraded mode. Some minor degradation in air data may be experienced under certain flight conditions.	Monitor Standby Flight Instrument	No Dispatch
AUTOPILOT FAIL	The autopilot function has failed	Crew Awareness Exit RVSM airspace	MEL
AUTOPILOT FAULT	An automatic pitch trim fault has been detected by the autopilot. If autopilot is disengaged, it may not be able to be reengaged.	Monitor autopilot operations. Notify maintenance following landing	No Dispatch
AVIONICS CONFIG	The avionics software/hardware configuration is not correct for the aircraft	Crew Awareness	No Dispatch

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

AVIONICS FAULT	One or more avionics components are reporting a need for maintenance. One or more displays may be dimmer than usual.	Verify all avionics SD cards are inserted properly. Notify maintenance following landing	MEL
COM 1(2) TEMP HIGH	Affected radio reporting a high internal temperature – effective range may be reduced	Consider using other radio for ATC communications	MEL
CONFIG BRAKE	Parking brake pressure is greater than 175 psi with the parking brake not set.	Crew Awareness	No Dispatch
CONFIG FLAP	Flaps not set to the correct takeoff position	Correct flap position	No Dispatch
CONFIG PITCH TRIM	Pitch trim not set in the takeoff band	Correct pitch trim	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

CONFIG SPEEDBRAKE	Speedbrake not in the takeoff position	Correct speedbrake position	No Dispatch
CONFIG WING ANTI-ICE	Wing anti-ice is active on the ground but wing temperature is less than 10 °C.	Allow wing to warm up before takeoff	No Dispatch
CPDLC TEMP HIGH	The CPDLC radio is reporting a high internal temperature – effective range may be reduced.	Consider operational impact	MEL
CRUISE SPD CTRL FAIL	Cruise speed control has failed	Control speed manually using the thrust levers	MEL
CVR FAIL	The Cockpit Voice Recorder has failed	Consider operational impact	MEL

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

DME FAIL	Distance measuring equipment has failed	Use other means of navigation. Note: GNSS may be substituted for DME per AC 90-108	MEL
FDR FAIL	The Flight Data Recorder has failed	Consider operational impact	MEL
GLIDE-SLOPE 1(2) FAIL	Affected glide slope has failed	Crew Awareness Use other navigation source	MEL
MAINTENANCE MODE	The avionics system is in maintenance mode	Exit Maintenance Mode	No Dispatch
L(R) MIC STUCK ON	Associated handmic or PTT switch is stuck on	If associated with handmic, unplug associated	MEL

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

RUDDER BIAS FAULT	Rudder Bias system has detected a fault but can still activate	Crew Awareness	No Dispatch
SATCOM FAIL	Satellite Communication system has failed including Connex TM Weather, or there has been a failure to receive data correctly during an update	Crew Awareness Another update may be attempted. A successful update will clear the message	MEL
STALL SHAKER FAIL	The stick shaker is unable to actuate a stall warning command	Crew Awareness	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

STALL WARN ICE ADVANCE	Stall Shaker and Pusher trigger thresholds and the approach reference speed (green circle) have been compensated due to ice detection or manual operation of wing anti-ice.	Monitor ITT during both automatic and manual operation of wing anti-ice. If the aircraft cannot be confirmed to be free of ice accumulation, use Flaps TO/APPR for landing. Reference Uncorrected Landing Field Length, Flaps TO/APPR – Icing (Section 5 – Performance).	Not applicable
STBY AHRS FAIL	The standby AHRS has failed	Crew Awareness	No Dispatch
STBY AIR DATA FAIL	Standby instrument air data computer is not providing proper data	Crew Awareness	No Dispatch
SURFACE WATCH FAIL	SurfaceWatch has failed.	Crew Awareness	MEL

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

TRANSPONDER 1(2) FAIL	The associated transponder has failed	Select other transponder (if installed) If no transponder is available, the aircraft no longer meets the requirements for flight in RVSM airspace and RVSM airspace must be exited	MEL
WEATHER RADAR FAIL	The weather radar has failed	Crew Awareness	MEL
WINDSHEAR FAIL	The windshear function has failed	Crew Awareness	MEL
XM DATALINK FAIL	SiriusXM [®] Datalink Weather and Music is failed or data is missing	Crew Awareness	MEL

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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AVIONICS / AFCS (continued)

YAW DAMPER FAULT	A fault has been detected in the yaw damper	Monitor yaw damper operations Notify maintenance following landing	No Dispatch
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DOORS

AFT BAG DOOR OPEN	Door open and parking brake set on ground	Crew Awareness	MEL
CABIN DOOR OPEN	Door open and parking brake set on ground	Crew Awareness	MEL
ELEC SRVC DOOR OPEN	Door open	Crew Awareness	MEL
EXT PWR DOOR OPEN	Door open and parking brake set on ground	Crew Awareness	Not Applicable
FWD BAG DOOR OPEN	Door open and parking brake set on ground	Crew Awareness	MEL

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ELECTRICAL SYSTEMS

BATTERY OFF	Battery selected off	Crew Awareness	No Dispatch
BATTERY 1(2) FAIL	The associated battery has failed	Crew Awareness	No Dispatch
BATTERY 1(2) FAULT	A fault has been detected in the battery or charging system, or the battery temperature is too low for charging.	Crew Awareness	No Dispatch
BUS TIE OPEN	BUS TIE switch is selected to OPEN	Crew Awareness	No Dispatch
CABIN BUS FAIL	The Cabin bus has failed	One solid state relay reset may be attempted, if applicable	MEL
CABIN POWER OFF	The Cabin bus has been manually turned off	Crew Awareness	Not Applicable
L(R) GENERATOR FAULT	Minor generator fault detected	Crew Awareness	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ELECTRICAL SYSTEMS (continued)

L(R) GENERATOR OFF	The affected GEN switch is selected OFF	Crew Awareness	No Dispatch
POWER OUTLETS MAN ON	Power outlets have been commanded on when auto mode would have commanded them off	Crew Awareness	No Dispatch

ENGINES

L(R) ENG FAULT	Minor engine fault detected	Crew Awareness	No Dispatch
L(R) ENG CHIP DETECTED	Engine oil chip detected	Continue planned flight	No Dispatch
L(R) ENG FUEL BYPASS	An impending bypass of the associated fuel filter has been detected	Continue planned flight	No Dispatch
L(R) ENG SHUTDOWN	Engine is shutdown	Crew Awareness	No Dispatch

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ENGINES (continued)

ENG SYNC FAIL	Engine sync has failed	Crew Awareness CSC will not be available	MEL
L(R) ENG TLD	Minor failure detected	Crew Awareness	MEL
L(R) ENG VIB DETECTOR FAIL	The associated engine vibration detector has failed	Crew Awareness	No Dispatch
L(R) FIRE BOTTLE DISCH	Affected fire bottle has been discharged	Crew Awareness	No Dispatch
L(R) FIRE BOTTLE LOW	The associated engine fire extinguisher pressure is low and may not be adequate to extinguish a fire	Continue planned flight	No Dispatch
L(R) FIRE EXT FAIL	Fire extinguisher has failed	Crew Awareness	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ENVIRONMENTAL / PRESSURIZATION SYSTEMS

CABIN ALT ABOVE 10K	Cabin altitude exceeds 10,000 feet when operating in High Field Mode	Crew Awareness	Inflight Only Message
CABIN ALT DUMP	Pressurization selected to DUMP mode	Crew Awareness	MEL
CABIN ALT FAULT	Fault in the cabin pressurization system	Crew Awareness	MEL
CABIN ALT HOLD	Pressurization selected to HOLD mode	Crew Awareness	No Dispatch
L(R) CABIN BLEED FAULT	A fault in the affected cabin bleed system detected	Crew Awareness	MEL
L(R) CABIN INFLOW OFF	Associated inflow selected OFF	Crew Awareness Maximum Pressure Altitude 25,000 feet	MEL
CABIN OXYGEN ON	Cabin oxygen system is on	Crew Awareness	No Dispatch

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October 30, 2016

HJ1-29000-003-001

Page 3B-14

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Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ENVIRONMENTAL / PRESSURIZATION SYSTEMS (continued)

ECS AIR COND FAULT	Either the cockpit or cabin evaporator fan has failed	Crew Awareness If cooling is insufficient, ECS Manual Mode may improve cooling	MEL
ECS TEMP CONTROL MANUAL	ECS manual mode has been selected	Crew Awareness	Yes
L(R) ENG BLEED FAULT	A fault in the affected engine bleed system detected	Crew Awareness	MEL
L(R) ENG BLEED OFF	Bleed selected off	Crew Awareness Maximum pressure altitude is 25,000 feet	MEL
OXYGEN QTY FAULT	A fault has been detected in the oxygen quantity display	Crew Awareness	No Dispatch

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3B-15

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Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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FUEL SYSTEMS

FUEL CROSSFEED OPEN	Fuel Crossfeed valve is open when commanded closed	Monitor fuel balance and crossfeed as required	No Dispatch
FUEL ISO VALVE CLOSED	Isolation valve has failed in the closed position	Crew Awareness Refuel system is inoperative	MEL
FUEL ISO VALVE OPEN	Fuel isolation SOV has failed in the open position	Crew Awareness	MEL
L(R) FUEL PUMP OFF	Affected FUEL PUMP switch selected OFF	Crew Awareness	No Dispatch
L(R) FUEL PUMP ON	Affected fuel pump is commanded on automatically or via the switch	Crew Awareness	No Dispatch
L(R) FUEL QTY DEGRADE	A single fuel quantity probe has failed in the affected fuel tank	Monitor affected fuel quantity	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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FUEL SYSTEMS (continued)

L(R) FUEL SOV FAIL	Fuel shutoff valve has failed to achieve commanded position	Crew Awareness	No Dispatch
L(R) FUEL SOV CLOSED	The Fuel SOV Valve is closed	Crew Awareness	No Dispatch

FLIGHT CONTROLS

FLAP DEGRADE	Minor flap fault detected	Crew Awareness	MEL
FLAP LEVER DISAGREE	The flap handle does not match the flap position after initial power-up	Select new flap position and then reset to current flap position	No Dispatch
PITCH TRIM FAULT	A fault has been detected by the pitch trim system One side of pitch trim indication displaying "X" indicates a failure of one trim tab. In this case the trim rate will be degraded	Crew Awareness If the trim rate is degraded, avoid rapid speed changes	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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FLIGHT CONTROLS (continued)

ROLL TRIM FAIL	Roll trim has failed or a stuck switch has been detected and disabled	In the event of a stuck trim switch, roll trim may be regained by cycling the ROLL TRIM POWER to OFF and then back to NORM	No Dispatch
ROLL TRIM FAULT	A fault has been detected by the roll trim system	Crew Awareness	MEL
ROLL TRIM OFF	Roll trim is selected OFF	Crew Awareness	No Dispatch
SPEEDBRAKE FAIL	Speedbrake has failed	Crew Awareness	MEL

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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FLIGHT CONTROLS (continued)

STBY PITCH TRIM ON	Standby pitch trim has been selected	Crew Awareness Exit RVSM airspace Normal pitch trim is not available Autopilot will not be available	No Dispatch
YAW TRIM FAIL	Yaw trim has failed or a stuck switch has been detected	Crew Awareness For a stuck trim switch, functionality might be regained by cycling the YAW TRIM POWER switch to OFF and back to NORM	No Dispatch
YAW TRIM FAULT	A fault has been detected by the yaw trim system	Crew Awareness	MEL

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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FLIGHT CONTROLS (continued)

YAW TRIM OFF	Yaw trim has been selected OFF	Crew Awareness	No Dispatch
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HYDRAULIC SYSTEMS

EMER BRAKE FAIL	Emergency brake accumulator pressure is less than 1650 psi	Crew Awareness	No Dispatch
HYD FLUID LEVEL LOW	The hydraulic reservoir fluid level is below allowable limits	Monitor hydraulic pressure and reservoir fluid level Service hydraulic fluid prior to next flight	MEL
HYD FLUID OVERFILL	The hydraulic reservoir fluid level is above allowable limits	Service hydraulic fluid prior to next flight	MEL
HYD PUMP FAIL ON	Hydraulic pump has failed on	Monitor hydraulic pressure	No Dispatch

Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ICE PROTECTION SYSTEMS

L(R) ENG ANTI- ICE FAULT	A fault has been detected in the affected engine anti-ice system	Crew Awareness	MEL
ICE DETECTED	Ice has been detected	Crew Awareness For planes that have NOT accomplished GHAE SB 72-0021: Notify maintenance following landing Maintenance action (LMM task 72-00-00-200-804) required within two flights	Not Applicable
ICE DETECT FAULT	One ice detector has failed	Exit icing conditions	MEL
TAIL DE-ICE FAIL ON	The tail de-ice system has failed on	Crew Awareness	MEL

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 3B-21

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Honda Aircraft Company

HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
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ICE PROTECTION SYSTEMS (continued)

TAIL DE-ICE FAULT	There is a degradation in tail de-icing capability	Exit icing conditions	MEL
TAIL DE-ICE OFF	The TAIL DE-ICE switch has been selected OFF	Crew Awareness	MEL
TAIL DE-ICE ON	The TAIL DE-ICE switch has been selected ON	Crew Awareness	Not Applicable
WING ANTI-ICE FAULT	A fault has been detected in the wing anti-ice system	Exit icing conditions	MEL
WING ANTI-ICE OFF	WING ANTI-ICE switch selected OFF	Crew Awareness	MEL
WING ANTI-ICE ON	WING ANTI-ICE switch selected ON	Monitor ITT during operation of wing anti-ice	Not Applicable
WING FLOW FROM L(R)	WING FLOW switch selected FROM L(R)	Crew Awareness	No Dispatch

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
---------	-------------	-------------	----------

ICE PROTECTION SYSTEMS (continued)

L(R) WSHD ZONE FAULT	Affected zone unable to select high heat mode	Exit icing conditions	MEL
L(R) WSHD ZONE OFF	Applicable windshield zone switch selected OFF	Crew Awareness	MEL

LANDING GEAR AND BRAKES

LDG GEAR DOOR FAIL	Landing gear door failed to close	Crew Awareness	No Dispatch
-----------------------	--------------------------------------	-------------------	----------------

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
---------	-------------	-------------	----------

LANDING GEAR AND BRAKES (continued)

NOSEWHEEL STEER FAULT	A fault has been detected in the Nosewheel Steering system	NOSE WHEEL STEERING switch should remain in the NORM position during flight and taxi operations to prevent a steering uncommand movement Note: When the aircraft is completely stopped, cycling the NOSE WHEEL STEERING switch may clear the message if the fault condition no longer exists	No Dispatch
NOSEWHEEL STEER OFF	NOSE WHEEL STEERING switch selected OFF	Crew Awareness	No Dispatch
WOW FAULT	WOW system disagree	Crew Awareness	No Dispatch

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HA-420 AFM

ADVISORY

CAS MSG	Description	Crew Action	Dispatch
---------	-------------	-------------	----------

LIGHTS

EXT LIGHTS MAN OFF	Some external lights have manually been selected OFF when automatic logic had determined they should be on	Crew Awareness	Yes
LIGHTING CONTROL 1(2) FAIL	The associated lighting controller has failed	Some lighting control may be unavailable Associated lights fail to a default configuration	MEL

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HA-420 AFM

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Page 3B-26

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Honda Aircraft Company

HA-420 AFM

NORMAL

SECTION 4 NORMAL PROCEDURES

TABLE OF CONTENTS

PREFLIGHT INSPECTION.....	4-3
BEFORE STARTING ENGINES.....	4-15
STARTING ENGINES	4-19
BEFORE TAXI.....	4-21
TAXI.....	4-22
BEFORE TAKEOFF	4-23
TAKEOFF.....	4-24
AFTER TAKEOFF	4-25
CLIMB	4-25
CRUISE.....	4-26
DESCENT	4-26
APPROACH	4-27
BEFORE LANDING.....	4-28
LANDING	4-28
BALKED LANDING	4-29
AFTER LANDING.....	4-30
SHUTDOWN	4-31

Honda Aircraft Company

HA-420 AFM

NORMAL

CROSSWIND PROCEDURES	4-32
COLD WEATHER OPERATIONS	4-37
OPERATING IN ICING CONDITIONS.....	4-43
OXYGEN DURATION TABLES	4-57

Honda Aircraft Company

HA-420 AFM

NORMAL

NORMAL PROCEDURES

GENERAL

The operating procedures contained in this section are recommended by Honda Aircraft Company for use in operating the airplane. These procedures are provided for guidance and should not be construed as prohibiting the owner/operator from developing equivalent procedures.

PREFLIGHT INSPECTION

PRELIMINARY EXTERIOR INSPECTION

1. Wheel Chocks.....As required
2. Engine Covers Removed
3. Static Wick Covers Removed
4. Pitot Covers Removed
5. External Rudder Lock Removed
6. Aircraft Tie-Downs Removed
7. External Power Unit Connect, as required

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HA-420 AFM

NORMAL

COCKPIT/CABIN INSPECTION

1. Emergency Exit Secure
2. Emergency Exit Handle Locking Mechanism..... Removed
3. Emergency Equipment Checked

NOTE Emergency equipment should be checked to ensure it is properly located, not damaged, and is serviceable.

4. Gust Lock Removed
5. Circuit Breakers..... Check
6. LANDING GEAR Handle DN
7. BATTERY..... ON

NOTE The avionics are not fully initialized until approximately 60 seconds after selecting the Battery to ON. Indications may be invalid or erroneous during that time.

8. Databases..... Verify current

CAUTION *Do not use outdated database information. Pilots using any outdated database do so entirely at their own risk.*

9. Fuel Quantity Check
10. PARKING BRAKE..... Set

Honda Aircraft Company

HA-420 AFM

NORMAL

COCKPIT/CABIN INSPECTION (continued)

11. Engine Oil Quantity..... Check

NOTE

If the oil quantity indicates LOW, the aircraft may dispatch if the oil quantity is visually checked to be above the ADD line on the affected engine's sight glass. If the oil quantity is low, add oil prior to engine operation. The minimum acceptable oil quantity as indicated by the ADD line is 5319 cc (5.62 quarts), and the maximum quantity as indicated by the FULL line is 5775 cc (6.10 quarts).

12. Hydraulic Reservoir Quantity..... Check

NOTE

Nominal hydraulic reservoir quantity is between 75 and 100%.

13. Oxygen Quantity Adequate for intended flight

14. Trims Set for takeoff

15. FLAPS UP

16. BATTERY..... As required

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION

NOTE

Prior to conducting flight operation in RVSM airspace, special emphasis should be placed on the condition of Air Data Probes. If any visible damage, deformation, obstruction or static port irregularities are noted, the airplane is no longer RVSM compliant until the discrepancy is corrected.

1. Cabin Door (Checked)
 - a. Door Seal..... Check condition
 - b. Upper fuselage antennas..... Check condition
2. Left Nose (Checked)
 - a. Windshield and Side Window Check condition
 - b. Left Air Data Probe Check condition
 - c. Ice Detector Check condition
 - d. Luggage Secured properly
 - e. Baggage Door..... Secure and locked
 - f. Oxygen Blowout Disk..... Green
 - g. Nosegear Doors Check condition
 - h. Nosewheel, Tire, and Strut Check condition
 - i. Nosewheel Steering Pin..... Installed

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

3. Right Nose (Checked)

- a. Standby Air Data Probe..... Check condition
- b. Right Air Data Probe..... Check condition
- c. Ice Detector Check condition
- d. Windshield and Side Window Check condition
- e. Lower Fuselage Antennas Check condition

4. Right Wing (Checked)

- a. Landing Light..... Check condition
- b. External Power Door Check condition and
Close if GPU not used
- c. Ice Light Check condition and operation
- d. Electrical Service Door ... Check condition and secure
- e. Air Inlet (belly panel) Check clear
- f. Center Tank Fuel Drains Drain and check (3)

NOTE

Fuel drain sumping is only required prior to the first flight of the day.

- g. Lower Fuselage Check condition
- h. Emergency Exit Check secure
- i. Engine Inlet Check condition
- j. T₁ Probe..... Check condition

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

- k. Engine Duct and Fan Check condition and clear
- l. Stall Strip Check condition
- m. Anti-ice Exhaust Holes Check clear
- n. Wing Tank Fwd Fuel Drain Drain and check

NOTE Fuel drain sumping is only required prior to the first flight of the day.

- o. Gear Doors Check condition
- p. Wheel, Tire, and Strut Check condition
- q. Upper Surface and Leading Edge Check condition
- r. Vortilon Checked
- s. Wing Bump Checked
- t. Wingtip Fence Checked
- u. Lower Surface Check condition
- v. Vortex Generators Check installed (14)
- w. Wingtip Light Assembly Check condition

NOTE Pay special attention to surface erosion or frosting of the lens as this may adversely affect the light pattern and brightness.

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

- x. Winglet..... Check condition
- y. Wingtip Triangles..... Check installed (18)
- z. Anti-ice Exhaust Vent Check clear
- aa. Static Wicks..... Check installed (3)
- bb. Aileron, Trim Tab, and Flap..... Check condition
- cc. Wing Tank Aft Fuel Drain Drain and check

NOTE Fuel drain sumping is only required prior to the first flight of the day.

5. Right Nacelle (Checked)

- a. Anti-Ice Exhaust Vent Check clear
- b. Oil Servicing Cap and Door Installed and secure
- c. Engine Drains Check clear
- d. Engine Exhaust..... Check clear
- e. Inboard Nacelle Access Door..... Closed
- f. Generator exhaust..... Check clear
- g. Fan Air Outlet..... Check clear
- h. Pylon NACA Scoop Check clear

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

6. Right Aft Fuselage (Checked)

- a. Wheel, Tire, and Strut Check condition

NOTE

The amount of exposed chrome on the landing gear strut should not be more than 1.5 inches difference between the left and right strut when the aircraft is on level ground with a balanced fuel load. An exposed chrome difference of more than 1.5 inches in this case may indicate an improperly serviced strut.

- b. Brake Check wear pins
- c. Beacon Check condition
- d. Fuel Vent Check clear
- e. Hydraulic Panel Check closed
- f. Lav Service Door Check closed
- g. ECS Heat Exchanger Exhaust Check clear
- h. Ram Air Intake Check clear
- i. Fuel Cap Installed and secure

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

7. Empennage (Checked)

- a. Vertical Tail Surface Check condition
- b. VOR Antenna Check condition
- c. Rudder, Trim Tab, and T-Strips Check condition
- d. Rudder Static Wicks Check installed (3)
- e. Position Light Checked
- f. Speedbrake Panels (if installed) Check condition
- g. Horizontal Tail Surface Check condition
- h. Logo Lights Checked
- i. Elevator, Cuffs, Trim Tabs,
and T-Strips Check condition
- j. Elevator Static Wicks Check installed (4)
- k. Vortex Generators Check installed (42)

8. Left Aft Fuselage (Checked)

- a. Environmental Air Intake Check clear
- b. Aft Fuselage Antennas Check condition
- c. Ram Air Intake Check clear
- d. Aft Baggage Compartment Properly loaded
and secure

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

- e. Aft Baggage DoorClosed and locked
- f. Fuel Vent Check clear
- g. Wheel, Tire, and Strut Check condition

NOTE

The amount of exposed chrome on the landing gear strut should not be more than 1.5 inches difference between the left and right strut when the aircraft is on level ground with a balanced fuel load. An exposed chrome difference of more than 1.5 inches in this case may indicate an improperly serviced strut.

- h. BrakeCheck wear pins
9. Left Nacelle (Checked)
- a. Fan Air Outlet..... Check clear
 - b. Pylon NACA Scoop Check clear
 - c. Generator Exhaust Check clear
 - d. Inboard Nacelle Access Door Closed
 - e. Engine Exhaust..... Check clear
 - f. Engine Drains Check clear
 - g. Oil Servicing Cap and DoorInstalled and secure
 - h. Anti-Ice Exhaust Vent Check clear

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

10. Left Wing (Checked)

- a. Wing Tank Aft Fuel Drain Drain and check

NOTE Fuel drain sumping is only required prior to the first flight of the day.

- b. Aileron, Trim Tab, and Flaps Check condition
- c. Static Wicks..... Check installed (3)
- d. Anti-ice Exhaust vent Check clear
- e. Winglet Check condition
- f. Wingtip Triangles..... Check installed (18)
- g. Wingtip Light Assembly Check condition

NOTE Pay special attention to surface erosion or frosting of the lens as this may adversely affect the light pattern and brightness.

- h. Vortex Generators Check installed (14)
- i. Lower Surface Check condition
- j. Wingtip Fence Checked
- k. Wing Bump Checked
- l. Vortilon Checked

Honda Aircraft Company

HA-420 AFM

NORMAL

EXTERIOR INSPECTION (continued)

- m. Upper Surface and Leading Edge..... Check condition
- n. Engine Inlet Check condition
- o. T₁ Probe..... Check condition
- p. Engine Duct and Fan Check condition
- q. Stall Strip..... Check condition
- r. Anti-ice Exhaust Holes..... Check condition
- s. Wing Tank Fwd Fuel Drain..... Drain and check

NOTE Fuel drain sumping is only required prior to the first flight of the day.

- t. Gear Doors Check condition
- u. Wheel, Tire, and Strut Check condition
- v. Ice Light Check condition and operation
- w. Electrical Service Door..... Check closed
- x. Landing Light..... Check condition

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE STARTING ENGINES

1. BATTERY.....ON
2. OXYGEN (Checked, ON and NORM)
 - a. OXYGEN SUPPLYPush ON
 - b. Oxygen MasksSet 100%
 - c. OXY MASK AUDIO EMER
 - d. Oxygen box TEST/RESETDepress and hold
verify blinker turns
yellow momentarily,
then turns black
 - e. Oxygen Mask PRESS TO TEST Simultaneously
Depress for
approximately 3 sec.,
verify blinker turns yellow
momentarily,
confirm speaker operation
by listening for oxygen flow
over speaker
 - f. Oxygen box TEST/RESETRelease
 - g. OXY MASK AUDIO NORM
 - h. CABIN OXYGEN..... NORM
3. PRESSURIZATION (NORM)
 - a. CONTROL MODE NORM
 - b. CABIN DUMP NORM

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE STARTING ENGINES (continued)

4. ELECTRICAL (ON and NORM)
 - a. BATTERY..... Verify ON
 - b. GENERATOR.....Both NORM
 - c. BUS TIE Normal (dark)
 - d. CABIN POWER..... Normal (dark)
5. ELT..... NORM
6. NOSE WHEEL STEERING NORM
7. LANDING GEAR..... DN
8. ALTERNATE GEAR RELEASE Handle.....Stowed (fully in)
9. PARKING BRAKE..... Set
10. FLAPS UP
11. Thrust Levers..... CUT OFF
12. SPEEDBRAKE (if installed).....RET
13. ICE PROTECTION (NORM and OFF)
 - a. WING ANTI-ICE..... NORM
 - b. WING FLOW NORM
 - c. ENGINE ANTI-ICE..... Both OFF
 - d. TAIL DE-ICE..... NORM
14. FUEL PANEL (NORM)
 - a. PUMPs.....Both NORM
 - b. CROSSFEED NORM

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE STARTING ENGINES (continued)

15. TRIM PANEL (Checked, set and NORM)
 - a. PITCH MODE STBY (check operation), then NORM
 - b. ROLL TRIM POWER..... NORM
 - c. YAW TRIM POWER NORM
 - d. Trims Checked and set for takeoff
16. WINDSHIELD HEAT Panel NORM
17. PNEUMATIC Panel (NORM)
 - a. CABIN INFLOWBoth NORM
 - b. ENGINE BLEEDBoth NORM
18. Glareshield Panels (Dark)
 - a. Pilot's DISPLAY REVERSION Normal (dark)
 - b. Pilot's CHIME..... Normal (dark)
 - c. AFCS PITCH SERVO POWER Normal (dark)
 - d. AFCS ROLL SERVO POWER Normal (dark)
 - e. AFCS YAW SERVO POWER Normal (dark)
 - f. Copilot's DISPLAY REVERSION..... Normal (dark)
 - g. Copilot's CHIME Normal (dark)
19. Standby Instrument..... Checked

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE STARTING ENGINES (continued)

20. AVIONICS INITIALIZATION (Complete)

- a. SYSTEM TESTSComplete
 - i. Fire Detection and Suppression.....Press
 - Verify both FIRE lights illuminate accompanied by FIRE aural alert
 - ii. Stall Warning and ProtectionPress
 - Verify shaker activates twice
 - iii. SwitchlightsPress
 - On GPU power, verify all switchlights are illuminated
 - On battery power, verify all switchlights illuminate, except AFCS Servo Power Panel and PUSH TO DISC switchlights.
 - iv. CVFDR.....Press
 - Verify tone can be heard in headset for 5 seconds

NOTE

When operating on battery power, the CVFDR is unpowered and the system test will fail. If necessary, the system can be retested once engine generator or GPU power is available.

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE STARTING ENGINES (continued)

- b. WEIGHT AND FUEL.....Entered
- c. FLIGHT PLANEntered
- d. V-speedsEntered
- e. ADS-B informationEntered

STARTING ENGINES

- 1. Passenger Briefing.....Complete
- 2. Rudder Pedals.....Adjusted
- 3. Seats, Safety Belts Set
- 4. Doors Closed

NOTE The pilot should verify the doors are closed by review of the CAS messages or synoptic page, and by verifying the main entry door indicators are green.

- 5. PARKING BRAKE..... Set
- 6. CAS Messages..... Review
- 7. ELEC VOLTS 23.5 Volts minimum

NOTE If the batteries have been cold-soaked as defined in Cold Weather Operations (Section 4 – Normal Procedures), engine start is permitted to 22.5 volts.

Honda Aircraft Company

HA-420 AFM

NORMAL

STARTING ENGINES (continued)

8. Engines (Start)

- a. ENGINE START (selected engine)Press
- b. Thrust Lever IDLE
- c. Engine IndicationsMONITOR
 - i. ITT Verify rise within 15 seconds
 - ii. Verify starter disengages at 45% N₂

NOTE

The FADEC will automatically terminate an abnormal start but it cannot ensure that ITT start limits (556 °C) are not exceeded. The Pilot must monitor ITT during start and ensure ITT remains within limits.

The FADEC will abort starts on the ground for the following conditions:

- No increase in ITT within 15 seconds after fuel is introduced
- No N₁ rotation
- N₂ not steadily increasing to Idle
- N₂ fails to reach 95% of Idle within 76 seconds after light off

Honda Aircraft Company

HA-420 AFM

NORMAL

STARTING ENGINES (continued)

- 9. Engine Instruments..... Check
- 10. ENGINE ANTI-ICE.....As required
- 11. External Power Disconnect
- 12. Flight Controls..... Free and correct

BEFORE TAXI

- 1. WING ANTI-ICE.....As required

NOTE Monitor ITT during operation of wing anti-ice.

- 2. Avionics / Flight Data (Set/Entered)
 - a. Transponder CodeEntered
 - b. Flight ID Confirmed
 - c. Flight PlanEntered
 - d. Runway.....Selected
 - e. Navigation source..... Set
- 3. AltimetersSet BARO, verify
altitude within 75 feet of
the field elevation and agree
within 75 feet of each other

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE TAXI (continued)

4. FLAPS Set for takeoff
5. SPEEDBRAKE (if installed).....Checked then RET
6. CABIN SIGNSAs required
7. PARKING BRAKE.....Release

TAXI

1. Brakes Check
2. Nosewheel Steering Check
3. Flight Instruments..... Check

NOTE

The recognition and landing lights may be used to supplement the taxi lights as needed in poorly lit and/or wet taxiway conditions.

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE TAKEOFF

1. FLAPS Set for takeoff
2. Trims Set for takeoff
3. SPEEDBRAKE (if installed).....RET
4. CAS Messages..... Review
5. Navigation Set
6. Flight Guidance Set
7. TOLD Data..... Confirmed

NOTE Crew must confirm that appropriate V-speeds are still posted. V-speeds may become deselected and revert back to default values after performing engine starts on battery power.

NOTE The takeoff field length provided in the performance section of the flight manual is based on a dry runway. If departing from a wet runway, it is recommended to increase the predicted takeoff field length by 30%.

8. Takeoff Briefing Complete
9. Radar As required

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE TAKEOFF (continued)

10. ICE PROTECTION (As required)

- a. WING ANTI-ICE.....As required

NOTE

Monitor ITT during operation of wing anti-ice.

- b. ENGINE ANTI-ICE.....As required

TAKEOFF

- 1. Thrust Levers..... TO
- 2. Engine Instruments..... Verify that N_1 pointers match N_1 target settings within 1%.
- 3. Brakes..... Release
- 4. Nose-up Pitch Attitude at Rotation..... 13° (Flaps UP)
 12° (Flaps TO/APPR)

Honda Aircraft Company

HA-420 AFM

NORMAL

AFTER TAKEOFF

1. LANDING GEARUP (positive rate of climb)
2. Yaw Damper.....Verify engaged
3. FLAPSUP (130 KIAS minimum)
4. Thrust Levers.....MCT
5. Flight GuidanceAs required
6. ICE PROTECTION (As required)
 - a. WING ANTI-ICE..... NORM
 - b. ENGINE ANTI-ICE.....As required
7. CABIN SIGNSAs required

CLIMB

1. Pressurization Checked
2. Altimeters (transition altitude)Set STD

PRIOR TO ENTERING RVSM AIRSPACE

3. Flight Guidance CPL to pilot flying,
select desired modes
4. Autopilot..... Engage

Honda Aircraft Company

HA-420 AFM

NORMAL

CRUISE

WHEN IN RVSM AIRSPACE, ONCE PER HOUR

1. Altimeters / AltitudeSet / Agree within 200 ft

DURING ALL PHASES OF CRUISE FLIGHT

2. Systems..... Monitor

DESCENT

1. Landing Field Elevation Verify set
2. ENGINE ANTI-ICEAs required
3. Altimeters (transition altitude) Set

Honda Aircraft Company

HA-420 AFM

NORMAL

APPROACH

1. Seats and Seat Belts..... Adjusted and secure
2. Passenger Briefing..... Complete
3. CABIN SIGNS As required
4. Avionics..... Set
5. Landing Data Set and confirmed
 - a. Radios and Navigation Set
 - b. V-speeds, FMS, and Flight Guidance..... Set,
programmed,
and modes selected
 - c. Landing Distance..... Confirm

NOTE

The landing distance provided in the performance section of the flight manual is based on a dry runway. If landing on a wet runway, it is recommended to increase the predicted landing distance by 30%.

6. CAS Messages..... Check
7. Approach Briefing..... Complete
8. FLAPS TO/APPR

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE LANDING

1. LANDING GEAR..... DN
2. SPEEDBRAKE (if installed).....RET
3. FLAPS LDG or TO/APPR (ice)

CAUTION

Do not extend the flaps to LDG unless the airframe can be confirmed free of ice, and icing conditions are not expected during approach and landing.

NOTE

If TAWS-A is installed, the TAWS Warnings will annunciate when landing with Flaps at TO/APPR, unless Flap Override is selected.

4. Airspeed..... V_{REF}
5. Autopilot/Yaw Damper Disengage

LANDING

1. Thrust Levers..... IDLE
2. Brakes.....Apply (after touchdown)

NOTE

Establish directional control using rudder and then apply brakes symmetrically during the initial part of the landing rollout.

3. SPEEDBRAKE (if installed)..... EXT

Honda Aircraft Company

HA-420 AFM

NORMAL

BALKED LANDING

1. TO/GA.....Press
2. Thrust Levers..... TO
3. Airspeed.....Maintain V_{REF}

WHEN CLEAR OF OBSTACLES

4. FLAPS TO/APPR
5. LANDING GEARUP (positive rate of climb)
6. Yaw Damper..... Engaged
7. Airspeed.....Maintain V_{AC}
8. FLAPSUP (130 KIAS minimum) or TO/APPR (ice)

CAUTION

Retracting the flaps to UP following an icing encounter with the flaps extended may result in damage to the flaps or airframe.

9. Thrust Levers.....MCT
10. Flight GuidanceAs required

Honda Aircraft Company

HA-420 AFM

NORMAL

AFTER LANDING

1. CABIN SIGNSAs required
2. ENGINE ANTI-ICEAs required
3. SPEEDBRAKE (if installed).....RET
4. FLAPS UP or TO/APPR (ice)

CAUTION

Retracting the flaps to UP following an icing encounter with the flaps extended may result in damage to the flaps or airframe. The flaps must be inspected, and any residual ice must be removed before retracting the flaps to UP.

5. Trims Set in green band

NOTE

The recognition and landing lights may be used to supplement the taxi lights as needed in poorly lit and/or wet taxiway conditions or when ice has accumulated on the wing taxi light. Reduce taxi speed as required.

Honda Aircraft Company

HA-420 AFM

NORMAL

SHUTDOWN

1. PARKING BRAKE..... Set
2. Wheel Chocks.....Installed
3. ENGINE ANTI-ICE..... OFF
4. External Power As desired
5. Thrust Levers..... CUT OFF

NOTE

Operate the engine at idle thrust for a minimum of 2 minutes before shutdown to thermally stabilize the engine hot section. Time of operation at or near idle such as taxiing can be considered idle thrust.

6. CABIN SIGNSAs required
7. OXYGEN SUPPLY Pull off
8. PARKING BRAKE.....Release
9. BATTERY..... OFF

Honda Aircraft Company

HA-420 AFM

NORMAL

CROSSWIND PROCEDURES

GENERAL

These procedures are applicable for crosswind takeoff and landing with or without the optional speedbrake.

CAUTION

Takeoffs and landings on contaminated runways can have significant additional risks due to the varied surface conditions likely to be encountered.

CAUTION

For takeoffs and landings on wet or icy runways, the maximum crosswind capability may be significantly reduced due to the reduced steering authority contributed by the nosewheel.

CAUTION

Operations with any tailwind component in conjunction with crosswinds, especially on contaminated runways, should be avoided due to the inherent hazard of operating on such runways.

CAUTION

Large and prompt aileron and rudder pedal inputs may be required in crosswind conditions close to or exceeding the maximum allowable crosswind or in gusty conditions.

Honda Aircraft Company

HA-420 AFM

NORMAL

CROSSWIND TAKEOFF PROCEDURE

The crosswind technique for takeoff requires that the pilot hold aileron into the wind to maintain wings level. Maintain nosewheel contact with the runway throughout the takeoff roll. Do not over control in aileron since it can result in heading deviations due to induced lift differences between the upwind and downwind wing. Rudder pedals must be used to maintain directional control.

The directional authority available through the rudder pedals is a function of groundspeed, airspeed, runway condition (wet, icy, etc.) and the magnitude of the crosswind.

CROSSWIND LANDING PROCEDURE

WARNING

The bank angle must be limited to 10 degrees in order to ensure wingtip clearance from the runway surface.

CAUTION

Not following the procedures provided in this section may result in large centerline deviations.

Honda Aircraft Company

HA-420 AFM

NORMAL

CROSSWIND LANDING PROCEDURE (continued)

The final approach phase of the crosswind landing should be flown wings-level, utilizing a crab angle into the wind sufficient to stabilize the aircraft path along the extended runway center line.

WARNING

For landings with a crosswind greater than 15 knots, the approach must be made using a crab technique. A landing in these conditions using a wing low method could result in a wingtip strike.

Avoid excessive airspeed during the final approach. Similar to a performance landing, minimum flare should be used. An extended flare will make drift correction more difficult and will increase pilot workload.

Just prior to touchdown, perform the de-crab maneuver by applying rudder to align the aircraft fuselage with the runway. Simultaneously apply aileron into the wind to minimize drift. Attempt to completely align the aircraft with the center line prior to touchdown. If drift cannot be controlled, a go-around should be performed.

Touch down with the upwind landing gear first, then set the other main wheel on the runway without delay followed by a prompt de-rotation.

Honda Aircraft Company

HA-420 AFM

NORMAL

CROSSWIND LANDING PROCEDURE (continued)

Upwind aileron should be applied immediately after touchdown to ensure the upwind landing gear remains firmly in contact with the surface. Directional control should be maintained by applying rudder to track the center line. Apply slight forward pressure on the yoke to ensure nosewheel contact with the runway. Symmetrical braking and speedbrake (if installed) should be applied as required to decelerate.

NOTE Lateral control during the ground roll has been shown to be relatively ineffective in countering wing rocking motions, which may occur after touchdown of one main gear prior to the opposite main gear. Use steady, upwind aileron input to maintain main gear firmly in contact with the ground rather than attempting a counter rocking motion.

NOTE Approximately 2 seconds after nosewheel has registered weight-on-wheels, the nosewheel steering becomes active, and the steering will then move to the position being commanded by the rudder pedals. This transition may introduce a yawing response, which should be promptly countered using rudder inputs.

NOTE Rudder and nosewheel steering are significantly more effective in maintaining directional control than differential braking. In addition, the amount of available differential braking may be reduced if anti-skid is active.

Honda Aircraft Company

HA-420 AFM

NORMAL

CROSSWIND LANDING PROCEDURE (continued)

NOTE

The contribution to directional stability from the main wheels is reduced with increased braking. If directional control is in question, release the brakes and apply rudder as required to correct. Once directional control has been re-established, apply symmetrical braking.

Honda Aircraft Company

HA-420 AFM

NORMAL

COLD WEATHER OPERATIONS

When operating in cold weather conditions, adherence to the procedures listed below in addition to the normal procedures, is recommended.

COLD SOAK

Cold soak is defined as a continuous period of longer than two hours at an outside air temperature below 0 °C in which the aircraft is not powered. Prior to aircraft cold-soak, perform the following:

General

- Retract flaps (after confirming free of ice and snow).
- Chock wheels. Release parking brake.
- Install engine, ice detector and air data probe covers.
- Remove beverage containers.
- Drain water system.
- Drain waste system.

Below -15 °C

- Remove crew oxygen masks and store at room temperature.

NOTE

Masks may be left in the aircraft if OXYGEN SUPPLY is left in the OFF position until the cockpit temperature is at or above -15 °C.

Honda Aircraft Company

HA-420 AFM

NORMAL

COLD SOAK (continued)

Below -15 °C (continued)

- Remove hand-held microphones and store at room temperature.

NOTE

Microphones may be left in the aircraft if not used until the cockpit temperature is at or above -15 °C.

Below -20 °C

- Remove both batteries and store at room temperature.

PREFLIGHT INSPECTION

During pre-flight inspection following a cold-soak, add or pay special attention to the following items.

Aircraft Preparation

- Install batteries (if removed).
- Service water system.
- Service waste system.

NOTE

If the water or waste system is serviced in below freezing temperatures, the cabin heating system must be operating within 30 minutes to prevent fluids from freezing.

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HA-420 AFM

NORMAL

PREFLIGHT INSPECTION (continued)

- Install crew oxygen masks (if removed).
- Install hand-held microphones (if removed).

Exterior Inspection

- Carefully inspect all surfaces of the airframe to ensure that aircraft critical areas are free of any ice, snow, slush, or frost. See ICING LIMITATIONS (Section 2 - Limitations) for definition of critical areas.
- Remove snow, slush and ice from landing gear struts, brakes, switches, doors and wheel wells.

NOTE Carefully inspect forward facing parts of the nose landing gear and door. Contamination of the nose landing gear and door may prevent successful gear retraction.

- Carefully inspect engine inlet, inlet lip, fan, inner fan case abradable liner (wear), spinner, and exhaust duct to ensure all ice and / or snow is removed. Rotate fan to ensure it rotates freely.
- Carefully inspect for any fluid leaks.
- Carefully inspect all intakes and vents for blockage.

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HA-420 AFM

NORMAL

BEFORE STARTING ENGINES

Cockpit displays may take up to one minute to power up. Colors may be faded, touch-screen response may be slower and dynamic items may appear blurred until the displays are fully warmed up.

NOTE

A “CHECKLIST FILE INVALID” CDU message may post. The checklist will be functional after the message has been acknowledged and all displays are powered.

Confirm seat adjustments have latched since they may not automatically spring back to the latched position before the cabin has been heated up.

If the batteries were cold-soaked, the following procedure must be followed:

Before starting engines, warm up the batteries by operating on battery power (no external power) with BUS TIE OPEN for a minimum of 25 minutes or when bus voltage drops to 22.5 V, whichever comes first.

The **FUEL TEMP LOW** message may post if the aircraft has been cold-soaked close to or below -40 °C. Refueling with warmer fuel or warming the aircraft in a hangar is recommended. Operating the engines for an extended period of time may also increase fuel temperature enough to clear the message.

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HA-420 AFM

NORMAL

STARTING ENGINES

Engine starts will be slower than normal and higher than normal oil pressure can be expected.

If the oil temperature is below the start limitation of -40°C , the engines should be preheated using warm air prior to start.

NOTE Oil pressure indications are dashed when the oil temperature is below -40°C .

Oil pressure may reach the maximum indication but should subside as oil temperature increases. If oil pressure remains above normal range after oil temperature stabilizes, the engine should be shut down and the cause investigated.

Delay selection of engine anti-ice to ON until 15 seconds after reaching IDLE thrust.

WARNING Do not use external heaters to heat the fuel.

Operate each trim axis to ensure proper operation, and perform a thorough, full deflection flight control sweep.

NOTE Flight control forces will be higher than usual due to increased friction.

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HA-420 AFM

NORMAL

BEFORE TAXI

Do not operate engines at MCT or above until oil temperature is at or above 10 °C.

Cockpit and Cabin temperature must be at or above -15 °C (5 °F) before dispatch. Operating both engines above IDLE thrust (at least 45 psi manifold pressure is recommended) will reduce time required to warm the cabin.

Cabin delta P of up to 0.5 psi may be indicated with the cabin unpressurized before the cabin has been heated.

TAXI

If taxiways are covered by ice, snow or slush, taxi at lower than usual speeds and allow greater distance for stopping. Avoid taxiing closely behind aircraft with running engines. Use the brakes frequently and taxi with flaps retracted. Monitor wingtips for clearance to snow banks.

TAKEOFF

Consider the effect of any runway contaminants on the takeoff distance.

Delay landing gear retraction after taking off from a snow or slush covered runway.

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HA-420 AFM

NORMAL

OPERATING IN ICING CONDITIONS

Adherence to the procedures listed below in addition to the normal procedures and limitations are required when operating in conditions susceptible to airframe icing.

ICING CONDITIONS

The aircraft has been approved to operate within the icing envelope defined by 14 CFR Part 25 Appendix C. Icing conditions that are outside this envelope may be encountered during flight operations. The ice detecting system detects those conditions requiring automatic activation of the ice protection systems. Some conditions that may result in a very light ice accretion on the airframe do not require ice protection and, therefore, will not be annunciated. Other conditions, such as freezing rain or freezing drizzle, may exceed the capabilities of the ice protection systems and can have a severe impact on aircraft handling characteristics and performance.

WARNING

The autopilot and CSC may not maintain airspeed if ice accretes on the airplane. Monitor airspeed closely.

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HA-420 AFM

NORMAL

ICING CONDITIONS (continued)

WARNING

If unusually extensive ice accretes on the wing or engine nacelle aft of the protected areas, or if unusual lateral trim requirements or autopilot trim warnings are encountered, or if ice accretes on the wing or engine protected areas, accomplish the following:

- If the flaps are extended, do not retract them until the airframe is verified clear of ice.
- Reduce the angle-of-attack by increasing speed as much as the airplane configuration and weather allow, without exceeding design maneuvering speed.
- If the autopilot is engaged, hold the control wheel firmly and disengage the autopilot. Do not re-engage the autopilot until the airframe is verified clear of ice.
- Exit the icing area immediately by changing altitude or course; and
- Report these weather conditions to air traffic control.

CAUTION

Flight in freezing rain or freezing drizzle, may exceed the capability of the ice protection system and result in hazardous ice build-up on protected surfaces, or may result in unusually extensive ice formation aft of the protected area. This ice may not be shed using the ice protection systems, and it may seriously degrade the performance and controllability of the airplane.

Honda Aircraft Company

HA-420 AFM

NORMAL

ICING CONDITIONS (continued)

NOTE

The prohibition on flight in freezing rain or freezing drizzle is not intended to prohibit purely inadvertent encounters with the specified meteorological conditions; however, pilots should make all reasonable efforts to avoid such encounters and must immediately exit the conditions if they are encountered.

Ice may form in freezing rain or freezing drizzle at temperatures as cold as -18 °C (0 °F), increased vigilance is warranted at temperatures around freezing with visible moisture present. In addition to the cues for determining icing conditions beyond the approved envelope described in ICING LIMITATIONS (Section 2 – Limitations) the following may be used to identify possible freezing rain or freezing drizzle:

- Visible rain when SAT is 5 °C or less.
- Droplets that splash or splatter on impact when SAT is 5 °C or less.

If freezing rain or freezing drizzle conditions are encountered, exit icing (by changing altitude and/or course) conditions immediately. Asking for priority to leave the area is fully justified under these conditions. The following actions are also recommended:

- Avoid abrupt and excessive maneuvering that may exacerbate control difficulties.
- Do not engage the autopilot, as it may mask unusual control system forces. If the autopilot is already engaged, hold the control wheel firmly and disengage.
- If an unusual roll response or uncommanded control movement is observed, reduce the angle-of-attack by increasing airspeed or rolling wings level, and apply additional power, if needed.

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HA-420 AFM

NORMAL

ICING CONDITIONS (continued)

- Avoid extending flaps. If the flaps are extended, do not retract them until the airframe is clear of ice.
- If erratic air data indications are observed, fly using known aircraft attitude and thrust settings, crosscheck GPS speed and altitude indications, etc., to maintain basic aircraft control.
- Report the weather conditions to ATC.

ICE DETECTION

While ice detectors are the primary means to identify icing conditions in-flight they are not capable of reliably detecting icing conditions during ramp and taxi operations. A combination of outside air temperature and atmospheric conditions have to be used, as described in ICING LIMITATIONS (Section 2 – Limitations), to determine whether ground icing conditions exist.

PREFLIGHT PLANNING

If icing conditions are expected along the planned route certain pre-flight planning considerations must be applied.

NOTE

When wing anti-ice is activated (automatically or manually), there will be a reduction in aircraft performance. See performance tables in the Airplane Flight Manual (Section 5 – Performance) and either the Quick Reference Handbook (Volume 1, Performance Section) or the Pilot's Operating Manual (Section 2 – Flight Planning).

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HA-420 AFM

NORMAL

PREFLIGHT PLANNING (continued)

Takeoff

Operating the engine and wing anti-ice systems has a negative impact on aircraft acceleration and climb capability. Performance tables for Wing Anti-ice ON must be used if taking off in ground icing conditions and should be considered if the planned takeoff profile penetrates known or forecast icing conditions. Refer to TAKEOFF – ANTI-ICE FLUID PERFORMANCE ADDITIVES (Section 5 – Performance) if Type II or IV fluids are used.

Enroute

Operating the engine and wing anti-ice systems has a detrimental effect on climb performance, altitude capability and cruise performance. The impact on fuel burn when operating with engine and wing anti-ice protection systems active must be considered.

Approach and Landing

If icing conditions are expected along the route of flight, landing performance planning must be based on using TO/APPR flaps and increased icing speeds. Reduced crosswind capability must also be considered. Additionally, operating the engine and wing anti-ice systems has a negative impact on go-around and climb capability in the event of a missed approach or balked landing.

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HA-420 AFM

NORMAL

PREFLIGHT PLANNING (continued)

Alternate

If icing conditions are expected at the destination airport it may not be possible to safely retract flaps to UP in case of a missed approach. The reduced approach climb capability must be considered, as well as the increased fuel burn to the alternate destination in this configuration.

NOTE See performance tables in the Airplane Flight Manual (Section 5 – Performance) and either the Quick Reference Handbook (Volume 1, Performance Section) or the Pilot's Operating Manual (Section 2 – Flight Planning).

EXTERIOR INSPECTION

The wing ice inspection lights must be operable prior to flight into known or forecast icing at night.

BEFORE TAXI

If wing anti-ice is required for takeoff, select it on after completion of the automatic pre-flight test to ensure adequate warm-up prior to takeoff. Up to 5 minutes of operation at idle power may be required.

NOTE Monitor ITT during operation of wing anti-ice.

If operating in ground icing conditions, a visual and tactile check of the wing leading edge and upper surfaces must be performed as defined in ICING LIMITATIONS (Section 2 – Limitations).

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HA-420 AFM

NORMAL

BEFORE TAXI (continued)

Ground De-icing

NOTE For instructions on how to de-ice the aircraft, see the Pilot's Operating Manual (Section 4 - Service, Handling, and Maintenance).

Following application of the de-icing fluid, perform a slow, full deflection flight control sweep to prevent pooling of fluid in flight control cavities.

If the aircraft has been de-iced, a de-icing inspection must be performed immediately following or during the ground de-icing/anti-icing process. All items below must be confirmed free of snow, ice, and frost accumulation.

- Wing, including leading edges, upper and lower surfaces and aileron control surfaces
- Flaps
- Vertical and horizontal stabilizers, including leading edges, horns, upper and lower surfaces, rudder, elevator and side panels
- Engine inlet and exhaust and nacelle precooler exhaust
- Windshield
- Antennas
- Fuselage
- Air data probes
- Ice Detectors

Honda Aircraft Company

HA-420 AFM

NORMAL

BEFORE TAXI (continued)

- Fuel tank vents
- Inlets and exhausts
- Landing gear, gear doors, and wheels and brake assemblies

A final inspection should be performed prior to takeoff to ensure effectiveness of the de-icing and/or anti-icing procedure. The inspection should be conducted within 5 minutes of takeoff and may be conducted from inside the airplane. If a visual inspection is not sufficient to determine whether ice is adhering, perform a tactile check. Indications of loss of effectiveness of de-icing/anti-icing fluid or contamination on airplane surfaces include the items that follow.

- Progressive surface freezing or snow accumulation
- Random snow accumulation
- Dulling of surface reflectivity (loss of gloss) caused by the gradual deterioration of the de-icing/anti-icing fluid to slush

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HA-420 AFM

NORMAL

TAXI

When operating in ground icing conditions, takeoff must be accomplished within 60 minutes of engine start. If an engine has been operated for more than 60 minutes in these conditions, the following ice shedding procedure must be performed prior to takeoff:

1. Over a span of 15 seconds, increase thrust from IDLE to 60% N_1 . Then over a span of 15 seconds, decrease thrust from 60% N_1 to IDLE.

NOTE

If airport surface conditions or the proximity of other aircraft do not permit the engine thrust to be increased to 60% N_1 , then use a thrust level as high as practical.

2. Repeat until exiting ground icing conditions or engine vibrations return to normal levels.
3. Shut down engine and inspect for ice accumulation.
4. Engines may be restarted if confirmed clear of ice accumulation.

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HA-420 AFM

NORMAL

TAKEOFF

WARNING

Do not take off with any ice, snow, slush, or frost (including polished) adhering to the aircraft critical areas. See ICING LIMITATIONS (Section 2 - Limitations) for definition of critical areas.

Wing anti-ice must be ON for takeoff in icing conditions. After takeoff, set the WING ANTI-ICE switch to NORM. Minimize time with flaps extended after takeoff. Accelerate to 180 KIAS as soon as practical.

NOTE

If the WING ANTI-ICE switch is not set to NORM, this may delay failure detection.

NOTE

Monitor ITT during operation of wing anti-ice.

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HA-420 AFM

NORMAL

ENROUTE

Minimize the duration of icing encounters as much as practical. Often a small change in altitude may be sufficient to exit icing conditions.

The requirement for maintaining a minimum thrust setting when in icing conditions may reduce the descent capability of the aircraft. Consider the reduction in descent and slow-down capability when planning a descent through potential icing conditions. Minimize time with thrust set below the steady state limit. Speedbrake (if installed) can be used to improve descent and slow-down capability.

NOTE Monitor ITT during operation of wing anti-ice.

NOTE Wing anti-ice must be manually selected on for flight in icing conditions above FL 340.

NOTE The outboard heating zone of each windshield may not remain clear in icing conditions.

The autopilot may mask tactile cues that indicate adverse changes in handling characteristics. When in icing conditions, consider not using the autopilot or disconnect it periodically to check for unusual control force or deflection, and to move the flight controls to check for evidence of ice accreting in control surface gaps or frozen actuators.

WARNING The autopilot and CSC may not maintain airspeed if ice accretes on the airplane. Monitor airspeed closely.

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HA-420 AFM

NORMAL

APPROACH AND LANDING

Minimum thrust setting when wing anti-ice is operating is 62% N_1 . Thrust settings as low as 50% N_1 are allowed if required for descent or deceleration, but these thrust reductions must be limited to less than 5 minutes.

NOTE

Thrust must be reset to 62% N_1 minimum prior to 5 minutes of operation at lower thrust setting to prevent **L-R WING A/I TEMP LOW** from posting and to prevent the stall warning ice advance from going to the failure schedule.

Use Flaps TO/APPR for landing whenever the aircraft cannot be confirmed clear of ice or if icing conditions may be encountered during approach and landing. Refer to Uncorrected Landing Field Length, Flaps TO/APPR - Icing (Section 5 – Performance).

CAUTION

Do not extend the flaps to LDG unless the airframe can be confirmed free of ice, and icing conditions are not expected during approach and landing.

NOTE

If TAWS-A is installed, the TAWS Warnings will annunciate when landing with Flaps at TO/APPR, unless Flap Override is selected.

Minimize time with thrust set below the minimum limit. Delay slow-down below 180 KIAS and flap and landing gear extension as long as possible, to minimize the build-up of ice on flaps and landing gear.

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HA-420 AFM

NORMAL

APPROACH AND LANDING (continued)

The stall warning ice advance must not be reset unless the airframe has been confirmed to be free of contamination. There is no visual indicator to confirm that the airframe is clean. Therefore an outside air temperature greater than 5 °C may be the only indication that the airframe is free of contamination.

BALKED LANDING

It may not be possible to safely retract flaps to UP in case of a missed approach, if icing conditions were encountered with the flaps extended. Climb performance capability and increased fuel burn during cruise to the alternate destination must be considered in this configuration.

CAUTION

Retracting the flaps to UP following an icing encounter with the flaps extended may result in damage to the flaps or airframe.

NOTE

See performance tables in the Airplane Flight Manual (Section 5 – Performance) and either the Quick Reference Handbook (Volume 1, Performance Section) or the Pilot's Operating Manual (Section 2 – Flight Planning).

Honda Aircraft Company

HA-420 AFM

NORMAL

AFTER LANDING

If icing was encountered with the flaps extended or after landing on snow or slush covered runway, do not retract flaps to UP until they can be verified to be free from ice and snow accumulation.

CAUTION

Retracting the flaps to UP following an icing encounter with the flaps extended may result in damage to the flaps or aircraft. The flaps must be inspected, and any residual ice must be removed before retracting the flaps to UP.

SHUTDOWN

If operating in ground icing conditions, perform the ice shedding procedure described in the preceding TAXI section prior to engine shutdown.

Following a flight with de-icing fluid applied, perform a slow, full deflection flight control sweep to prevent pooling of fluid in flight control cavities.

POSTFLIGHT INSPECTION

Carefully inspect engine inlet, inlet lip, fan, inner fan case abradable (wear), spinner, and exhaust duct to ensure all ice or snow is removed. Rotate fan to ensure it spins freely.

Inspect the airframe for signs of damage due to ice shedding.

Honda Aircraft Company

HA-420 AFM

NORMAL

OXYGEN DURATION TABLES

The following tables are provided for information and mission planning.

OXYGEN CONSUMPTION

The 50 cubic ft oxygen cylinder provides the following consumption for a pressurized aircraft. The usable oxygen quantity is 1,243 liters with a fully charged system.

NOTE

Once the lanyard for a cabin mask has been pulled that mask will continue to flow at the listed rate until oxygen is depleted. To preserve crew oxygen, flow to the cabin can be stopped by selecting the CABIN OXYGEN knob to OFF.

Honda Aircraft Company

HA-420 AFM

NORMAL

OXYGEN CONSUMPTION (LITERS/HOUR) – CRUISE FLIGHT

NOTE

If a passenger is seated in the Copilot seat, use data for Crew Mask for that passenger.

Table 4-1. Oxygen Consumption – Cruise Flight

Aircraft Altitude (Feet)	Each Crew Mask (Liters per Hour)		Each Cabin Mask (Liters per Hour)
	Normal	100%	
43,000	114	114	186
41,000	138	138	186
40,000	150	150	186
35,000	210	210	189
30,000	270	270	192
25,000	300	414	195
20,000	198	474	198
15,000	150	570	204
10,000	150	714	210

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HA-420 AFM

NORMAL

OXYGEN CONSUMPTION (LITERS) – EMERGENCY DESCENT

The following tables provide the amount of oxygen required to execute an emergency descent from 43,000 ft to the listed final altitude. Table 4-2 indicates oxygen consumption when the copilot seat is not occupied, and Table 4-3 indicates consumption when the copilot seat is occupied.

Table 4-2. Oxygen Consumption – If Copilot Seat Is Not Occupied

Final Altitude (Feet)	Number of Passengers in Cabin					
	0	1	2	3	4	5
25,000	25	38	50	63	76	88
20,000	36	53	68	87	103	120
15,000	53	75	95	118	139	159
10,000	79	106	130	161	186	212

Table 4-3. Oxygen Consumption – If Copilot Seat Is Occupied

Final Altitude (Feet)	Number of Passengers in Cabin					
	0	1	2	3	4	5
25,000	51	63	77	89	102	114
20,000	74	90	110	126	142	158
15,000	110	130	155	175	196	216
10,000	164	188	222	247	272	297

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HA-420 AFM

NORMAL

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HJ1-29000-003-001

Page 4-60

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

SECTION 5 PERFORMANCE

TABLE OF CONTENTS

GENERAL	5-3
PERFORMANCE DEFINITIONS.....	5-7
REFERENCE INFORMATION	5-15
ACOUSTIC LEVELS	5-25
CERTIFICATION NOISE LEVELS	5-25
GENERAL INFORMATION	5-26
TAKEOFF	5-36
TAKEOFF – ANTI-ICE FLUID PERFORMANCE ADDITIVES	5-38
TAKEOFF – FLAPS TO/APPR	5-39
TAKEOFF – FLAPS UP	5-107
OBSTACLE CLEARANCE.....	5-184
ENROUTE CLIMB	5-212
APPROACH AND LANDING.....	5-232
TURNAROUND.....	5-261

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-2

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

INTRODUCTION

GENERAL

This document presents the performance information for the HondaJet Model HA-420 airplane fitted with two Honda GE HF-120 engines. Information in this section specifies the conditions, configurations, and operating modes in which the appropriate performance criteria were calculated.

All performance data in this section are based on flight test data and the following performance conditions:

1. Minimum guaranteed thrust rating less installation, bleed air, and accessory losses.
2. Full temperature and altitude accountability within the operational limits for which the airplane has been certified.
3. Thrust settings (N_1) as indicated by the N_1 pointers.
4. All takeoff and landing performance is based on a paved, hard runway surface.
5. ICAO standard atmosphere conditions (ISA), with corrections for non-standard conditions, when applicable.
6. Wind speed is measured at a height of 10 meters (33 feet) above the surface.
7. The performance data have been calculated based on the following
 - a. A minimum takeoff safety speed (V_2) of $1.13 \times V_{SR1}$
 - b. A minimum landing reference speed (V_{REF}) of $1.23 \times V_{SR0}$

NOTE When necessary, interpolation may be used between table values to determine performance for the current ambient conditions and aircraft weight. If interpolation is not used, the next higher ambient temperature and weight should be used for the determination of aircraft performance.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Standard Operating Configurations

Flight Segment	Engines	Thrust	Flaps	Gear	Speed
1 st Segment Climb	1	TO	UP or TO/APPR	DN	V_{LOF} to V_2
2 nd Segment Climb	1	TO	UP or TO/APPR	UP	V_2
Final Segment/ Enroute Climb	1	MCT	UP	UP	140 KIAS
Approach Climb	1	TO	TO/APPR	UP	V_{AC}
Landing Climb/ Landing	2	TO	LDG	DN	V_{REF}
Landing Climb/ Landing (Icing)	2	TO	TO/APPR	DN	V_{REF}

Standard Performance Procedures

Normal All Engine Takeoff

1. Set the thrust levers to the takeoff position.
2. Once the engines have stabilized at takeoff power, release the brakes.
3. At V_R , rotate smoothly to a pitch attitude of 12° (Flaps TO/APPR) or 13° (Flaps UP).
4. Adjust the pitch attitude as required to achieve a minimum of V_2+10 to 15 KIAS at a height of 35 feet above the runway.
5. Retract the landing gear when a positive rate of climb is achieved.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Takeoff with One Engine Inoperative (Accelerate Go)

1. Follow the procedures for a Normal All Engine Takeoff.
2. The pilot recognizes an engine failure at or above V_1 .
3. Continue the acceleration to V_R .
4. At V_R , rotate smoothly to a pitch attitude of 10° (Flaps TO/APPR) or 11° (Flaps UP).
5. Adjust the pitch attitude as required to achieve a minimum of V_2 at a height of 35 feet above the runway.
6. Retract the landing gear when a positive rate of climb is achieved.

NOTE A momentary decrease in rate of climb may occur due to the landing gear transitioning to the up position.

Rejected Takeoff (Accelerate Stop)

1. Follow the procedures for a Normal All Engine Takeoff.
2. The pilot recognizes an engine failure or other malfunction prior to V_1 .
3. Reduce both thrust levers to IDLE.
4. Simultaneously initiate maximum braking.
5. Continue maximum braking until the airplane comes to a complete stop.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Landing

1. Prior to 50 feet above the runway, establish the airplane on approach at V_{REF} with the Flaps set as required for landing and gear down using thrust to maintain a 3 degree glideslope.
2. At 50 feet above the runway, rapidly reduce both thrust levers to IDLE.
3. Touchdown firmly with little or no flare.
4. Upon touchdown, promptly derotate to nose touchdown and then apply maximum braking.
5. Deploy speedbrakes (if installed).
6. Continue maximum braking until the airplane comes to a complete stop.

NOTE The distances provided in this AFM do not take credit for the decrease in stopping distance when deploying the speedbrakes.

Variable Factors Affecting Performance

Some performance data presented in this section exceeds the weight temperature/altitude limits of the aircraft. These data are provided for interpolation purposes only.

Details of variables affecting performance are given with tables and charts to which they apply. Conditions which relate to all performance computations are:

1. Cabin pressurization ON;
2. Effect of wind corrections have been applied per the applicable FAA regulations.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

PERFORMANCE DEFINITIONS

Term	Definition
1st Segment Climb	The OEI takeoff climb with the remaining engine at takeoff thrust and the landing gear extended from V_{LOF} to V_2 .
2nd Segment Climb	The OEI takeoff climb with the remaining engine at takeoff thrust and the landing gear retracted at V_2 .
Accelerate-Stop Distance	The horizontal distance from brake release to the point at which the airplane comes to a complete stop during a takeoff when the first action to stop (apply brakes) was initiated no later than V_1 .
Actual Landing Distance (ALD)	The distance from a point 50 feet above the runway surface to the point at which the airplane would come to a complete stop. This distance assumes a dry, hard surfaced runway.
AEO	All engines operating.
Approach Climb	The OEI climb in the approach configuration with the engine at takeoff thrust, the flaps in the approach position and the landing gear retracted at V_{AC} .
Balked Landing Climb	The AEO climb in the landing configuration with the engines at takeoff thrust, the flaps in the landing position and the landing gear down at V_{REF} .

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Term	Definition
Brake Energy	The portion of total airplane energy that is absorbed by the brakes during deceleration.
Buffet Boundary	The speed of buffet onset in maneuvering flight with flaps and gear up.
CAS	Calibrated airspeed. Indicated airspeed corrected for position and instrument error. KCAS is calibrated airspeed expressed in knots.
Demonstrated Crosswind	The 90° crosswind velocity component for which adequate control of the airplane was actually demonstrated during takeoff and landing dry runway certification flight testing.
Departure Airport Icing Conditions	Icing conditions at the departure airport are defined as a temperature of 5°C SAT or colder with visible moisture present, whether or not icing has been reported or forecasted.
Engine-Out Accelerate-Go Distance	The horizontal distance from brake release to the point at which the airplane is 35 feet above the runway surface during a takeoff when one engine is failed at the critical engine failure speed. Also referred to as OEI takeoff distance.
Enroute Climb	The OEI climb with the remaining engine at maximum continuous thrust, the flaps up and the landing gear retracted at 140 KIAS.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Term	Definition
Factored Landing Distance	The factored dry landing distance is equal to the actual landing distance increased by an operational factor.
Final Segment Climb	The OEI climb with the remaining engine at maximum continuous thrust, the flaps up and the landing gear retracted at 140 KIAS.
Gradient of Climb	The ratio of the change in height during a portion of the climb to the horizontal distance traveled in the same time interval.
Gross Climb Gradient	See gradient of climb.
IAS	Indicated airspeed. The airspeed indicator reading as installed in the airplane. The information in this manual is presented in terms of knots indicated airspeed (KIAS) unless otherwise stated and assumes zero instrument error.
Ice Protection On	Engine Anti-Ice on and Wing Anti-Ice on
ISA	International standard atmosphere.
M	Mach number. The ratio of true airspeed to the speed of sound.
M _I	Indicated Mach number. The Mach number reading as installed in the airplane. Zero instrument error is assumed.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Term	Definition
M_{MO}	Maximum operating Mach number. The Mach number that may not be deliberately exceeded in any flight condition.
Net Climb Gradient	The gross gradient of climb reduced by 0.8 % during the takeoff phase. The net climb gradient is used for calculating the takeoff flight path.
OAT	Outside air temperature. (OAT = SAT)
OEI	One engine inoperative.
Position Error Correction	A correction applied to indicated airspeed, Mach number and altitude to eliminate the effect of the location of the pressure sources on the instrument reading.
Pressure Altitude	Altitude measured from standard sea level pressure (29.92 in Hg) by a pressure altimeter. It is the indicated pressure altitude corrected for position and instrument error. In this manual, altimeter instrument errors are assumed to be zero. Position errors may be obtained from the correction tables/charts.
RTO	Rejected takeoff. RTO distance = accelerate-stop distance.
Runway Gradient	The change in runway elevation per 100 feet of runway length. A positive gradient refers to an uphill gradient.

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HA-420 AFM

PERFORMANCE

Term	Definition
SAT	Static air temperature. The ambient free air static temperature obtained from either 1) ground meteorological sources or 2) from the total air temperature obtained from onboard temperature measurement adjusted for compressibility effects. (SAT = OAT)
Takeoff Flight Path	The climb profile of the airplane beginning at 35 feet above the runway and continuing to a height of at least 1,500 feet, and the aircraft being configured in its enroute configuration and airspeed. The profile is constructed assuming the failure of one engine at the critical engine failure speed during takeoff. The takeoff flight path information may be used to ensure clearance of obstacles during departure.
Takeoff Speed Schedule	The values of V_1 , V_R and V_2 for a particular takeoff considering the aircraft configuration, ambient conditions and the runway.
TAT	Total air temperature. The onboard measurement of temperature not corrected for compressibility effects on the temperature probe. TAT will always be greater than SAT in proportion to Mach number.

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HA-420 AFM

PERFORMANCE

Term	Definition
TOFL	Takeoff field length. The takeoff field length for a specific weight, altitude, temperature, wind and runway gradient that is the longer of: 115 % of the AEO takeoff distance The OEI RTO distance The OEI takeoff distance.
Turnaround Time	The minimum waiting time after landing that will ensure that the brakes have cooled enough to provide sufficient braking capacity to perform an RTO and meet the published takeoff performance.
V_A	Maneuvering speed. The maximum speed at which application of full aerodynamic control will not overstress the airplane.
V_1	Takeoff decision speed. The maximum speed at which the pilot must initiate the first action to discontinue the takeoff.
V_2	Takeoff safety speed (OEI). The actual speed attained at 35 feet above the runway with one engine inoperative as demonstrated during certification flight testing.
V_{35}	Takeoff safety speed (AEO). The actual speed attained at 35 feet above the runway with all engines operating.
V_{AC}	Approach climb speed. $V_{AC} = V_{REF} + 5$

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PERFORMANCE

Term	Definition
V_{EF}	Engine failure speed. The speed at which the critical engine is assumed to fail during takeoff.
V_{FE}	Maximum flaps extended speed. The maximum airspeed allowed for operation with the flap setting.
V_{FTO}	Final takeoff speed. Also called enroute climb speed, this is the OEI climb speed with flaps up and maximum continuous thrust.
Visible Moisture	Visible moisture includes, but is not limited to, the following conditions: fog or clouds with visibility less than one mile, falling snow and rain.
V_{LE}	Maximum landing gear extended speed. The maximum airspeed allowed for aircraft operation with the landing gear extended.
V_{LO}	Maximum landing gear operating speed. The maximum airspeed at which the landing gear can safely be extended or retracted.
V_{LOF}	Liftoff speed. The speed at which the airplane first becomes airborne during the takeoff roll.
V_{MCA}	Minimum control speed, air. The minimum flight speed out of ground effect, in the takeoff configuration, at which the airplane is controllable with up to 5° of bank when one engine suddenly becomes inoperative and the remaining engine is operating at takeoff thrust.

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HA-420 AFM

PERFORMANCE

Term	Definition
V_{MCL}	Minimum control speed, landing. The minimum flight speed out of ground effect, in the landing configuration, at which the airplane is controllable with up to 5° of bank when one engine suddenly becomes inoperative and the remaining engine is operating at takeoff thrust.
V_{MO}	Maximum operating airspeed. The speed that may not be deliberately exceeded in any flight condition.
V_R	Rotation speed. The speed at which rotation is initiated during the takeoff roll.
V_{REF}	Reference landing speed. The landing approach/climb airspeed.
V_{SR}	Reference stall speed. The lowest airspeed at which level flight can be sustained by the airplane's wings.
Wind	The wind velocity, in knots, recorded as variables in the tables/charts of this section are to be understood as the headwind or tailwind components of the actual winds at 10 meters (33 feet) above the runway surface (tower winds).

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PERFORMANCE

REFERENCE INFORMATION

ISA Conversion

Relationship of Static Air Temperature to ISA

ISA Conversion											
Alt [ft]	Static Air Temperature [°C]										
	-55	-50	-40	-30	-20	-10	0	10	20	30	40
0	-40.0	-35.0	-25.0	-15.0	-5.0	5.0	15.0	25.0	35.0	45.0	55.0
1000	-42.0	-37.0	-27.0	-17.0	-7.0	3.0	13.0	23.0	33.0	43.0	53.0
2000	-44.0	-39.0	-29.0	-19.0	-9.0	1.0	11.0	21.0	31.0	41.0	51.0
3000	-45.9	-40.9	-30.9	-20.9	-10.9	-0.9	9.1	19.1	29.1	39.1	49.1
4000	-47.9	-42.9	-32.9	-22.9	-12.9	-2.9	7.1	17.1	27.1	37.1	47.1
5000	-49.9	-44.9	-34.9	-24.9	-14.9	-4.9	5.1	15.1	25.1	35.1	45.1
6000	-51.9	-46.9	-36.9	-26.9	-16.9	-6.9	3.1	13.1	23.1	33.1	43.1
7000	-53.9	-48.9	-38.9	-28.9	-18.9	-8.9	1.1	11.1	21.1	31.1	41.1
8000	-55.8	-50.8	-40.8	-30.8	-20.8	-10.8	-0.8	9.2	19.2	29.2	39.2
9000	-57.8	-52.8	-42.8	-32.8	-22.8	-12.8	-2.8	7.2	17.2	27.2	37.2
10000	-59.8	-54.8	-44.8	-34.8	-24.8	-14.8	-4.8	5.2	15.2	25.2	35.2
11000	-61.8	-56.8	-46.8	-36.8	-26.8	-16.8	-6.8	3.2	13.2	23.2	33.2
12000	-63.8	-58.8	-48.8	-38.8	-28.8	-18.8	-8.8	1.2	11.2	21.2	31.2
13000	-65.8	-60.8	-50.8	-40.8	-30.8	-20.8	-10.8	-0.8	9.2	19.2	29.2
14000	-67.7	-62.7	-52.7	-42.7	-32.7	-22.7	-12.7	-2.7	7.3	17.3	27.3
15000	-69.7	-64.7	-54.7	-44.7	-34.7	-24.7	-14.7	-4.7	5.3	15.3	25.3
16000	-71.7	-66.7	-56.7	-46.7	-36.7	-26.7	-16.7	-6.7	3.3	13.3	23.3
17000	-73.7	-68.7	-58.7	-48.7	-38.7	-28.7	-18.7	-8.7	1.3	11.3	21.3
18000	-75.7	-70.7	-60.7	-50.7	-40.7	-30.7	-20.7	-10.7	-0.7	9.3	19.3
19000	-77.6	-72.6	-62.6	-52.6	-42.6	-32.6	-22.6	-12.6	-2.6	7.4	17.4
20000	-79.6	-74.6	-64.6	-54.6	-44.6	-34.6	-24.6	-14.6	-4.6	5.4	15.4
21000	-81.6	-76.6	-66.6	-56.6	-46.6	-36.6	-26.6	-16.6	-6.6	3.4	13.4
22000	-83.6	-78.6	-68.6	-58.6	-48.6	-38.6	-28.6	-18.6	-8.6	1.4	11.4
23000	-85.6	-80.6	-70.6	-60.6	-50.6	-40.6	-30.6	-20.6	-10.6	-0.6	9.4
24000	-87.5	-82.5	-72.5	-62.5	-52.5	-42.5	-32.5	-22.5	-12.5	-2.5	7.5
25000	-89.5	-84.5	-74.5	-64.5	-54.5	-44.5	-34.5	-24.5	-14.5	-4.5	5.5
26000	-91.5	-86.5	-76.5	-66.5	-56.5	-46.5	-36.5	-26.5	-16.5	-6.5	3.5
27000	-93.5	-88.5	-78.5	-68.5	-58.5	-48.5	-38.5	-28.5	-18.5	-8.5	1.5
28000	-95.5	-90.5	-80.5	-70.5	-60.5	-50.5	-40.5	-30.5	-20.5	-10.5	-0.5
29000	-97.5	-92.5	-82.5	-72.5	-62.5	-52.5	-42.5	-32.5	-22.5	-12.5	-2.5
30000	-99.4	-94.4	-84.4	-74.4	-64.4	-54.4	-44.4	-34.4	-24.4	-14.4	-4.4
31000	-101.4	-96.4	-86.4	-76.4	-66.4	-56.4	-46.4	-36.4	-26.4	-16.4	-6.4
32000	-103.4	-98.4	-88.4	-78.4	-68.4	-58.4	-48.4	-38.4	-28.4	-18.4	-8.4
33000	-105.4	-100.4	-90.4	-80.4	-70.4	-60.4	-50.4	-40.4	-30.4	-20.4	-10.4
34000	-107.4	-102.4	-92.4	-82.4	-72.4	-62.4	-52.4	-42.4	-32.4	-22.4	-12.4
35000	-109.3	-104.3	-94.3	-84.3	-74.3	-64.3	-54.3	-44.3	-34.3	-24.3	-14.3
36000	-111.3	-106.3	-96.3	-86.3	-76.3	-66.3	-56.3	-46.3	-36.3	-26.3	-16.3
37000	-111.5	-106.5	-96.5	-86.5	-76.5	-66.5	-56.5	-46.5	-36.5	-26.5	-16.5
43000	-111.5	-106.5	-96.5	-86.5	-76.5	-66.5	-56.5	-46.5	-36.5	-26.5	-16.5

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HJ1-29000-003-001

Page 5-15

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PERFORMANCE

Temperature Conversion

To convert from Celsius to Fahrenheit, find, in bold face columns, the number representing the Celsius temperature to be converted. The equivalent Fahrenheit temperature is read in the adjacent column headed °F.

To convert from Fahrenheit to Celsius, find, in bold face columns, the number representing the Fahrenheit temperature to be converted. The equivalent Celsius temperature is read in the adjacent column headed °C.

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Fahrenheit-Celsius Temperature Conversion														
°C	◀▶	°F	°C	◀▶	°F	°C	◀▶	°F	°C	◀▶	°F	°C	◀▶	°F
-73.3	-100	-148.0	-45.6	-50	-58.0	-17.8	0	32.0	10.0	50	122.0	37.8	100	212.0
-72.8	-99	-146.2	-45.0	-49	-56.2	-17.2	1	33.8	10.6	51	123.8	38.3	101	213.8
-72.2	-98	-144.4	-44.4	-48	-54.4	-16.7	2	35.6	11.1	52	125.6	38.9	102	215.6
-71.7	-97	-142.6	-43.9	-47	-52.6	-16.1	3	37.4	11.7	53	127.4	39.4	103	217.4
-71.1	-96	-140.8	-43.3	-46	-50.8	-15.6	4	39.2	12.2	54	129.2	40.0	104	219.2
-70.6	-95	-139.0	-42.8	-45	-49.0	-15.0	5	41.0	12.8	55	131.0	40.6	105	221.0
-70.0	-94	-137.2	-42.2	-44	-47.2	-14.4	6	42.8	13.3	56	132.8	41.1	106	222.8
-69.4	-93	-135.4	-41.7	-43	-45.4	-13.9	7	44.6	13.9	57	134.6	41.7	107	224.6
-68.9	-92	-133.6	-41.1	-42	-43.6	-13.3	8	46.4	14.4	58	136.4	42.2	108	226.4
-68.3	-91	-131.8	-40.6	-41	-41.8	-12.8	9	48.2	15.0	59	138.2	42.8	109	228.2
-67.8	-90	-130.0	-40.0	-40	-40.0	-12.2	10	50.0	15.6	60	140.0	43.3	110	230.0
-67.2	-89	-128.2	-39.4	-39	-38.2	-11.7	11	51.8	16.1	61	141.8	43.9	111	231.8
-66.7	-88	-126.4	-38.9	-38	-36.4	-11.1	12	53.6	16.7	62	143.6	44.4	112	233.6
-66.1	-87	-124.6	-38.3	-37	-34.6	-10.6	13	55.4	17.2	63	145.4	45.0	113	235.4
-65.6	-86	-122.8	-37.8	-36	-32.8	-10.0	14	57.2	17.8	64	147.2	45.6	114	237.2
-65.0	-85	-121.0	-37.2	-35	-31.0	-9.4	15	59.0	18.3	65	149.0	46.1	115	239.0
-64.4	-84	-119.2	-36.7	-34	-29.2	-8.9	16	60.8	18.9	66	150.8	46.7	116	240.8
-63.9	-83	-117.4	-36.1	-33	-27.4	-8.3	17	62.6	19.4	67	152.6	47.2	117	242.6
-63.3	-82	-115.6	-35.6	-32	-25.6	-7.8	18	64.4	20.0	68	154.4	47.8	118	244.4
-62.8	-81	-113.8	-35.0	-31	-23.8	-7.2	19	66.2	20.6	69	156.2	48.3	119	246.2
-62.2	-80	-112.0	-34.4	-30	-22.0	-6.7	20	68.0	21.1	70	158.0	48.9	120	248.0
-61.7	-79	-110.2	-33.9	-29	-20.2	-6.1	21	69.8	21.7	71	159.8	49.4	121	249.8
-61.1	-78	-108.4	-33.3	-28	-18.4	-5.6	22	71.6	22.2	72	161.6	50.0	122	251.6
-60.6	-77	-106.6	-32.8	-27	-16.6	-5.0	23	73.4	22.8	73	163.4	50.6	123	253.4
-60.0	-76	-104.8	-32.2	-26	-14.8	-4.4	24	75.2	23.3	74	165.2	51.1	124	255.2
-59.4	-75	-103.0	-31.7	-25	-13.0	-3.9	25	77.0	23.9	75	167.0	51.7	125	257.0
-58.9	-74	-101.2	-31.1	-24	-11.2	-3.3	26	78.8	24.4	76	168.8	52.2	126	258.8
-58.3	-73	-99.4	-30.6	-23	-9.4	-2.8	27	80.6	25.0	77	170.6	52.8	127	260.6
-57.8	-72	-97.6	-30.0	-22	-7.6	-2.2	28	82.4	25.6	78	172.4	53.3	128	262.4
-57.2	-71	-95.8	-29.4	-21	-5.8	-1.7	29	84.2	26.1	79	174.2	53.9	129	264.2
-56.7	-70	-94.0	-28.9	-20	-4.0	-1.1	30	86.0	26.7	80	176.0	54.4	130	266.0
-56.1	-69	-92.2	-28.3	-19	-2.2	-0.6	31	87.8	27.2	81	177.8	55.0	131	267.8
-55.6	-68	-90.4	-27.8	-18	-0.4	0.0	32	89.6	27.8	82	179.6	55.6	132	269.6
-55.0	-67	-88.6	-27.2	-17	1.4	0.6	33	91.4	28.3	83	181.4	56.1	133	271.4
-54.4	-66	-86.8	-26.7	-16	3.2	1.1	34	93.2	28.9	84	183.2	56.7	134	273.2
-53.9	-65	-85.0	-26.1	-15	5.0	1.7	35	95.0	29.4	85	185.0	57.2	135	275.0
-53.3	-64	-83.2	-25.6	-14	6.8	2.2	36	96.8	30.0	86	186.8	57.8	136	276.8
-52.8	-63	-81.4	-25.0	-13	8.6	2.8	37	98.6	30.6	87	188.6	58.3	137	278.6
-52.2	-62	-79.6	-24.4	-12	10.4	3.3	38	100.4	31.1	88	190.4	58.9	138	280.4
-51.7	-61	-77.8	-23.9	-11	12.2	3.9	39	102.2	31.7	89	192.2	59.4	139	282.2
-51.1	-60	-76.0	-23.3	-10	14.0	4.4	40	104.0	32.2	90	194.0	60.0	140	284.0
-50.6	-59	-74.2	-22.8	-9	15.8	5.0	41	105.8	32.8	91	195.8	60.6	141	285.8
-50.0	-58	-72.4	-22.2	-8	17.6	5.6	42	107.6	33.3	92	197.6	61.1	142	287.6
-49.4	-57	-70.6	-21.7	-7	19.4	6.1	43	109.4	33.9	93	199.4	61.7	143	289.4
-48.9	-56	-68.8	-21.1	-6	21.2	6.7	44	111.2	34.4	94	201.2	62.2	144	291.2
-48.3	-55	-67.0	-20.6	-5	23.0	7.2	45	113.0	35.0	95	203.0	62.8	145	293.0
-47.8	-54	-65.2	-20.0	-4	24.8	7.8	46	114.8	35.6	96	204.8	63.3	146	294.8
-47.2	-53	-63.4	-19.4	-3	26.6	8.3	47	116.6	36.1	97	206.6	63.9	147	296.6
-46.7	-52	-61.6	-18.9	-2	28.4	8.9	48	118.4	36.7	98	208.4	64.4	148	298.4
-46.1	-51	-59.8	-18.3	-1	30.2	9.4	49	120.2	37.2	99	210.2	65.0	149	300.2

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Page 5-17

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PERFORMANCE

Weight Conversion

To convert from kilograms to pounds, find, in bold face columns, the number representing the kilogram weight to be converted. The equivalent pound weight is read in the adjacent column headed pounds.

To convert from pounds to kilograms, find, in bold face columns, the number representing the pounds weight to be converted. The equivalent kilogram weight is read in the adjacent column headed kilograms.

Honda Aircraft Company

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PERFORMANCE

Weight Conversion								
Kg	◀ ▶	Lb	Kg	◀ ▶	Lb	Kg	◀ ▶	Lb
0.5	1	2.2	204.1	450	992.1	2404.0	5300	11684.5
0.9	2	4.4	208.7	460	1014.1	2449.4	5400	11905.0
1.4	3	6.6	213.2	470	1036.2	2494.8	5500	12125.4
1.8	4	8.8	217.7	480	1058.2	2540.1	5600	12345.9
2.3	5	11.0	222.3	490	1080.3	2585.5	5700	12566.3
2.7	6	13.2	226.8	500	1102.3	2630.8	5800	12786.8
3.2	7	15.4	272.2	600	1322.8	2676.2	5900	13007.3
3.6	8	17.6	317.5	700	1543.2	2721.6	6000	13227.7
4.1	9	19.8	362.9	800	1763.7	2766.9	6100	13448.2
4.5	10	22.0	408.2	900	1984.2	2812.3	6200	13668.7
9.1	20	44.1	453.6	1000	2204.6	2857.6	6300	13889.1
13.6	30	66.1	499.0	1100	2425.1	2903.0	6400	14109.6
18.1	40	88.2	544.3	1200	2645.5	2948.4	6500	14330.0
22.7	50	110.2	589.7	1300	2866.0	2993.7	6600	14550.5
27.2	60	132.3	635.0	1400	3086.5	3039.1	6700	14771.0
31.8	70	154.3	680.4	1500	3306.9	3084.4	6800	14991.4
36.3	80	176.4	725.7	1600	3527.4	3129.8	6900	15211.9
40.8	90	198.4	771.1	1700	3747.9	3175.1	7000	15432.4
45.4	100	220.5	816.5	1800	3968.3	3220.5	7100	15652.8
49.9	110	242.5	861.8	1900	4188.8	3265.9	7200	15873.3
54.4	120	264.6	907.2	2000	4409.2	3311.2	7300	16093.7
59.0	130	286.6	952.5	2100	4629.7	3356.6	7400	16314.2
63.5	140	308.6	997.9	2200	4850.2	3401.9	7500	16534.7
68.0	150	330.7	1043.3	2300	5070.6	3447.3	7600	16755.1
72.6	160	352.7	1088.6	2400	5291.1	3492.7	7700	16975.6
77.1	170	374.8	1134.0	2500	5511.6	3538.0	7800	17196.1
81.6	180	396.8	1179.3	2600	5732.0	3583.4	7900	17416.5
86.2	190	418.9	1224.7	2700	5952.5	3628.7	8000	17637.0
90.7	200	440.9	1270.1	2800	6172.9	3674.1	8100	17857.4
95.3	210	463.0	1315.4	2900	6393.4	3719.5	8200	18077.9
99.8	220	485.0	1360.8	3000	6613.9	3764.8	8300	18298.4
104.3	230	507.1	1406.1	3100	6834.3	3810.2	8400	18518.8
108.9	240	529.1	1451.5	3200	7054.8	3855.5	8500	18739.3
113.4	250	551.2	1496.9	3300	7275.3	3900.9	8600	18959.8
117.9	260	573.2	1542.2	3400	7495.7	3946.3	8700	19180.2
122.5	270	595.2	1587.6	3500	7716.2	3991.6	8800	19400.7
127.0	280	617.3	1632.9	3600	7936.6	4037.0	8900	19621.1
131.5	290	639.3	1678.3	3700	8157.1	4082.3	9000	19841.6
136.1	300	661.4	1723.7	3800	8377.6	4127.7	9100	20062.1
140.6	310	683.4	1769.0	3900	8598.0	4173.0	9200	20282.5
145.1	320	705.5	1814.4	4000	8818.5	4218.4	9300	20503.0
149.7	330	727.5	1859.7	4100	9039.0	4263.8	9400	20723.5
154.2	340	749.6	1905.1	4200	9259.4	4309.1	9500	20943.9
158.8	350	771.6	1950.4	4300	9479.9	4354.5	9600	21164.4
163.3	360	793.7	1995.8	4400	9700.3	4399.8	9700	21384.8
167.8	370	815.7	2041.2	4500	9920.8	4445.2	9800	21605.3
172.4	380	837.8	2086.5	4600	10141.3	4490.6	9900	21825.8
176.9	390	859.8	2131.9	4700	10361.7	4535.9	10000	22046.2
181.4	400	881.8	2177.2	4800	10582.2	4581.3	10100	22266.7
186.0	410	903.9	2222.6	4900	10802.7	4626.6	10200	22487.2
190.5	420	925.9	2268.0	5000	11023.1	4672.0	10300	22707.6
195.0	430	948.0	2313.3	5100	11243.6	4717.4	10400	22928.1
199.6	440	970.0	2358.7	5200	11464.0	4762.7	10500	23148.5

WT_CONVERSION

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October 30, 2016

HJ1-29000-003-001

Page 5-19

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HA-420 AFM

PERFORMANCE

Volume Conversion

To convert from liters to gallons, find, in bold face columns, the number representing the liters volume to be converted. The equivalent gallons volume is read in the adjacent column headed gallons.

To convert from gallons to liters, find, in bold face columns, the number representing the gallons volume to be converted. The equivalent liters volume is read in the adjacent column headed liters.

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Volume Conversion								
Liters	◀ ▶	Gallons	Liters	◀ ▶	Gallons	Liters	◀ ▶	Gallons
3.8	1	0.3	1324.9	350	92.5	2952.6	780	206.1
7.6	2	0.5	1362.7	360	95.1	2990.5	790	208.7
11.4	3	0.8	1400.6	370	97.7	3028.3	800	211.3
15.1	4	1.1	1438.5	380	100.4	3066.2	810	214.0
18.9	5	1.3	1476.3	390	103.0	3104.0	820	216.6
22.7	6	1.6	1514.2	400	105.7	3141.9	830	219.3
26.5	7	1.8	1552.0	410	108.3	3179.7	840	221.9
30.3	8	2.1	1589.9	420	111.0	3217.6	850	224.5
34.1	9	2.4	1627.7	430	113.6	3255.5	860	227.2
37.9	10	2.6	1665.6	440	116.2	3293.3	870	229.8
75.7	20	5.3	1703.4	450	118.9	3331.2	880	232.5
113.6	30	7.9	1741.3	460	121.5	3369.0	890	235.1
151.4	40	10.6	1779.1	470	124.2	3406.9	900	237.8
189.3	50	13.2	1817.0	480	126.8	3444.7	910	240.4
227.1	60	15.9	1854.9	490	129.4	3482.6	920	243.0
265.0	70	18.5	1892.7	500	132.1	3520.4	930	245.7
302.8	80	21.1	1930.6	510	134.7	3558.3	940	248.3
340.7	90	23.8	1968.4	520	137.4	3596.1	950	251.0
378.5	100	26.4	2006.3	530	140.0	3634.0	960	253.6
416.4	110	29.1	2044.1	540	142.7	3671.8	970	256.2
454.2	120	31.7	2082.0	550	145.3	3709.7	980	258.9
492.1	130	34.3	2119.8	560	147.9	3747.6	990	261.5
530.0	140	37.0	2157.7	570	150.6	3785.4	1000	264.2
567.8	150	39.6	2195.5	580	153.2	4164.0	1100	290.6
605.7	160	42.3	2233.4	590	155.9	4542.5	1200	317.0
643.5	170	44.9	2271.2	600	158.5	4921.0	1300	343.4
681.4	180	47.6	2309.1	610	161.1	5299.6	1400	369.8
719.2	190	50.2	2347.0	620	163.8	5678.1	1500	396.3
757.1	200	52.8	2384.8	630	166.4	6056.7	1600	422.7
794.9	210	55.5	2422.7	640	169.1	6435.2	1700	449.1
832.8	220	58.1	2460.5	650	171.7	6813.7	1800	475.5
870.6	230	60.8	2498.4	660	174.4	7192.3	1900	501.9
908.5	240	63.4	2536.2	670	177.0	7570.8	2000	528.3
946.4	250	66.0	2574.1	680	179.6	7949.4	2100	554.8
984.2	260	68.7	2611.9	690	182.3	8327.9	2200	581.2
1022.1	270	71.3	2649.8	700	184.9	8706.4	2300	607.6
1059.9	280	74.0	2687.6	710	187.6	9085.0	2400	634.0
1097.8	290	76.6	2725.5	720	190.2	9463.5	2500	660.4
1135.6	300	79.3	2763.4	730	192.8	9842.1	2600	686.8
1173.5	310	81.9	2801.2	740	195.5	10220.6	2700	713.3
1211.3	320	84.5	2839.1	750	198.1	10599.2	2800	739.7
1249.2	330	87.2	2876.9	760	200.8	10977.7	2900	766.1
1287.0	340	89.8	2914.8	770	203.4	11356.2	3000	792.5

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October 30, 2016

HJ1-29000-003-001

Page 5-21

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Length Conversion

To convert from meters to feet, find, in bold face columns, the number representing the meters length to be converted. The equivalent feet are read in the adjacent column headed, Feet.

To convert from feet to meters, find, in bold face columns, the number representing the feet length to be converted. The equivalent meters are read in the adjacent column headed, Meters.

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PERFORMANCE

Length Conversion					
Meters	◀ ▶	Feet	Meters	◀ ▶	Feet
0.3	1	3	365.8	1200	3937
0.6	2	7	396.2	1300	4265
0.9	3	10	426.7	1400	4593
1.2	4	13	457.2	1500	4921
1.5	5	16	487.7	1600	5249
1.8	6	20	518.2	1700	5577
2.1	7	23	548.6	1800	5906
2.4	8	26	579.1	1900	6234
2.7	9	30	609.6	2000	6562
3.0	10	33	640.1	2100	6890
6.1	20	66	670.6	2200	7218
9.1	30	98	701.0	2300	7546
12.2	40	131	731.5	2400	7874
15.2	50	164	762.0	2500	8202
18.3	60	197	792.5	2600	8530
21.3	70	230	823.0	2700	8858
24.4	80	262	853.4	2800	9186
27.4	90	295	883.9	2900	9514
30.5	100	328	914.4	3000	9843
45.7	150	492	944.9	3100	10171
61.0	200	656	975.4	3200	10499
76.2	250	820	1005.8	3300	10827
91.4	300	984	1036.3	3400	11155
106.7	350	1148	1066.8	3500	11483
121.9	400	1312	1097.3	3600	11811
137.2	450	1476	1127.8	3700	12139
152.4	500	1640	1158.2	3800	12467
167.6	550	1804	1188.7	3900	12795
182.9	600	1969	1219.2	4000	13123
198.1	650	2133	1249.7	4100	13451
213.4	700	2297	1280.2	4200	13780
228.6	750	2461	1310.6	4300	14108
243.8	800	2625	1341.1	4400	14436
259.1	850	2789	1371.6	4500	14764
274.3	900	2953	1402.1	4600	15092
289.6	950	3117	1432.6	4700	15420
304.8	1000	3281	1463.0	4800	15748
335.3	1100	3609	1493.5	4900	16076

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October 30, 2016

HJ1-29000-003-001

Page 5-23

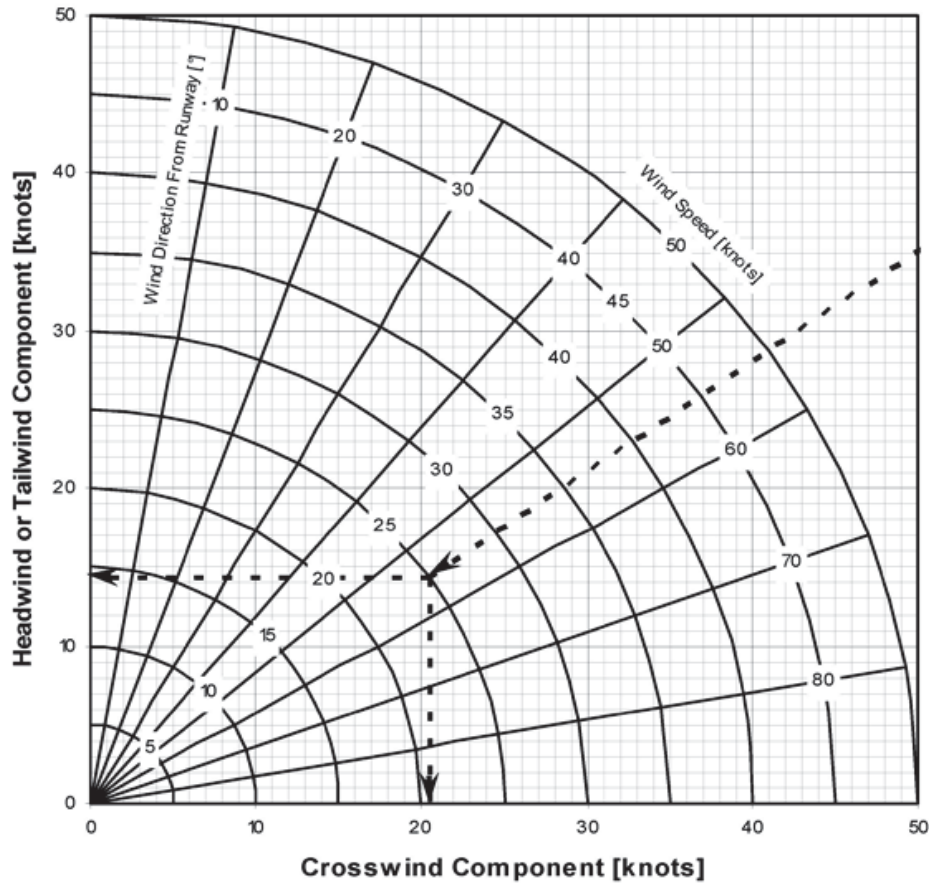
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PERFORMANCE

Wind Components



Example:

Ambient Conditions:

Wind Speed	25 knots
Wind Direction	175°

Using the Chart:

- | | |
|-------------------------------|------------|
| 1. Runway Heading | 230° |
| 2. Wind Direction from Runway | 55° |
| 3. Wind Velocity | 25 knots |
| 4. Headwind Component | 14.3 knots |
| 5. Crosswind Component | 20.5 knots |

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-24

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

ACOUSTIC LEVELS

The following noise levels comply with part 36, Appendix B, Stage 4 maximum noise level requirements and were obtained by analysis of approved data from noise tests conducted under the provisions of part 36, Amendment 36-29. The noise measurement and evaluation procedures used to obtain these noise levels are considered by the FAA to be equivalent to the Chapter 4 noise level required by the International Civil Aviation Organization (ICAO) in Annex 16, Volume I, Appendix 2, Amendment 10, effective July 2011.

No determination has been made by the Federal Aviation Administration that the noise levels of this aircraft are or should be acceptable or unacceptable for operation at, into, or out of any airport.

These noise values are stated for reference conditions per a standard acoustic atmospheric day, defined as ISA sea level pressure, 25 °C ambient temperature (ISA + 10 °C), 70 % relative humidity, no runway slope and zero wind.

Flyover and lateral noise levels were determined for the maximum takeoff weight of 10,600 lbs., TO/APPR flap setting, climb speed of $V_2 + 15$ KIAS. For Flyover, a thrust cutback altitude of 3,300 ft AGL (nominally) was used. Approach noise levels were determined for the maximum landing weight of 9,860 lbs., LDG flap setting, approach speed of $V_{REF} + 10$ KIAS, landing gear down (DN) and a glideslope of 3 degrees.

CERTIFICATION NOISE LEVELS

The demonstrated effective perceived noise levels (EPNdb), noise limits and margins of compliance to Title 14 CFR 36, Stage 4, and ICAO Annex 16, Volume I, Chapter 4 are listed below:

Reference Condition	Certification Noise Level [EPNdb]	Maximum Allowable Requirement [EPNdb]
Flyover	72.9	89.0
Lateral	85.4	94.0
Approach	87.5	98.0

ICAO Annex 16, Volume I, Chapter 4 and 14 CFR 36 Stage 4 compliance has been demonstrated with a minimum -8.6 EPNdb margin.

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October 30, 2016

HJ1-29000-003-001

Page 5-25

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

GENERAL INFORMATION

PRIMARY AIRSPEED POSITION ERROR CORRECTION - dV	
INDICATED AIRSPEED [KIAS]	dV [knots]
100	-1
125	-1
150	-1
175	-1
200	-1
225	-1
250	-1
270	-1

CSP-SSEC-AS-03

dV = KCAS - KIAS

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-26

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HA-420 AFM

PERFORMANCE

Airspeed Correction (continued)

STDBY AIRSPEED POSITION ERROR CORRECTION - dV			
INDICATED AIRSPEED [KIAS]	dV [knots]		
	Flaps UP	Flaps TO/APPR	Flaps LDG
100	-5	-2	-2
125	-3	-1	-1
150	-1	-1	-1
175	-1	-1	-1
200	-1	-1	-1
225	-1	-1	-1
250	-1	-1	-1
270	-1	-1	-1

STB-SSEC-AS-03

dV = KCAS - KIAS

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HA-420 AFM

PERFORMANCE

Altimeter Correction

PRIMARY ALTITUDE POSITION ERROR CORRECTION - dH			
INDICATED AIRSPEED [KIAS]	dH [feet]		
	Up to FL200	FL300	FL400
100	10	5	-15
125	5	-5	-25
150	-5	-15	-35
175	-5	-20	-40
200	-10	-25	-30
225	-10	-25	-30
250	-15	-20	
270	-15	-30	

CSP-SSEC-HP-03

dH = Calibrated - Indicated Altitude

Note: Altitude correction is less than 10 feet for
Flaps TO/APPR and LDG

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HA-420 AFM

PERFORMANCE

Altimeter Correction (continued)

STDBY ALTITUDE POSITION ERROR CORRECTION - dH			
INDICATED AIRSPEED [KIAS]	dH [feet]		
	Flaps UP	Flaps TO/APPR	Flaps LDG
100	-70	-25	-15
125	-40	-5	0
150	-10	5	-5
175	0	-5	-25
200	0	-10	
225	-10		
250	-25		
270	-45		

STB-SSEC-HP-03

dH = Calibrated - Indicated Altitude

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October 30, 2016

HJ1-29000-003-001

Page 5-29

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HA-420 AFM

PERFORMANCE

Stall Speeds

Reference Stall Speed [KIAS]*				
FLAPS UP				
Weight [lb]	Bank Angle [deg]			
	0	30	45	60
7000	89	95	106	126
7800	94	101	111	133
8200	96	103	114	136
8600	98	106	117	139
8800	100	107	118	141
9000	101	108	120	142
9400	103	110	122	145
9800	105	113	125	148
10200	107	115	127	151
10300	107	115	128	152
10600	109	117	129	154

Reference Stall Speed [KIAS]*				
Flaps TO/APPR				
Weight [lb]	Bank Angle [deg]			
	0	30	45	60
7000	81	87	96	114
7800	85	92	101	120
8200	87	94	104	123
8600	89	96	106	126
8800	90	97	108	128
9000	92	98	109	129
9400	94	100	111	132
9800	95	103	113	135
10200	97	105	116	138
10300	98	105	116	138
10600	99	107	118	140

Reference Stall Speed [KIAS]*				
Flaps LDG				
Weight [lb]	Bank Angle [deg]			
	0	30	45	60
7000	78	84	93	110
7800	82	88	98	116
8200	84	91	100	119
8600	86	93	103	122
8800	87	94	104	123
9000	88	95	105	125
9400	90	97	107	127
9800	92	99	109	130
10200	92	99	110	131
10300	94	101	112	133
10600	96	103	114	135

SS_ALL_0_LO_06

* Valid from Sea Level to 20,000 feet.

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October 30, 2016

HJ1-29000-003-001

Page 5-30

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HA-420 AFM

PERFORMANCE

High Altitude Stall Speeds

Reference Stall Speed [KIAS]						
FLAPS UP						
Weight [lb]	Altitude [1000 ft]					
	20	25	30	35	40	43
7000	89	91	94	97	102	106
7500	92	95	97	101	106	110
8000	95	98	100	104	110	113
8500	98	101	104	107	113	117
9000	101	104	107	111	116	121
9500	103	106	109	113	120	124
10000	106	109	112	116	123	127
10500	108	111	115	119	125	130

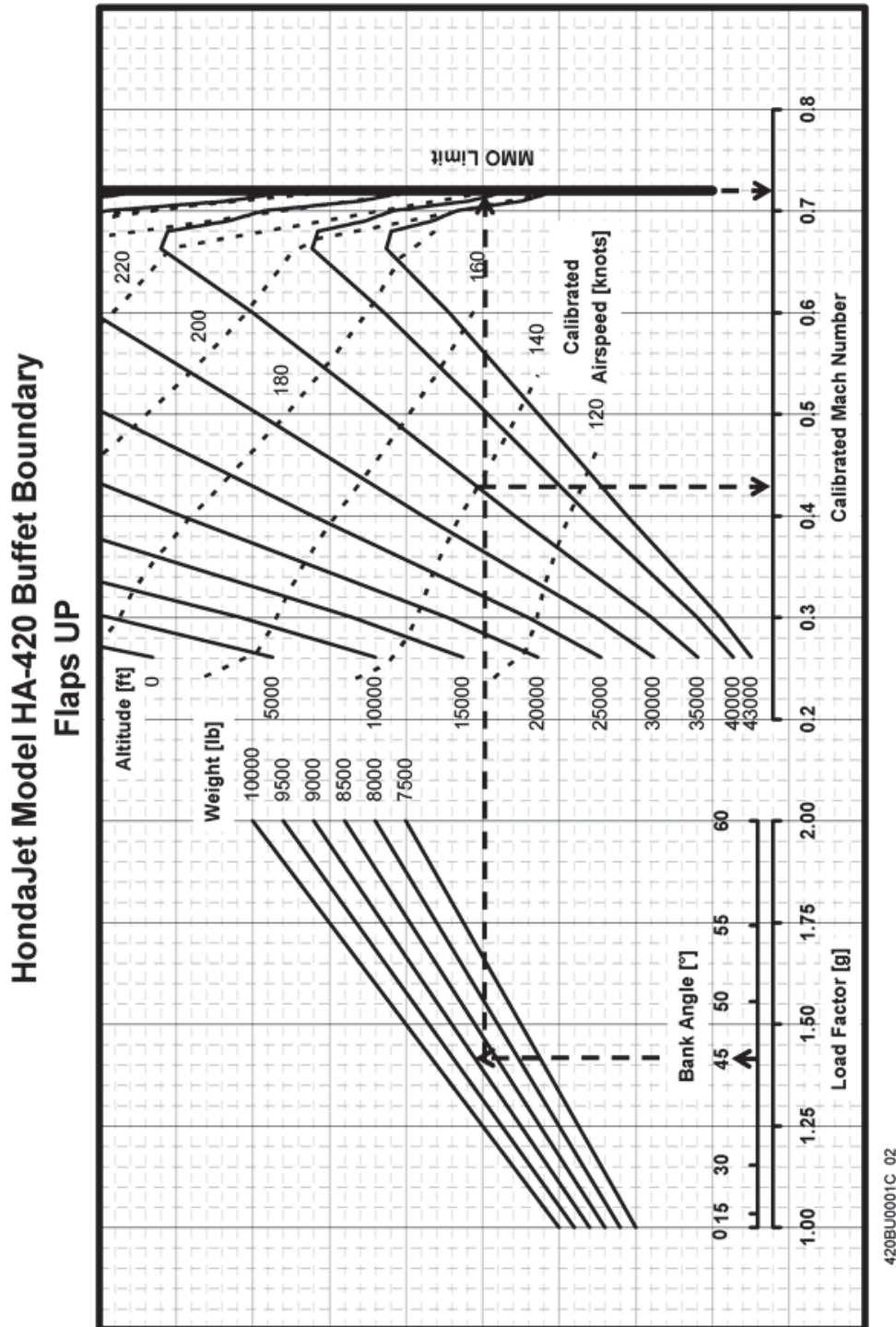
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HA-420 AFM

PERFORMANCE

Buffet Boundary



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October 30, 2016

HJ1-29000-003-001

Page 5-32

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HA-420 AFM

PERFORMANCE

Buffet Boundary (continued)

Example:

Initial Conditions:

Bank Angle	45°
Weight	9000 lbs
Altitude	35000 ft

Using the Chart:

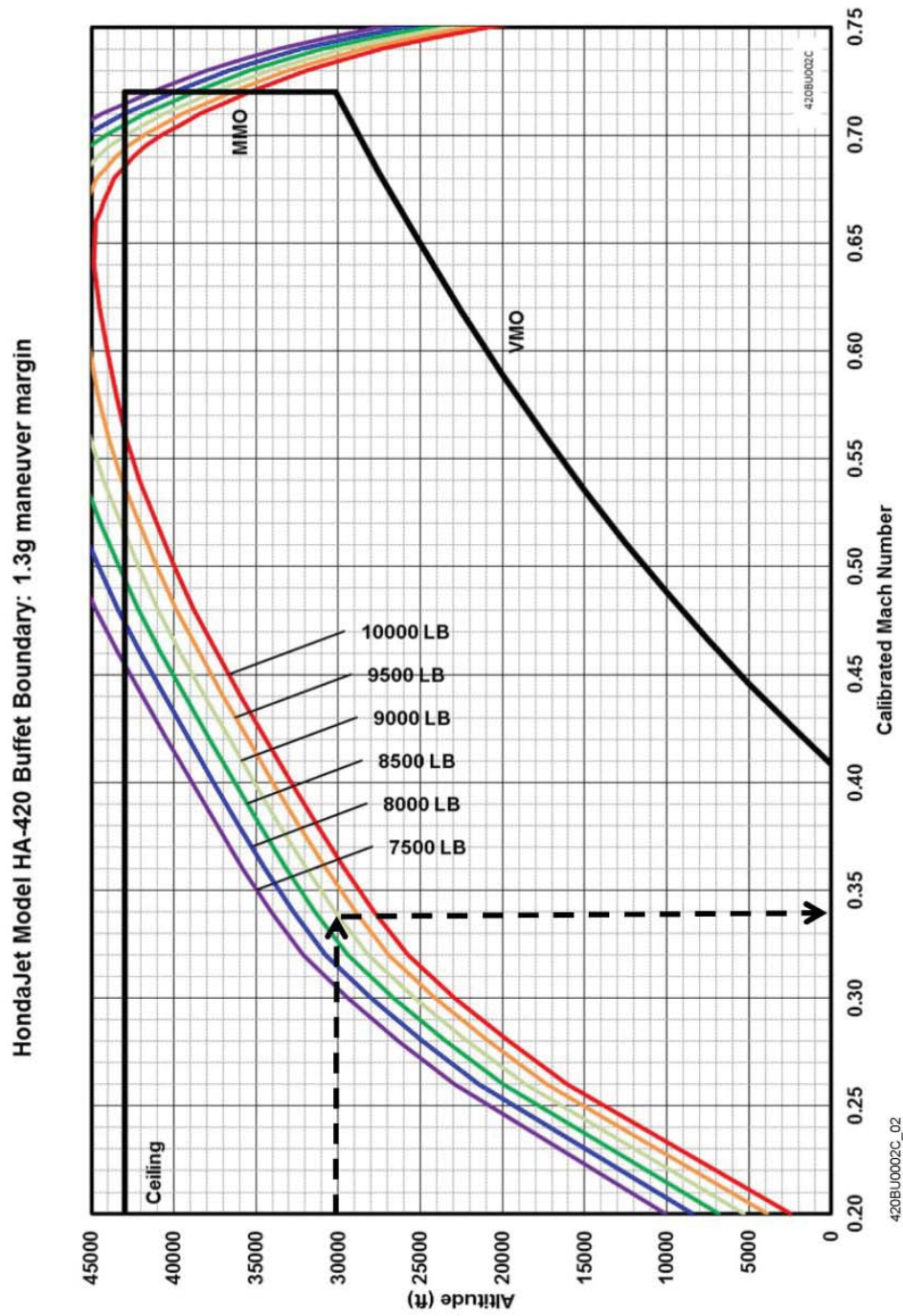
1. Low Speed Buffet Boundary 140 kcas / 0.43M
2. High Speed Buffet Boundary 270 kcas / 0.72M

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PERFORMANCE

1.3g Maneuver Margin



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-34

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HA-420 AFM

PERFORMANCE

1.3g Maneuver Margin (continued)

Example:

Initial Conditions:

Weight	9000 lbs
Altitude	30000 ft

Using the Chart:

1. 1.3 g maneuver margin	0.34 Mach
--------------------------	-----------

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PERFORMANCE

TAKEOFF

Use of takeoff information tables:

1. Determine the airplane gross weight for the desired loading and the planned airplane configuration (flaps, ice protection).
2. Obtain airport information:
 - active runway
 - available runway length
 - runway gradient
 - obstacles in the takeoff flight path
 - ambient temperature
 - Verify that the temperature is within the ambient temperature limits found in the limitations section.
 - pressure altitude
 - wind
 - Determine the wind component parallel to active runway from the crosswind component chart.
3. Check the maximum takeoff weight permitted by climb requirements and brake energy limits for the planned airplane configuration using “Maximum Takeoff Weight” charts. If any of these limitations restrict the takeoff gross weight, the pilot must off load weight until the requirement is met.
4. Using the takeoff gross weight determined in step 3 and the planned airplane takeoff configuration, determine V_1 , V_R , V_2 , attitude and uncorrected takeoff field length using the tables on pages 48 – 59 (Flaps TO/APPR, Ice Protection Off), pages 71 – 82 (Flaps TO/APPR, Ice Protection On), pages 116 – 127 (Flaps UP, Ice Protection Off) or 145 – 156 (Flaps UP, Ice Protection On).
5. For wind and runway gradients, the takeoff field length must be corrected using the correction tables on pages 63 – 67 (Flaps TO/APPR, Ice Protection Off) and 86 – 90 (Flaps TO/APPR, Ice Protection On) or 131 – 138 (Flaps UP, Ice Protection Off) and 160 – 167 (Flaps UP, Ice Protection On).

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HA-420 AFM

PERFORMANCE

Use of takeoff information tables (continued):

6. If the available runway length is less than the required takeoff field length, the airplane weight must be reduced until this requirement can be met.
7. If obstacle clearance is a factor, use the Obstacle Clearance section to ensure adequate performance is available. If the required obstacle clearance is not achieved, the gross weight must be adjusted until the obstacle can be cleared.
8. The second segment takeoff net climb tables are provided on pages 91 – 97 (Flaps TO/APPR, Ice Protection Off and On) and 168 – 174 (Flaps UP, Ice Protection Off and On) with wind correction tables on pages 94, 98, 171 and 175.
9. Final segment climb tables are provided on pages 99 – 105 (Flaps TO/APPR, Ice Protection Off and On) and 176 – 182 (Flaps UP, Ice Protection Off and On) with wind correction tables on pages 102, 106, 179 and 183.

NOTE If MTOW is not limited by brake energy for a given wind, a chart is not provided.

NOTE Ice Protection ON assumes both ENG and WING A/I are ON. If only ENG A/I is on, the Ice Protection ON tables must be used.

NOTE The takeoff field length provided in the performance section of the flight manual is based on a dry runway. If departing from a wet runway, it is recommended to increase the predicted takeoff field length by 30 %.

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HA-420 AFM

PERFORMANCE

Takeoff Pitch Trim Settings

Pitch trim for takeoff should be set per the following table.

TOGW C.G. [% M.A.C.]	Pitch Trim Setting	
	Flaps UP	Flaps TO/APPR
20 – 24	3.2	4.3
24 – 27	2.9	4.0
27 – 31	2.5	3.6

TO_PITCH_TRIM_03

TAKEOFF – ANTI-ICE FLUID PERFORMANCE ADDITIVES

After the use of anti-ice fluids:

1. Increase V_R and V_2 by 5
2. Increase the calculated field length by 30%.

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HA-420 AFM

PERFORMANCE

TAKEOFF – FLAPS TO/APPR

Max Takeoff Weight – Climb and Brake Energy Limited

Red shading on the takeoff tables indicates conditions where the airplane does not meet the climb requirements or exceeds the brake energy limits, but can be used for interpolation.

Example:

Ambient Conditions:

Temperature	35 °C
Airport Altitude	5500 ft
Wind	10 kts Tailwind
Runway Gradient	-2.0 %

Aircraft Configuration:

Flaps	TO/APPR
-------	---------

Bleed Setting:

Ice Protection	Off
----------------	-----

Using the Weight Limit Charts:

Climb Weight Limit	9500 lbs
--------------------	----------

Using the Tables:

Uncorrected

• V_1	119 KIAS
• V_R	121.5 KIAS
• V_2	122.5 KIAS
• Field Length	6967 ft
• Climb Gradient	1.5 %

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October 30, 2016

HJ1-29000-003-001

Page 5-39

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HA-420 AFM

PERFORMANCE

Example: (continued)

Using the Tables (continued):

Wind Correction:

- V_1 118 KIAS
- Field Length 7960 ft
- Climb Gradient 1.4 %

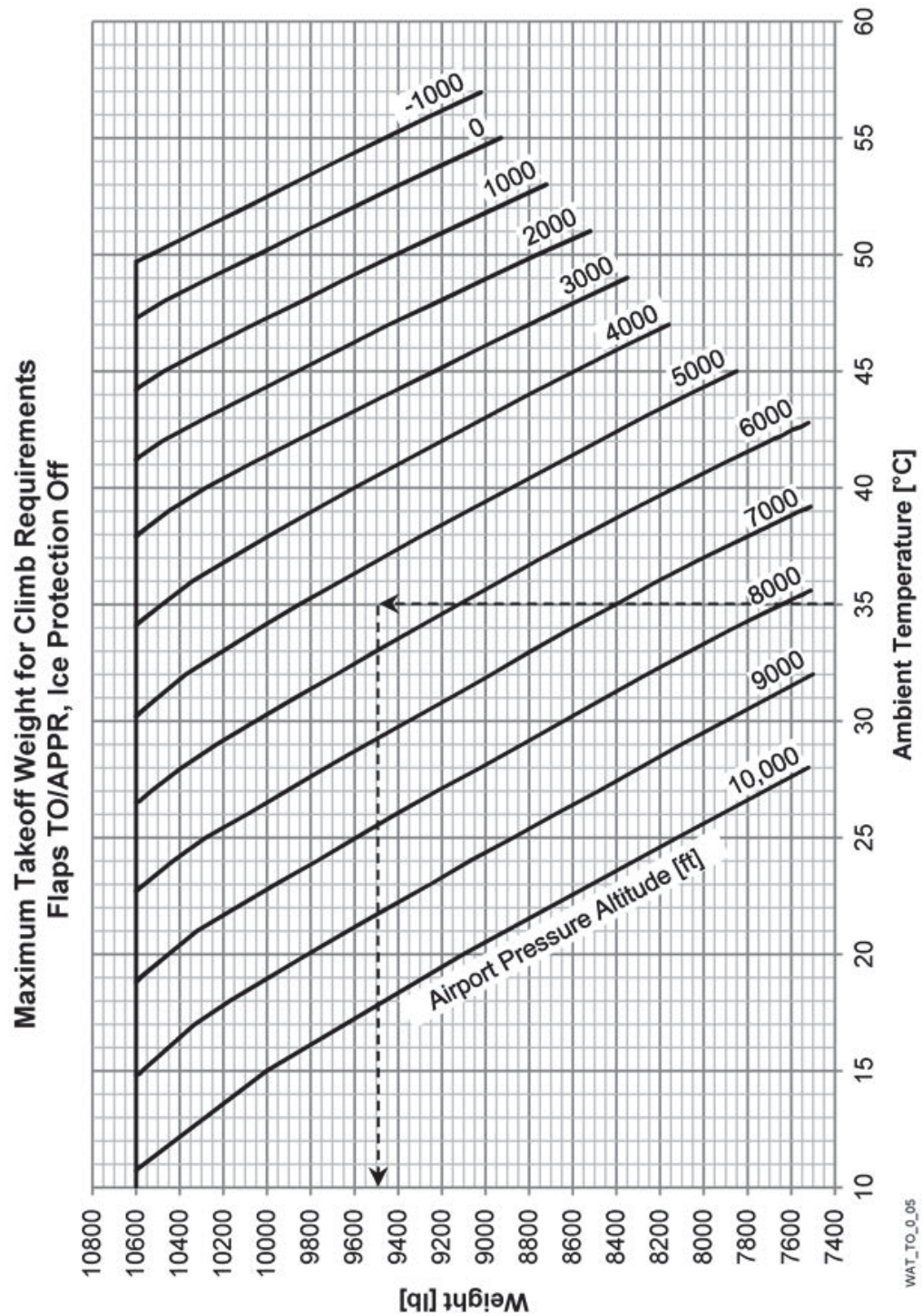
Slope Correction:

- V_1 116 KIAS
- V_2 124.5 KIAS
- Field Length 7565 ft

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HA-420 AFM

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October 30, 2016

HJ1-29000-003-001

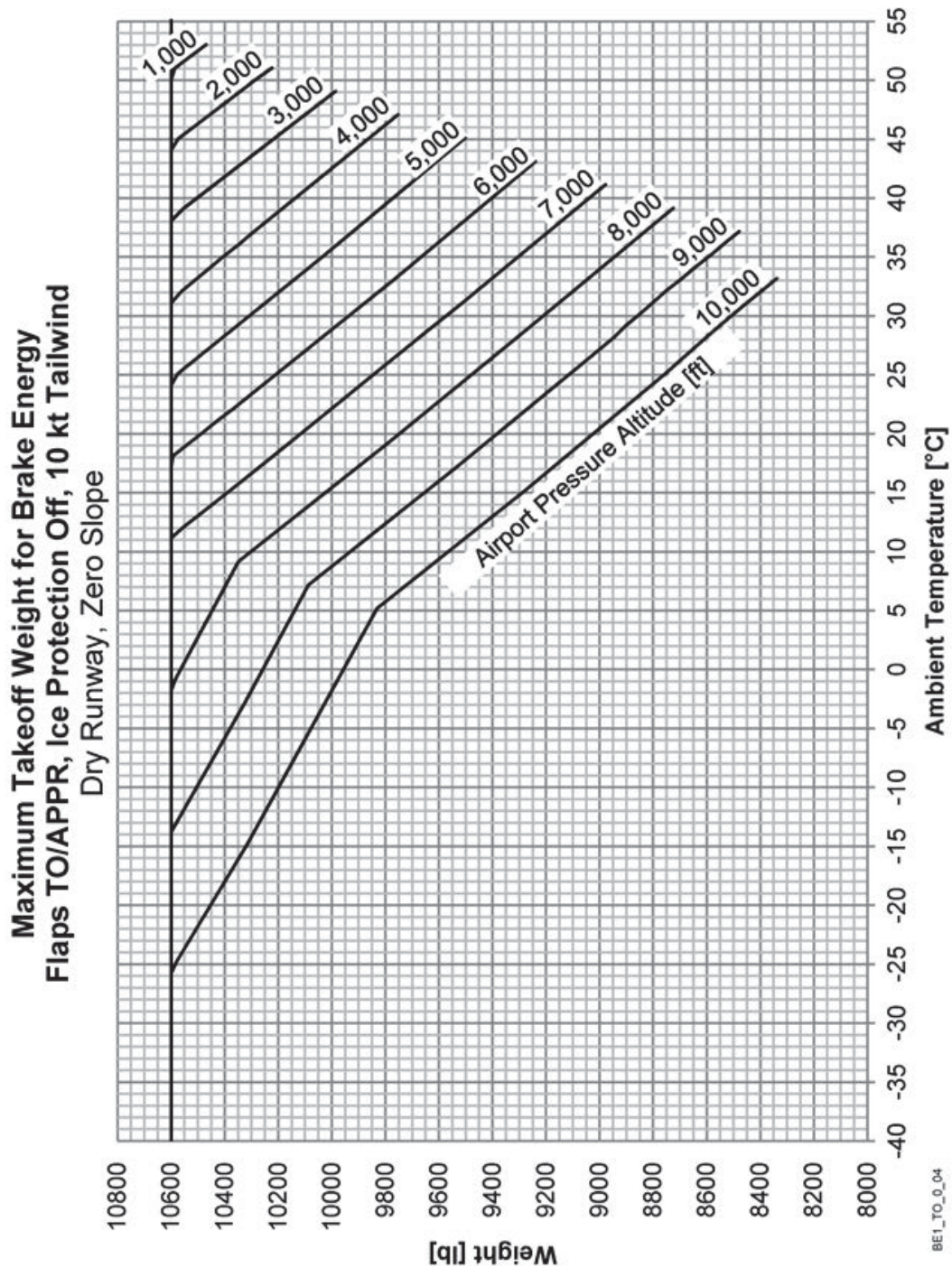
Page 5-41

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HA-420 AFM

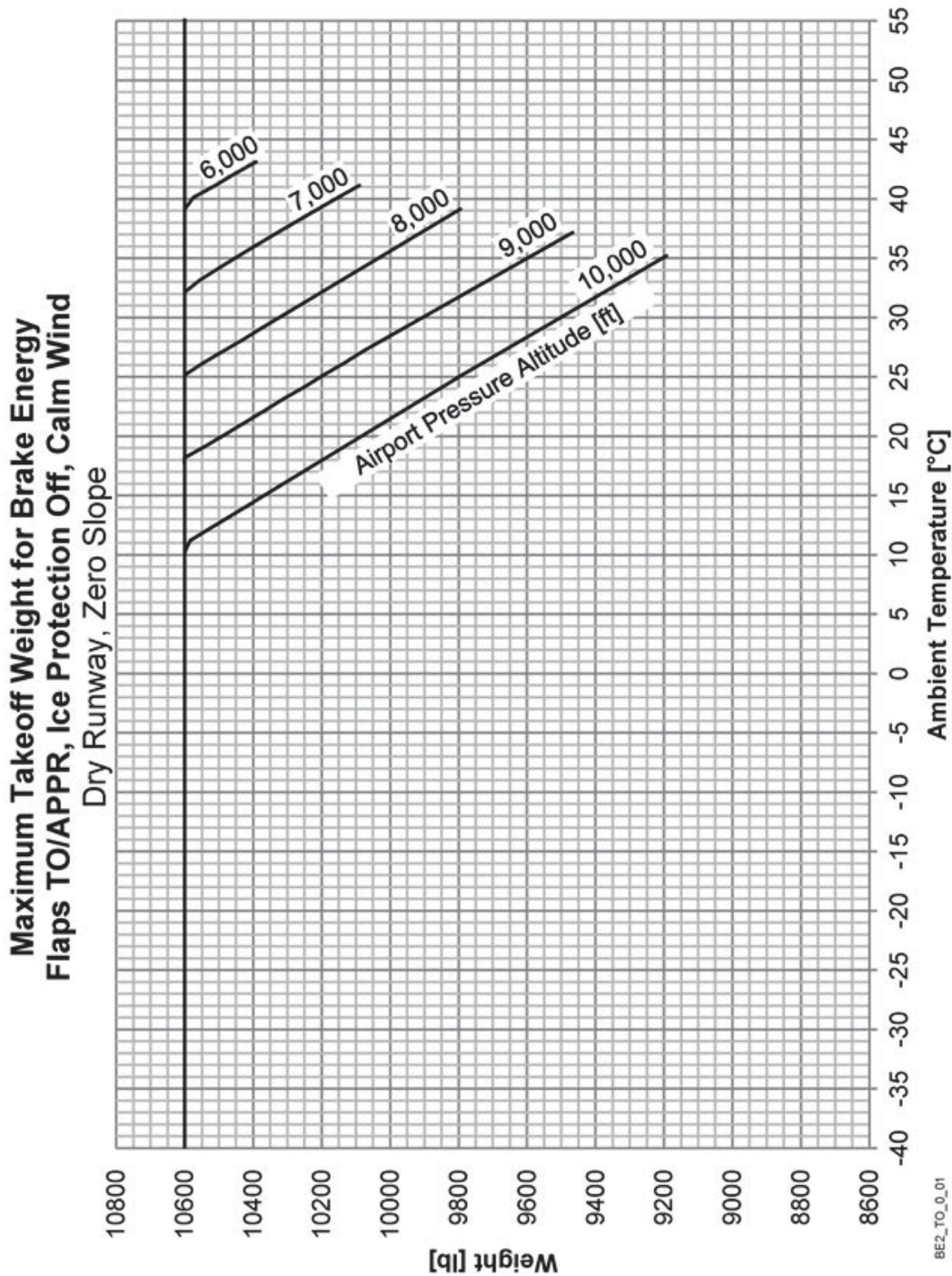
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HA-420 AFM

PERFORMANCE



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October 30, 2016

HJ1-29000-003-001

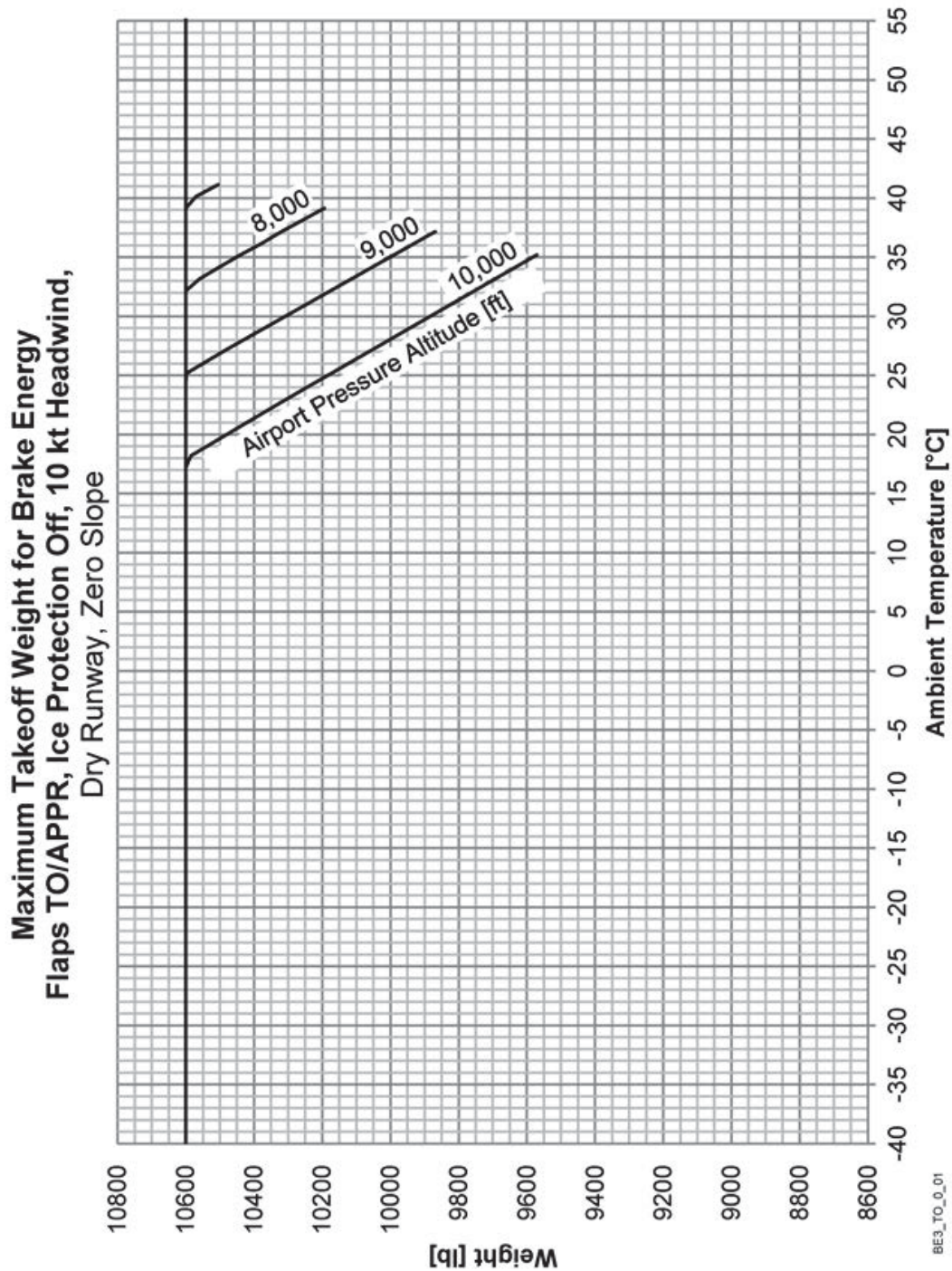
Page 5-43

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE



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October 30, 2016

HJ1-29000-003-001

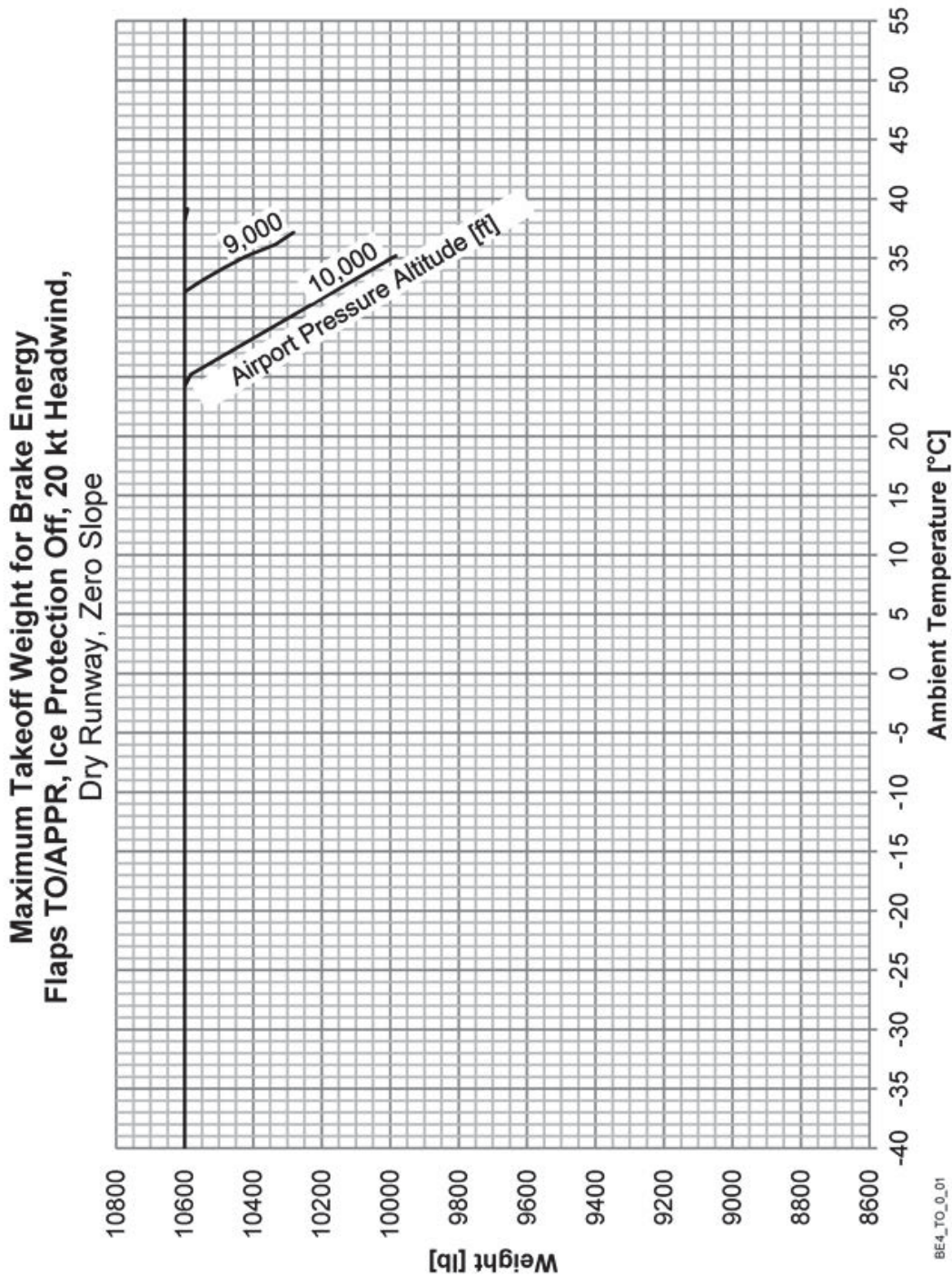
Page 5-44

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

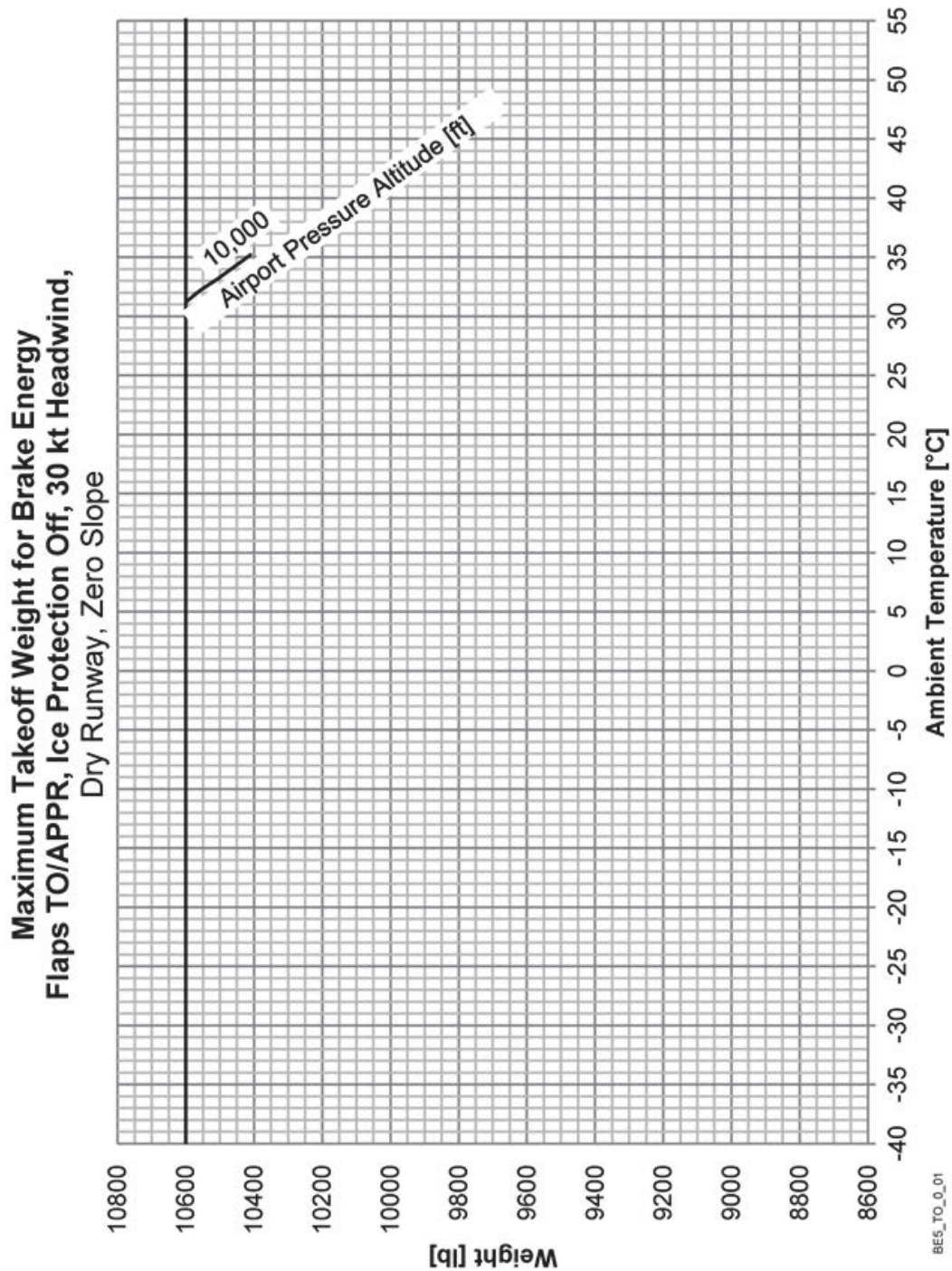
Page 5-45

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HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-46

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Maximum Takeoff Weight for Brake Energy [lb]				
Flaps TO/APPR, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
7564	7698	7800	8110	8605
7659	7795	7900	8202	8683
7753	7891	8000	8294	8760
7848	7988	8100	8386	8838
7942	8084	8200	8479	8915
8037	8181	8300	8571	8993
8131	8277	8400	8663	9070
8226	8374	8500	8755	9148
8320	8470	8600	8847	9225
8415	8567	8700	8939	9303
8509	8663	8800	9031	9380
8604	8760	8900	9124	9458
8698	8856	9000	9216	9535
8793	8953	9100	9308	9613
8887	9049	9200	9400	9690
8982	9146	9300	9492	9768
9076	9242	9400	9584	9845
9171	9339	9500	9676	9923
9265	9435	9600	9769	10000
9360	9532	9700	9861	10078
9454	9628	9800	9953	10155
9549	9725	9900	10045	10233
9643	9821	10000	10137	10310
9738	9918	10100	10229	10388
9832	10014	10200	10321	10465
9927	10111	10300	10414	10543
10021	10207	10400	10506	10600
10116	10304	10500	10598	10600
10210	10400	10600	10600	10600

BESC1_TO_0_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-47

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude -1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	109	110	111
	VR	113	113	113	113	113	114	115
	V2	124	123	122	121	120	120	120
	TOFL	2698	2726	2797	2868	2939	3081	3204
15	V1	106	106	107	107	108	109	110
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	120	119	120
	TOFL	3182	3214	3304	3393	3482	3627	3809
25	V1	106	106	106	107	107	109	110
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	120	119	120
	TOFL	3275	3300	3393	3485	3577	3720	3913
35	V1	106	107	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	121	120	120	119	120	120	120
	TOFL	3532	3574	3679	3852	4030	4199	4438
40	V1	107	108	110	111	112	114	115
	VR	113	113	115	115	116	118	119
	V2	120	119	120	120	120	121	121
	TOFL	3715	3772	4011	4160	4366	4581	4840
45	V1	110	110	112	113	115	116	117
	VR	115	115	116	117	118	119	120
	V2	120	119	120	120	121	121	122
	TOFL	4072	4123	4336	4565	4798	5033	5316
50	V1	112	113	114	116	117	118	120
	VR	117	117	118	119	120	121	122
	V2	120	120	121	121	122	122	123
	TOFL	4458	4559	4812	5055	5312	5569	5882
55	V1	115	116	117	118	119	121	122
	VR	119	119	120	121	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	4977	5087	5364	5643	5915	6201	6610

TOFL_TO_0_-1_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-48

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude Sea Level								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	109	110	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	2781	2810	2884	2958	3032	3187	3318
15	V1	105	106	106	107	108	109	110
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	119	119	120
	TOFL	3282	3314	3407	3500	3593	3750	3934
25	V1	105	106	106	107	107	109	110
	VR	113	113	113	113	113	114	115
	V2	122	122	121	120	119	119	120
	TOFL	3378	3404	3500	3597	3693	3849	4045
35	V1	107	107	108	110	111	112	114
	VR	113	113	114	115	115	117	118
	V2	120	120	119	120	120	120	121
	TOFL	3698	3742	3891	4111	4265	4472	4728
40	V1	108	109	110	112	113	115	116
	VR	114	115	115	116	117	118	120
	V2	119	119	119	120	120	121	122
	TOFL	3959	4062	4227	4448	4677	4908	5186
45	V1	110	111	113	114	116	117	118
	VR	115	116	117	118	119	120	121
	V2	119	120	120	121	121	122	122
	TOFL	4312	4413	4661	4901	5151	5404	5709
50	V1	114	114	116	117	118	119	120
	VR	118	118	119	120	121	122	123
	V2	121	121	121	122	122	123	123
	TOFL	4807	4916	5187	5460	5726	6005	6343
55	V1	117	117	118	120	121	122	123
	VR	120	120	121	122	123	123	124
	V2	122	122	122	123	123	123	124
	TOFL	5387	5506	5805	6107	6400	6714	7142

TOFL_TO_0_0_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-49

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	109	111	112
	VR	113	113	113	113	114	115	116
	V2	123	122	121	120	120	120	120
	TOFL	2892	2923	3002	3081	3197	3352	3493
15	V1	106	106	106	107	108	110	111
	VR	113	113	113	113	113	115	115
	V2	122	122	121	120	119	120	120
	TOFL	3410	3450	3548	3646	3756	3957	4127
25	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	114	115	116
	V2	121	121	120	119	119	120	120
	TOFL	3548	3590	3694	3798	3959	4152	4347
30	V1	106	107	107	109	110	112	113
	VR	113	113	113	114	115	116	117
	V2	120	120	119	119	119	120	120
	TOFL	3721	3766	3879	4094	4254	4455	4710
35	V1	107	108	110	111	112	114	115
	VR	113	113	115	116	117	118	119
	V2	119	119	120	120	120	121	121
	TOFL	3914	3998	4236	4404	4626	4856	5133
40	V1	110	110	112	113	115	116	117
	VR	115	115	116	117	118	119	120
	V2	120	119	120	120	121	121	122
	TOFL	4290	4345	4583	4829	5078	5329	5631
45	V1	112	113	114	116	117	118	120
	VR	117	117	118	119	120	121	122
	V2	120	120	121	121	122	122	123
	TOFL	4707	4814	5084	5354	5617	5891	6225
50	V1	115	116	117	119	120	121	122
	VR	119	119	120	121	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	5257	5375	5670	5968	6257	6562	6961

TOFL_TO_0_1_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-50

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 2000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	110	111	112
	VR	113	113	113	113	114	115	116
	V2	122	122	121	120	120	120	120
	TOFL	3008	3041	3124	3208	3365	3495	3674
10	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	114	115	116
	V2	122	121	120	119	119	120	120
	TOFL	3501	3542	3644	3746	3900	4094	4281
20	V1	105	106	106	107	108	110	111
	VR	113	113	113	113	114	115	116
	V2	121	121	120	119	119	120	120
	TOFL	3598	3641	3747	3853	4007	4213	4402
30	V1	107	107	109	110	111	113	115
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	120	120	121
	TOFL	3941	3990	4218	4391	4609	4839	5117
35	V1	109	110	111	112	114	115	117
	VR	114	115	116	117	118	119	120
	V2	119	120	119	120	121	121	122
	TOFL	4263	4357	4545	4782	5031	5280	5582
40	V1	111	112	113	115	116	117	119
	VR	116	116	117	118	119	120	121
	V2	120	120	120	121	121	122	122
	TOFL	4625	4725	5000	5259	5530	5803	6133
45	V1	114	115	116	117	118	120	121
	VR	118	118	119	120	121	122	123
	V2	121	121	121	122	122	123	123
	TOFL	5141	5258	5550	5844	6129	6430	6795
50	V1	117	118	119	120	121	122	123
	VR	120	120	121	122	123	124	124
	V2	122	122	122	123	123	124	124
	TOFL	5761	5889	6211	6536	6852	7187	7614

TOFL_TO_0_2_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-51

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 3000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	109	110	111	113
	VR	113	113	113	113	115	115	117
	V2	122	121	120	119	120	120	120
	TOFL	3132	3167	3255	3370	3552	3670	3870
10	V1	106	106	107	107	109	110	112
	VR	113	113	113	113	114	115	116
	V2	121	121	120	119	119	120	120
	TOFL	3648	3691	3799	3908	4117	4276	4513
20	V1	106	106	107	107	109	110	112
	VR	113	113	113	113	115	115	117
	V2	121	120	120	119	119	119	120
	TOFL	3776	3821	3935	4064	4293	4448	4702
25	V1	107	107	108	110	111	112	114
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	119	120	121
	TOFL	3970	4019	4199	4423	4607	4836	5115
30	V1	108	109	110	112	113	114	116
	VR	114	114	115	116	117	118	120
	V2	119	119	119	120	120	121	121
	TOFL	4241	4353	4536	4776	5026	5277	5580
35	V1	110	111	112	114	115	116	118
	VR	115	116	117	118	119	120	121
	V2	119	119	120	121	121	122	122
	TOFL	4587	4697	4964	5225	5496	5768	6098
40	V1	113	113	115	116	117	119	120
	VR	117	118	119	120	121	121	122
	V2	120	120	121	122	122	122	123
	TOFL	5052	5169	5460	5752	6036	6335	6695
45	V1	116	116	118	119	120	121	122
	VR	119	120	121	121	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	5627	5754	6073	6393	6705	7034	7436

TOFL_TO_0_3_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-52

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 4000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	109	111	112	113
	VR	113	113	113	114	115	116	117
	V2	121	121	120	120	120	120	121
	TOFL	3263	3301	3395	3555	3712	3867	4083
10	V1	106	106	107	108	110	111	112
	VR	113	113	113	114	115	115	117
	V2	121	120	119	119	120	119	120
	TOFL	3806	3852	3967	4125	4351	4510	4769
20	V1	106	106	107	109	110	112	113
	VR	113	113	113	115	115	116	118
	V2	120	120	118	119	119	120	121
	TOFL	4001	4050	4192	4447	4609	4836	5116
25	V1	107	108	110	111	112	114	115
	VR	113	114	115	116	117	118	119
	V2	118	119	119	119	120	121	121
	TOFL	4224	4333	4568	4776	5024	5276	5580
30	V1	110	110	112	113	115	116	117
	VR	115	115	116	117	119	120	121
	V2	119	119	120	120	121	121	122
	TOFL	4621	4693	4964	5226	5499	5773	6104
35	V1	112	113	114	115	117	118	119
	VR	117	117	118	119	120	121	122
	V2	120	120	121	121	122	122	123
	TOFL	5037	5154	5447	5730	6026	6325	6687
40	V1	115	115	117	118	119	120	121
	VR	118	119	120	121	122	122	123
	V2	121	121	122	122	123	123	123
	TOFL	5561	5688	6005	6325	6636	6963	7362
45	V1	117	118	119	120	121	122	
	VR	120	121	122	122	123	124	
	V2	122	122	123	123	123	124	
	TOFL	6182	6320	6668	7020	7360	7723	

TOFL_TO_0_4_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-53

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 5000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	110	111	112	114
	VR	113	113	113	115	115	116	118
	V2	121	120	119	120	120	120	121
	TOFL	3402	3442	3558	3764	3896	4080	4308
5	V1	106	106	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	120	120	119	119	119	120	120
	TOFL	3918	3966	4087	4313	4487	4699	4971
15	V1	106	106	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	120	120	119	119	119	120	120
	TOFL	4040	4090	4215	4453	4630	4853	5136
25	V1	109	110	111	113	114	115	117
	VR	115	115	116	117	118	119	120
	V2	119	119	119	120	121	121	122
	TOFL	4659	4720	4957	5225	5500	5775	6108
30	V1	111	112	114	115	116	117	119
	VR	116	117	118	119	120	121	122
	V2	120	120	120	121	122	122	123
	TOFL	5032	5155	5450	5735	6034	6334	6698
35	V1	114	114	116	117	118	119	121
	VR	118	118	119	120	121	122	123
	V2	121	121	121	122	122	123	123
	TOFL	5542	5670	5990	6311	6622	6951	7349
40	V1	117	117	118	120	121	122	123
	VR	120	120	121	122	123	123	124
	V2	122	122	122	123	123	123	124
	TOFL	6145	6284	6632	6984	7325	7687	8128
45	V1	119	120	121	122	123		
	VR	122	122	123	124	124		
	V2	123	123	123	124	124		
	TOFL	6857	7009	7392	7780	8156		

TOFL_TO_0_5_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-54

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 6000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	120	120	119	120	120	120	121
	TOFL	3549	3591	3754	3942	4106	4305	4547
0	V1	106	107	108	110	110	112	114
	VR	113	113	114	115	115	117	118
	V2	120	119	119	119	119	120	121
	TOFL	4036	4086	4249	4497	4668	4900	5183
10	V1	106	106	108	109	110	112	114
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	119	120	121
	TOFL	4166	4218	4394	4646	4829	5070	5365
20	V1	108	109	110	112	113	115	116
	VR	114	115	115	117	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4642	4757	4957	5221	5497	5774	6108
25	V1	110	111	113	114	116	117	118
	VR	116	116	117	118	119	120	121
	V2	119	119	120	121	121	122	122
	TOFL	5024	5137	5439	5727	6027	6328	6693
30	V1	113	114	115	117	118	119	120
	VR	118	118	119	120	121	122	123
	V2	120	121	121	122	122	123	123
	TOFL	5541	5669	5991	6315	6629	6959	7359
35	V1	116	116	118	119	120	121	122
	VR	119	120	121	122	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	6126	6266	6616	6969	7311	7674	8115
40	V1	118	119	120	121	122	123	
	VR	121	122	122	123	124	124	
	V2	122	122	123	123	124	124	
	TOFL	6819	6972	7355	7743	8119	8521	

TOFL_TO_0_6_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-55

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 7000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	109	110	112	113	115
	VR	113	113	115	115	116	117	119
	V2	120	119	120	119	120	121	121
	TOFL	3704	3749	3973	4127	4331	4544	4801
0	V1	106	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	119	119	119	119	120	120	121
	TOFL	4227	4281	4523	4718	4952	5202	5504
10	V1	106	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	119	119	119	119	120	120	121
	TOFL	4365	4421	4683	4876	5127	5388	5703
15	V1	108	109	110	111	113	114	116
	VR	114	115	115	116	117	118	120
	V2	119	119	119	120	120	121	121
	TOFL	4628	4755	4958	5225	5503	5782	6118
20	V1	110	110	112	114	115	116	118
	VR	115	116	117	118	119	120	121
	V2	119	119	120	120	121	122	122
	TOFL	5010	5133	5431	5720	6021	6324	6690
25	V1	112	113	115	116	117	118	120
	VR	117	118	119	120	121	121	122
	V2	120	120	121	121	122	122	123
	TOFL	5529	5659	5983	6308	6624	6956	7357
30	V1	115	116	117	118	119	120	122
	VR	119	119	120	121	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	6117	6257	6609	6963	7308	7671	8114
35	V1	118	118	119	121	122	122	
	VR	121	121	122	123	123	124	
	V2	122	122	123	123	123	124	
	TOFL	6782	6936	7321	7709	8085	8486	

TOFL_TO_0_7_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-56

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 8000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	108	110	111	112	114	115
	VR	113	114	115	116	117	118	119
	V2	119	119	120	120	120	121	121
	TOFL	3867	3955	4183	4353	4572	4798	5070
-10	V1	107	107	109	110	112	113	115
	VR	113	113	115	115	117	118	119
	V2	119	118	119	119	120	120	121
	TOFL	4283	4368	4636	4823	5071	5327	5635
0	V1	107	107	109	110	112	113	115
	VR	113	113	115	116	117	118	119
	V2	119	118	119	119	120	120	121
	TOFL	4432	4530	4805	5004	5264	5531	5852
10	V1	107	108	109	111	112	114	115
	VR	113	114	115	116	117	118	119
	V2	118	119	119	119	120	121	121
	TOFL	4627	4739	4994	5232	5509	5790	6128
15	V1	109	110	112	113	114	116	117
	VR	115	115	116	118	119	120	121
	V2	119	119	120	120	121	121	122
	TOFL	5051	5137	5439	5731	6034	6339	6708
20	V1	112	112	114	115	117	118	119
	VR	117	117	118	119	120	121	122
	V2	120	120	121	121	122	122	123
	TOFL	5517	5647	5973	6288	6618	6950	7353
25	V1	114	115	116	118	119	120	121
	VR	118	119	120	121	122	122	123
	V2	121	121	122	122	122	123	123
	TOFL	6101	6243	6598	6954	7300	7664	8107
30	V1	117	118	119	120	121	122	
	VR	120	121	122	122	123	124	
	V2	122	122	122	123	123	124	
	TOFL	6769	6923	7310	7700	8079	8481	

TOFL_TO_0_8_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-57

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 9000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	108	109	110	111	113	114	116
	VR	114	114	115	116	117	118	119
	V2	119	119	119	120	120	121	121
	TOFL	4080	4175	4374	4590	4828	5067	5356
-10	V1	107	108	110	111	113	114	116
	VR	114	114	115	116	117	118	119
	V2	118	119	119	120	120	121	121
	TOFL	4535	4654	4871	5120	5391	5663	5992
0	V1	107	108	109	111	113	114	116
	VR	114	114	115	116	117	118	119
	V2	118	119	119	119	120	121	121
	TOFL	4695	4821	5039	5305	5588	5872	6214
10	V1	109	109	111	112	114	115	117
	VR	115	115	116	117	118	119	120
	V2	119	119	119	120	121	121	122
	TOFL	5094	5164	5439	5741	6047	6354	6725
15	V1	111	112	113	115	116	117	119
	VR	116	117	118	119	120	121	122
	V2	119	120	120	121	121	122	123
	TOFL	5520	5660	5989	6307	6640	6974	7380
20	V1	114	114	116	117	118	119	121
	VR	118	118	120	120	121	122	123
	V2	120	121	121	122	122	123	123
	TOFL	6092	6234	6591	6949	7297	7663	8108
25	V1	116	117	118	119	120	121	123
	VR	120	120	121	122	123	123	124
	V2	121	122	122	123	123	123	124
	TOFL	6755	6911	7300	7691	8071	8474	8965
30	V1	119	119	121	122	123		
	VR	122	122	123	123	124		
	V2	122	123	123	123	124		
	TOFL	7517	7687	8113	8543	8960		

TOFL_TO_0_9_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-58

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off, Altitude 10,000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	109	109	110	112	113	115	116
	VR	114	115	115	117	118	119	120
	V2	119	120	119	120	121	121	122
	TOFL	4329	4442	4620	4862	5115	5369	5676
-15	V1	108	109	110	112	113	115	116
	VR	114	115	115	117	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4750	4868	5073	5345	5628	5912	6255
-5	V1	108	109	110	112	113	115	116
	VR	114	115	116	117	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4914	5028	5251	5534	5830	6126	6483
5	V1	108	109	110	112	113	115	116
	VR	115	115	116	117	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	5101	5207	5445	5743	6051	6360	6733
10	V1	110	111	113	114	116	117	118
	VR	116	116	118	119	120	121	122
	V2	119	120	120	121	121	122	122
	TOFL	5550	5685	6018	6339	6676	7014	7422
15	V1	113	114	115	117	118	119	120
	VR	118	118	119	120	121	122	123
	V2	120	120	121	122	122	123	123
	TOFL	6102	6246	6606	6967	7319	7687	8134
20	V1	116	116	118	119	120	121	122
	VR	119	120	121	122	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	6744	6900	7291	7684	8067	8471	8963
25	V1	118	119	120	121	122	123	
	VR	121	122	122	123	124	124	
	V2	122	122	123	123	124	124	
	TOFL	7520	7691	8120	8553	8972	9420	

TOFL_TO_0_10_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-59

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected V1 [KIAS]		
FLAPS TO/APPR		
Ice Protection Off		
Tailwind		Headwind
10	◀ REF [0] ▶	30
104	105	106
105	106	107
106	107	108
107	108	109
108	109	110
109	110	111
110	111	112
111	112	113
112	113	114
113	114	115
114	115	116
115	116	117
116	117	118
117	118	119
118	119	119
119	120	120
121	121	121
122	122	122
123	123	123
124	124	124

V1WC_TO_0_05

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-60

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected V1 [KIAS]				
FLAPS TO/APPR, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
103	104	104	105	106
104	105	105	106	107
105	106	106	107	108
106	106	107	108	109
107	107	108	109	110
108	108	109	110	111
109	109	110	111	112
110	110	111	112	113
111	111	112	113	114
111	112	113	114	115
112	113	114	115	116
113	114	115	116	116
114	115	116	117	117
115	116	117	118	118
116	117	118	119	119
117	118	119	120	120
118	119	120	121	121
119	120	121	122	122
120	121	122	123	123
121	122	123	124	124
122	123	124	125	125
123	124	125	126	126

V1SC_TO_0_05

Takeoff Rotation Speed (V_R) Slope Correction, Ice Protection Off

NOTE V_R with flaps TO/APPR and Ice Protection Off does not require slope correction.

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HA-420 AFM

PERFORMANCE

Slope Corrected V2 [KIAS]				
FLAPS TO/APPR, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
120	119	118	117	116
121	120	119	118	117
122	121	120	119	118
123	122	121	120	119
124	123	122	121	120
125	124	123	122	121
126	125	124	123	122

V2SC_TO_0_05

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-62

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
3181	2600	2437	2295	2155
3291	2700	2533	2388	2244
3400	2800	2630	2481	2333
3510	2900	2726	2573	2423
3619	3000	2823	2666	2512
3729	3100	2919	2759	2601
3838	3200	3016	2852	2691
3947	3300	3112	2944	2780
4057	3400	3209	3037	2869
4166	3500	3306	3130	2959
4276	3600	3402	3222	3048
4385	3700	3499	3315	3137
4495	3800	3595	3408	3227
4604	3900	3692	3500	3316
4714	4000	3788	3593	3405
4823	4100	3885	3686	3494
4933	4200	3981	3778	3584
5042	4300	4078	3871	3673
5151	4400	4174	3964	3762
5261	4500	4271	4056	3852
5370	4600	4367	4149	3941
5480	4700	4464	4242	4030
5589	4800	4560	4334	4120
5699	4900	4657	4427	4209
5808	5000	4753	4520	4298
5918	5100	4850	4613	4388
6027	5200	4946	4705	4477
6137	5300	5043	4798	4566
6246	5400	5139	4891	4656
6356	5500	5236	4983	4745
6465	5600	5332	5076	4834
6574	5700	5429	5169	4924
6684	5800	5525	5261	5013
6793	5900	5622	5354	5102
6903	6000	5718	5447	5192
7012	6100	5815	5539	5281

TOWCT_TO_0_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-63

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
7012	6100	5815	5539	5281
7122	6200	5911	5632	5370
7231	6300	6008	5725	5460
7341	6400	6104	5817	5549
7450	6500	6201	5910	5638
7560	6600	6297	6003	5728
7669	6700	6394	6096	5817
7779	6800	6491	6188	5906
7888	6900	6587	6281	5995
7997	7000	6684	6374	6085
8107	7100	6780	6466	6174
8216	7200	6877	6559	6263
8326	7300	6973	6652	6353
8435	7400	7070	6744	6442
8545	7500	7166	6837	6531
8654	7600	7263	6930	6621
8764	7700	7359	7022	6710
8873	7800	7456	7115	6799
8983	7900	7552	7208	6889
9092	8000	7649	7300	6978
9202	8100	7745	7393	7067
9311	8200	7842	7486	7157
9420	8300	7938	7578	7246
9530	8400	8035	7671	7335
9639	8500	8131	7764	7425
9749	8600	8228	7857	7514
9858	8700	8324	7949	7603
9968	8800	8421	8042	7693
10077	8900	8517	8135	7782
10187	9000	8614	8227	7871
10296	9100	8710	8320	7961
10406	9200	8807	8413	8050
10515	9300	8903	8505	8139
10624	9400	9000	8598	8228

TOWC2_TO_0_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-64

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
2361	2222	2100	2072	2064
2450	2316	2200	2176	2173
2538	2411	2300	2281	2283
2627	2505	2400	2385	2392
2716	2599	2500	2489	2502
2805	2693	2600	2593	2612
2894	2787	2700	2697	2721
2982	2882	2800	2802	2831
3071	2976	2900	2906	2940
3160	3070	3000	3010	3050
3249	3164	3100	3114	3160
3338	3258	3200	3218	3269
3426	3353	3300	3323	3379
3515	3447	3400	3427	3488
3604	3541	3500	3531	3598
3693	3635	3600	3635	3708
3782	3729	3700	3739	3817
3870	3824	3800	3844	3927
3959	3918	3900	3948	4036
4048	4012	4000	4052	4146
4137	4106	4100	4156	4256
4226	4200	4200	4260	4365
4314	4295	4300	4365	4475
4403	4389	4400	4469	4584
4492	4483	4500	4573	4694
4581	4577	4600	4677	4804
4670	4671	4700	4781	4913
4758	4766	4800	4886	5023
4847	4860	4900	4990	5132
4936	4954	5000	5094	5242
5025	5048	5100	5198	5352
5114	5142	5200	5302	5461
5202	5237	5300	5407	5571
5291	5331	5400	5511	5680
5380	5425	5500	5615	5790
5469	5519	5600	5719	5900

10SC1_16_0_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-65

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
5469	5519	5600	5719	5900
5558	5613	5700	5823	6009
5646	5708	5800	5928	6119
5735	5802	5900	6032	6228
5824	5896	6000	6136	6338
5913	5990	6100	6240	6448
6002	6084	6200	6344	6557
6090	6179	6300	6449	6667
6179	6273	6400	6553	6776
6268	6367	6500	6657	6886
6357	6461	6600	6761	6996
6446	6555	6700	6865	7105
6534	6650	6800	6970	7215
6623	6744	6900	7074	7324
6712	6838	7000	7178	7434
6801	6932	7100	7282	7544
6890	7026	7200	7386	7653
6978	7121	7300	7491	7763
7067	7215	7400	7595	7872
7156	7309	7500	7699	7982
7245	7403	7600	7803	8092
7334	7497	7700	7907	8201
7422	7592	7800	8012	8311
7511	7686	7900	8116	8420
7600	7780	8000	8220	8530
7689	7874	8100	8324	8640
7778	7968	8200	8428	8749
7866	8063	8300	8533	8859
7955	8157	8400	8637	8968
8044	8251	8500	8741	9078
8133	8345	8600	8845	9188
8222	8439	8700	8949	9297
8310	8534	8800	9054	9407
8399	8628	8900	9158	9516
8488	8722	9000	9262	9626
8577	8816	9100	9366	9736

TOSC2_TO_0_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-66

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
8577	8816	9100	9366	9736
8666	8910	9200	9470	9845
8754	9005	9300	9575	9955
8843	9099	9400	9679	10064
8932	9193	9500	9783	10174
9021	9287	9600	9887	10284
9110	9381	9700	9991	10393
9198	9476	9800	10096	10503
9287	9570	9900	10200	10612
9376	9664	10000	10304	10722
9465	9758	10100	10408	10832
9554	9852	10200	10512	10941
9642	9947	10300	10617	11051
9731	10041	10400	10721	11160
9820	10135	10500	10825	11270
9909	10229	10600	10929	11380
9998	10323	10700	11033	11489

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October 30, 2016

HJ1-29000-003-001

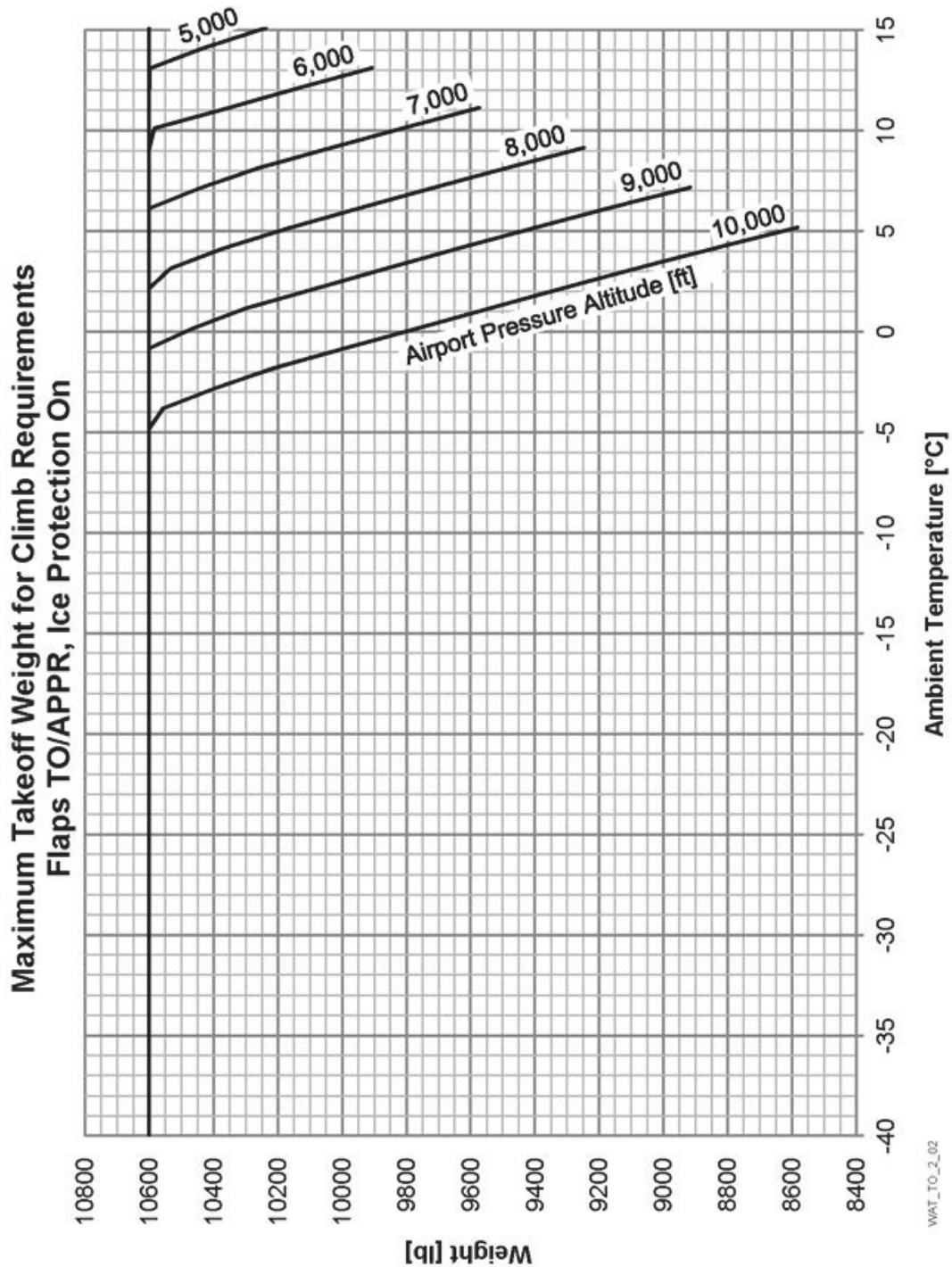
Page 5-67

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

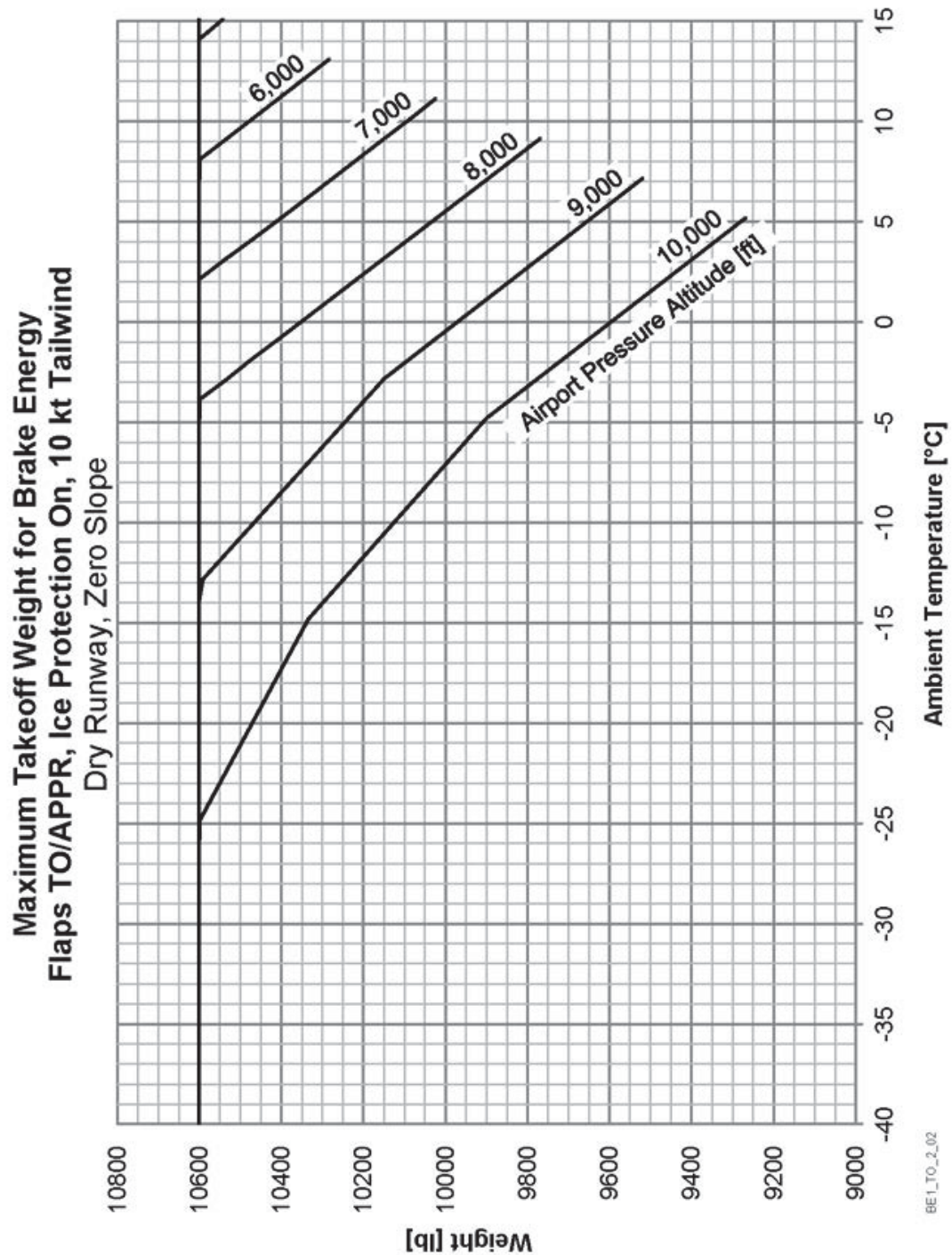
Page 5-68

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

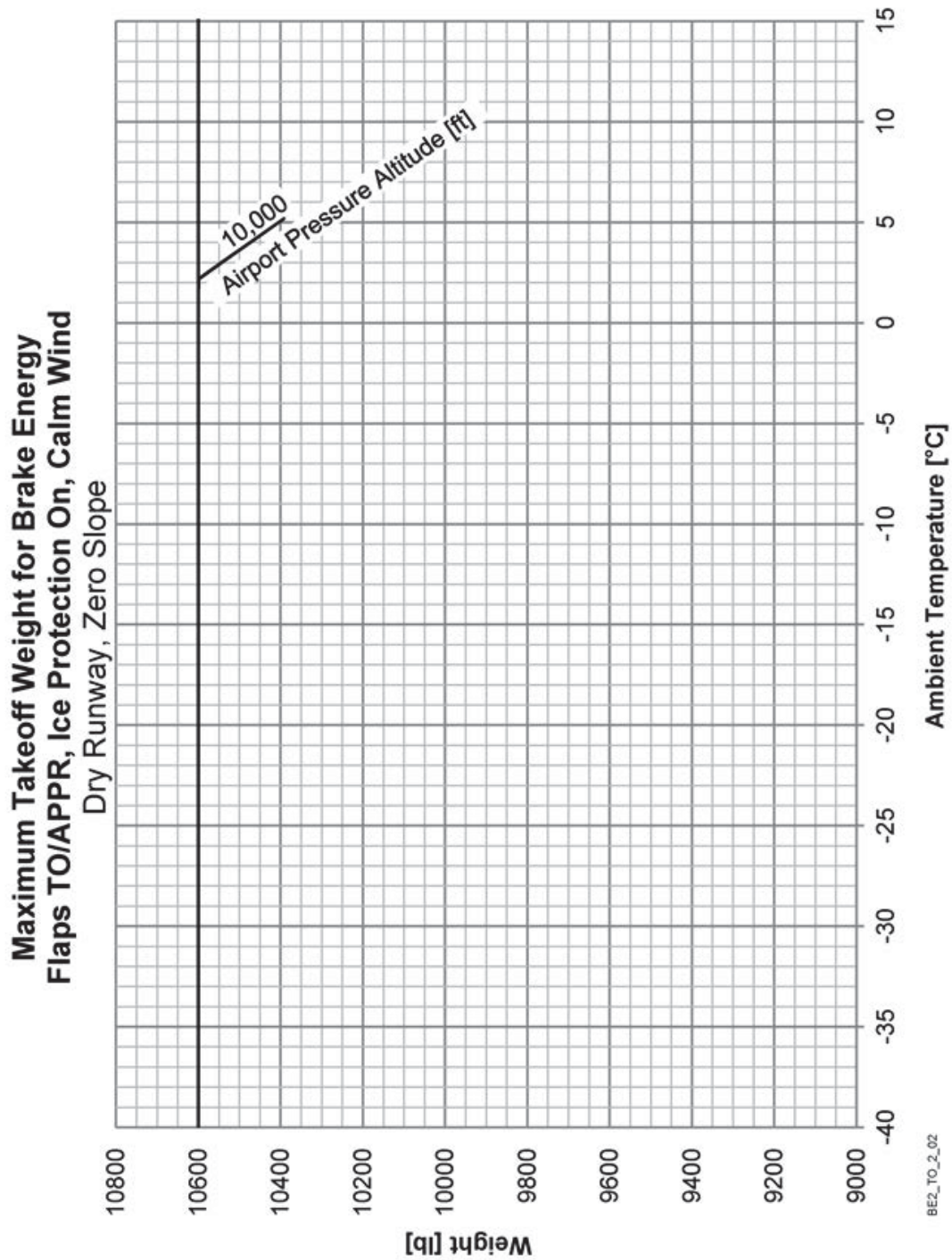
Page 5-69

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-70

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude -1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	109	110	111
	VR	113	113	113	113	113	114	115
	V2	124	123	122	121	120	120	120
	TOFL	2700	2729	2799	2870	2941	3081	3205
-30	V1	106	107	107	108	108	110	111
	VR	113	113	113	113	113	114	115
	V2	124	123	122	121	120	120	120
	TOFL	2789	2819	2893	2967	3042	3186	3318
-20	V1	106	107	107	108	108	110	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	2876	2907	2985	3062	3140	3287	3429
-10	V1	106	106	107	108	108	110	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	2965	2998	3079	3160	3241	3389	3541
-5	V1	106	106	107	107	108	109	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	3008	3041	3124	3206	3289	3434	3594
0	V1	106	106	107	107	108	109	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	3050	3084	3168	3252	3337	3480	3647
5	V1	106	106	107	107	108	109	111
	VR	113	113	113	113	113	114	115
	V2	123	122	121	121	120	119	120
	TOFL	3097	3127	3213	3299	3385	3526	3700
10	V1	106	106	107	107	108	109	110
	VR	113	113	113	113	113	114	115
	V2	123	122	121	121	120	119	120
	TOFL	3143	3170	3258	3345	3433	3572	3754

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-71

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude Sea Level								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	107	108	109	110	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	2782	2811	2885	2959	3034	3186	3318
-30	V1	106	107	107	108	108	110	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	2877	2908	2985	3062	3140	3296	3435
-20	V1	106	106	107	108	108	110	111
	VR	113	113	113	113	113	114	115
	V2	123	123	122	121	120	120	120
	TOFL	2964	2996	3078	3158	3240	3398	3540
-10	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	113	114	115
	V2	123	122	121	121	120	120	120
	TOFL	3055	3089	3174	3258	3343	3500	3655
-5	V1	106	106	107	107	108	109	111
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	120	120	120
	TOFL	3100	3134	3220	3306	3393	3548	3710
0	V1	106	106	107	107	108	109	111
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	120	120	120
	TOFL	3144	3179	3267	3354	3442	3595	3765
5	V1	106	106	107	107	108	109	110
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	120	119	120
	TOFL	3193	3224	3313	3403	3492	3644	3821
10	V1	106	106	106	107	108	109	110
	VR	113	113	113	113	113	114	115
	V2	123	122	121	120	120	119	120
	TOFL	3239	3268	3360	3451	3542	3694	3876

TOFL_TO_2_0_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-72

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	109	111	112
	VR	113	113	113	113	114	115	116
	V2	123	122	121	120	120	120	120
	TOFL	2892	2923	3002	3081	3195	3352	3491
-30	V1	106	107	107	108	109	111	112
	VR	113	113	113	113	114	115	116
	V2	123	122	121	120	119	120	120
	TOFL	2990	3023	3105	3187	3302	3470	3611
-20	V1	106	107	107	108	109	110	111
	VR	113	113	113	113	113	115	116
	V2	122	122	121	120	119	120	120
	TOFL	3083	3117	3203	3289	3403	3582	3726
-10	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	113	115	115
	V2	122	122	121	120	119	120	120
	TOFL	3177	3213	3302	3391	3502	3687	3839
-5	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	113	115	115
	V2	122	122	121	120	119	120	120
	TOFL	3223	3260	3351	3442	3551	3739	3895
0	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	113	115	115
	V2	122	122	121	120	119	120	120
	TOFL	3269	3307	3400	3493	3600	3791	3950
5	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	113	115	115
	V2	122	122	121	120	119	120	120
	TOFL	3316	3354	3449	3543	3650	3843	4007
10	V1	106	106	107	107	109	110	112
	VR	113	113	113	113	114	115	116
	V2	122	121	120	119	119	120	120
	TOFL	3381	3419	3515	3611	3770	3939	4128

TOFL_TO_2_1_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-73

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 2000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	108	110	111	112
	VR	113	113	113	113	114	115	116
	V2	122	122	121	120	120	120	120
	TOFL	3007	3040	3124	3207	3362	3494	3671
-30	V1	106	107	107	108	109	111	112
	VR	113	113	113	113	114	115	116
	V2	122	122	121	120	120	120	120
	TOFL	3107	3142	3229	3317	3470	3616	3795
-20	V1	106	107	107	108	109	111	112
	VR	113	113	113	113	114	115	116
	V2	122	122	121	120	119	120	120
	TOFL	3205	3241	3332	3423	3575	3735	3914
-10	V1	106	106	107	108	109	110	112
	VR	113	113	113	113	114	115	116
	V2	122	121	121	120	119	120	120
	TOFL	3303	3341	3436	3530	3680	3853	4033
-5	V1	106	106	107	107	109	110	111
	VR	113	113	113	113	114	115	116
	V2	122	121	120	120	119	120	120
	TOFL	3353	3391	3488	3584	3733	3913	4093
0	V1	106	106	107	107	109	110	111
	VR	113	113	113	113	114	115	116
	V2	122	121	120	120	119	120	120
	TOFL	3401	3441	3539	3638	3786	3972	4153
5	V1	106	106	107	107	108	110	111
	VR	113	113	113	113	114	115	116
	V2	122	121	120	120	119	120	120
	TOFL	3451	3491	3591	3692	3840	4032	4214
10	V1	106	107	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	121	120	119	119	120	120	120
	TOFL	3538	3579	3679	3845	4021	4183	4415

TOFL_TO_2_2_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-74

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 3000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	109	110	111	113
	VR	113	113	113	113	115	115	117
	V2	122	121	120	119	120	120	120
	TOFL	3130	3165	3254	3365	3546	3666	3865
-30	V1	106	107	107	108	110	111	112
	VR	113	113	113	113	115	115	116
	V2	122	121	120	119	120	120	120
	TOFL	3240	3276	3368	3476	3663	3790	3998
-20	V1	106	107	107	108	110	111	112
	VR	113	113	113	113	115	115	116
	V2	121	121	120	119	120	120	120
	TOFL	3334	3373	3469	3572	3766	3895	4116
-10	V1	106	106	107	108	109	110	112
	VR	113	113	113	113	114	115	116
	V2	121	121	120	119	120	120	120
	TOFL	3438	3478	3579	3679	3878	4021	4245
-5	V1	106	106	107	107	109	110	112
	VR	113	113	113	113	114	115	116
	V2	121	121	120	119	119	120	120
	TOFL	3490	3531	3633	3736	3936	4083	4309
0	V1	106	106	107	107	109	110	112
	VR	113	113	113	113	114	115	116
	V2	121	121	120	119	119	120	120
	TOFL	3542	3584	3688	3793	3994	4147	4375
5	V1	106	106	107	108	110	110	112
	VR	113	113	113	113	115	115	117
	V2	121	121	120	119	120	119	120
	TOFL	3604	3646	3752	3878	4093	4237	4475
10	V1	107	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	120	120	121
	TOFL	3715	3758	3938	4116	4296	4501	4748

TOFL_TO_2_3_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-75

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 4000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	109	111	112	113
	VR	113	113	113	114	115	116	117
	V2	121	121	120	120	120	120	121
	TOFL	3262	3299	3393	3549	3710	3861	4077
-30	V1	107	107	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	121	121	120	119	120	120	120
	TOFL	3371	3410	3508	3660	3838	3987	4211
-20	V1	106	107	107	108	110	111	113
	VR	113	113	113	114	115	116	117
	V2	121	121	120	119	120	120	120
	TOFL	3475	3516	3618	3768	3961	4110	4342
-10	V1	106	106	107	108	110	111	113
	VR	113	113	113	114	115	116	117
	V2	121	121	120	119	120	120	120
	TOFL	3585	3628	3734	3884	4090	4240	4481
-5	V1	106	106	107	108	110	111	112
	VR	113	113	113	114	115	115	117
	V2	121	120	120	119	120	120	120
	TOFL	3640	3683	3792	3941	4155	4305	4550
0	V1	106	106	107	108	110	111	112
	VR	113	113	113	114	115	115	117
	V2	121	120	120	119	120	119	120
	TOFL	3695	3739	3850	4000	4219	4371	4620
5	V1	106	107	108	110	110	112	114
	VR	113	113	113	115	115	116	118
	V2	120	120	119	120	119	120	121
	TOFL	3779	3824	3959	4193	4338	4545	4801
10	V1	108	109	110	112	113	114	116
	VR	114	114	115	116	117	118	119
	V2	119	119	119	120	120	121	121
	TOFL	3967	4065	4228	4436	4655	4874	5138

TOFL_TO_2_4_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-76

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 5000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	110	111	112	114
	VR	113	113	113	115	115	116	118
	V2	121	120	119	120	120	120	121
	TOFL	3399	3439	3541	3752	3886	4069	4297
-30	V1	107	107	107	109	110	112	113
	VR	113	113	113	115	115	116	117
	V2	121	120	119	120	119	120	121
	TOFL	3518	3560	3663	3875	4017	4208	4445
-20	V1	106	107	107	109	110	112	113
	VR	113	113	113	114	115	116	117
	V2	121	120	119	119	119	120	121
	TOFL	3623	3667	3775	3986	4138	4334	4581
-10	V1	106	107	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	120	120	119	119	119	120	120
	TOFL	3739	3784	3897	4108	4275	4473	4729
-5	V1	106	106	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	120	120	119	119	119	120	120
	TOFL	3797	3843	3959	4172	4343	4544	4805
0	V1	106	106	107	109	110	111	113
	VR	113	113	113	114	115	116	117
	V2	120	120	119	119	119	120	120
	TOFL	3856	3903	4020	4235	4412	4615	4881
5	V1	107	108	110	111	112	113	115
	VR	113	113	115	115	117	118	119
	V2	119	119	120	119	120	121	121
	TOFL	3969	4047	4283	4447	4665	4890	5161
10	V1	110	110	112	113	115	116	118
	VR	115	115	116	118	119	120	121
	V2	119	119	120	120	121	122	122
	TOFL	4298	4370	4595	4830	5065	5299	5583

TOFL_TO_2_5_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-77

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 6000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	108	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	120	120	119	120	120	120	121
	TOFL	3545	3587	3741	3937	4094	4292	4534
-30	V1	107	107	108	110	111	112	114
	VR	113	113	114	115	116	117	118
	V2	120	120	119	120	120	120	121
	TOFL	3666	3710	3861	4074	4230	4436	4687
-20	V1	107	107	108	110	111	112	114
	VR	113	113	113	115	115	117	118
	V2	120	120	119	120	119	120	121
	TOFL	3781	3827	3977	4207	4363	4576	4837
-10	V1	106	107	108	110	111	112	114
	VR	113	113	113	115	115	117	118
	V2	120	120	119	120	119	120	121
	TOFL	3904	3952	4103	4343	4505	4727	4999
-5	V1	106	107	108	110	110	112	114
	VR	113	113	113	115	115	117	118
	V2	120	120	119	120	119	120	121
	TOFL	3967	4016	4169	4414	4580	4806	5083
0	V1	107	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	119	119	119	119	120	120	121
	TOFL	4054	4104	4334	4514	4735	4970	5253
5	V1	109	110	111	112	114	115	117
	VR	115	115	116	117	118	119	120
	V2	119	119	119	120	121	121	122
	TOFL	4339	4408	4603	4838	5079	5321	5612
10	V1	112	112	114	115	117	118	119
	VR	116	117	118	119	120	121	122
	V2	120	120	121	121	122	122	123
	TOFL	4699	4800	5054	5299	5555	5811	6120

TOFL_TO_2_6_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-78

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 7000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	107	109	110	112	113	115
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	120	120	121
	TOFL	3702	3747	3958	4117	4318	4531	4787
-30	V1	107	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	120	120	121
	TOFL	3832	3878	4088	4264	4466	4687	4954
-20	V1	107	107	109	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	120	119	119	119	120	120	121
	TOFL	3952	4001	4205	4404	4608	4839	5116
-10	V1	107	107	108	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	119	119	119	119	120	120	121
	TOFL	4085	4136	4347	4555	4768	5008	5297
-5	V1	106	107	108	110	111	113	114
	VR	113	113	114	115	116	117	118
	V2	119	119	119	119	120	120	121
	TOFL	4155	4207	4434	4635	4855	5100	5396
0	V1	108	109	110	112	113	114	116
	VR	114	115	115	116	117	118	120
	V2	119	119	119	120	120	121	121
	TOFL	4338	4454	4634	4874	5124	5373	5675
5	V1	111	111	113	114	116	117	118
	VR	116	116	117	118	119	120	121
	V2	119	120	120	121	121	122	122
	TOFL	4670	4768	5031	5280	5539	5798	6112
10	V1	114	114	116	117	118	120	121
	VR	118	118	119	120	121	122	123
	V2	121	121	121	122	122	123	123
	TOFL	5180	5290	5566	5831	6110	6389	6729

TOFL_TO_2_7_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-79

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 8000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	107	108	110	111	112	114	115
	VR	113	113	115	115	117	118	119
	V2	119	119	120	120	120	121	121
	TOFL	3865	3943	4177	4342	4560	4785	5056
-30	V1	107	107	110	110	112	113	115
	VR	113	113	115	115	116	118	119
	V2	119	119	120	119	120	121	121
	TOFL	4000	4072	4327	4490	4716	4951	5234
-20	V1	107	107	109	110	112	113	115
	VR	113	113	115	115	116	118	119
	V2	119	118	119	119	120	120	121
	TOFL	4134	4207	4473	4644	4881	5126	5421
-10	V1	107	107	109	110	112	113	115
	VR	113	113	115	115	117	118	119
	V2	119	118	119	119	120	120	121
	TOFL	4279	4359	4636	4815	5063	5318	5626
-5	V1	107	108	110	111	112	114	115
	VR	113	114	115	116	117	118	119
	V2	118	119	119	119	120	121	121
	TOFL	4369	4482	4730	4942	5199	5460	5775
0	V1	110	110	112	113	115	116	118
	VR	115	115	117	118	119	120	121
	V2	119	119	120	120	121	122	122
	TOFL	4684	4778	5043	5299	5566	5833	6155
5	V1	113	113	115	116	118	119	120
	VR	117	118	119	120	121	122	122
	V2	120	120	121	121	122	122	123
	TOFL	5126	5237	5515	5782	6062	6343	6683
10	V1	116	117	118	119	120	121	122
	VR	120	120	121	122	122	123	124
	V2	121	122	122	123	123	123	124
	TOFL	5719	5839	6138	6439	6731	7038	7412

TOFL_TO_2_8_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-80

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 9000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	108	108	110	111	113	114	116
	VR	113	114	115	116	117	118	119
	V2	119	119	119	120	120	121	121
	TOFL	4068	4164	4371	4580	4816	5055	5343
-30	V1	107	108	110	111	113	114	115
	VR	113	114	115	116	117	118	119
	V2	119	119	119	120	120	121	121
	TOFL	4210	4310	4532	4746	4992	5241	5542
-20	V1	107	108	110	111	113	114	116
	VR	113	114	115	116	117	118	119
	V2	118	119	119	120	120	121	121
	TOFL	4361	4465	4693	4921	5179	5440	5754
-10	V1	107	108	110	111	113	114	116
	VR	114	114	115	116	117	118	119
	V2	118	119	119	120	120	121	121
	TOFL	4528	4646	4866	5113	5384	5656	5985
-5	V1	109	109	111	113	114	115	117
	VR	115	115	116	117	118	119	120
	V2	119	119	119	120	121	121	122
	TOFL	4768	4837	5081	5355	5632	5909	6244
0	V1	112	112	114	115	117	118	119
	VR	116	117	118	119	120	121	122
	V2	120	120	121	121	122	122	123
	TOFL	5110	5223	5507	5781	6067	6354	6700
5	V1	115	116	117	118	119	120	122
	VR	119	119	120	121	122	123	124
	V2	121	121	122	122	123	123	124
	TOFL	5656	5777	6079	6381	6673	6981	7354
10	V1	118	119	120	121	122	123	
	VR	121	121	122	123	124	124	
	V2	122	122	123	123	124	124	
	TOFL	6324	6454	6780	7109	7427	7764	

TOFL_TO_2_9_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-81

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection On, Altitude 10,000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	108	109	110	112	113	115	116
	VR	114	115	115	116	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4314	4434	4612	4855	5108	5362	5669
-30	V1	108	109	110	112	113	115	116
	VR	114	115	115	116	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4474	4600	4787	5040	5305	5570	5890
-20	V1	108	109	110	112	113	115	116
	VR	114	115	115	117	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4653	4775	4971	5236	5513	5791	6126
-10	V1	108	109	110	112	113	115	116
	VR	114	115	116	117	118	119	120
	V2	119	119	119	120	120	121	122
	TOFL	4832	4943	5161	5440	5729	6020	6370
-5	V1	110	111	113	114	116	117	118
	VR	116	116	117	118	119	120	121
	V2	119	119	120	121	121	122	122
	TOFL	5107	5215	5515	5797	6091	6385	6741
0	V1	114	114	116	117	118	119	121
	VR	118	118	120	120	121	122	123
	V2	121	121	121	122	122	123	123
	TOFL	5596	5718	6021	6312	6619	6927	7300
5	V1	117	118	119	120	121	122	
	VR	120	121	122	122	123	124	
	V2	122	122	122	123	123	124	
	TOFL	6264	6395	6723	7053	7373	7711	
10	V1	120	121	122	123			
	VR	122	123	123	124			
	V2	123	123	123	124			
	TOFL	7018	7160	7516	7875			

TOFL_TO_2_10_03

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HA-420 AFM

PERFORMANCE

Wind Corrected V1 [KIAS]		
FLAPS TO/APPR Ice Protection On		
Tailwind		Headwind
10	◀ REF [0] ▶	30
104	105	107
105	106	108
106	107	109
107	108	109
108	109	110
109	110	111
110	111	112
111	112	113
112	113	114
113	114	115
114	115	116
115	116	117
116	117	118
117	118	119
118	119	120
119	120	121
120	121	121
121	122	122
122	123	123
123	124	124

V1WC_TO_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-83

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HA-420 AFM

PERFORMANCE

Slope Corrected V1 [KIAS]				
FLAPS TO/APPR, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
103	104	104	105	106
104	105	105	106	107
105	106	106	107	108
106	106	107	108	109
107	107	108	109	110
108	108	109	110	111
109	109	110	111	112
110	110	111	112	113
111	111	112	113	114
112	112	113	114	115
112	113	114	115	115
113	114	115	116	116
114	115	116	117	117
115	116	117	118	118
116	117	118	119	119
117	118	119	120	120
118	119	120	121	121
119	120	121	122	122
120	121	122	123	123
121	122	123	124	124
122	123	124	125	125

V1SC_TO_2_05

Takeoff Rotation Speed (V_R) Slope Correction, Ice Protection On

NOTE V_R with flaps TO/APPR and Ice Protection On does not require slope correction.

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HA-420 AFM

PERFORMANCE

Slope Corrected V2 [KIAS]				
FLAPS TO/APPR, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
120	119	118	117	116
121	120	119	118	117
122	121	120	119	118
123	122	121	120	119
124	123	122	121	120
125	124	123	122	121
126	125	124	123	122

V2SC_TO_2_06

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October 30, 2016

HJ1-29000-003-001

Page 5-85

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HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
3180	2600	2447	2293	2159
3289	2700	2543	2385	2248
3399	2800	2640	2478	2337
3509	2900	2736	2570	2426
3618	3000	2833	2663	2515
3728	3100	2929	2755	2603
3837	3200	3026	2848	2692
3947	3300	3122	2940	2781
4057	3400	3219	3033	2870
4166	3500	3315	3125	2959
4276	3600	3412	3218	3048
4386	3700	3509	3311	3137
4495	3800	3605	3403	3226
4605	3900	3702	3496	3315
4715	4000	3798	3588	3404
4824	4100	3895	3681	3493
4934	4200	3991	3773	3581
5043	4300	4088	3866	3670
5153	4400	4184	3958	3759
5263	4500	4281	4051	3848
5372	4600	4377	4143	3937
5482	4700	4474	4236	4026
5592	4800	4571	4329	4115
5701	4900	4667	4421	4204
5811	5000	4764	4514	4293
5921	5100	4860	4606	4382
6030	5200	4957	4699	4471
6140	5300	5053	4791	4559
6249	5400	5150	4884	4648
6359	5500	5246	4976	4737
6469	5600	5343	5069	4826
6578	5700	5439	5161	4915
6688	5800	5536	5254	5004
6798	5900	5633	5347	5093
6907	6000	5729	5439	5182
7017	6100	5826	5532	5271

TOWC1_TO_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-86

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HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
7017	6100	5826	5532	5271
7127	6200	5922	5624	5360
7236	6300	6019	5717	5449
7346	6400	6115	5809	5537
7455	6500	6212	5902	5626
7565	6600	6308	5994	5715
7675	6700	6405	6087	5804
7784	6800	6501	6179	5893
7894	6900	6598	6272	5982
8004	7000	6695	6365	6071
8113	7100	6791	6457	6160
8223	7200	6888	6550	6249
8333	7300	6984	6642	6338
8442	7400	7081	6735	6427
8552	7500	7177	6827	6515
8661	7600	7274	6920	6604
8771	7700	7370	7012	6693
8881	7800	7467	7105	6782
8990	7900	7563	7197	6871
9100	8000	7660	7290	6960
9210	8100	7757	7383	7049
9319	8200	7853	7475	7138
9429	8300	7950	7568	7227
9539	8400	8046	7660	7316
9648	8500	8143	7753	7405
9758	8600	8239	7845	7493
9867	8700	8336	7938	7582
9977	8800	8432	8030	7671
10087	8900	8529	8123	7760
10196	9000	8625	8215	7849
10306	9100	8722	8308	7938
10416	9200	8819	8401	8027
10525	9300	8915	8493	8116
10635	9400	9012	8586	8205

TOWC2_TO_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-87

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
2318	2209	2100	2096	2075
2409	2304	2200	2199	2186
2499	2400	2300	2303	2297
2590	2495	2400	2406	2409
2680	2590	2500	2510	2520
2770	2685	2600	2614	2631
2861	2780	2700	2717	2743
2951	2876	2800	2821	2854
3042	2971	2900	2924	2965
3132	3066	3000	3028	3077
3222	3161	3100	3131	3188
3313	3256	3200	3235	3299
3403	3352	3300	3338	3411
3494	3447	3400	3442	3522
3584	3542	3500	3546	3633
3674	3637	3600	3649	3745
3765	3732	3700	3753	3856
3855	3828	3800	3856	3967
3946	3923	3900	3960	4079
4036	4018	4000	4063	4190
4126	4113	4100	4167	4301
4217	4208	4200	4270	4413
4307	4304	4300	4374	4524
4398	4399	4400	4478	4635
4488	4494	4500	4581	4747
4578	4589	4600	4685	4858
4669	4684	4700	4788	4969
4759	4780	4800	4892	5081
4850	4875	4900	4995	5192
4940	4970	5000	5099	5303
5030	5065	5100	5202	5415
5121	5160	5200	5306	5526
5211	5256	5300	5410	5637
5302	5351	5400	5513	5749
5392	5446	5500	5617	5860
5482	5541	5600	5720	5971

TOSC1_TO_2_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-88

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
5482	5541	5600	5720	5971
5573	5636	5700	5824	6083
5663	5732	5800	5927	6194
5754	5827	5900	6031	6305
5844	5922	6000	6134	6417
5934	6017	6100	6238	6528
6025	6112	6200	6342	6639
6115	6208	6300	6445	6751
6206	6303	6400	6549	6862
6296	6398	6500	6652	6973
6386	6493	6600	6756	7085
6477	6588	6700	6859	7196
6567	6684	6800	6963	7307
6658	6779	6900	7066	7419
6748	6874	7000	7170	7530
6838	6969	7100	7274	7641
6929	7064	7200	7377	7753
7019	7160	7300	7481	7864
7110	7255	7400	7584	7975
7200	7350	7500	7688	8087
7290	7445	7600	7791	8198
7381	7540	7700	7895	8309
7471	7636	7800	7998	8421
7562	7731	7900	8102	8532
7652	7826	8000	8206	8643
7742	7921	8100	8309	8755
7833	8016	8200	8413	8866
7923	8112	8300	8516	8977
8014	8207	8400	8620	9089
8104	8302	8500	8723	9200
8194	8397	8600	8827	9311
8285	8492	8700	8930	9423
8375	8588	8800	9034	9534
8466	8683	8900	9138	9645

TOSC2_TO_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-89

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS TO/APPR, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
8466	8683	8900	9138	9645
8556	8778	9000	9241	9757
8646	8873	9100	9345	9868
8737	8968	9200	9448	9979
8827	9064	9300	9552	10091
8918	9159	9400	9655	10202
9008	9254	9500	9759	10313

TOSC3_TO_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-90

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps TO/APPR, Gear UP, Ice Protection Off, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	10.3	9.9	8.9	7.9	7.0	6.1	5.1
	15	10.6	10.2	9.1	8.1	7.2	6.3	5.3
	25	10.7	10.3	9.2	8.2	7.3	6.4	5.4
	35	8.9	8.5	7.5	6.5	5.7	4.8	3.9
	40	7.7	7.3	6.3	5.4	4.6	3.8	3.0
	45	6.5	6.1	5.2	4.3	3.6	2.9	2.1
	50	5.2	4.9	4.0	3.2	2.5	1.9	1.2
	55	4.0	3.6	2.9	2.2	1.6	1.0	0.3
Sea Level	-40	10.1	9.7	8.6	7.7	6.8	5.9	5.0
	15	10.4	10.0	8.9	8.0	7.0	6.2	5.2
	25	10.5	10.1	9.0	8.0	7.1	6.2	5.2
	35	8.3	7.9	6.9	6.0	5.1	4.3	3.4
	40	7.1	6.6	5.7	4.9	4.1	3.3	2.5
	45	5.8	5.5	4.6	3.8	3.0	2.4	1.6
	50	4.6	4.2	3.4	2.7	2.0	1.4	0.7
	55	3.4	3.0	2.3	1.7	1.1	0.5	-0.1
1000	-40	9.7	9.3	8.3	7.4	6.5	5.6	4.7
	15	10.1	9.7	8.6	7.6	6.7	5.9	4.9
	25	9.7	9.3	8.3	7.3	6.4	5.5	4.6
	30	8.6	8.2	7.2	6.3	5.4	4.6	3.7
	35	7.5	7.1	6.1	5.3	4.4	3.7	2.8
	40	6.3	5.9	5.0	4.2	3.4	2.7	2.0
	45	5.1	4.7	3.9	3.1	2.4	1.8	1.1
	50	3.9	3.5	2.8	2.1	1.5	0.9	0.2
2000	-40	9.4	9.0	8.0	7.0	6.1	5.3	4.4
	10	9.7	9.3	8.3	7.3	6.4	5.5	4.6
	20	9.8	9.3	8.3	7.4	6.4	5.6	4.6
	30	7.8	7.4	6.4	5.5	4.7	3.9	3.1
	35	6.7	6.3	5.4	4.6	3.8	3.1	2.3
	40	5.6	5.2	4.3	3.5	2.8	2.2	1.4
	45	4.4	4.0	3.2	2.5	1.9	1.2	0.6
	50	3.2	2.9	2.2	1.5	0.9	0.4	-0.3

TCG2_TO_0_1_06

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October 30, 2016

HJ1-29000-003-001

Page 5-91

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps TO/APPR, Gear UP, Ice Protection Off, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	9.0	8.6	7.6	6.7	5.8	5.0	4.1
	10	9.4	8.9	7.9	7.0	6.1	5.2	4.3
	20	9.2	8.8	7.8	6.8	5.9	5.1	4.2
	25	8.1	7.7	6.7	5.8	4.9	4.2	3.3
	30	7.0	6.6	5.7	4.8	4.0	3.3	2.5
	35	5.9	5.5	4.6	3.8	3.1	2.4	1.7
	40	4.8	4.4	3.6	2.9	2.2	1.6	0.9
	45	3.7	3.3	2.6	1.9	1.3	0.7	0.1
4000	-40	8.7	8.3	7.3	6.4	5.5	4.7	3.8
	10	9.0	8.6	7.6	6.6	5.7	4.9	4.0
	20	8.4	7.9	7.0	6.0	5.2	4.4	3.5
	25	7.3	6.9	5.9	5.0	4.2	3.5	2.7
	30	6.1	5.8	4.9	4.0	3.3	2.6	1.8
	35	5.1	4.7	3.9	3.1	2.4	1.8	1.1
	40	4.0	3.7	2.9	2.2	1.6	1.0	0.3
	45	3.0	2.7	2.0	1.3	0.8	0.2	
5000	-40	8.3	7.9	7.0	6.0	5.2	4.4	3.5
	5	8.6	8.2	7.2	6.2	5.4	4.6	3.7
	15	8.6	8.1	7.2	6.2	5.4	4.6	3.7
	25	6.4	6.0	5.1	4.3	3.5	2.8	2.0
	30	5.3	4.9	4.1	3.3	2.6	2.0	1.2
	35	4.3	3.9	3.2	2.4	1.8	1.2	0.5
	40	3.2	2.9	2.2	1.6	1.0	0.4	-0.2
	45	2.2	2.0	1.3	0.7	0.2		
6000	-40	8.0	7.6	6.7	5.7	4.9	4.1	3.3
	0	8.2	7.8	6.8	5.9	5.0	4.2	3.4
	10	8.2	7.7	6.8	5.8	5.0	4.2	3.3
	20	6.7	6.3	5.4	4.5	3.7	3.0	2.2
	25	5.6	5.2	4.4	3.6	2.8	2.2	1.4
	30	4.5	4.2	3.4	2.6	2.0	1.4	0.7
	35	3.5	3.2	2.4	1.8	1.2	0.6	0.0
	40	2.5	2.2	1.6	0.9	0.4	-0.1	

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October 30, 2016

HJ1-29000-003-001

Page 5-92

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps TO/APPR, Gear UP, Ice Protection Off, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	7.7	7.3	6.3	5.4	4.6	3.9	3.0
	0	7.8	7.4	6.4	5.5	4.7	3.9	3.1
	10	7.7	7.3	6.4	5.5	4.6	3.9	3.0
	15	6.9	6.5	5.6	4.7	3.9	3.2	2.4
	20	5.9	5.5	4.6	3.8	3.1	2.4	1.6
	25	4.8	4.4	3.6	2.9	2.2	1.6	0.9
	30	3.7	3.4	2.7	2.0	1.4	0.8	0.1
	35	2.8	2.5	1.8	1.2	0.6	0.1	
8000	-40	7.4	7.0	6.0	5.2	4.3	3.6	2.8
	-10	7.4	7.1	6.1	5.2	4.4	3.6	2.8
	0	7.4	7.0	6.0	5.1	4.3	3.6	2.8
	10	7.2	6.8	5.8	4.9	4.1	3.4	2.6
	15	6.0	5.7	4.8	4.0	3.2	2.5	1.8
	20	5.0	4.7	3.8	3.1	2.4	1.7	1.0
	25	4.0	3.6	2.9	2.2	1.6	1.0	0.3
	30	3.0	2.7	2.0	1.3	0.8	0.2	
9000	-40	7.1	6.7	5.7	4.9	4.1	3.3	2.5
	-10	7.0	6.6	5.7	4.8	4.0	3.3	2.5
	0	7.0	6.6	5.7	4.8	4.0	3.3	2.5
	10	6.3	5.9	5.0	4.2	3.4	2.7	1.9
	15	5.2	4.9	4.0	3.2	2.5	1.9	1.2
	20	4.2	3.9	3.1	2.4	1.7	1.1	0.5
	25	3.2	2.9	2.2	1.5	0.9	0.4	-0.2
	30	2.3	2.0	1.3	0.7	0.2		
10,000	-40	6.7	6.3	5.4	4.5	3.7	3.0	2.2
	-15	6.6	6.2	5.3	4.5	3.7	3.0	2.2
	-5	6.6	6.2	5.3	4.5	3.7	3.0	2.2
	5	6.5	6.2	5.3	4.4	3.6	2.9	2.1
	10	5.4	5.1	4.2	3.4	2.7	2.0	1.3
	15	4.4	4.1	3.3	2.6	1.9	1.3	0.6
	20	3.4	3.1	2.4	1.7	1.1	0.6	-0.1
	25	2.5	2.2	1.5	0.9	0.4	-0.2	

TCG2_TO_0_7_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-93

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Climb Gradient [%]				
FLAPS UP and TO/APPR, V2, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.0	0.0	0.0	0.0	0.0
0.5	0.5	0.5	0.5	0.5
0.9	1.0	1.0	1.0	1.1
1.4	1.5	1.5	1.6	1.6
1.8	2.0	2.1	2.1	2.2
2.3	2.5	2.6	2.7	2.8
2.7	3.0	3.1	3.2	3.4
3.2	3.5	3.6	3.8	3.9
3.6	4.0	4.2	4.3	4.5
4.0	4.5	4.7	4.9	5.1
4.5	5.0	5.2	5.4	5.7
4.9	5.5	5.7	6.0	6.2
5.4	6.0	6.2	6.5	6.8
5.8	6.5	6.8	7.1	7.4
6.3	7.0	7.3	7.6	7.9
6.7	7.5	7.8	8.2	8.5
7.1	8.0	8.3	8.7	9.1
7.6	8.5	8.9	9.2	9.7
8.0	9.0	9.4	9.8	10.2
8.5	9.5	9.9	10.3	10.8
8.9	10.0	10.4	10.9	11.4
9.4	10.5	10.9	11.4	12.0
9.8	11.0	11.5	12.0	12.5
10.2	11.5	12.0	12.5	13.1
10.7	12.0	12.5	13.1	13.7

CGWC_TO_ALL_05

NOTE Use this table when determining the ground reference flight path for obstacle clearance with Ice Protection Off.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps TO/APPR, Gear UP, Ice Protection On, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	10.3	9.9	8.9	7.9	7.0	6.1	5.2
	-30	10.4	9.9	8.9	7.9	7.0	6.1	5.2
	-20	10.4	10.0	8.9	8.0	7.0	6.2	5.2
	-10	10.5	10.0	9.0	8.0	7.1	6.2	5.2
	-5	10.5	10.1	9.0	8.1	7.1	6.2	5.3
	0	10.6	10.1	9.1	8.1	7.2	6.3	5.3
	5	10.6	10.2	9.1	8.1	7.2	6.3	5.3
	10	10.6	10.2	9.1	8.2	7.2	6.3	5.3
Sea Level	-40	10.1	9.7	8.7	7.7	6.8	5.9	5.0
	-30	10.1	9.7	8.7	7.7	6.8	6.0	5.0
	-20	10.2	9.8	8.8	7.8	6.9	6.0	5.0
	-10	10.3	9.9	8.8	7.8	6.9	6.1	5.1
	-5	10.3	9.9	8.9	7.9	7.0	6.1	5.1
	0	10.4	10.0	8.9	7.9	7.0	6.1	5.1
	5	10.4	10.0	8.9	8.0	7.0	6.2	5.2
	10	10.5	10.0	9.0	8.0	7.1	6.2	5.2
1000	-40	9.8	9.3	8.3	7.4	6.5	5.6	4.7
	-30	9.8	9.4	8.4	7.4	6.5	5.7	4.7
	-20	9.9	9.5	8.4	7.5	6.6	5.7	4.8
	-10	10.0	9.5	8.5	7.6	6.6	5.8	4.8
	-5	10.0	9.6	8.6	7.6	6.7	5.8	4.8
	0	10.1	9.6	8.6	7.6	6.7	5.8	4.9
	5	10.1	9.6	8.6	7.6	6.7	5.8	4.9
	10	9.6	9.2	8.2	7.2	6.3	5.5	4.5
2000	-40	9.4	9.0	8.0	7.1	6.2	5.3	4.4
	-30	9.5	9.1	8.1	7.1	6.2	5.4	4.4
	-20	9.6	9.2	8.1	7.2	6.3	5.4	4.5
	-10	9.7	9.2	8.2	7.3	6.4	5.5	4.6
	-5	9.7	9.3	8.2	7.3	6.4	5.5	4.6
	0	9.7	9.3	8.3	7.3	6.4	5.5	4.6
	5	9.7	9.3	8.3	7.3	6.4	5.6	4.6
	10	8.8	8.4	7.4	6.5	5.6	4.8	3.9

TCG2_TO_2_1_02

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October 30, 2016

HJ1-29000-003-001

Page 5-95

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps TO/APPR, Gear UP, Ice Protection On, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	9.1	8.7	7.7	6.8	5.8	5.0	4.1
	-30	9.2	8.7	7.7	6.8	5.9	5.1	4.2
	-20	9.2	8.8	7.8	6.9	6.0	5.2	4.2
	-10	9.3	8.9	7.9	6.9	6.0	5.2	4.3
	-5	9.3	8.9	7.9	7.0	6.1	5.2	4.3
	0	9.4	8.9	7.9	7.0	6.1	5.2	4.3
	5	9.2	8.7	7.7	6.8	5.9	5.1	4.1
	10	7.9	7.5	6.6	5.7	4.8	4.0	3.2
4000	-40	8.7	8.3	7.3	6.4	5.5	4.7	3.8
	-30	8.8	8.4	7.4	6.5	5.6	4.8	3.9
	-20	8.9	8.5	7.5	6.6	5.7	4.9	3.9
	-10	9.0	8.5	7.5	6.6	5.7	4.9	4.0
	-5	9.0	8.6	7.6	6.6	5.7	4.9	4.0
	0	9.0	8.6	7.6	6.7	5.7	4.9	4.0
	5	8.3	7.9	7.0	6.0	5.2	4.4	3.5
	10	7.0	6.6	5.7	4.8	4.0	3.3	2.5
5000	-40	8.4	8.0	7.1	6.1	5.3	4.5	3.6
	-30	8.5	8.1	7.1	6.2	5.3	4.5	3.6
	-20	8.5	8.1	7.2	6.2	5.4	4.6	3.7
	-10	8.6	8.2	7.2	6.3	5.4	4.6	3.7
	-5	8.6	8.2	7.2	6.3	5.4	4.6	3.7
	0	8.6	8.2	7.2	6.3	5.4	4.6	3.7
	5	7.5	7.1	6.1	5.2	4.4	3.6	2.8
	10	6.1	5.7	4.8	4.0	3.2	2.5	1.8
6000	-40	8.1	7.7	6.7	5.8	5.0	4.2	3.3
	-30	8.2	7.7	6.8	5.9	5.0	4.2	3.4
	-20	8.2	7.8	6.9	5.9	5.1	4.3	3.4
	-10	8.2	7.8	6.9	5.9	5.1	4.3	3.4
	-5	8.2	7.8	6.9	5.9	5.1	4.3	3.4
	0	7.8	7.4	6.4	5.5	4.7	3.9	3.1
	5	6.5	6.1	5.2	4.4	3.6	2.9	2.1
	10	5.1	4.8	3.9	3.2	2.5	1.8	1.1

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October 30, 2016

HJ1-29000-003-001

Page 5-96

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PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps TO/APPR, Gear UP, Ice Protection On, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	7.8	7.4	6.4	5.5	4.7	3.9	3.1
	-30	7.8	7.4	6.5	5.6	4.7	4.0	3.1
	-20	7.9	7.5	6.5	5.6	4.8	4.0	3.1
	-10	7.9	7.5	6.5	5.6	4.8	4.0	3.1
	-5	7.8	7.4	6.5	5.6	4.7	4.0	3.1
	0	6.9	6.5	5.6	4.7	3.9	3.2	2.4
	5	5.6	5.2	4.3	3.5	2.8	2.2	1.4
	10	4.2	3.9	3.1	2.4	1.8	1.2	0.5
8000	-40	7.4	7.1	6.1	5.2	4.4	3.6	2.8
	-30	7.5	7.1	6.1	5.3	4.4	3.7	2.9
	-20	7.5	7.1	6.1	5.3	4.4	3.7	2.9
	-10	7.5	7.1	6.1	5.2	4.4	3.6	2.8
	-5	7.3	6.8	5.9	5.0	4.2	3.5	2.6
	0	6.0	5.6	4.7	3.9	3.1	2.5	1.7
	5	4.7	4.3	3.5	2.8	2.1	1.5	0.8
	10	3.4	3.1	2.4	1.7	1.1	0.5	-0.1
9000	-40	7.1	6.7	5.8	4.9	4.1	3.4	2.5
	-30	7.2	6.7	5.8	4.9	4.1	3.4	2.6
	-20	7.1	6.7	5.8	4.9	4.1	3.4	2.5
	-10	7.1	6.6	5.7	4.8	4.0	3.3	2.5
	-5	6.3	5.9	5.0	4.2	3.5	2.8	2.0
	0	5.1	4.7	3.9	3.1	2.4	1.8	1.1
	5	3.8	3.5	2.7	2.0	1.4	0.8	0.2
	10	2.6	2.3	1.6	1.0	0.5	-0.1	
10,000	-40	6.7	6.3	5.4	4.6	3.8	3.1	2.3
	-30	6.7	6.3	5.4	4.6	3.8	3.1	2.3
	-20	6.7	6.2	5.3	4.5	3.7	3.0	2.2
	-10	6.6	6.2	5.3	4.5	3.7	3.0	2.2
	-5	5.5	5.1	4.2	3.5	2.8	2.1	1.4
	0	4.2	3.8	3.1	2.4	1.7	1.1	0.5
	5	2.9	2.7	2.0	1.3	0.7	0.2	-0.4
	10	1.8	1.6	0.9	0.4	-0.2		

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October 30, 2016

HJ1-29000-003-001

Page 5-97

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PERFORMANCE

Wind Corrected Takeoff Climb Gradient [%]				
FLAPS UP and TO/APPR, V2, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.1	0.0	0.0	0.0	0.0
0.5	0.5	0.5	0.5	0.4
1.0	1.0	1.0	1.0	1.0
1.4	1.5	1.5	1.6	1.6
1.9	2.0	2.0	2.1	2.2
2.3	2.5	2.6	2.6	2.7
2.7	3.0	3.1	3.2	3.3
3.2	3.5	3.6	3.7	3.9
3.6	4.0	4.1	4.3	4.4
4.1	4.5	4.7	4.8	5.0
4.5	5.0	5.2	5.4	5.6
5.0	5.5	5.7	5.9	6.2
5.4	6.0	6.2	6.5	6.7
5.8	6.5	6.8	7.0	7.3
6.3	7.0	7.3	7.6	7.9
6.7	7.5	7.8	8.1	8.5
7.2	8.0	8.3	8.7	9.0
7.6	8.5	8.8	9.2	9.6
8.1	9.0	9.4	9.8	10.2
8.5	9.5	9.9	10.3	10.8
8.9	10.0	10.4	10.9	11.3
9.4	10.5	10.9	11.4	11.9
9.8	11.0	11.5	11.9	12.5
10.3	11.5	12.0	12.5	13.1
10.7	12.0	12.5	13.0	13.6

CGWC_UP_2_03

NOTE Use this table when determining the ground reference flight path for obstacle clearance with Ice Protection On.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	9.8	9.4	8.5	7.6	6.9	6.1	5.3
	15	10.2	9.8	8.9	8.0	7.2	6.4	5.6
	25	10.3	9.9	8.9	8.0	7.2	6.5	5.7
	35	8.6	8.2	7.3	6.5	5.8	5.1	4.4
	40	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	45	6.5	6.2	5.4	4.7	4.1	3.5	2.9
	50	5.3	5.1	4.4	3.7	3.2	2.6	2.0
	55	4.2	4.0	3.3	2.8	2.3	1.8	1.2
Sea Level	-40	9.5	9.1	8.2	7.3	6.6	5.9	5.1
	15	9.9	9.5	8.5	7.7	6.9	6.2	5.4
	25	9.9	9.5	8.6	7.7	6.9	6.2	5.4
	35	7.8	7.4	6.6	5.8	5.2	4.5	3.8
	40	6.8	6.5	5.7	5.0	4.3	3.7	3.1
	45	5.8	5.5	4.8	4.1	3.5	3.0	2.3
	50	4.7	4.4	3.8	3.2	2.6	2.1	1.5
	55	3.6	3.4	2.8	2.2	1.7	1.3	0.7
1000	-40	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	15	9.5	9.1	8.2	7.4	6.6	5.9	5.1
	25	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	30	8.0	7.7	6.8	6.1	5.4	4.7	4.0
	35	7.0	6.7	5.9	5.2	4.5	3.9	3.2
	40	6.0	5.7	5.0	4.3	3.7	3.2	2.5
	45	5.1	4.8	4.1	3.5	2.9	2.4	1.8
	50	4.1	3.8	3.2	2.6	2.1	1.6	1.1
2000	-40	8.8	8.4	7.5	6.7	6.0	5.3	4.5
	10	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	20	9.2	8.8	7.9	7.1	6.3	5.6	4.8
	30	7.2	6.9	6.1	5.4	4.7	4.1	3.4
	35	6.2	5.9	5.2	4.5	3.9	3.3	2.7
	40	5.3	5.0	4.3	3.7	3.1	2.6	2.0
	45	4.4	4.1	3.5	2.9	2.4	1.9	1.3
	50	3.5	3.2	2.6	2.1	1.6	1.2	0.6

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October 30, 2016

HJ1-29000-003-001

Page 5-99

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PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	8.4	8.0	7.2	6.4	5.7	5.0	4.3
	10	8.8	8.4	7.5	6.7	6.0	5.3	4.5
	20	8.6	8.2	7.3	6.5	5.8	5.1	4.4
	25	7.4	7.1	6.3	5.6	4.9	4.3	3.6
	30	6.4	6.1	5.3	4.7	4.0	3.5	2.8
	35	5.5	5.2	4.5	3.9	3.3	2.7	2.1
	40	4.6	4.3	3.7	3.1	2.5	2.0	1.5
	45	3.7	3.5	2.9	2.4	1.9	1.4	0.9
4000	-40	8.1	7.7	6.9	6.1	5.4	4.8	4.0
	10	8.3	8.0	7.1	6.3	5.6	5.0	4.2
	20	7.7	7.3	6.5	5.8	5.1	4.5	3.8
	25	6.6	6.3	5.5	4.9	4.2	3.6	3.0
	30	5.6	5.3	4.6	4.0	3.4	2.9	2.3
	35	4.8	4.5	3.8	3.2	2.7	2.2	1.6
	40	3.9	3.7	3.1	2.5	2.0	1.5	1.0
	45	3.1	2.9	2.3	1.8	1.3	0.9	0.5
5000	-40	7.7	7.4	6.6	5.8	5.1	4.5	3.8
	5	7.9	7.6	6.7	6.0	5.3	4.6	3.9
	15	7.9	7.5	6.7	6.0	5.3	4.6	3.9
	25	5.9	5.5	4.8	4.2	3.6	3.0	2.4
	30	4.9	4.7	4.0	3.4	2.8	2.3	1.7
	35	4.1	3.8	3.2	2.7	2.1	1.7	1.1
	40	3.3	3.0	2.5	2.0	1.5	1.0	0.5
	45	2.5	2.3	1.8	1.3	0.9	0.6	0.3
6000	-40	7.4	7.1	6.3	5.5	4.9	4.2	3.6
	0	7.5	7.2	6.4	5.6	5.0	4.3	3.6
	10	7.5	7.2	6.3	5.6	4.9	4.3	3.6
	20	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	25	5.2	4.9	4.2	3.6	3.0	2.5	1.9
	30	4.3	4.0	3.4	2.8	2.3	1.8	1.3
	35	3.5	3.2	2.7	2.1	1.6	1.2	0.7
	40	2.7	2.5	1.9	1.5	1.0	0.6	0.3

TCG3_TO_03_06

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October 30, 2016

HJ1-29000-003-001

Page 5-100

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	7.1	6.8	6.0	5.3	4.6	4.0	3.3
	0	7.1	6.8	6.0	5.3	4.6	4.0	3.3
	10	7.1	6.8	6.0	5.3	4.6	4.0	3.3
	15	6.4	6.1	5.3	4.6	4.0	3.4	2.8
	20	5.4	5.1	4.4	3.8	3.2	2.7	2.1
	25	4.5	4.2	3.6	3.0	2.5	2.0	1.4
	30	3.7	3.4	2.8	2.3	1.8	1.3	0.8
	35	2.9	2.7	2.1	1.6	1.2	0.7	
8000	-40	6.8	6.5	5.7	5.0	4.3	3.8	3.1
	-10	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	0	6.8	6.4	5.7	5.0	4.3	3.7	3.1
	10	6.6	6.3	5.5	4.8	4.2	3.6	3.0
	15	5.7	5.4	4.7	4.0	3.4	2.9	2.3
	20	4.7	4.5	3.8	3.2	2.7	2.2	1.6
	25	3.9	3.6	3.0	2.5	2.0	1.5	1.0
	30	3.1	2.9	2.3	1.8	1.3	0.9	
9000	-40	6.4	6.1	5.4	4.7	4.1	3.5	2.8
	-10	6.4	6.1	5.3	4.7	4.0	3.5	2.8
	0	6.4	6.1	5.4	4.7	4.0	3.5	2.8
	10	5.9	5.6	4.9	4.2	3.6	3.1	2.4
	15	5.0	4.7	4.0	3.4	2.9	2.3	1.8
	20	4.1	3.8	3.2	2.7	2.1	1.7	1.1
	25	3.3	3.0	2.5	2.0	1.5	1.0	0.5
	30	2.5	2.3	1.8	1.3	0.9		
10,000	-40	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	-15	6.0	5.7	5.0	4.4	3.7	3.2	2.5
	-5	6.1	5.7	5.0	4.4	3.8	3.2	2.6
	5	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	10	5.2	4.9	4.2	3.6	3.1	2.5	1.9
	15	4.3	4.1	3.4	2.9	2.3	1.8	1.3
	20	3.5	3.2	2.7	2.1	1.6	1.2	0.7
	25	2.7	2.5	1.9	1.5	1.0	0.6	

TCG3_TO_0_7_06

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October 30, 2016

HJ1-29000-003-001

Page 5-101

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HA-420 AFM

PERFORMANCE

Wind Corrected Enroute Climb Gradient [%]				
FLAPS UP, 140 [KIAS], Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.1	0.0	0.0	0.0	0.1
0.5	0.5	0.5	0.5	0.7
1.0	1.0	1.0	1.0	1.2
1.4	1.5	1.5	1.6	1.8
1.9	2.0	2.1	2.1	2.3
2.3	2.5	2.6	2.6	2.9
2.8	3.0	3.1	3.2	3.5
3.2	3.5	3.6	3.7	4.0
3.7	4.0	4.1	4.3	4.6
4.1	4.5	4.6	4.8	5.2
4.6	5.0	5.2	5.3	5.7
5.0	5.5	5.7	5.9	6.3
5.5	6.0	6.2	6.4	6.8
5.9	6.5	6.7	7.0	7.4
6.4	7.0	7.2	7.5	8.0
6.8	7.5	7.8	8.1	8.5
7.3	8.0	8.3	8.6	9.1
7.7	8.5	8.8	9.1	9.7
8.1	9.0	9.3	9.7	10.2
8.6	9.5	9.8	10.2	10.8
9.0	10.0	10.4	10.8	11.4
9.5	10.5	10.9	11.3	11.9
9.9	11.0	11.4	11.8	12.5
10.4	11.5	11.9	12.4	13.0
10.8	12.0	12.4	12.9	13.6

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October 30, 2016

HJ1-29000-003-001

Page 5-102

FOR TRAINING PURPOSES ONLY

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	9.9	9.5	8.5	7.7	6.9	6.2	5.4
	-30	9.9	9.5	8.6	7.7	6.9	6.2	5.4
	-20	10.0	9.6	8.6	7.8	7.0	6.3	5.5
	-10	10.1	9.7	8.7	7.9	7.1	6.3	5.5
	-5	10.1	9.7	8.8	7.9	7.1	6.4	5.6
	0	10.2	9.8	8.8	7.9	7.1	6.4	5.6
	5	10.2	9.8	8.8	8.0	7.2	6.4	5.6
	10	10.3	9.8	8.9	8.0	7.2	6.5	5.6
Sea Level	-40	9.5	9.1	8.2	7.3	6.6	5.9	5.1
	-30	9.6	9.2	8.2	7.4	6.6	5.9	5.1
	-20	9.7	9.3	8.3	7.5	6.7	6.0	5.2
	-10	9.8	9.4	8.4	7.6	6.8	6.1	5.3
	-5	9.8	9.4	8.5	7.6	6.8	6.1	5.3
	0	9.8	9.4	8.5	7.6	6.9	6.1	5.3
	5	9.9	9.5	8.5	7.7	6.9	6.2	5.4
	10	9.9	9.5	8.6	7.7	6.9	6.2	5.4
1000	-40	9.2	8.8	7.9	7.0	6.3	5.6	4.8
	-30	9.2	8.8	7.9	7.1	6.4	5.7	4.9
	-20	9.3	8.9	8.0	7.2	6.4	5.7	5.0
	-10	9.4	9.0	8.1	7.3	6.5	5.8	5.0
	-5	9.4	9.1	8.1	7.3	6.5	5.8	5.0
	0	9.5	9.1	8.2	7.3	6.6	5.9	5.1
	5	9.5	9.1	8.2	7.4	6.6	5.9	5.1
	10	9.0	8.6	7.7	6.9	6.2	5.5	4.7
2000	-40	8.8	8.4	7.6	6.7	6.0	5.3	4.6
	-30	8.9	8.5	7.6	6.8	6.1	5.4	4.6
	-20	9.0	8.6	7.7	6.9	6.2	5.5	4.7
	-10	9.1	8.7	7.8	7.0	6.2	5.5	4.8
	-5	9.1	8.7	7.8	7.0	6.2	5.6	4.8
	0	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	5	9.2	8.8	7.9	7.0	6.3	5.6	4.8
	10	8.1	7.7	6.9	6.1	5.4	4.8	4.0

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October 30, 2016

HJ1-29000-003-001

Page 5-103

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	8.5	8.1	7.3	6.5	5.7	5.1	4.3
	-30	8.6	8.2	7.3	6.5	5.8	5.1	4.4
	-20	8.6	8.3	7.4	6.6	5.9	5.2	4.5
	-10	8.7	8.3	7.5	6.7	5.9	5.3	4.5
	-5	8.7	8.4	7.5	6.7	6.0	5.3	4.5
	0	8.8	8.4	7.5	6.7	6.0	5.3	4.5
	5	8.5	8.1	7.3	6.5	5.8	5.1	4.4
	10	7.2	6.8	6.0	5.3	4.7	4.0	3.4
4000	-40	8.2	7.8	7.0	6.2	5.5	4.8	4.1
	-30	8.2	7.9	7.0	6.2	5.5	4.9	4.2
	-20	8.3	7.9	7.1	6.3	5.6	4.9	4.2
	-10	8.4	8.0	7.1	6.4	5.6	5.0	4.2
	-5	8.4	8.0	7.1	6.4	5.7	5.0	4.3
	0	8.4	8.0	7.2	6.4	5.7	5.0	4.3
	5	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	10	6.3	5.9	5.2	4.5	3.9	3.3	2.7
5000	-40	7.8	7.4	6.6	5.9	5.2	4.5	3.8
	-30	7.9	7.5	6.7	5.9	5.2	4.6	3.9
	-20	7.9	7.6	6.8	6.0	5.3	4.7	3.9
	-10	8.0	7.6	6.8	6.0	5.3	4.7	4.0
	-5	8.0	7.6	6.8	6.0	5.3	4.7	4.0
	0	8.0	7.6	6.8	6.0	5.3	4.7	4.0
	5	6.7	6.3	5.6	4.9	4.2	3.7	3.0
	10	5.4	5.1	4.4	3.8	3.2	2.7	2.1
6000	-40	7.5	7.1	6.3	5.6	4.9	4.3	3.6
	-30	7.5	7.2	6.4	5.6	5.0	4.3	3.6
	-20	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	-10	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	-5	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	0	7.1	6.7	6.0	5.2	4.6	4.0	3.3
	5	5.8	5.5	4.8	4.1	3.5	3.0	2.4
	10	4.6	4.3	3.7	3.1	2.6	2.1	1.5

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October 30, 2016

HJ1-29000-003-001

Page 5-104

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	7.1	6.8	6.0	5.3	4.6	4.0	3.4
	-30	7.2	6.9	6.1	5.4	4.7	4.1	3.4
	-20	7.2	6.9	6.1	5.4	4.7	4.1	3.4
	-10	7.2	6.9	6.1	5.3	4.7	4.1	3.4
	-5	7.2	6.8	6.0	5.3	4.7	4.1	3.4
	0	6.2	5.9	5.2	4.5	3.9	3.3	2.7
	5	5.0	4.7	4.0	3.4	2.9	2.3	1.8
	10	3.9	3.6	3.0	2.5	2.0	1.5	0.9
8000	-40	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	-30	6.9	6.5	5.8	5.1	4.4	3.8	3.2
	-20	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	-10	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	-5	6.6	6.3	5.5	4.8	4.2	3.6	2.9
	0	5.4	5.1	4.4	3.8	3.2	2.7	2.1
	5	4.2	4.0	3.3	2.8	2.3	1.8	1.2
	10	3.2	2.9	2.4	1.8	1.4	0.9	0.4
9000	-40	6.5	6.2	5.4	4.7	4.1	3.5	2.9
	-30	6.5	6.2	5.4	4.8	4.1	3.5	2.9
	-20	6.5	6.1	5.4	4.7	4.1	3.5	2.9
	-10	6.4	6.1	5.4	4.7	4.1	3.5	2.8
	-5	5.8	5.5	4.8	4.1	3.5	3.0	2.3
	0	4.6	4.3	3.7	3.1	2.6	2.1	1.5
	5	3.5	3.3	2.7	2.2	1.7	1.2	0.7
	10	2.5	2.3	1.7	1.3	0.8	0.4	
10,000	-40	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	-30	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	-20	6.1	5.8	5.0	4.4	3.8	3.2	2.6
	-10	6.1	5.8	5.0	4.4	3.8	3.2	2.6
	-5	5.0	4.7	4.0	3.4	2.9	2.4	1.8
	0	3.9	3.6	3.0	2.5	1.9	1.5	0.9
	5	2.8	2.6	2.0	1.5	1.1	0.6	0.2
	10	1.8	1.6	1.2	0.7	0.3		

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October 30, 2016

HJ1-29000-003-001

Page 5-105

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PERFORMANCE

Wind Corrected Enroute Climb Gradient [%]				
FLAPS UP, 140 [KIAS], Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.1	0.0	0.0	0.0	0.0
0.5	0.5	0.5	0.5	0.5
1.0	1.0	1.0	1.0	1.0
1.4	1.5	1.5	1.6	1.6
1.9	2.0	2.1	2.1	2.2
2.3	2.5	2.6	2.6	2.7
2.8	3.0	3.1	3.2	3.3
3.2	3.5	3.6	3.7	3.9
3.6	4.0	4.1	4.3	4.4
4.1	4.5	4.7	4.8	5.0
4.5	5.0	5.2	5.4	5.6
5.0	5.5	5.7	5.9	6.1
5.4	6.0	6.2	6.5	6.7
5.9	6.5	6.7	7.0	7.3
6.3	7.0	7.3	7.5	7.8
6.8	7.5	7.8	8.1	8.4
7.2	8.0	8.3	8.6	9.0
7.7	8.5	8.8	9.2	9.5
8.1	9.0	9.3	9.7	10.1
8.6	9.5	9.9	10.3	10.7
9.0	10.0	10.4	10.8	11.2
9.4	10.5	10.9	11.3	11.8
9.9	11.0	11.4	11.9	12.4
10.3	11.5	11.9	12.4	12.9
10.8	12.0	12.5	13.0	13.5

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October 30, 2016

HJ1-29000-003-001

Page 5-106

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HA-420 AFM

PERFORMANCE

TAKEOFF – FLAPS UP

Max Takeoff Weight – Climb and Brake Energy Limited

Red shading on the takeoff tables indicates conditions where the airplane does not meet the climb requirements or exceeds the brake energy limit, but can be used for interpolation.

Example:

Ambient Conditions:

Temperature	-10 °C
Airport Altitude	7500 ft
Wind	15 kts Headwind
Runway Gradient	+1.0 %

Aircraft Configuration:

Flaps	UP
Weight	10600 lbs

Bleed Setting:

Ice Protection	On
----------------	----

Using the Weight Limit Charts:

BE Weight Limit 15kt Headwind	Not Limited
BE Weight Limit +1% Runway Gradient	Not Limited

Using the Tables:

Uncorrected

- V_1 126 KIAS
- V_R 126 KIAS
- V_2 131.5 KIAS
- Field Length 7642 ft
- Climb Gradient 4.5 %

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October 30, 2016

HJ1-29000-003-001

Page 5-107

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HA-420 AFM

PERFORMANCE

Using the Tables: (continued)

Wind Correction:

- V_1 126 KIAS
- Field Length 7216 ft
- Climb Gradient 4.8 %

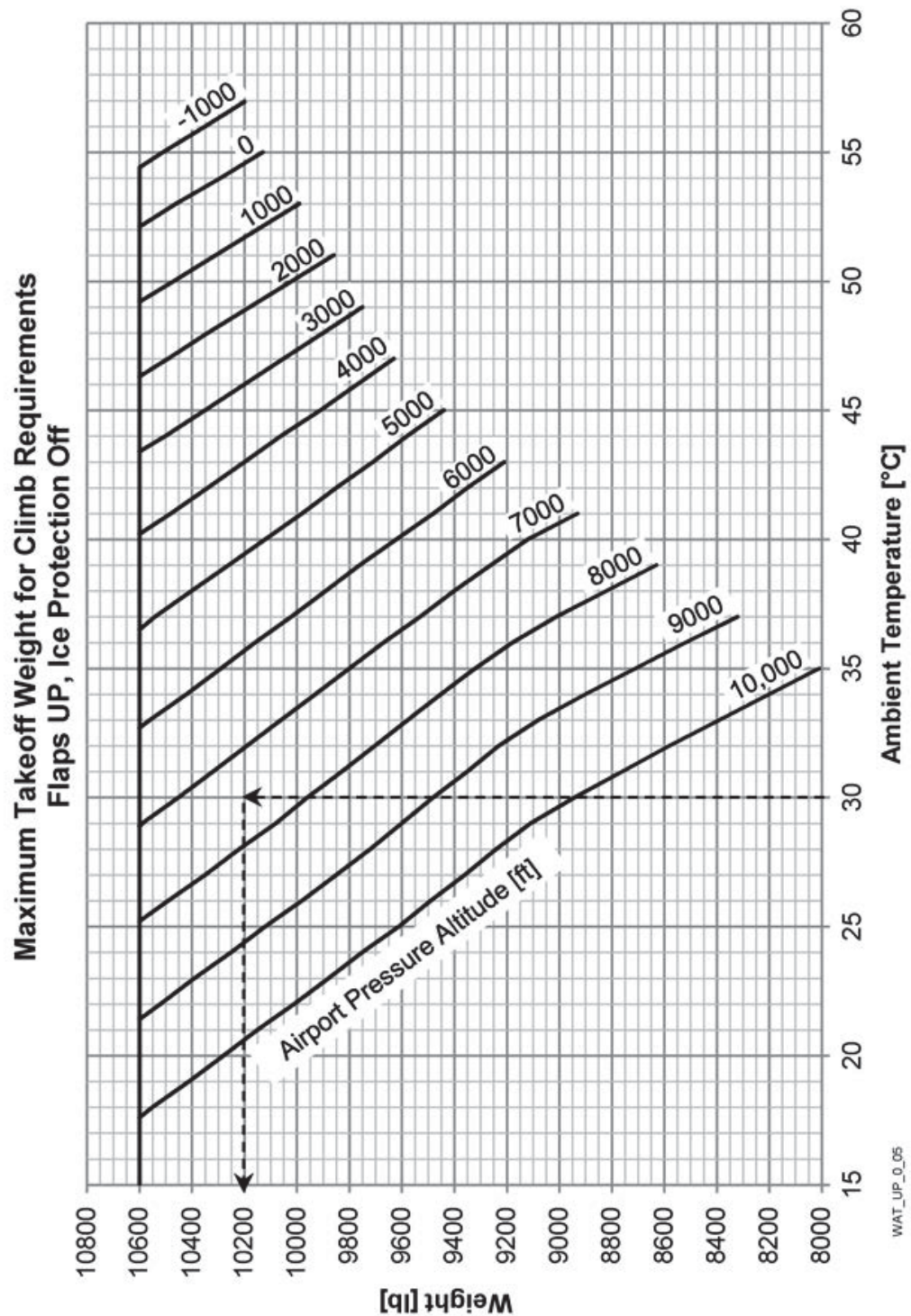
Slope Correction:

- V_1 126 KIAS
- V_2 130.5 KIAS
- Field Length 8249 ft

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HA-420 AFM

PERFORMANCE



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October 30, 2016

HJ1-29000-003-001

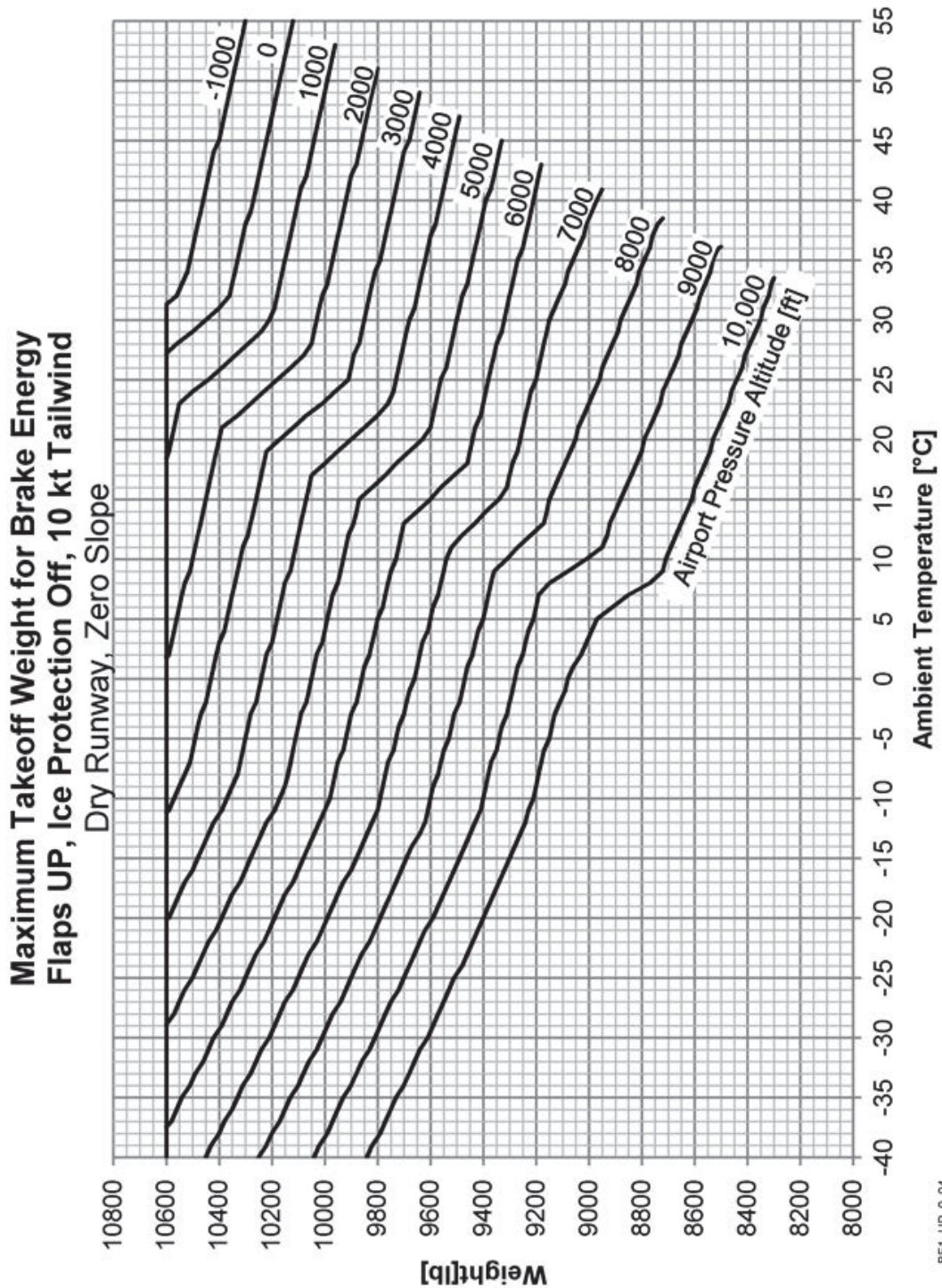
Page 5-109

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HA-420 AFM

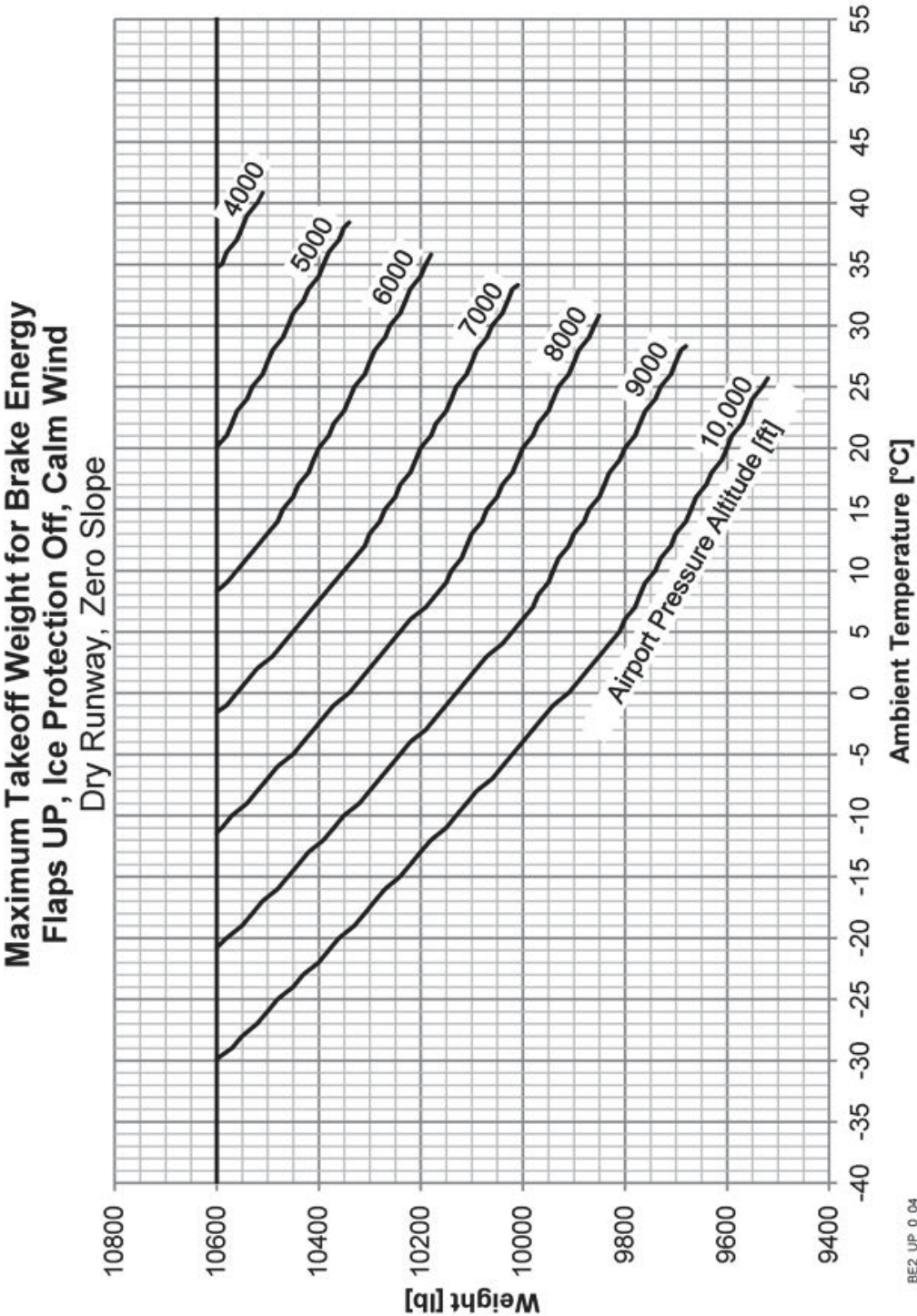
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HA-420 AFM

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October 30, 2016

HJ1-29000-003-001

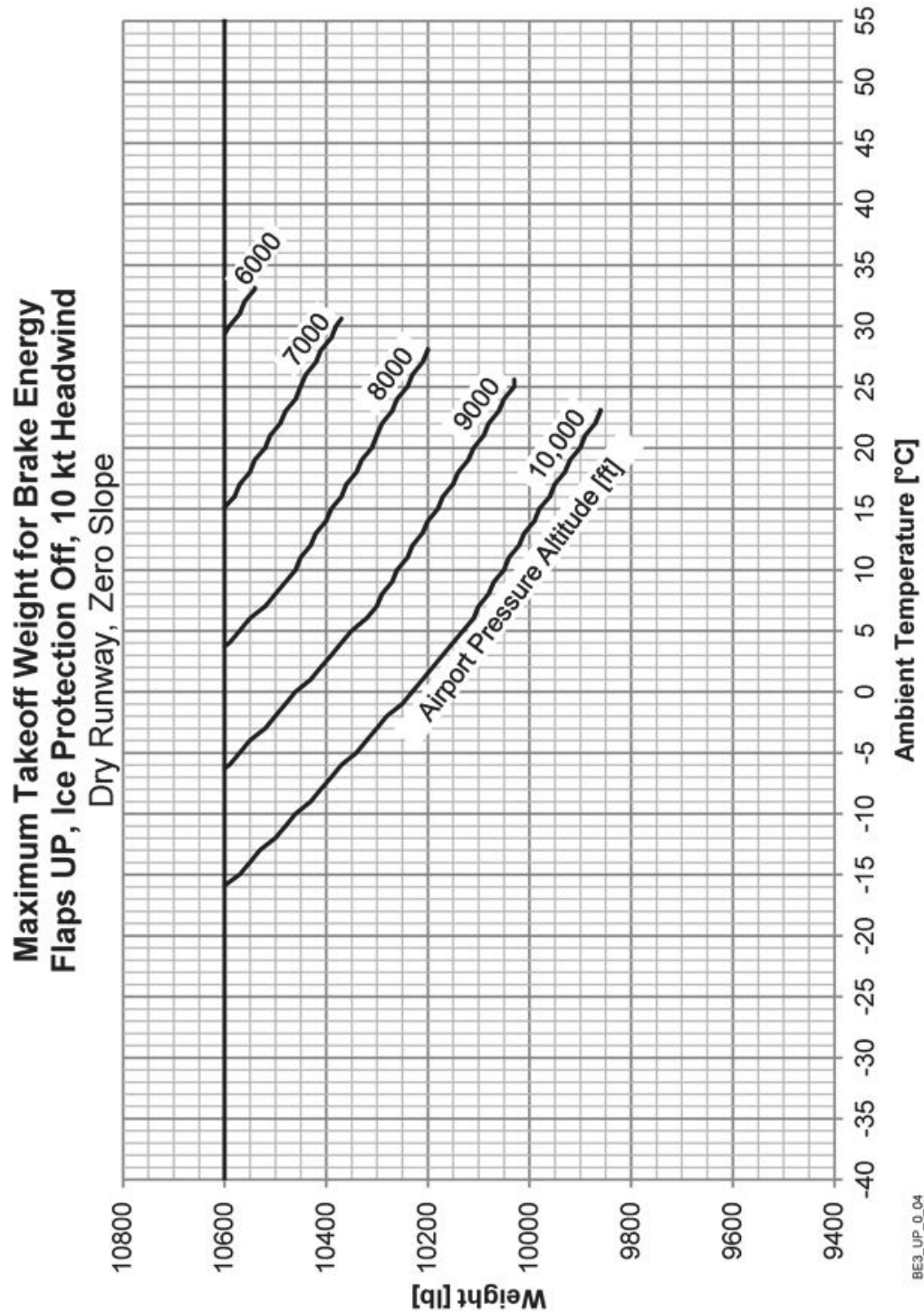
Page 5-111

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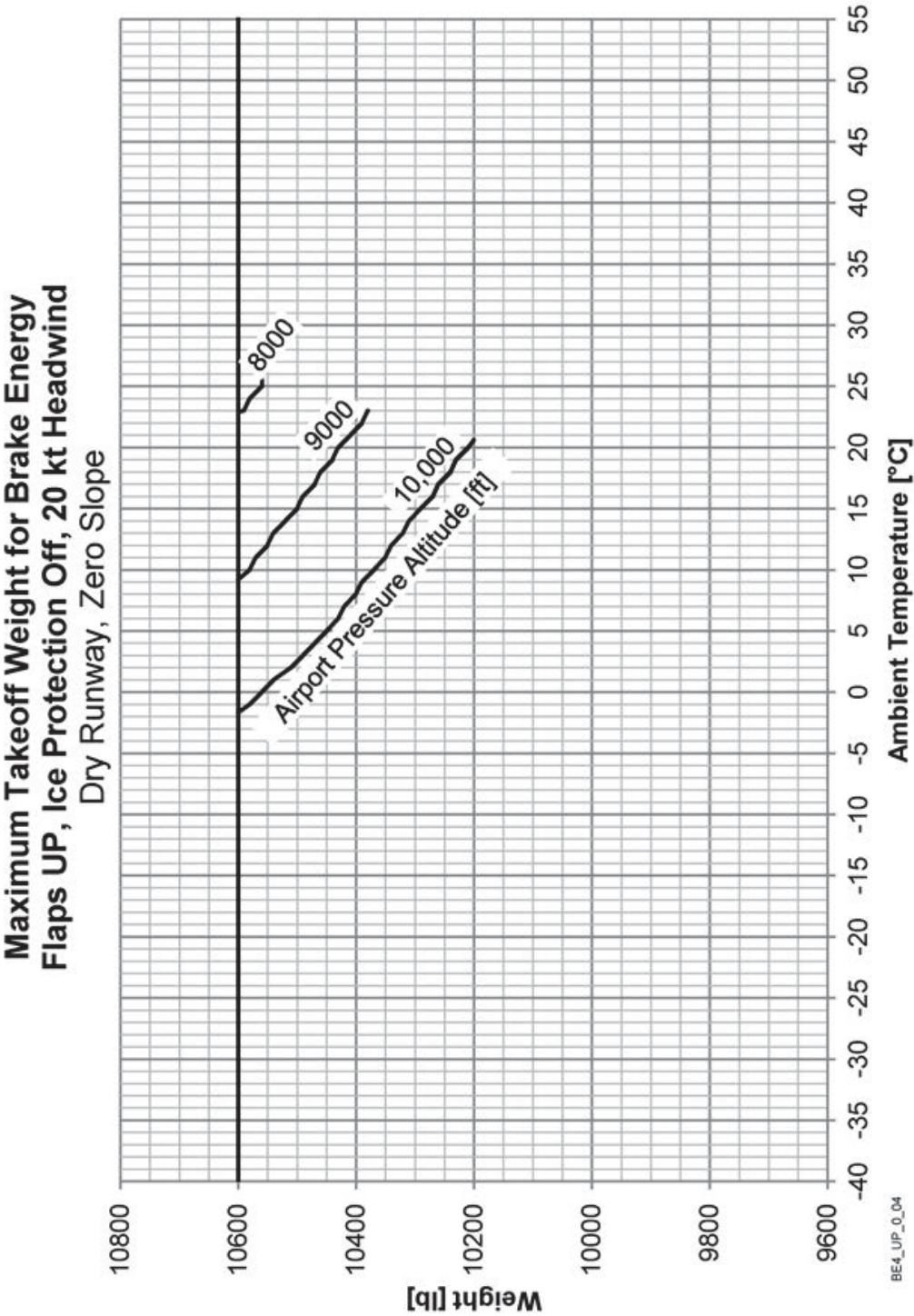
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HA-420 AFM

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October 30, 2016

HJ1-29000-003-001

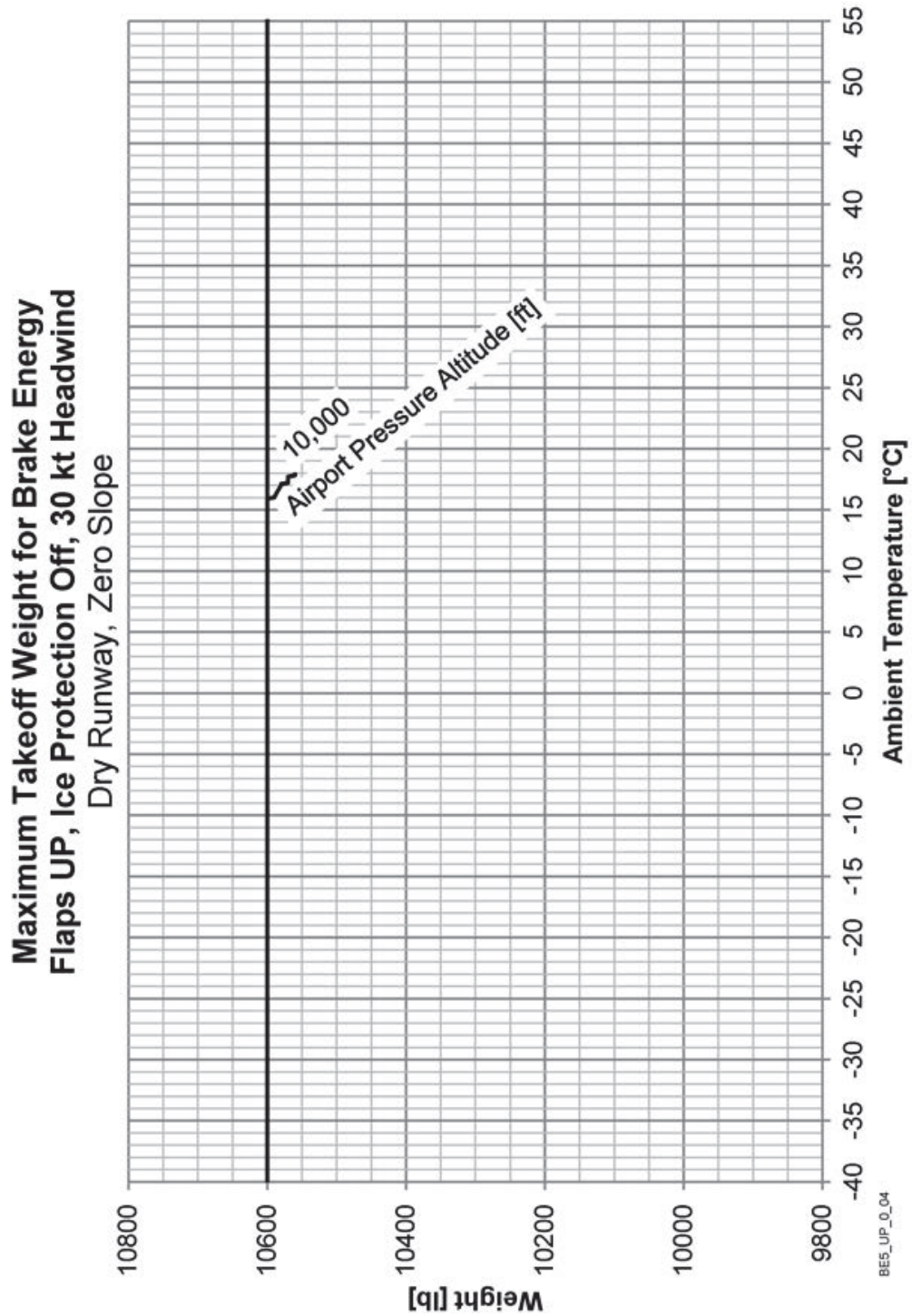
Page 5-113

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HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-114

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HA-420 AFM

PERFORMANCE

Slope Corrected Maximum Takeoff Weight for Brake Energy [lb]				
Flaps UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
7440	7640	7800	7940	8083
7539	7739	7900	8042	8187
7637	7837	8000	8144	8290
7736	7936	8100	8246	8393
7834	8034	8200	8349	8497
7933	8133	8300	8451	8600
8031	8231	8400	8553	8703
8130	8330	8500	8655	8807
8229	8429	8600	8757	8910
8327	8527	8700	8859	9013
8426	8626	8800	8961	9117
8524	8724	8900	9064	9220
8623	8823	9000	9166	9323
8721	8921	9100	9268	9427
8820	9020	9200	9370	9530
8919	9119	9300	9472	9633
9017	9217	9400	9574	9737
9116	9316	9500	9676	9840
9214	9414	9600	9779	9943
9313	9513	9700	9881	10047
9411	9611	9800	9983	10150
9510	9710	9900	10085	10253
9609	9809	10000	10187	10357
9707	9907	10100	10289	10460
9806	10006	10200	10391	10563
9904	10104	10300	10494	10600
10003	10203	10400	10596	10600
10101	10301	10500	10600	10600
10200	10400	10600	10600	10600

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October 30, 2016

HJ1-29000-003-001

Page 5-115

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude -1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	111	111	113	114	117	121	125
	VR	119	119	119	119	121	123	126
	V2	131	131	130	130	130	132	134
	TOFL	2736	2790	2923	3050	3269	3530	3898
15	V1	109	110	112	113	116	119	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	129	130	131	134
	TOFL	3210	3277	3439	3594	3858	4175	4565
25	V1	109	110	111	113	116	119	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	130	131	133
	TOFL	3291	3360	3527	3687	3959	4285	4688
35	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3706	3776	3951	4120	4410	4901	5986
40	V1	114	115	116	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	126	128	129	132
	TOFL	3982	4055	4232	4404	5006	5946	7055
45	V1	116	116	118	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4281	4355	4624	5234	6102	7047	8169
50	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	4775	5053	5727	6371	7236	8194	9338
55	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	125	127	130
	TOFL	5985	6253	6900	7523	8391	9367	10541

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October 30, 2016

HJ1-29000-003-001

Page 5-116

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude Sea Level								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	111	111	113	114	117	121	126
	VR	119	119	119	119	121	123	126
	V2	131	131	130	129	130	132	134
	TOFL	2825	2881	3018	3150	3376	3647	4042
15	V1	109	110	112	113	116	119	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3316	3385	3552	3711	3985	4312	4717
25	V1	109	110	111	113	116	119	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3402	3473	3645	3810	4092	4430	4847
35	V1	113	113	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3918	3992	4173	4349	4679	5477	6603
40	V1	115	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4220	4295	4479	4779	5639	6599	7734
45	V1	116	117	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	126	125	126	128	131
	TOFL	4534	4649	5193	5864	6747	7717	8871
50	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	125	125	124	126	128	130
	TOFL	5444	5722	6396	7042	7925	8908	10088
55	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6683	6950	7600	8226	9116	10123	11340

TOFL_UP_0_0_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-117

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	111	112	113	115	118	121	126
	VR	119	119	119	119	121	123	126
	V2	131	130	130	129	130	132	134
	TOFL	2966	3024	3165	3303	3538	3820	4408
15	V1	110	110	112	113	116	120	124
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3480	3551	3723	3889	4175	4516	5018
25	V1	110	111	112	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	130	133
	TOFL	3653	3726	3903	4078	4373	4729	5359
30	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3915	3988	4174	4353	4659	5219	6359
35	V1	114	114	116	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4201	4278	4466	4648	5311	6285	7435
40	V1	116	116	118	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	129	131
	TOFL	4511	4589	4893	5528	6427	7409	8576
45	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	5025	5314	6013	6681	7578	8574	9763
50	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6271	6548	7220	7866	8769	9785	11008

TOFL_UP_0_1_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-118

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 2000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	112	112	114	115	118	121	126
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3109	3168	3314	3459	3703	3996	4775
10	V1	110	111	112	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	131	133
	TOFL	3601	3673	3847	4019	4311	4661	5273
20	V1	110	111	112	114	116	120	125
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	130	133
	TOFL	3699	3773	3953	4131	4432	4793	5400
30	V1	113	114	115	116	120	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4203	4281	4473	4659	5118	6060	7225
35	V1	115	115	116	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4502	4582	4778	5211	6129	7125	8306
40	V1	117	118	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	4847	5015	5650	6341	7255	8262	9463
45	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	125	127	130
	TOFL	5857	6144	6841	7508	8425	9450	10681
50	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	124	124	124	125	127	129
	TOFL	7120	7397	8069	8718	9645	10697	11968

TOFL_UP_0_2_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-119

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 3000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	112	113	114	115	118	122	126
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3261	3322	3475	3625	3879	4225	5166
10	V1	111	111	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	127	128	130	133
	TOFL	3780	3854	4034	4215	4518	4884	5700
20	V1	111	112	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3927	4003	4191	4377	4691	5078	5999
25	V1	113	113	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	126	127	129	132
	TOFL	4212	4292	4488	4677	5058	5885	7066
30	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4517	4598	4798	5104	5951	6959	8153
35	V1	116	116	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	131
	TOFL	4830	4913	5370	6083	7014	8034	9248
40	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	125	125	124	126	128	130
	TOFL	5469	5767	6488	7178	8111	9149	10392
45	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6713	7000	7695	8364	9305	10365	11643

TOFL_UP_0_3_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-120

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 4000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	113	113	115	116	118	123	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	130	133
	TOFL	3422	3486	3646	3801	4065	4555	5585
10	V1	111	112	113	114	117	121	126
	VR	119	119	119	119	121	123	126
	V2	128	128	128	127	128	130	132
	TOFL	3974	4051	4240	4427	4743	5190	6187
20	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	126	128	129	132
	TOFL	4224	4305	4504	4697	5028	5719	6916
25	V1	114	114	115	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4533	4616	4819	5049	5784	6806	8016
30	V1	115	116	118	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	126	125	126	128	131
	TOFL	4855	4939	5281	5932	6877	7909	9137
35	V1	118	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	125	125	124	126	128	130
	TOFL	5326	5557	6299	7008	7954	9001	10254
40	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6404	6701	7420	8110	9063	10131	11417
45	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	123	125	127	129
	TOFL	7599	7886	8584	9258	10225	11324	12656

TOFL_UP_0_4_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-121

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 5000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	113	114	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3591	3658	3824	3985	4261	4963	6016
5	V1	112	112	114	115	117	122	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	4120	4198	4394	4584	4908	5472	6617
15	V1	112	112	114	115	117	122	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	4249	4330	4532	4729	5064	5649	6806
25	V1	115	115	117	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	126	125	126	128	131
	TOFL	4878	4965	5230	5764	6725	7771	9015
30	V1	117	118	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	125	125	125	126	128	130
	TOFL	5290	5466	6158	6883	7845	8908	10178
35	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6208	6513	7250	7957	8925	10006	11306
40	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	123	125	127	129
	TOFL	7389	7684	8402	9094	10073	11182	12524
45	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	123	123	124	126	129
	TOFL	8595	8883	9585	10266	11264	12409	13803

TOFL_UP_0_5_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-122

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 6000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	113	114	115	116	120	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3768	3839	4011	4177	4530	5371	6449
0	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4270	4352	4552	4746	5091	5859	7056
10	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4410	4495	4703	4903	5269	6047	7273
20	V1	114	115	116	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	126	128	131
	TOFL	4899	4987	5203	5610	6564	7624	8884
25	V1	116	117	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	5242	5405	5965	6709	7686	8761	10044
30	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6051	6364	7119	7842	8825	9919	11234
35	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	124	125	127	129
	TOFL	7220	7522	8258	8965	9958	11079	12433
40	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	123	123	124	126	129
	TOFL	8419	8714	9433	10129	11140	12296	13702

TOFL_UP_0_8_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-123

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 7000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3954	4027	4206	4379	4853	5789	6893
0	V1	113	113	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	127	126	127	129	131
	TOFL	4502	4587	4795	4997	5471	6392	7619
10	V1	113	113	114	115	119	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4653	4741	4956	5165	5669	6610	7866
15	V1	114	114	115	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	128	131
	TOFL	4924	5014	5233	5570	6422	7498	8774
20	V1	115	116	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	5266	5358	5800	6538	7532	8621	9920
25	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	125	127	130
	TOFL	5871	6179	6954	7695	8693	9800	11129
30	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	124	124	124	125	127	129
	TOFL	7058	7369	8122	8845	9853	10987	12355
35	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	123	123	124	127	129
	TOFL	8239	8542	9278	9990	11014	12182	13599

TOFL_UP_0_7_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-124

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 8000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	114	115	116	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4150	4225	4410	4590	5269	6229	7362
0	V1	113	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4750	4839	5054	5264	5903	6968	8228
10	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	129	131
	TOFL	4948	5040	5263	5527	6270	7360	8651
15	V1	115	115	118	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	5300	5394	5771	6412	7421	8524	9839
20	V1	118	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	125	128	130
	TOFL	5810	6002	6778	7537	8550	9671	11016
25	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	124	124	124	125	127	130
	TOFL	6877	7196	7968	8708	9731	10880	12264
30	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	123	125	127	129
	TOFL	8092	8402	9155	9883	10921	12102	13535
35	V1	118	118	118	118	120	122	125
	VR	118	118	118	118	120	122	125
	V2	122	122	122	122	123	125	128
	TOFL	9308	9612	10353	11073	12133	13354	14846

TOFL_UP_0_8_07

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-125

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 9000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4354	4432	4624	4891	5693	6675	7838
0	V1	114	114	115	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	128	131
	TOFL	5003	5095	5318	5636	6449	7541	8836
10	V1	115	115	117	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	5333	5428	5738	6273	7297	8413	9742
15	V1	117	118	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	126	128	130
	TOFL	5788	5978	6658	7434	8462	9597	10955
20	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6698	7025	7817	8575	9613	10773	12169
25	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	123	125	127	129
	TOFL	7935	8253	9025	9768	10821	12014	13458
30	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	123	123	123	124	126	129
	TOFL	9163	9474	10233	10968	12042	13275	14779
35	V1	118	118	118	118	120	122	125
	VR	118	118	118	118	120	122	125
	V2	122	122	122	121	123	125	127
	TOFL	10453	10759	11508	12240	13344	14630	16208

TOFL_UP_0_9_07

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-126

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection Off, Altitude 10,000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	115	115	116	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4577	4659	4859	5238	6159	7164	8355
-5	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	126	125	126	128	131
	TOFL	5181	5274	5503	5931	6890	7995	9308
5	V1	114	115	115	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	131
	TOFL	5361	5457	5695	6158	7151	8282	9626
10	V1	116	117	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	126	128	130
	TOFL	5780	5977	6568	7361	8404	9553	10928
15	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6561	6896	7706	8481	9534	10709	12121
20	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	123	125	127	129
	TOFL	7774	8100	8890	9651	10718	11926	13386
25	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	123	123	123	124	126	129
	TOFL	9056	9374	10149	10900	11988	13234	14752
30	V1	118	118	118	118	120	122	125
	VR	118	118	118	118	120	122	125
	V2	122	122	122	121	123	125	127
	TOFL	10315	10628	11395	12141	13258	14552	16140

TOFL_UP_0_10_07

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October 30, 2016

HJ1-29000-003-001

Page 5-127

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HA-420 AFM

PERFORMANCE

Wind Corrected V1 [KIAS]		
FLAPS UP		
Ice Protection Off		
Tailwind		Headwind
10	◀ REF [0] ▶	30
108	109	111
109	110	112
110	111	113
111	112	114
112	113	114
113	114	115
114	115	116
115	116	117
116	117	118
117	118	119
118	119	120
119	120	121
120	121	122
121	122	123
122	123	124
123	124	125
124	125	126
125	126	126

V1WC_UP_0_04

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October 30, 2016

HJ1-29000-003-001

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Page 5-128

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected V1 [KIAS]				
FLAPS UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
105	106	108	110	113
106	107	109	111	114
107	108	110	112	114
108	109	111	113	115
109	110	112	114	116
110	111	113	115	116
110	112	114	116	117
111	113	115	117	118
112	114	116	117	119
113	115	117	118	119
114	117	118	119	120
115	118	119	120	121
116	119	120	121	121
117	120	121	122	122
118	121	122	123	123
119	122	123	124	123
120	123	124	124	124
121	124	125	125	125
122	125	126	126	126

V1SC_UP_0_04

Takeoff Rotation Speed (V_R) Slope Correction, Ice Protection Off

NOTE V_R with flaps UP and Ice Protection Off does not require slope correction.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected V2 [KIAS]				
FLAPS UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
124	124	123	122	122
125	125	124	123	123
126	126	125	124	124
127	127	126	125	125
128	128	127	126	126
129	129	128	127	127
130	130	129	128	128
131	131	130	129	129
132	132	131	130	130
133	133	132	131	131
134	134	133	132	132
135	135	134	133	133

V2SC_UP_0_05

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October 30, 2016

HJ1-29000-003-001

Page 5-130

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HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
3244	2700	2538	2396	2274
3352	2800	2635	2491	2366
3461	2900	2733	2585	2458
3570	3000	2830	2680	2550
3679	3100	2927	2775	2642
3788	3200	3025	2869	2734
3896	3300	3122	2964	2826
4005	3400	3220	3058	2918
4114	3500	3317	3153	3010
4223	3600	3414	3248	3102
4331	3700	3512	3342	3194
4440	3800	3609	3437	3286
4549	3900	3706	3532	3378
4658	4000	3804	3626	3470
4766	4100	3901	3721	3562
4875	4200	3999	3815	3654
4984	4300	4096	3910	3746
5093	4400	4193	4005	3838
5202	4500	4291	4099	3930
5310	4600	4388	4194	4022
5419	4700	4486	4288	4114
5528	4800	4583	4383	4206
5637	4900	4680	4478	4298
5745	5000	4778	4572	4390
5854	5100	4875	4667	4482
5963	5200	4972	4762	4574
6072	5300	5070	4856	4666
6180	5400	5167	4951	4758
6289	5500	5265	5045	4850
6398	5600	5362	5140	4942
6507	5700	5459	5235	5034
6616	5800	5557	5329	5126
6724	5900	5654	5424	5218
6833	6000	5752	5518	5310
6942	6100	5849	5613	5402
7051	6200	5946	5708	5494

TOWC1_UP_0_02

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October 30, 2016

HJ1-29000-003-001

Page 5-131

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
7051	6200	5946	5708	5494
7159	6300	6044	5802	5586
7268	6400	6141	5897	5678
7377	6500	6238	5992	5770
7486	6600	6336	6086	5862
7594	6700	6433	6181	5954
7703	6800	6531	6275	6046
7812	6900	6628	6370	6138
7921	7000	6725	6465	6230
8030	7100	6823	6559	6322
8138	7200	6920	6654	6414
8247	7300	7018	6748	6506
8356	7400	7115	6843	6598
8465	7500	7212	6938	6690
8573	7600	7310	7032	6782
8682	7700	7407	7127	6874
8791	7800	7504	7222	6966
8900	7900	7602	7316	7058
9008	8000	7699	7411	7150
9117	8100	7797	7505	7242
9226	8200	7894	7600	7334
9335	8300	7991	7695	7426
9444	8400	8089	7789	7518
9552	8500	8186	7884	7610
9661	8600	8284	7978	7702
9770	8700	8381	8073	7794
9879	8800	8478	8168	7886
9987	8900	8576	8262	7978
10096	9000	8673	8357	8070
10205	9100	8770	8452	8162
10314	9200	8868	8546	8254
10422	9300	8965	8641	8346
10531	9400	9063	8735	8438
10640	9500	9160	8830	8530
10749	9600	9257	8925	8622
10858	9700	9355	9019	8714

TOWC2_UP_0_02

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October 30, 2016

HJ1-29000-003-001

Page 5-132

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
10858	9700	9355	9019	8714
10966	9800	9452	9114	8806
11075	9900	9550	9208	8898
11184	10000	9647	9303	8990
11293	10100	9744	9398	9082
11401	10200	9842	9492	9174
11510	10300	9939	9587	9266
11619	10400	10036	9682	9358
11728	10500	10134	9776	9450
11836	10600	10231	9871	9542
11945	10700	10329	9965	9634
12054	10800	10426	10060	9726
12163	10900	10523	10155	9818
12272	11000	10621	10249	9910
12380	11100	10718	10344	10002
12489	11200	10816	10438	10094
12598	11300	10913	10533	10186
12707	11400	11010	10628	10278
12815	11500	11108	10722	10370
12924	11600	11205	10817	10462
13033	11700	11302	10912	10554
13142	11800	11400	11006	10646
13250	11900	11497	11101	10738
13359	12000	11595	11195	10830
13468	12100	11692	11290	10922
13577	12200	11789	11385	11014
13686	12300	11887	11479	11106
13794	12400	11984	11574	11198
13903	12500	12082	11668	11290
14012	12600	12179	11763	11382
14121	12700	12276	11858	11474
14229	12800	12374	11952	11566
14338	12900	12471	12047	11658
14447	13000	12568	12142	11750
14556	13100	12666	12236	11842
14664	13200	12763	12331	11934

TOWCS_UP_0_02

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October 30, 2016

HJ1-29000-003-001

Page 5-133

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
14664	13200	12763	12331	11934
14773	13300	12861	12425	12026
14882	13400	12958	12520	12118
14991	13500	13055	12615	12210
15100	13600	13153	12709	12302
15208	13700	13250	12804	12394
15317	13800	13348	12898	12486
15426	13900	13445	12993	12578

TOWC4_UP_0_02

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October 30, 2016

HJ1-29000-003-001

Page 5-134

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
2200	2200	2200	2237	2273
2300	2300	2300	2338	2377
2400	2400	2400	2440	2480
2500	2500	2500	2542	2583
2600	2600	2600	2643	2687
2700	2700	2700	2745	2790
2800	2800	2800	2847	2893
2900	2900	2900	2948	2997
3000	3000	3000	3050	3100
3109	3109	3100	3197	3303
3219	3219	3200	3344	3506
3328	3328	3300	3492	3708
3422	3426	3400	3639	3911
3505	3516	3500	3786	4114
3588	3606	3600	3933	4317
3671	3697	3700	4080	4520
3755	3787	3800	4227	4723
3838	3877	3900	4375	4925
3921	3967	4000	4522	5128
4004	4057	4100	4669	5331
4088	4148	4200	4816	5534
4171	4238	4300	4963	5737
4254	4328	4400	5111	5939
4337	4418	4500	5254	6134
4421	4509	4600	5369	6256
4504	4599	4700	5484	6378
4587	4689	4800	5599	6500
4670	4779	4900	5714	6622
4754	4869	5000	5828	6744
4837	4960	5100	5943	6866
4920	5050	5200	6058	6988
5003	5140	5300	6173	7110
5087	5230	5400	6287	7232
5170	5320	5500	6402	7354
5253	5411	5600	6517	7476
5336	5501	5700	6632	7598

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October 30, 2016

HJ1-29000-003-001

Page 5-135

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
5336	5501	5700	6632	7598
5417	5588	5800	6747	7720
5476	5648	5900	6861	7842
5535	5707	6000	6966	7964
5594	5766	6100	7069	8086
5653	5825	6200	7171	8208
5712	5885	6300	7274	8330
5771	5944	6400	7377	8443
5830	6003	6500	7479	8555
5889	6062	6600	7582	8668
5948	6122	6700	7684	8780
6006	6181	6800	7787	8893
6065	6240	6900	7890	9005
6124	6299	7000	7992	9118
6183	6359	7100	8095	9230
6242	6418	7200	8197	9343
6301	6477	7300	8300	9455
6360	6537	7400	8403	9567
6419	6596	7500	8507	9672
6478	6655	7600	8611	9778
6537	6714	7700	8715	9883
6596	6774	7800	8820	9989
6655	6833	7900	8924	10095
6713	6935	8000	9028	10200
6772	7041	8100	9132	10306
6831	7146	8200	9237	10411
6890	7252	8300	9341	10517
6949	7358	8400	9445	10622
7008	7463	8500	9550	10728
7067	7569	8600	9654	10833
7126	7674	8700	9758	10939
7185	7780	8800	9863	11045
7244	7882	8900	9967	11150
7303	7976	9000	10071	11256
7361	8071	9100	10175	11361
7420	8165	9200	10280	11467

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October 30, 2016

HJ1-29000-003-001

Page 5-136

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HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
7420	8165	9200	10280	11467
7479	8259	9300	10384	11572
7538	8353	9400	10488	11678
7597	8447	9500	10593	11783
7656	8541	9600	10697	11889
7715	8635	9700	10801	11995
7774	8729	9800	10905	12100
7833	8823	9900	11010	12206
7892	8918	10000	11114	12311
7951	9012	10100	11218	12417
8010	9106	10200	11323	12522
8068	9200	10300	11427	12628
8127	9294	10400	11531	12733
8186	9388	10500	11636	12839
8245	9482	10600	11740	12944
8304	9576	10700	11844	13050
8363	9671	10800	11948	13156
8422	9765	10900	12053	13261
8481	9859	11000	12157	13367
8540	9953	11100	12261	13472
8599	10047	11200	12366	13578
8658	10141	11300	12470	13683
8717	10235	11400	12574	13789
8775	10329	11500	12679	13894
8834	10424	11600	12783	14000
8893	10518	11700	12887	14106
8952	10612	11800	12991	14211
9011	10706	11900	13096	14317
9070	10800	12000	13200	14422
9129	10894	12100	13304	14528
9188	10988	12200	13409	14633
9247	11082	12300	13513	14739
9306	11176	12400	13617	14844
9365	11271	12500	13721	14950
9423	11365	12600	13826	15056
9482	11459	12700	13930	15161

TOSC3_UP_0_02

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October 30, 2016

HJ1-29000-003-001

Page 5-137

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HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection Off				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
9482	11459	12700	13930	15161
9541	11553	12800	14034	15267
9600	11647	12900	14139	15372
9659	11741	13000	14243	15478
9718	11835	13100	14347	15583
9777	11929	13200	14452	15689
9836	12024	13300	14556	15794
9895	12118	13400	14660	15900
9954	12212	13500	14764	16005
10013	12306	13600	14869	16111
10072	12400	13700	14973	16217
10130	12494	13800	15077	16322
10189	12588	13900	15182	16428
10248	12682	14000	15286	16533
10307	12777	14100	15390	16639
10366	12871	14200	15495	16744
10425	12965	14300	15599	16850
10484	13059	14400	15703	16955
10543	13153	14500	15807	17061
10602	13247	14600	15912	17167
10661	13341	14700	16016	17272
10720	13435	14800	16120	17378
10779	13529	14900	16225	17483
10837	13624	15000	16329	17589
10896	13718	15100	16433	17694
10955	13812	15200	16537	17800
11014	13906	15300	16642	17905
11073	14000	15400	16746	18011
11132	14094	15500	16850	18117

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HJ1-29000-003-001

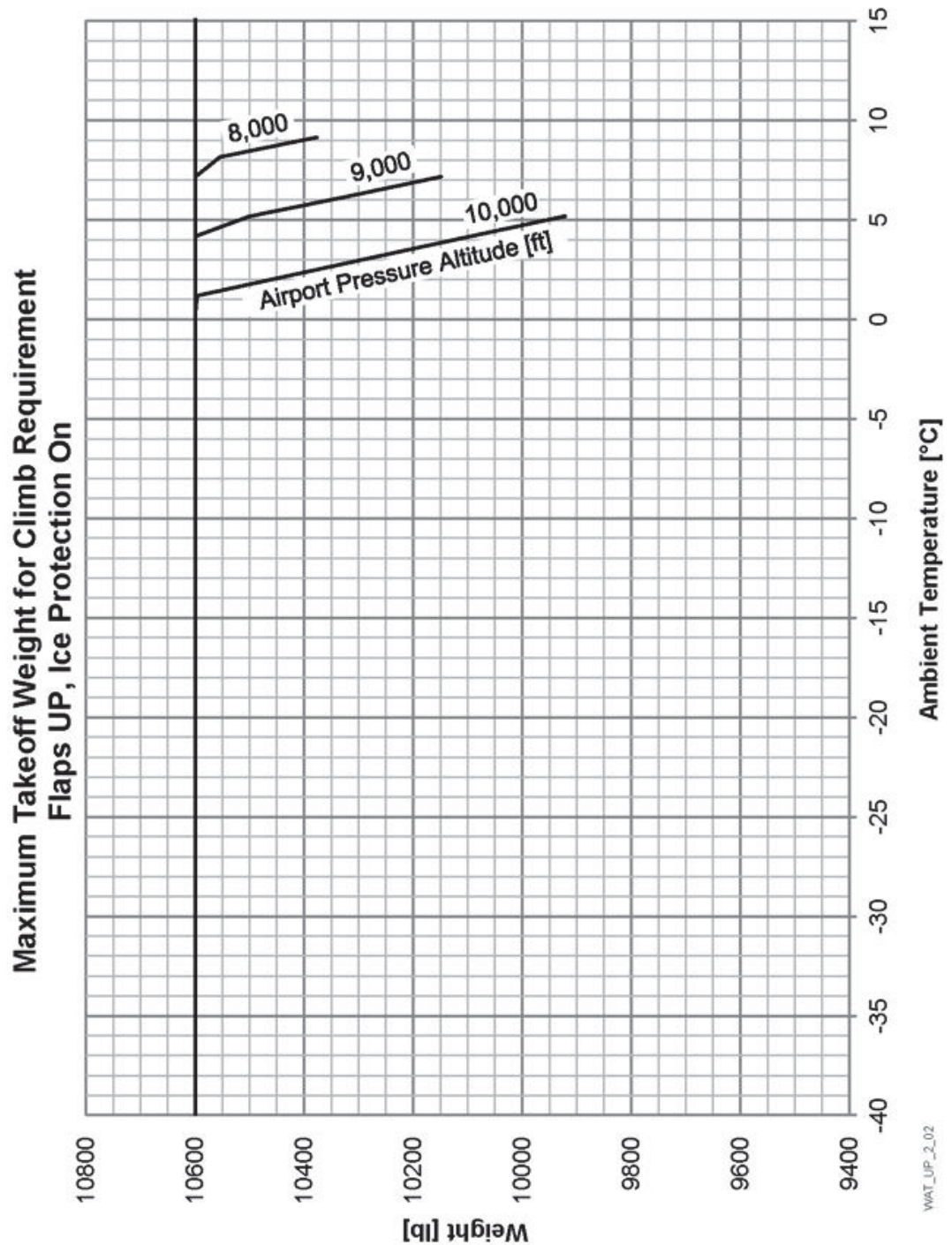
Page 5-138

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE



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October 30, 2016

HJ1-29000-003-001

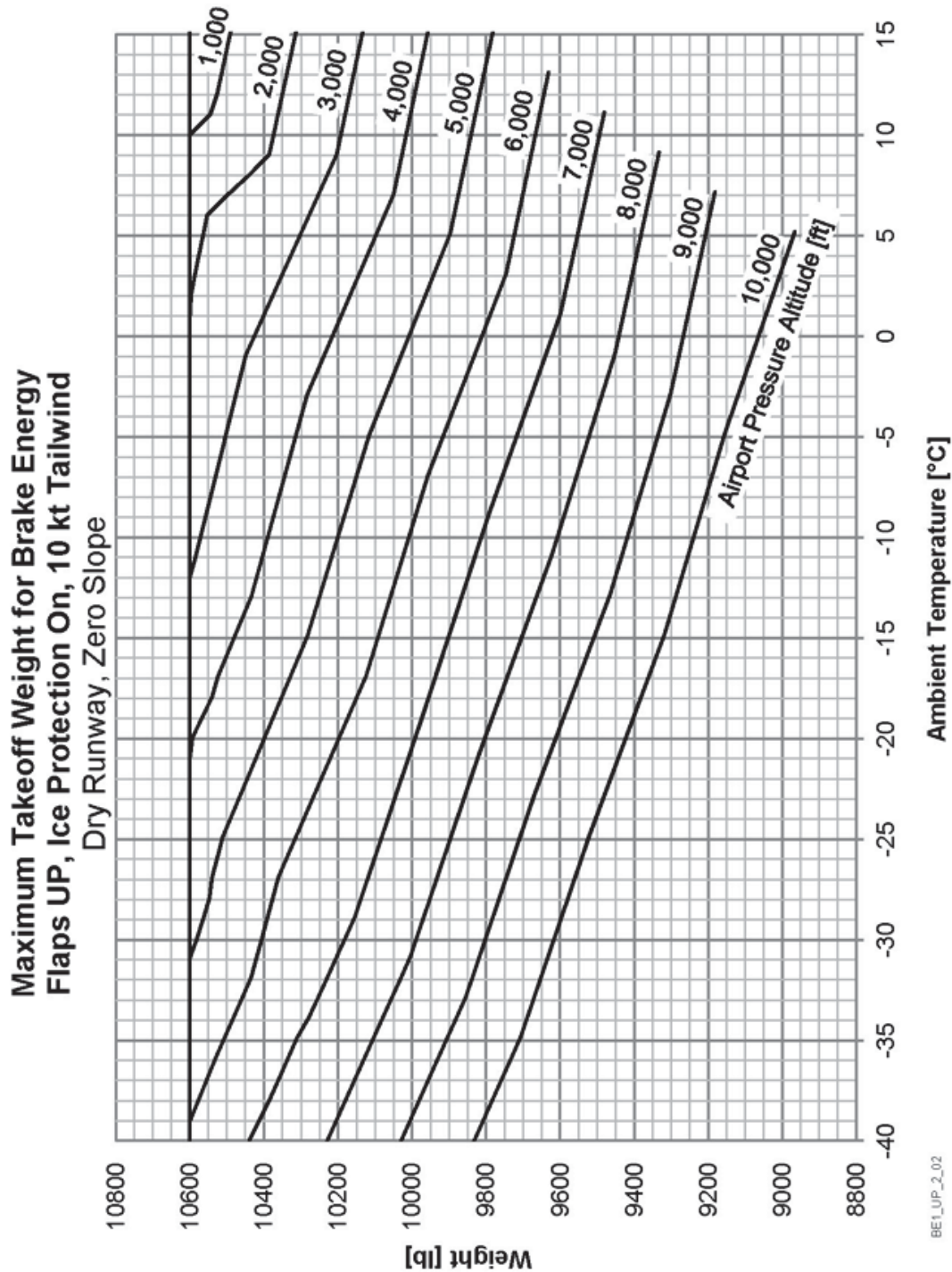
Page 5-139

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HA-420 AFM

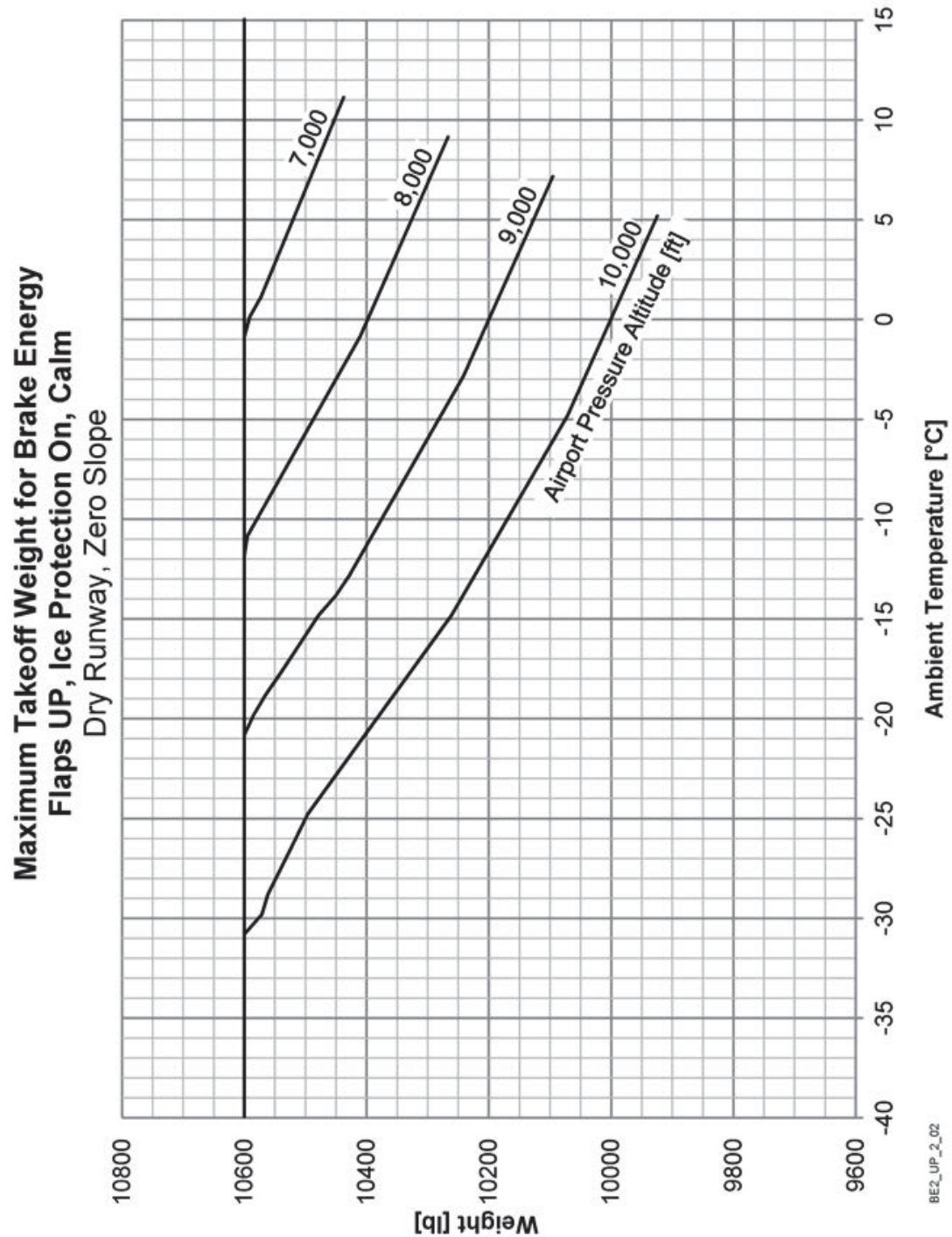
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HA-420 AFM

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October 30, 2016

HJ1-29000-003-001

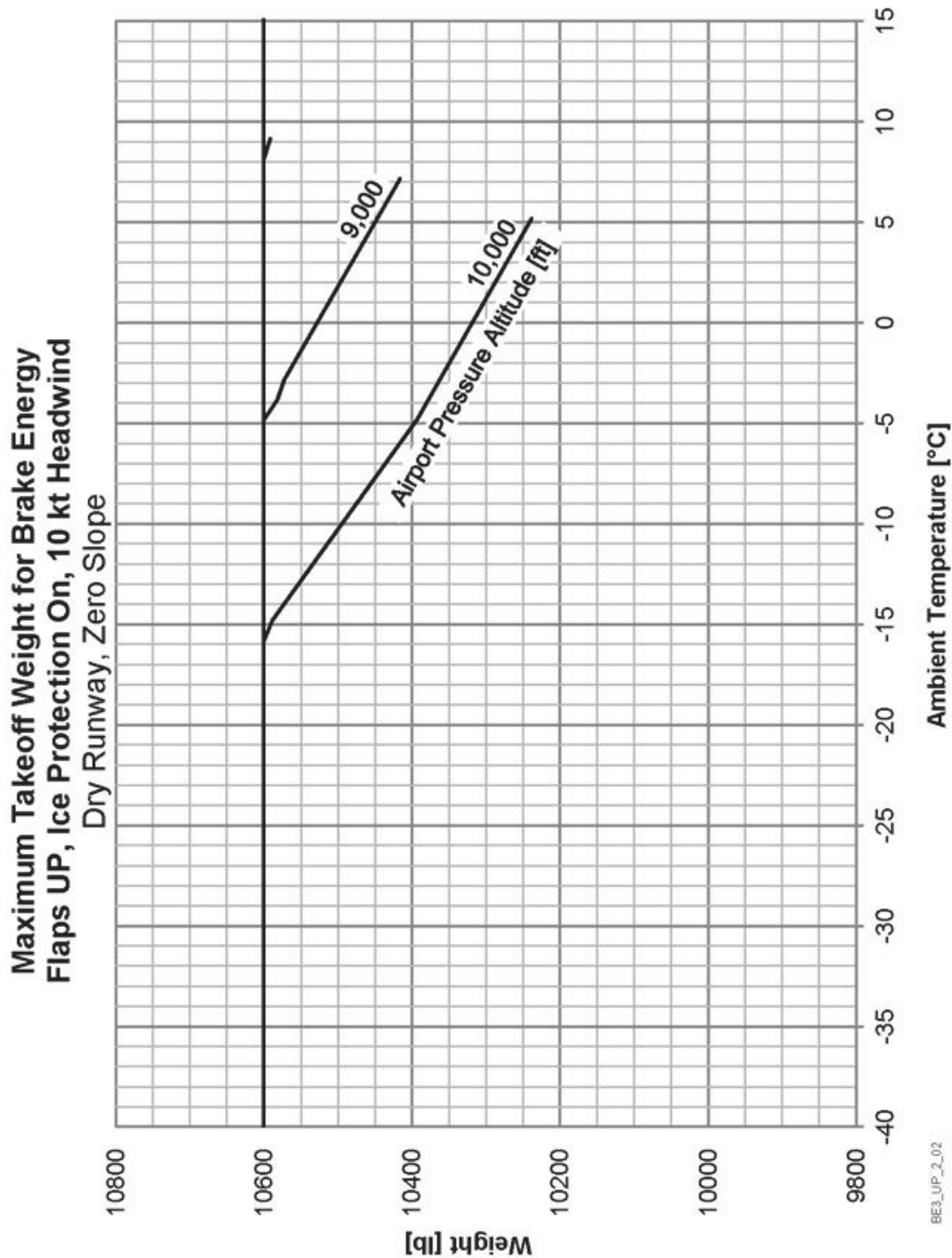
Page 5-141

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HA-420 AFM

PERFORMANCE



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October 30, 2016

HJ1-29000-003-001

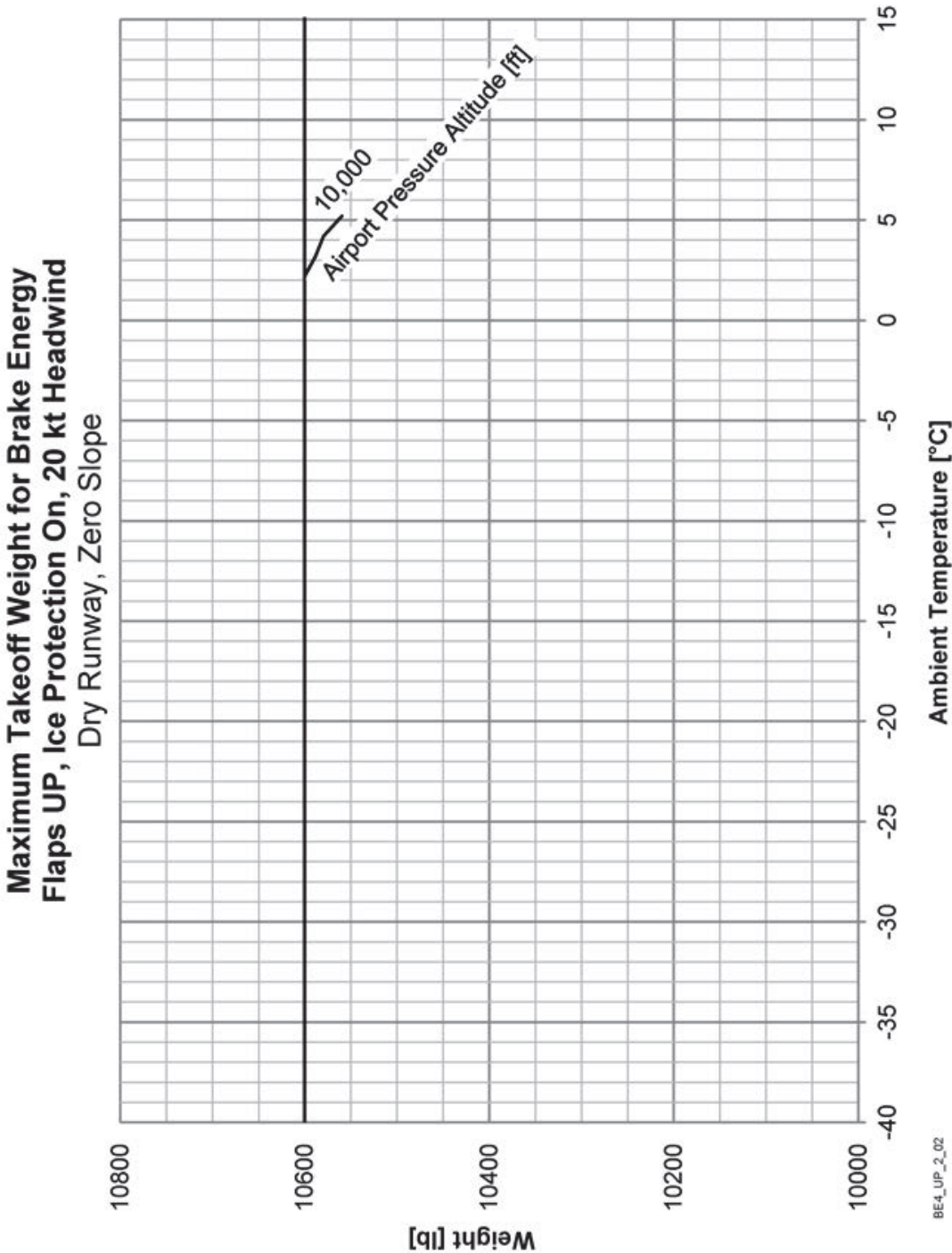
Page 5-142

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HA-420 AFM

PERFORMANCE



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October 30, 2016

HJ1-29000-003-001

Page 5-143

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Maximum Takeoff Weight for Brake Energy [lb]				
Flaps UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
7471	7629	7800	7974	8237
7568	7728	7900	8075	8334
7666	7827	8000	8175	8431
7763	7926	8100	8276	8528
7861	8025	8200	8377	8625
7958	8124	8300	8478	8722
8055	8223	8400	8578	8819
8153	8322	8500	8679	8916
8250	8421	8600	8780	9013
8348	8520	8700	8881	9110
8445	8619	8800	8981	9207
8543	8718	8900	9082	9304
8640	8817	9000	9183	9401
8738	8916	9100	9284	9499
8835	9015	9200	9384	9596
8933	9114	9300	9485	9693
9030	9213	9400	9586	9790
9127	9312	9500	9686	9887
9225	9411	9600	9787	9984
9322	9510	9700	9888	10081
9420	9609	9800	9989	10178
9517	9708	9900	10089	10275
9615	9807	10000	10190	10372
9712	9906	10100	10291	10469
9810	10005	10200	10392	10566
9907	10104	10300	10492	10600
10005	10203	10400	10593	10600
10102	10302	10500	10600	10600
10199	10400	10600	10600	10600

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October 30, 2016

HJ1-29000-003-001

Page 5-144

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HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude -1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	111	111	113	114	117	121	125
	VR	119	119	119	119	121	123	126
	V2	131	131	130	130	130	132	134
	TOFL	2736	2791	2923	3051	3270	3531	3892
-30	V1	110	111	113	114	117	121	125
	VR	119	119	119	119	121	123	126
	V2	131	131	130	129	130	132	134
	TOFL	2826	2883	3021	3153	3381	3653	4020
-20	V1	110	111	113	114	117	120	125
	VR	119	119	119	119	121	123	126
	V2	131	131	130	129	130	132	134
	TOFL	2913	2973	3116	3253	3489	3771	4143
-10	V1	110	111	112	114	117	120	124
	VR	119	119	119	119	121	123	126
	V2	131	130	130	129	130	132	134
	TOFL	3001	3063	3211	3354	3598	3891	4263
-5	V1	110	110	112	113	116	120	124
	VR	119	119	119	119	121	123	126
	V2	131	130	130	129	130	132	134
	TOFL	3042	3104	3256	3401	3649	3946	4312
0	V1	109	110	112	113	116	120	124
	VR	119	119	119	119	121	123	126
	V2	131	130	130	129	130	132	134
	TOFL	3082	3146	3300	3447	3699	4001	4374
5	V1	109	110	112	113	116	120	124
	VR	119	119	119	119	121	123	126
	V2	131	130	129	129	130	132	134
	TOFL	3123	3188	3344	3494	3750	4057	4436
10	V1	109	110	112	113	116	120	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	129	130	131	134
	TOFL	3163	3229	3389	3541	3801	4113	4498

TOFL_UP_2-1_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-145

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude Sea Level								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	111	111	113	114	117	121	125
	VR	119	119	119	119	121	123	126
	V2	131	131	130	129	130	132	134
	TOFL	2825	2881	3018	3150	3376	3647	4035
-30	V1	110	111	113	114	117	120	125
	VR	119	119	119	119	121	123	126
	V2	131	131	130	129	130	132	134
	TOFL	2920	2979	3121	3258	3493	3775	4168
-20	V1	110	111	112	114	117	120	125
	VR	119	119	119	119	121	123	126
	V2	131	130	130	129	130	132	134
	TOFL	3007	3068	3216	3358	3603	3895	4290
-10	V1	110	110	112	114	117	120	124
	VR	119	119	119	119	121	123	126
	V2	131	130	129	129	130	132	134
	TOFL	3097	3160	3314	3461	3714	4016	4410
-5	V1	110	110	112	113	116	120	124
	VR	119	119	119	119	121	123	126
	V2	130	130	129	129	130	131	134
	TOFL	3139	3203	3360	3509	3766	4074	4463
0	V1	109	110	112	113	116	120	124
	VR	119	119	119	119	121	123	126
	V2	130	130	129	129	130	131	134
	TOFL	3181	3247	3406	3558	3819	4131	4517
5	V1	109	110	112	113	116	120	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	129	130	131	134
	TOFL	3224	3291	3453	3608	3873	4190	4582
10	V1	109	110	112	113	116	119	123
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	130	131	133
	TOFL	3267	3336	3500	3657	3927	4249	4647

TOFL_UP_2_0_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-146

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 1000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	111	112	113	115	118	121	126
	VR	119	119	119	119	121	123	126
	V2	131	130	130	129	130	132	134
	TOFL	2964	3023	3163	3302	3537	3818	4394
-30	V1	111	112	113	114	117	121	126
	VR	119	119	119	119	121	123	126
	V2	130	130	129	129	130	131	134
	TOFL	3063	3123	3269	3413	3658	3950	4499
-20	V1	111	111	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	130	131	133
	TOFL	3155	3218	3370	3519	3773	4077	4590
-10	V1	110	111	113	114	117	120	125
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3247	3312	3470	3624	3887	4202	4697
-5	V1	110	111	112	114	117	120	125
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3292	3358	3519	3676	3943	4263	4754
0	V1	110	111	112	114	116	120	125
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3338	3405	3569	3727	3999	4324	4814
5	V1	110	111	112	113	116	120	125
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3383	3452	3619	3780	4056	4387	4876
10	V1	111	111	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	131	133
	TOFL	3484	3552	3716	3879	4156	4488	5154

TOFL_UP_2_1_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-147

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 2000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	112	112	114	115	118	121	126
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3107	3167	3312	3458	3702	3995	4767
-30	V1	111	112	113	115	118	121	126
	VR	119	119	119	119	121	123	126
	V2	130	130	129	128	129	131	133
	TOFL	3207	3269	3421	3572	3825	4130	4867
-20	V1	111	112	113	114	117	121	126
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3303	3368	3525	3682	3945	4262	4954
-10	V1	111	111	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3400	3468	3631	3792	4065	4393	5039
-5	V1	111	111	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3449	3518	3684	3848	4125	4459	5085
0	V1	110	111	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	131	133
	TOFL	3498	3568	3737	3903	4185	4524	5133
5	V1	110	111	112	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	131	133
	TOFL	3547	3618	3790	3959	4246	4591	5190
10	V1	112	113	114	115	118	122	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3704	3772	3942	4107	4390	4850	5898

TOFL_UP_2_2_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-148

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 3000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	112	113	114	115	118	122	126
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3257	3319	3471	3621	3875	4215	5149
-30	V1	112	112	114	115	118	121	126
	VR	119	119	119	119	121	123	126
	V2	130	129	129	128	129	131	133
	TOFL	3365	3429	3587	3744	4008	4339	5244
-20	V1	111	112	113	115	117	121	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	131	133
	TOFL	3460	3526	3690	3853	4126	4456	5332
-10	V1	111	112	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	131	133
	TOFL	3564	3633	3802	3971	4254	4596	5434
-5	V1	111	112	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	130	133
	TOFL	3616	3687	3858	4030	4319	4666	5489
0	V1	111	112	113	114	117	120	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	127	129	130	133
	TOFL	3669	3741	3915	4090	4384	4737	5548
5	V1	111	112	113	114	117	121	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	127	128	130	132
	TOFL	3748	3820	3997	4173	4469	4844	5768
10	V1	113	114	115	116	120	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	126	128	129	132
	TOFL	3946	4017	4189	4356	4743	5605	6713

TOFL_UP_2_3_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-149

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 4000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	113	113	114	116	118	123	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	130	133
	TOFL	3417	3481	3641	3797	4061	4520	5552
-30	V1	112	113	114	115	118	122	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	128	129	130	133
	TOFL	3526	3593	3758	3920	4195	4627	5655
-20	V1	112	112	114	115	118	122	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	127	128	130	133
	TOFL	3631	3700	3872	4040	4325	4751	5761
-10	V1	112	112	114	115	117	122	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3743	3815	3992	4167	4462	4888	5884
-5	V1	111	112	113	115	117	121	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3799	3872	4052	4230	4530	4956	5946
0	V1	111	112	113	115	117	121	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3855	3929	4112	4293	4599	5026	6008
5	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3989	4063	4246	4423	4727	5428	6568
10	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4218	4289	4464	4727	5526	6480	7605

TOFL_UP_2_4_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-150

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 5000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	113	113	115	116	118	123	126
	VR	119	119	119	119	121	123	126
	V2	129	129	128	127	128	130	132
	TOFL	3585	3651	3818	3979	4255	4924	5977
-30	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3704	3772	3946	4113	4400	5006	6091
-20	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	129	128	128	127	128	130	132
	TOFL	3812	3883	4063	4237	4534	5103	6216
-10	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3930	4004	4191	4371	4679	5225	6353
-5	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3990	4066	4255	4439	4752	5295	6424
0	V1	112	112	114	115	117	122	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	4051	4127	4320	4507	4826	5373	6501
5	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	127	126	127	129	131
	TOFL	4246	4322	4505	4683	5272	6251	7404
10	V1	116	116	118	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	126	125	126	128	131
	TOFL	4507	4579	4899	5556	6447	7413	8556

TOFL_UP_2_5_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-151

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 6000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	113	114	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	128	127	128	130	132
	TOFL	3759	3830	4003	4170	4504	5317	6396
-30	V1	113	113	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3881	3955	4135	4308	4634	5422	6532
-20	V1	113	113	115	116	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	4000	4076	4263	4443	4762	5529	6669
-10	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	4128	4206	4400	4588	4909	5660	6829
-5	V1	112	113	114	115	118	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	126	128	129	132
	TOFL	4194	4274	4471	4662	4989	5735	6918
0	V1	113	114	115	116	120	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	127	126	127	129	132
	TOFL	4317	4397	4593	4782	5237	6155	7340
5	V1	115	115	116	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	128	131
	TOFL	4547	4623	4829	5297	6212	7203	8374
10	V1	118	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	125	125	125	126	128	130
	TOFL	4940	5109	5827	6511	7410	8397	9570

TOFL_UP_2_6_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-152

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 7000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	114	114	115	116	120	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	127	128	130	132
	TOFL	3946	4020	4198	4372	4808	5731	6837
-30	V1	113	114	115	116	120	123	126
	VR	119	119	119	119	121	123	126
	V2	128	128	127	126	128	129	132
	TOFL	4079	4156	4342	4522	4948	5853	6990
-20	V1	113	114	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4205	4284	4478	4666	5091	5985	7154
-10	V1	113	113	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4345	4427	4628	4823	5259	6147	7345
-5	V1	113	113	115	116	119	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	127	126	127	129	132
	TOFL	4420	4504	4709	4907	5356	6253	7466
0	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	125	127	129	131
	TOFL	4606	4686	4884	5199	6045	7061	8264
5	V1	116	117	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	130
	TOFL	4860	5009	5550	6258	7179	8185	9380
10	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	5831	6123	6829	7504	8417	9430	10639

TOFL_UP_2_7_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-153

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 8000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	128	127	127	126	127	129	132
	TOFL	4143	4219	4404	4584	5217	6180	7316
-30	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	127	126	127	129	132
	TOFL	4283	4362	4554	4742	5329	6319	7487
-20	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	127	126	127	129	131
	TOFL	4425	4507	4708	4903	5478	6495	7695
-10	V1	113	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4581	4666	4875	5077	5665	6707	7938
-5	V1	114	114	115	116	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4687	4773	4982	5212	5930	6980	8221
0	V1	115	116	118	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	131
	TOFL	4913	4995	5361	6042	6987	8018	9242
5	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	125	127	130
	TOFL	5504	5807	6537	7234	8166	9194	10419
10	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	124	125	127	129
	TOFL	6853	7141	7838	8507	9438	10482	11734

TOFL_UP_2_8_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-154

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 9000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	114	115	116	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4347	4425	4618	4868	5644	6627	7789
-30	V1	114	114	115	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4501	4583	4783	5030	5787	6798	7995
-20	V1	114	114	115	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4659	4744	4952	5217	5989	7028	8258
-10	V1	114	114	115	117	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	129	131
	TOFL	4831	4919	5135	5426	6220	7284	8545
-5	V1	115	115	117	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	125	125	126	128	131
	TOFL	4993	5080	5354	5878	6853	7914	9173
0	V1	117	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	125	124	126	128	130
	TOFL	5366	5538	6278	7002	7956	9006	10258
5	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	124	124	124	125	127	129
	TOFL	6536	6834	7556	8247	9195	10252	11519
10	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	123	123	124	126	129
	TOFL	7880	8165	8857	9525	10478	11556	12856

TOFL_UP_2_9_02

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October 30, 2016

HJ1-29000-003-001

Page 5-155

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Takeoff Field Length [feet] and Speeds [KIAS]								
Dry Runway, Zero Slope, No Wind								
Flaps UP, Ice Protection On, Altitude 10,000 feet								
Temp. [°C]	Data	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-40	V1	115	115	116	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	127	126	126	127	129	131
	TOFL	4573	4654	4855	5218	6127	7132	8323
-30	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	127	129	131
	TOFL	4741	4827	5035	5403	6310	7344	8570
-20	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	127	126	126	125	126	128	131
	TOFL	4914	5003	5220	5614	6552	7615	8875
-10	V1	114	115	116	118	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	126	126	125	126	128	131
	TOFL	5094	5186	5411	5832	6793	7885	9182
-5	V1	116	117	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	126	125	125	124	126	128	130
	TOFL	5302	5478	6031	6783	7764	8839	10118
0	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	125	125	124	124	125	127	130
	TOFL	6211	6521	7267	7981	8946	10017	11298
5	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	124	124	124	123	125	127	129
	TOFL	7610	7905	8619	9307	10275	11366	12682
10	V1	119	119	119	119	121	123	126
	VR	119	119	119	119	121	123	126
	V2	123	123	123	123	124	126	129
	TOFL	8954	9236	9926	10594	11572	12692	14048

TOFL_UP_2_10_02

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October 30, 2016

HJ1-29000-003-001

Page 5-156

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected V1 [KIAS]		
FLAPS UP Ice Protection On		
Tailwind		Headwind
10	◀ REF [0] ▶	30
108	109	111
109	110	112
110	111	113
111	112	114
112	113	114
113	114	115
114	115	116
115	116	117
116	117	118
117	118	119
118	119	120
119	120	121
120	121	122
121	122	123
122	123	124
123	124	125
124	125	126
125	126	126

V1WC_UP_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-157

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HA-420 AFM

PERFORMANCE

Slope Corrected V1 [KIAS]				
FLAPS UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
106	107	108	110	112
106	107	109	111	113
107	108	110	112	114
107	109	111	113	115
108	110	112	114	116
109	111	113	115	116
110	112	114	116	117
111	113	115	117	118
112	114	116	118	119
114	115	117	119	120
115	116	118	120	120
116	117	119	121	121
117	118	120	121	122
118	119	121	122	123
119	121	122	123	124
120	122	123	124	125
121	123	124	125	125
122	124	125	126	126
124	125	126	126	126

V1SC_UP_2_04

Takeoff Rotation Speed (V_R) Slope Correction, Ice Protection On

NOTE V_R with flaps up and Ice Protection On does not require slope correction.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected V2 [KIAS]				
FLAPS UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
123	123	122	121	121
124	124	123	122	122
125	125	124	123	123
126	126	125	124	124
127	127	126	125	125
128	128	127	126	126
129	129	128	127	127
130	130	129	128	128
131	131	130	129	129
132	132	131	130	130
133	133	132	131	131
134	134	133	132	132
135	135	134	133	133

V2SC_UP_2_05

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October 30, 2016

HJ1-29000-003-001

Page 5-159

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
3311	2700	2535	2409	2265
3417	2800	2633	2503	2357
3523	2900	2730	2598	2449
3629	3000	2828	2692	2541
3734	3100	2925	2787	2634
3840	3200	3023	2881	2726
3946	3300	3121	2975	2818
4051	3400	3218	3070	2911
4157	3500	3316	3164	3003
4263	3600	3413	3259	3095
4369	3700	3511	3353	3187
4474	3800	3608	3448	3280
4580	3900	3706	3542	3372
4686	4000	3804	3636	3464
4791	4100	3901	3731	3557
4897	4200	3999	3825	3649
5003	4300	4096	3920	3741
5109	4400	4194	4014	3833
5214	4500	4291	4109	3926
5320	4600	4389	4203	4018
5426	4700	4487	4297	4110
5531	4800	4584	4392	4203
5637	4900	4682	4486	4295
5743	5000	4779	4581	4387
5849	5100	4877	4675	4479
5954	5200	4974	4770	4572
6060	5300	5072	4864	4664
6166	5400	5170	4958	4756
6271	5500	5267	5053	4849
6377	5600	5365	5147	4941
6483	5700	5462	5242	5033
6589	5800	5560	5336	5125
6694	5900	5657	5431	5218
6800	6000	5755	5525	5310

TOWC1_UP_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-160

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
6800	6000	5755	5525	5310
6906	6100	5853	5619	5402
7011	6200	5950	5714	5495
7117	6300	6048	5808	5587
7223	6400	6145	5903	5679
7329	6500	6243	5997	5771
7434	6600	6340	6092	5864
7540	6700	6438	6186	5956
7646	6800	6536	6280	6048
7751	6900	6633	6375	6141
7857	7000	6731	6469	6233
7963	7100	6828	6564	6325
8069	7200	6926	6658	6417
8174	7300	7023	6753	6510
8280	7400	7121	6847	6602
8386	7500	7219	6941	6694
8491	7600	7316	7036	6787
8597	7700	7414	7130	6879
8703	7800	7511	7225	6971
8809	7900	7609	7319	7063
8914	8000	7706	7414	7156
9020	8100	7804	7508	7248
9126	8200	7902	7602	7340
9231	8300	7999	7697	7433
9337	8400	8097	7791	7525
9443	8500	8194	7886	7617
9549	8600	8292	7980	7709
9654	8700	8389	8075	7802
9760	8800	8487	8169	7894
9866	8900	8585	8263	7986
9971	9000	8682	8358	8079
10077	9100	8780	8452	8171
10183	9200	8877	8547	8263
10289	9300	8975	8641	8355

TOWC2_UP_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-161

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
10289	9300	8975	8641	8355
10394	9400	9072	8736	8448
10500	9500	9170	8830	8540
10606	9600	9268	8924	8632
10711	9700	9365	9019	8725
10817	9800	9463	9113	8817
10923	9900	9560	9208	8909
11029	10000	9658	9302	9001
11134	10100	9755	9397	9094
11240	10200	9853	9491	9186
11346	10300	9951	9585	9278
11451	10400	10048	9680	9371
11557	10500	10146	9774	9463
11663	10600	10243	9869	9555
11769	10700	10341	9963	9647
11874	10800	10438	10058	9740
11980	10900	10536	10152	9832
12086	11000	10634	10246	9924
12191	11100	10731	10341	10017
12297	11200	10829	10435	10109
12403	11300	10926	10530	10201
12509	11400	11024	10624	10293
12614	11500	11121	10719	10386
12720	11600	11219	10813	10478
12826	11700	11317	10907	10570
12931	11800	11414	11002	10663
13037	11900	11512	11096	10755
13143	12000	11609	11191	10847
13249	12100	11707	11285	10939
13354	12200	11804	11380	11032
13460	12300	11902	11474	11124
13566	12400	12000	11568	11216
13671	12500	12097	11663	11309
13777	12600	12195	11757	11401

TOWC3_UP_2_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-162

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
13777	12600	12195	11757	11401
13883	12700	12292	11852	11493
13989	12800	12390	11946	11585
14094	12900	12487	12041	11678

TOWC4_UP_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-163

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
2205	2205	2200	2278	2357
2308	2308	2300	2416	2536
2410	2410	2400	2555	2715
2513	2513	2500	2694	2894
2615	2615	2600	2833	3072
2718	2718	2700	2971	3251
2821	2821	2800	3110	3430
2923	2923	2900	3249	3609
3026	3026	3000	3388	3787
3128	3128	3100	3526	3966
3231	3231	3200	3665	4145
3315	3322	3300	3804	4323
3399	3414	3400	3943	4502
3483	3505	3500	4082	4681
3567	3596	3600	4220	4860
3651	3687	3700	4359	5038
3735	3778	3800	4498	5217
3819	3870	3900	4637	5396
3903	3961	4000	4775	5574
3987	4052	4100	4896	5717
4071	4143	4200	5006	5839
4155	4234	4300	5116	5962
4239	4326	4400	5226	6085
4323	4417	4500	5336	6207
4407	4508	4600	5446	6330
4491	4599	4700	5557	6453
4575	4690	4800	5667	6576
4659	4781	4900	5777	6698
4743	4873	5000	5887	6821
4827	4964	5100	5997	6935
4911	5055	5200	6107	7049
4995	5146	5300	6217	7162
5079	5237	5400	6328	7276
5156	5326	5500	6438	7389

TOSC1_UP_2_03

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October 30, 2016

HJ1-29000-003-001

Page 5-164

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
5156	5326	5500	6438	7389
5217	5406	5600	6548	7503
5278	5487	5700	6658	7616
5339	5567	5800	6768	7730
5400	5647	5900	6878	7844
5461	5728	6000	6982	7957
5522	5808	6100	7087	8071
5583	5889	6200	7191	8184
5644	5969	6300	7295	8298
5705	6050	6400	7399	8412
5766	6130	6500	7503	8525
5827	6211	6600	7608	8639
5888	6291	6700	7712	8752
5949	6371	6800	7816	8866
6010	6452	6900	7920	8980
6071	6532	7000	8024	9093
6132	6613	7100	8129	9201
6193	6693	7200	8233	9307
6254	6774	7300	8337	9414
6315	6854	7400	8441	9521
6376	6935	7500	8545	9627
6437	7015	7600	8650	9734
6498	7095	7700	8754	9840
6559	7176	7800	8858	9947
6620	7256	7900	8962	10054
6681	7337	8000	9066	10160
6742	7417	8100	9170	10267
6803	7498	8200	9275	10373
6864	7578	8300	9379	10480
6925	7659	8400	9483	10587
6986	7739	8500	9587	10693
7047	7819	8600	9691	10800
7108	7900	8700	9796	10906
7169	7980	8800	9900	11013

TOSC2_UP_2_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-165

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
7169	7980	8800	9900	11013
7230	8061	8900	10002	11117
7291	8141	9000	10097	11213
7352	8222	9100	10192	11309
7412	8302	9200	10287	11405
7473	8382	9300	10382	11500
7534	8463	9400	10478	11596
7595	8543	9500	10573	11692
7656	8624	9600	10668	11788
7717	8704	9700	10763	11884
7778	8785	9800	10858	11980
7839	8865	9900	10953	12076
7900	8946	10000	11048	12171
7961	9026	10100	11144	12267
8022	9106	10200	11239	12363
8083	9187	10300	11334	12459
8144	9267	10400	11429	12555
8205	9348	10500	11524	12651
8266	9428	10600	11619	12747
8327	9509	10700	11715	12842
8388	9589	10800	11810	12938
8449	9670	10900	11905	13034
8510	9750	11000	12000	13130
8571	9830	11100	12095	13226
8632	9911	11200	12190	13322
8693	9991	11300	12285	13418
8754	10072	11400	12381	13513
8815	10152	11500	12476	13609
8876	10233	11600	12571	13705
8937	10313	11700	12666	13801
8998	10394	11800	12761	13897
9059	10474	11900	12856	13993
9120	10554	12000	12952	14089
9181	10635	12100	13047	14184

TOSC3_UP_2_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-166

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Takeoff Field Length [feet]				
FLAPS UP, Ice Protection On				
Runway Gradient [%]				
-2	-1	◀ REF [0] ▶	1	2
9181	10635	12100	13047	14184
9242	10715	12200	13142	14280
9303	10796	12300	13237	14376
9364	10876	12400	13332	14472
9425	10957	12500	13427	14568
9486	11037	12600	13522	14664
9547	11118	12700	13618	14760
9608	11198	12800	13713	14855
9668	11278	12900	13808	14951
9729	11359	13000	13903	15047
9790	11439	13100	13998	15143
9851	11520	13200	14093	15239
9912	11600	13300	14189	15335
9973	11681	13400	14284	15431
10034	11761	13500	14379	15526
10095	11841	13600	14474	15622
10156	11922	13700	14569	15718
10217	12002	13800	14664	15814
10278	12083	13900	14759	15910
10339	12163	14000	14855	16006
10400	12244	14100	14950	16102

TOSC4_UP_2_03

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-167

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	11.9	11.4	10.3	9.3	8.4	7.5	6.5
	15	12.2	11.7	10.6	9.6	8.6	7.7	6.7
	25	12.2	11.8	10.7	9.6	8.6	7.7	6.7
	35	10.4	9.9	8.9	8.0	7.1	6.2	5.3
	40	9.2	8.8	7.8	6.9	6.1	5.3	4.4
	45	8.0	7.6	6.7	5.9	5.1	4.3	3.5
	50	6.8	6.5	5.6	4.8	4.1	3.4	2.6
	55	5.7	5.3	4.6	3.8	3.1	2.5	1.8
Sea Level	-40	11.6	11.2	10.1	9.1	8.2	7.3	6.3
	15	12.0	11.5	10.4	9.4	8.4	7.5	6.5
	25	12.0	11.6	10.5	9.5	8.5	7.6	6.6
	35	9.7	9.3	8.3	7.4	6.5	5.7	4.8
	40	8.5	8.1	7.2	6.4	5.5	4.8	3.9
	45	7.4	7.0	6.1	5.3	4.6	3.8	3.1
	50	6.2	5.9	5.1	4.3	3.6	2.9	2.2
	55	5.1	4.8	4.0	3.3	2.7	2.0	1.4
1000	-40	11.3	10.8	9.8	8.8	7.9	7.0	6.0
	15	11.6	11.2	10.1	9.1	8.1	7.2	6.3
	25	11.2	10.8	9.7	8.7	7.8	6.9	6.0
	30	10.1	9.7	8.7	7.7	6.9	6.0	5.1
	35	9.0	8.6	7.6	6.7	5.9	5.1	4.3
	40	7.8	7.4	6.6	5.7	4.9	4.2	3.4
	45	6.7	6.3	5.5	4.7	4.0	3.3	2.5
	50	5.6	5.2	4.5	3.7	3.1	2.4	1.7
2000	-40	10.9	10.5	9.4	8.5	7.6	6.7	5.8
	10	11.2	10.8	9.7	8.7	7.8	6.9	6.0
	20	11.3	10.8	9.8	8.8	7.8	7.0	6.0
	30	9.3	8.9	7.9	7.0	6.2	5.4	4.5
	35	8.2	7.8	6.9	6.1	5.3	4.5	3.7
	40	7.1	6.8	5.9	5.1	4.3	3.6	2.9
	45	6.0	5.7	4.9	4.1	3.4	2.8	2.1
	50	5.0	4.6	3.9	3.2	2.5	1.9	1.2

TCG2_UP_0-1_06

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October 30, 2016

HJ1-29000-003-001

Page 5-168

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	10.5	10.1	9.1	8.1	7.2	6.4	5.5
	10	10.9	10.4	9.4	8.4	7.5	6.6	5.7
	20	10.7	10.3	9.2	8.3	7.3	6.5	5.6
	25	9.6	9.2	8.2	7.3	6.4	5.6	4.7
	30	8.5	8.1	7.2	6.3	5.5	4.7	3.9
	35	7.4	7.1	6.2	5.4	4.6	3.9	3.1
	40	6.4	6.1	5.3	4.5	3.8	3.1	2.4
	45	5.4	5.1	4.3	3.6	2.9	2.3	1.6
4000	-40	10.2	9.7	8.7	7.8	6.9	6.1	5.2
	10	10.5	10.0	9.0	8.1	7.2	6.3	5.4
	20	9.8	9.4	8.4	7.5	6.6	5.8	4.9
	25	8.7	8.4	7.4	6.5	5.7	4.9	4.1
	30	7.7	7.3	6.4	5.6	4.8	4.1	3.3
	35	6.7	6.3	5.5	4.7	4.0	3.3	2.6
	40	5.7	5.4	4.6	3.9	3.2	2.5	1.8
	45	4.8	4.4	3.7	3.0	2.4	1.8	1.1
5000	-40	9.8	9.4	8.4	7.5	6.6	5.8	4.9
	5	10.1	9.6	8.6	7.7	6.8	6.0	5.1
	15	10.1	9.6	8.6	7.7	6.8	6.0	5.1
	25	7.9	7.6	6.7	5.8	5.0	4.3	3.5
	30	6.9	6.5	5.7	4.9	4.2	3.5	2.7
	35	5.9	5.6	4.8	4.1	3.4	2.7	2.0
	40	5.0	4.7	3.9	3.2	2.6	2.0	1.3
	45	4.1	3.8	3.1	2.4	1.8	1.2	0.6
6000	-40	9.5	9.1	8.1	7.2	6.3	5.6	4.7
	0	9.7	9.3	8.3	7.4	6.5	5.7	4.8
	10	9.7	9.2	8.2	7.3	6.5	5.7	4.8
	20	8.2	7.8	6.9	6.0	5.2	4.5	3.7
	25	7.2	6.8	5.9	5.1	4.4	3.7	2.9
	30	6.2	5.8	5.0	4.3	3.5	2.9	2.2
	35	5.2	4.9	4.1	3.4	2.8	2.1	1.5
	40	4.3	4.0	3.3	2.6	2.0	1.4	0.8

TCG2_UP_0_3_06

FAA APPROVED
March 3, 2017

HJ1-29000-003-001

Page 5-169

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	9.2	8.8	7.8	6.9	6.1	5.3	4.4
	0	9.3	8.9	7.9	7.0	6.1	5.4	4.5
	10	9.2	8.8	7.9	7.0	6.1	5.3	4.5
	15	8.4	8.0	7.1	6.2	5.4	4.7	3.8
	20	7.4	7.0	6.2	5.4	4.6	3.9	3.1
	25	6.4	6.1	5.2	4.5	3.7	3.1	2.3
	30	5.5	5.1	4.3	3.6	2.9	2.3	1.6
	35	4.6	4.3	3.5	2.8	2.2	1.6	1.0
8000	-40	8.8	8.4	7.5	6.6	5.8	5.0	4.2
	-10	8.9	8.5	7.6	6.7	5.8	5.1	4.2
	0	8.9	8.5	7.5	6.6	5.8	5.0	4.2
	10	8.7	8.3	7.3	6.4	5.6	4.9	4.0
	15	7.6	7.2	6.3	5.5	4.7	4.0	3.2
	20	6.6	6.3	5.4	4.7	3.9	3.2	2.5
	25	5.7	5.3	4.6	3.8	3.1	2.5	1.8
	30	4.8	4.5	3.7	3.0	2.4	1.8	1.1
9000	-40	8.5	8.1	7.2	6.4	5.5	4.8	4.0
	-10	8.5	8.1	7.2	6.3	5.5	4.8	3.9
	0	8.5	8.1	7.2	6.3	5.5	4.7	3.9
	10	7.8	7.4	6.5	5.7	4.9	4.2	3.4
	15	6.8	6.5	5.6	4.8	4.1	3.4	2.6
	20	5.9	5.6	4.8	4.0	3.3	2.7	2.0
	25	5.0	4.7	3.9	3.2	2.5	1.9	1.3
	30	4.1	3.8	3.1	2.4	1.8	1.2	0.6
10,000	-40	8.1	7.7	6.8	6.0	5.2	4.4	3.6
	-5	8.1	7.7	6.8	6.0	5.2	4.4	3.6
	5	8.1	7.7	6.8	5.9	5.1	4.4	3.6
	10	7.0	6.6	5.8	5.0	4.2	3.5	2.8
	15	6.1	5.7	4.9	4.2	3.5	2.8	2.1
	20	5.2	4.9	4.1	3.4	2.7	2.1	1.4
	25	4.3	4.0	3.3	2.6	2.0	1.4	0.8
	30	3.5	3.2	2.5	1.9	1.3	0.7	0.1

TCG2_UP_0_7_07

FAA APPROVED
March 3, 2017

HJ1-29000-003-001

Page 5-170

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HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Climb Gradient [%]				
FLAPS UP and TO/APPR, V2, Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.0	0.0	0.0	0.0	0.0
0.5	0.5	0.5	0.5	0.5
0.9	1.0	1.0	1.0	1.1
1.4	1.5	1.5	1.6	1.6
1.8	2.0	2.1	2.1	2.2
2.3	2.5	2.6	2.7	2.8
2.7	3.0	3.1	3.2	3.4
3.2	3.5	3.6	3.8	3.9
3.6	4.0	4.2	4.3	4.5
4.0	4.5	4.7	4.9	5.1
4.5	5.0	5.2	5.4	5.7
4.9	5.5	5.7	6.0	6.2
5.4	6.0	6.2	6.5	6.8
5.8	6.5	6.8	7.1	7.4
6.3	7.0	7.3	7.6	7.9
6.7	7.5	7.8	8.2	8.5
7.1	8.0	8.3	8.7	9.1
7.6	8.5	8.9	9.2	9.7
8.0	9.0	9.4	9.8	10.2
8.5	9.5	9.9	10.3	10.8
8.9	10.0	10.4	10.9	11.4
9.4	10.5	10.9	11.4	12.0
9.8	11.0	11.5	12.0	12.5
10.2	11.5	12.0	12.5	13.1
10.7	12.0	12.5	13.1	13.7

CGWC_TO_ALL_05

NOTE Use this table when determining the ground reference flight path for obstacle clearance with Ice Protection Off.

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-171

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	11.9	11.4	10.3	9.4	8.4	7.5	6.5
	-30	11.9	11.5	10.4	9.4	8.4	7.5	6.5
	-20	12.0	11.5	10.4	9.4	8.4	7.5	6.6
	-10	12.0	11.5	10.5	9.5	8.5	7.6	6.6
	-5	12.1	11.6	10.5	9.5	8.5	7.6	6.6
	0	12.1	11.6	10.5	9.5	8.6	7.6	6.7
	5	12.1	11.7	10.6	9.6	8.6	7.7	6.7
	10	12.2	11.7	10.6	9.6	8.6	7.7	6.7
Sea Level	-40	11.6	11.2	10.1	9.1	8.2	7.3	6.3
	-30	11.7	11.2	10.2	9.2	8.2	7.3	6.4
	-20	11.7	11.3	10.2	9.2	8.3	7.4	6.4
	-10	11.8	11.4	10.3	9.3	8.3	7.4	6.5
	-5	11.9	11.4	10.3	9.3	8.4	7.5	6.5
	0	11.9	11.5	10.4	9.4	8.4	7.5	6.5
	5	12.0	11.5	10.4	9.4	8.4	7.5	6.5
	10	12.0	11.5	10.4	9.4	8.4	7.5	6.6
1000	-40	11.3	10.8	9.8	8.8	7.9	7.0	6.1
	-30	11.3	10.9	9.8	8.9	7.9	7.0	6.1
	-20	11.4	11.0	9.9	8.9	8.0	7.1	6.1
	-10	11.5	11.0	10.0	9.0	8.0	7.1	6.2
	-5	11.5	11.1	10.0	9.0	8.1	7.2	6.2
	0	11.6	11.1	10.0	9.0	8.1	7.2	6.2
	5	11.6	11.1	10.1	9.1	8.1	7.2	6.3
	10	11.1	10.7	9.6	8.7	7.7	6.9	5.9
2000	-40	10.9	10.5	9.4	8.5	7.6	6.7	5.8
	-30	11.0	10.6	9.5	8.6	7.6	6.8	5.8
	-20	11.1	10.6	9.6	8.6	7.7	6.8	5.9
	-10	11.2	10.7	9.7	8.7	7.8	6.9	5.9
	-5	11.2	10.8	9.7	8.7	7.8	6.9	6.0
	0	11.2	10.8	9.7	8.7	7.8	6.9	6.0
	5	11.3	10.8	9.7	8.8	7.8	7.0	6.0
	10	10.3	9.9	8.8	7.9	7.0	6.2	5.3

TCG2_UP_2-1_02

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October 30, 2016

HJ1-29000-003-001

Page 5-172

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	10.6	10.1	9.1	8.2	7.3	6.4	5.5
	-30	10.7	10.2	9.2	8.3	7.3	6.5	5.6
	-20	10.8	10.3	9.3	8.3	7.4	6.6	5.6
	-10	10.8	10.4	9.3	8.4	7.5	6.6	5.7
	-5	10.9	10.4	9.4	8.4	7.5	6.6	5.7
	0	10.9	10.4	9.4	8.4	7.5	6.6	5.7
	5	10.7	10.2	9.2	8.2	7.3	6.5	5.6
	10	9.4	9.0	8.0	7.1	6.3	5.5	4.6
4000	-40	10.2	9.8	8.8	7.9	7.0	6.1	5.2
	-30	10.3	9.9	8.9	7.9	7.0	6.2	5.3
	-20	10.4	10.0	8.9	8.0	7.1	6.3	5.4
	-10	10.4	10.0	9.0	8.1	7.1	6.3	5.4
	-5	10.5	10.0	9.0	8.1	7.2	6.3	5.4
	0	10.5	10.1	9.0	8.1	7.2	6.3	5.4
	5	9.8	9.4	8.4	7.5	6.6	5.8	4.9
	10	8.5	8.1	7.2	6.3	5.5	4.8	3.9
5000	-40	9.9	9.5	8.5	7.6	6.7	5.9	5.0
	-30	10.0	9.5	8.6	7.6	6.7	5.9	5.0
	-20	10.0	9.6	8.6	7.7	6.8	6.0	5.1
	-10	10.1	9.7	8.7	7.7	6.8	6.0	5.1
	-5	10.1	9.7	8.7	7.7	6.9	6.0	5.1
	0	10.1	9.7	8.7	7.8	6.9	6.0	5.1
	5	8.9	8.5	7.6	6.7	5.9	5.1	4.3
	10	7.6	7.2	6.4	5.5	4.8	4.0	3.3
6000	-40	9.6	9.1	8.2	7.3	6.4	5.6	4.7
	-30	9.6	9.2	8.2	7.3	6.5	5.7	4.8
	-20	9.7	9.3	8.3	7.4	6.5	5.7	4.8
	-10	9.7	9.3	8.3	7.4	6.5	5.7	4.9
	-5	9.7	9.3	8.3	7.4	6.5	5.7	4.9
	0	9.3	8.9	7.9	7.0	6.1	5.4	4.5
	5	8.0	7.6	6.7	5.9	5.1	4.4	3.6
	10	6.8	6.4	5.6	4.8	4.0	3.4	2.6

TCG2_UP_2_3_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-173

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Second Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, V2								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	9.2	8.8	7.9	7.0	6.1	5.3	4.5
	-30	9.3	8.9	7.9	7.0	6.2	5.4	4.5
	-20	9.4	8.9	8.0	7.1	6.2	5.4	4.6
	-10	9.4	8.9	8.0	7.1	6.2	5.4	4.6
	-5	9.3	8.9	8.0	7.1	6.2	5.4	4.5
	0	8.4	8.0	7.1	6.2	5.4	4.6	3.8
	5	7.1	6.8	5.9	5.1	4.4	3.7	2.9
	10	5.9	5.6	4.8	4.0	3.3	2.7	2.0
8000	-40	8.9	8.5	7.6	6.7	5.9	5.1	4.2
	-30	9.0	8.6	7.6	6.7	5.9	5.1	4.3
	-20	9.0	8.6	7.6	6.7	5.9	5.1	4.3
	-10	8.9	8.5	7.6	6.7	5.9	5.1	4.3
	-5	8.7	8.3	7.4	6.5	5.7	4.9	4.1
	0	7.5	7.1	6.3	5.4	4.7	4.0	3.2
	5	6.3	6.0	5.1	4.4	3.7	3.0	2.3
	10	5.1	4.8	4.1	3.4	2.7	2.1	1.4
9000	-40	8.6	8.2	7.3	6.4	5.6	4.8	4.0
	-30	8.6	8.2	7.3	6.4	5.6	4.8	4.0
	-20	8.6	8.2	7.3	6.4	5.6	4.8	4.0
	-10	8.5	8.1	7.2	6.3	5.5	4.8	3.9
	-5	7.9	7.5	6.6	5.8	5.0	4.2	3.4
	0	6.7	6.3	5.5	4.7	4.0	3.3	2.6
	5	5.5	5.2	4.4	3.7	3.0	2.4	1.7
	10	4.4	4.1	3.4	2.7	2.1	1.5	0.9
10,000	-40	8.2	7.8	6.9	6.1	5.3	4.5	3.7
	-30	8.2	7.8	6.9	6.1	5.3	4.5	3.7
	-20	8.2	7.8	6.9	6.0	5.2	4.5	3.7
	-10	8.1	7.7	6.8	6.0	5.2	4.4	3.6
	-5	7.1	6.7	5.8	5.0	4.3	3.6	2.8
	0	5.9	5.5	4.7	4.0	3.3	2.6	1.9
	5	4.7	4.4	3.7	3.0	2.3	1.7	1.1
	10	3.7	3.4	2.7	2.1	1.5	0.9	0.3

TCG2_UP_2_7_02

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October 30, 2016

HJ1-29000-003-001

Page 5-174

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Wind Corrected Takeoff Climb Gradient [%]				
FLAPS UP and TO/APPR, V2, Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.1	0.0	0.0	0.0	0.0
0.5	0.5	0.5	0.5	0.4
1.0	1.0	1.0	1.0	1.0
1.4	1.5	1.5	1.6	1.6
1.9	2.0	2.0	2.1	2.2
2.3	2.5	2.6	2.6	2.7
2.7	3.0	3.1	3.2	3.3
3.2	3.5	3.6	3.7	3.9
3.6	4.0	4.1	4.3	4.4
4.1	4.5	4.7	4.8	5.0
4.5	5.0	5.2	5.4	5.6
5.0	5.5	5.7	5.9	6.2
5.4	6.0	6.2	6.5	6.7
5.8	6.5	6.8	7.0	7.3
6.3	7.0	7.3	7.6	7.9
6.7	7.5	7.8	8.1	8.5
7.2	8.0	8.3	8.7	9.0
7.6	8.5	8.8	9.2	9.6
8.1	9.0	9.4	9.8	10.2
8.5	9.5	9.9	10.3	10.8
8.9	10.0	10.4	10.9	11.3
9.4	10.5	10.9	11.4	11.9
9.8	11.0	11.5	11.9	12.5
10.3	11.5	12.0	12.5	13.1
10.7	12.0	12.5	13.0	13.6

CGWC_UP_2_03

NOTE Use this table when determining the ground reference flight path for obstacle clearance with Ice Protection On.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	9.8	9.4	8.5	7.6	6.9	6.1	5.3
	15	10.2	9.8	8.9	8.0	7.2	6.4	5.6
	25	10.3	9.9	8.9	8.0	7.2	6.5	5.7
	35	8.6	8.2	7.3	6.5	5.8	5.1	4.4
	40	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	45	6.5	6.2	5.4	4.7	4.1	3.5	2.9
	50	5.3	5.1	4.4	3.7	3.2	2.6	2.0
	55	4.2	4.0	3.3	2.8	2.3	1.8	1.2
Sea Level	-40	9.5	9.1	8.2	7.3	6.6	5.9	5.1
	15	9.9	9.5	8.5	7.7	6.9	6.2	5.4
	25	9.9	9.5	8.6	7.7	6.9	6.2	5.4
	35	7.8	7.4	6.6	5.8	5.2	4.5	3.8
	40	6.8	6.5	5.7	5.0	4.3	3.7	3.1
	45	5.8	5.5	4.8	4.1	3.5	3.0	2.3
	50	4.7	4.4	3.8	3.2	2.6	2.1	1.5
	55	3.6	3.4	2.8	2.2	1.7	1.3	0.7
1000	-40	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	15	9.5	9.1	8.2	7.4	6.6	5.9	5.1
	25	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	30	8.0	7.7	6.8	6.1	5.4	4.7	4.0
	35	7.0	6.7	5.9	5.2	4.5	3.9	3.2
	40	6.0	5.7	5.0	4.3	3.7	3.2	2.5
	45	5.1	4.8	4.1	3.5	2.9	2.4	1.8
	50	4.1	3.8	3.2	2.6	2.1	1.6	1.1
2000	-40	8.8	8.4	7.5	6.7	6.0	5.3	4.5
	10	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	20	9.2	8.8	7.9	7.1	6.3	5.6	4.8
	30	7.2	6.9	6.1	5.4	4.7	4.1	3.4
	35	6.2	5.9	5.2	4.5	3.9	3.3	2.7
	40	5.3	5.0	4.3	3.7	3.1	2.6	2.0
	45	4.4	4.1	3.5	2.9	2.4	1.9	1.3
	50	3.5	3.2	2.6	2.1	1.6	1.2	0.6

TCG3_UP_0-1_06

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October 30, 2016

HJ1-29000-003-001

Page 5-176

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	8.4	8.0	7.2	6.4	5.7	5.0	4.3
	10	8.8	8.4	7.5	6.7	6.0	5.3	4.5
	20	8.6	8.2	7.3	6.5	5.8	5.1	4.4
	25	7.4	7.1	6.3	5.6	4.9	4.3	3.6
	30	6.4	6.1	5.3	4.7	4.0	3.5	2.8
	35	5.5	5.2	4.5	3.9	3.3	2.7	2.1
	40	4.6	4.3	3.7	3.1	2.5	2.0	1.5
	45	3.7	3.5	2.9	2.4	1.9	1.4	0.9
4000	-40	8.1	7.7	6.9	6.1	5.4	4.8	4.0
	10	8.3	8.0	7.1	6.3	5.6	5.0	4.2
	20	7.7	7.3	6.5	5.8	5.1	4.5	3.8
	25	6.6	6.3	5.5	4.9	4.2	3.6	3.0
	30	5.6	5.3	4.6	4.0	3.4	2.9	2.3
	35	4.8	4.5	3.8	3.2	2.7	2.2	1.6
	40	3.9	3.7	3.1	2.5	2.0	1.5	1.0
	45	3.1	2.9	2.3	1.8	1.3	0.9	0.4
5000	-40	7.7	7.4	6.6	5.8	5.1	4.5	3.8
	5	7.9	7.6	6.7	6.0	5.3	4.6	3.9
	15	7.9	7.5	6.7	6.0	5.3	4.6	3.9
	25	5.9	5.5	4.8	4.2	3.6	3.0	2.4
	30	4.9	4.7	4.0	3.4	2.8	2.3	1.7
	35	4.1	3.8	3.2	2.7	2.1	1.7	1.1
	40	3.3	3.0	2.5	2.0	1.5	1.0	0.5
	45	2.5	2.3	1.8	1.3	0.9	0.4	0.0
6000	-40	7.4	7.1	6.3	5.5	4.9	4.2	3.6
	0	7.5	7.2	6.4	5.6	5.0	4.3	3.6
	10	7.5	7.2	6.3	5.6	4.9	4.3	3.6
	20	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	25	5.2	4.9	4.2	3.6	3.0	2.5	1.9
	30	4.3	4.0	3.4	2.8	2.3	1.8	1.3
	35	3.5	3.2	2.7	2.1	1.6	1.2	0.7
	40	2.7	2.5	1.9	1.5	1.0	0.6	0.1

TCG3_UP_0_3_06

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March 3, 2017

HJ1-29000-003-001

Page 5-177

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PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection Off, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	7.1	6.8	6.0	5.3	4.6	4.0	3.3
	0	7.1	6.8	6.0	5.3	4.6	4.0	3.3
	10	7.1	6.8	6.0	5.3	4.6	4.0	3.3
	15	6.4	6.1	5.3	4.6	4.0	3.4	2.8
	20	5.4	5.1	4.4	3.8	3.2	2.7	2.1
	25	4.5	4.2	3.6	3.0	2.5	2.0	1.4
	30	3.7	3.4	2.8	2.3	1.8	1.3	0.8
	35	2.9	2.7	2.1	1.6	1.2	0.7	0.2
8000	-40	6.8	6.5	5.7	5.0	4.3	3.8	3.1
	-10	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	0	6.8	6.4	5.7	5.0	4.3	3.7	3.1
	10	6.6	6.3	5.5	4.8	4.2	3.6	3.0
	15	5.7	5.4	4.7	4.0	3.4	2.9	2.3
	20	4.7	4.5	3.8	3.2	2.7	2.2	1.6
	25	3.9	3.6	3.0	2.5	2.0	1.5	1.0
	30	3.1	2.9	2.3	1.8	1.3	0.9	0.4
9000	-40	6.4	6.1	5.4	4.7	4.1	3.5	2.8
	-10	6.4	6.1	5.3	4.7	4.0	3.5	2.8
	0	6.4	6.1	5.4	4.7	4.0	3.5	2.8
	10	5.9	5.6	4.9	4.2	3.6	3.1	2.4
	15	5.0	4.7	4.0	3.4	2.9	2.3	1.8
	20	4.1	3.8	3.2	2.7	2.1	1.7	1.1
	25	3.3	3.0	2.5	2.0	1.5	1.0	0.5
	30	2.5	2.3	1.8	1.3	0.9	0.4	0.0
10,000	-40	6.0	5.7	5.0	4.4	3.7	3.2	2.5
	-5	6.1	5.7	5.0	4.4	3.8	3.2	2.6
	5	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	10	5.2	4.9	4.2	3.6	3.1	2.5	1.9
	15	4.3	4.1	3.4	2.9	2.3	1.8	1.3
	20	3.5	3.2	2.7	2.1	1.6	1.2	0.7
	25	2.7	2.5	1.9	1.5	1.0	0.6	0.1
	30	2.0	1.8	1.3	0.8	0.4	0.0	-0.4

TCG3_UP_0_7_07

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March 3, 2017

HJ1-29000-003-001

Page 5-178

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PERFORMANCE

Wind Corrected Enroute Climb Gradient [%]				
FLAPS UP, 140 [KIAS], Ice Protection Off				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.1	0.0	0.0	0.0	0.1
0.5	0.5	0.5	0.5	0.7
1.0	1.0	1.0	1.0	1.2
1.4	1.5	1.5	1.6	1.8
1.9	2.0	2.1	2.1	2.3
2.3	2.5	2.6	2.6	2.9
2.8	3.0	3.1	3.2	3.5
3.2	3.5	3.6	3.7	4.0
3.7	4.0	4.1	4.3	4.6
4.1	4.5	4.6	4.8	5.2
4.6	5.0	5.2	5.3	5.7
5.0	5.5	5.7	5.9	6.3
5.5	6.0	6.2	6.4	6.8
5.9	6.5	6.7	7.0	7.4
6.4	7.0	7.2	7.5	8.0
6.8	7.5	7.8	8.1	8.5
7.3	8.0	8.3	8.6	9.1
7.7	8.5	8.8	9.1	9.7
8.1	9.0	9.3	9.7	10.2
8.6	9.5	9.8	10.2	10.8
9.0	10.0	10.4	10.8	11.4
9.5	10.5	10.9	11.3	11.9
9.9	11.0	11.4	11.8	12.5
10.4	11.5	11.9	12.4	13.0
10.8	12.0	12.4	12.9	13.6

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October 30, 2016

HJ1-29000-003-001

Page 5-179

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PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
-1000	-40	9.9	9.5	8.5	7.7	6.9	6.2	5.4
	-30	9.9	9.5	8.6	7.7	6.9	6.2	5.4
	-20	10.0	9.6	8.6	7.8	7.0	6.3	5.5
	-10	10.1	9.7	8.7	7.9	7.1	6.3	5.5
	-5	10.1	9.7	8.8	7.9	7.1	6.4	5.6
	0	10.2	9.8	8.8	7.9	7.1	6.4	5.6
	5	10.2	9.8	8.8	8.0	7.2	6.4	5.6
	10	10.3	9.8	8.9	8.0	7.2	6.5	5.6
Sea Level	-40	9.5	9.1	8.2	7.3	6.6	5.9	5.1
	-30	9.6	9.2	8.2	7.4	6.6	5.9	5.1
	-20	9.7	9.3	8.3	7.5	6.7	6.0	5.2
	-10	9.8	9.4	8.4	7.6	6.8	6.1	5.3
	-5	9.8	9.4	8.5	7.6	6.8	6.1	5.3
	0	9.8	9.4	8.5	7.6	6.9	6.1	5.3
	5	9.9	9.5	8.5	7.7	6.9	6.2	5.4
	10	9.9	9.5	8.6	7.7	6.9	6.2	5.4
1000	-40	9.2	8.8	7.9	7.0	6.3	5.6	4.8
	-30	9.2	8.8	7.9	7.1	6.4	5.7	4.9
	-20	9.3	8.9	8.0	7.2	6.4	5.7	5.0
	-10	9.4	9.0	8.1	7.3	6.5	5.8	5.0
	-5	9.4	9.1	8.1	7.3	6.5	5.8	5.0
	0	9.5	9.1	8.2	7.3	6.6	5.9	5.1
	5	9.5	9.1	8.2	7.4	6.6	5.9	5.1
	10	9.0	8.6	7.7	6.9	6.2	5.5	4.7
2000	-40	8.8	8.4	7.6	6.7	6.0	5.3	4.6
	-30	8.9	8.5	7.6	6.8	6.1	5.4	4.6
	-20	9.0	8.6	7.7	6.9	6.2	5.5	4.7
	-10	9.1	8.7	7.8	7.0	6.2	5.5	4.8
	-5	9.1	8.7	7.8	7.0	6.2	5.6	4.8
	0	9.1	8.7	7.8	7.0	6.3	5.6	4.8
	5	9.2	8.8	7.9	7.0	6.3	5.6	4.8
	10	8.1	7.7	6.9	6.1	5.4	4.8	4.0

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October 30, 2016

HJ1-29000-003-001

Page 5-180

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
3000	-40	8.5	8.1	7.3	6.5	5.7	5.1	4.3
	-30	8.6	8.2	7.3	6.5	5.8	5.1	4.4
	-20	8.6	8.3	7.4	6.6	5.9	5.2	4.5
	-10	8.7	8.3	7.5	6.7	5.9	5.3	4.5
	-5	8.7	8.4	7.5	6.7	6.0	5.3	4.5
	0	8.8	8.4	7.5	6.7	6.0	5.3	4.5
	5	8.5	8.1	7.3	6.5	5.8	5.1	4.4
	10	7.2	6.8	6.0	5.3	4.7	4.0	3.4
4000	-40	8.2	7.8	7.0	6.2	5.5	4.8	4.1
	-30	8.2	7.9	7.0	6.2	5.5	4.9	4.2
	-20	8.3	7.9	7.1	6.3	5.6	4.9	4.2
	-10	8.4	8.0	7.1	6.4	5.6	5.0	4.2
	-5	8.4	8.0	7.1	6.4	5.7	5.0	4.3
	0	8.4	8.0	7.2	6.4	5.7	5.0	4.3
	5	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	10	6.3	5.9	5.2	4.5	3.9	3.3	2.7
5000	-40	7.8	7.4	6.6	5.9	5.2	4.5	3.8
	-30	7.9	7.5	6.7	5.9	5.2	4.6	3.9
	-20	7.9	7.6	6.8	6.0	5.3	4.7	3.9
	-10	8.0	7.6	6.8	6.0	5.3	4.7	4.0
	-5	8.0	7.6	6.8	6.0	5.3	4.7	4.0
	0	8.0	7.6	6.8	6.0	5.3	4.7	4.0
	5	6.7	6.3	5.6	4.9	4.2	3.7	3.0
	10	5.4	5.1	4.4	3.8	3.2	2.7	2.1
6000	-40	7.5	7.1	6.3	5.6	4.9	4.3	3.6
	-30	7.5	7.2	6.4	5.6	5.0	4.3	3.6
	-20	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	-10	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	-5	7.6	7.2	6.4	5.7	5.0	4.4	3.7
	0	7.1	6.7	6.0	5.2	4.6	4.0	3.3
	5	5.8	5.5	4.8	4.1	3.5	3.0	2.4
	10	4.6	4.3	3.7	3.1	2.6	2.1	1.5

TCG3_UP_2_3_02

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October 30, 2016

HJ1-29000-003-001

Page 5-181

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HA-420 AFM

PERFORMANCE

Net Takeoff Climb Gradient, Final Segment [%]								
Zero Slope, No Wind								
Flaps UP, Gear UP, Ice Protection On, 140 KIAS								
F. ALT [ft]	Temp. [°C]	Takeoff Weight [lb]						
		7800	8000	8500	9000	9500	10000	10600
7000	-40	7.1	6.8	6.0	5.3	4.6	4.0	3.4
	-30	7.2	6.9	6.1	5.4	4.7	4.1	3.4
	-20	7.2	6.9	6.1	5.4	4.7	4.1	3.4
	-10	7.2	6.9	6.1	5.3	4.7	4.1	3.4
	-5	7.2	6.8	6.0	5.3	4.7	4.1	3.4
	0	6.2	5.9	5.2	4.5	3.9	3.3	2.7
	5	5.0	4.7	4.0	3.4	2.9	2.3	1.8
	10	3.9	3.6	3.0	2.5	2.0	1.5	0.9
8000	-40	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	-30	6.9	6.5	5.8	5.1	4.4	3.8	3.2
	-20	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	-10	6.8	6.5	5.7	5.0	4.4	3.8	3.1
	-5	6.6	6.3	5.5	4.8	4.2	3.6	2.9
	0	5.4	5.1	4.4	3.8	3.2	2.7	2.1
	5	4.2	4.0	3.3	2.8	2.3	1.8	1.2
	10	3.2	2.9	2.4	1.8	1.4	0.9	0.4
9000	-40	6.5	6.2	5.4	4.7	4.1	3.5	2.9
	-30	6.5	6.2	5.4	4.8	4.1	3.5	2.9
	-20	6.5	6.1	5.4	4.7	4.1	3.5	2.9
	-10	6.4	6.1	5.4	4.7	4.1	3.5	2.8
	-5	5.8	5.5	4.8	4.1	3.5	3.0	2.3
	0	4.6	4.3	3.7	3.1	2.6	2.1	1.5
	5	3.5	3.3	2.7	2.2	1.7	1.2	0.7
	10	2.5	2.3	1.7	1.3	0.8	0.4	-0.1
10,000	-40	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	-30	6.1	5.8	5.1	4.4	3.8	3.2	2.6
	-20	6.1	5.8	5.0	4.4	3.8	3.2	2.6
	-10	6.1	5.8	5.0	4.4	3.8	3.2	2.6
	-5	5.0	4.7	4.0	3.4	2.9	2.4	1.8
	0	3.9	3.6	3.0	2.5	1.9	1.5	0.9
	5	2.8	2.6	2.0	1.5	1.1	0.6	0.2
	10	1.8	1.6	1.2	0.7	0.3	-0.1	-0.5

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October 30, 2016

HJ1-29000-003-001

Page 5-182

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PERFORMANCE

Wind Corrected Enroute Climb Gradient [%]				
FLAPS UP, 140 [KIAS], Ice Protection On				
Tailwind		Headwind		
10	◀ REF [0] ▶	10	20	30
0.1	0.0	0.0	0.0	0.0
0.5	0.5	0.5	0.5	0.5
1.0	1.0	1.0	1.0	1.0
1.4	1.5	1.5	1.6	1.6
1.9	2.0	2.1	2.1	2.2
2.3	2.5	2.6	2.6	2.7
2.8	3.0	3.1	3.2	3.3
3.2	3.5	3.6	3.7	3.9
3.6	4.0	4.1	4.3	4.4
4.1	4.5	4.7	4.8	5.0
4.5	5.0	5.2	5.4	5.6
5.0	5.5	5.7	5.9	6.1
5.4	6.0	6.2	6.5	6.7
5.9	6.5	6.7	7.0	7.3
6.3	7.0	7.3	7.5	7.8
6.8	7.5	7.8	8.1	8.4
7.2	8.0	8.3	8.6	9.0
7.7	8.5	8.8	9.2	9.5
8.1	9.0	9.3	9.7	10.1
8.6	9.5	9.9	10.3	10.7
9.0	10.0	10.4	10.8	11.2
9.4	10.5	10.9	11.3	11.8
9.9	11.0	11.4	11.9	12.4
10.3	11.5	11.9	12.4	12.9
10.8	12.0	12.5	13.0	13.5

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March 3, 2017

HJ1-29000-003-001

Page 5-183

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HA-420 AFM

PERFORMANCE

OBSTACLE CLEARANCE

Obstacle Clearance Procedure

1. Prior to determining obstacle clearance information, determine the takeoff data for the desired weight. (Configuration, TOFL, airport altitude, temperature, wind)
2. Determine the minimum climb gradient for departure. (Min Grad = Obstacle Height / (Rwy Length – TOFL + Distance to Obstacle from DER).
3. Determine the height of the obstacle (Clearance Height MSL).
4. Convert the Clearance Height to pressure altitude using the Height and Temperature in the Pressure Altitude table.

NOTE Temperature decreases 1°C per 500 ft altitude.

5. Determine the climb gradient at the pressure altitude of the obstacle.
6. Correct the climb gradient for wind.
7. Compare the corrected climb gradient to the minimum climb gradient.
 - i) If the corrected climb gradient is equal to or greater than the minimum climb gradient, the aircraft meets the performance requirement.
 - ii) If the corrected climb gradient is less than the minimum climb gradient, the aircraft does NOT meet the performance requirement and configuration must be changed to meet the requirement. (Less weight, alternate flap setting, alternate runway).
8. Determine indicated altitude at Clearance Height.
 - i) Convert Airport Elevation to pressure altitude using the Pressure Altitude table. Do not use altimeter reading for this.
 - ii) Subtract the airport pressure altitude (B) from the pressure altitude of the obstacle (M).
 - iii) Add the result to the airport elevation (A) to get indicated altitude at the obstacle height. This assumes the correct baro-setting on the altimeter and represents the minimum level-off height

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PERFORMANCE

The example worksheet shows a situation that meets the performance requirement after an engine failure.

Obstacle Clearance Worksheet

Step 1	A	Airport Elevation	
	B	Airport Pressure Altitude from Table	
	C	Airport Temperature	
	D	Airport Winds	
	E	Available Runway Length (ft)	
	F	Obstacle Height Above Runway (ft)	
	G	Obstacle Distance from Runway (ft)	
	H	Takeoff Flap and Ice Protection Setting	
	I	Aircraft Weight	
	J	Corrected TOFL	
Step 2	K	Min Grad = $[F / (E - J + G)] * 100$	
Step 3	L	Clearance Height MSL = $(A + F)$	
Step 4	M	CH Pressure Altitude from Table (L, C)	
Step 5	N	Uncorrected Climb Gradient	
Step 6	O	Corrected Climb Gradient (N, D)	
Step 7		Compare K and O; $O \geq K$	
Step 8		Altimeter Reading at CH = $(A + M - B)$	

OCW_02

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PERFORMANCE

Example Worksheet

Step 1	A	Airport Elevation	3000
	B	Airport Pressure Altitude from Table	3346
	C	Airport Temperature	-20
	D	Airport Winds	10 kt HW
	E	Available Runway Length (ft)	7000
	F	Obstacle Height Above Runway (ft)	1000
	G	Obstacle Distance from Runway (ft)	4nm x 6076
	H	Takeoff Flap and Ice Protection Setting	TO/APPR, Ice Protection On
	I	Aircraft Weight	10600
	J	Wind Corrected TOFL	3910
Step 2	K	Min Grad = $[F / (E - J + G)] * 100$	3.7
Step 3	L	Clearance Height MSL = (A + F)	4000
Step 4	M	CH Pressure Altitude from Table (L, C)	4428
Step 5	N	Uncorrected Climb Gradient	3.9
Step 6	O	Corrected Climb Gradient (N, D)	4.0
Step 7		Compare K and O; $O \geq K$	$4.0 > 3.7$
Step 8		Altimeter Reading at CH = (A + M - B)	4082

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Airport Pressure Altitude Conversion

To convert from Airport Elevation to Pressure Altitude, find, in bold face numbers in the first column, the number representing the airport elevation to be converted. The equivalent pressure altitude is read in the adjacent column headed temperature.

For example, for an airport with an elevation of 900 feet and a temperature of 0 °C, read 945 feet pressure altitude from the table.

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Ht [ft MSL]	Pressure Altitude [feet] Ambient Temperature °C							
	-40	-20	0	10	20	30	40	55
0	0	0	0	0	0	0	0	0
100	124	114	106	102	98	95	92	88
200	247	228	211	203	196	190	184	175
300	370	341	316	305	295	285	276	263
400	494	455	421	406	392	379	367	350
500	617	568	526	507	490	474	459	438
600	739	681	631	609	588	568	550	525
700	862	794	736	709	685	663	642	612
800	984	907	840	810	783	757	733	699
900	1107	1019	945	911	880	851	824	786
1000	1229	1132	1049	1011	977	945	915	873
1100	1351	1244	1153	1112	1074	1038	1006	959
1200	1472	1356	1257	1212	1171	1132	1097	1046
1300	1594	1468	1361	1312	1268	1226	1187	1133
1400	1715	1580	1464	1412	1365	1319	1278	1219
1500	1836	1691	1568	1511	1461	1412	1368	1305
1600	1957	1803	1671	1611	1557	1505	1458	1391
1700	2078	1914	1774	1710	1654	1598	1548	1477
1800	2199	2025	1877	1810	1750	1691	1638	1563
1900	2319	2136	1980	1909	1846	1784	1728	1649
2000	2439	2247	2083	2008	1941	1876	1818	1735
2100	2559	2358	2186	2107	2037	1969	1908	1820
2200	2679	2468	2288	2205	2133	2061	1997	1906
2300	2799	2578	2390	2304	2228	2153	2087	1991
2400	2918	2689	2493	2402	2323	2245	2176	2077
2500	3038	2799	2595	2500	2419	2337	2265	2162
2600	3157	2908	2696	2599	2514	2429	2354	2247
2700	3276	3018	2798	2697	2609	2521	2443	2332
2800	3395	3128	2900	2795	2703	2612	2532	2417
2900	3513	3237	3001	2893	2798	2704	2621	2502

FPPA_0_03

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October 30, 2016

HJ1-29000-003-001

Page 5-188

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Ht [ft MSL]	Pressure Altitude [feet] Ambient Temperature °C							
	-40	-20	0	10	20	30	40	55
3000	3632	3346	3103	2991	2892	2795	2709	2586
3100	3750	3455	3204	3088	2987	2887	2798	2671
3200	3868	3564	3305	3186	3081	2978	2886	2755
3300	3986	3672	3406	3283	3175	3069	2974	2839
3400	4103	3781	3506	3381	3269	3160	3062	2924
3500	4221	3889	3607	3478	3363	3251	3150	3008
3600	4338	3997	3707	3575	3457	3342	3238	3092
3700	4455	4105	3807	3672	3550	3432	3326	3176
3800	4572	4213	3908	3768	3644	3523	3414	3259
3900	4689	4321	4007	3865	3737	3613	3501	3343
4000	4805	4428	4107	3962	3830	3704	3589	3427
4100	4921	4536	4207	4058	3923	3794	3676	3510
4200	5038	4643	4306	4154	4016	3884	3763	3594
4300	5154	4750	4406	4250	4109	3974	3850	3677
4400	5269	4857	4505	4346	4201	4064	3937	3760
4500	5385	4963	4604	4442	4294	4153	4024	3843
4600	5500	5070	4703	4538	4386	4243	4110	3926
4700	5615	5176	4802	4633	4479	4332	4197	4009
4800	5730	5282	4900	4729	4571	4422	4283	4091
4900	5845	5388	4999	4824	4663	4511	4370	4174
5000	5960	5494	5097	4919	4754	4600	4456	4256
5100	6074	5599	5195	5014	4846	4689	4542	4339
5200	6188	5705	5293	5109	4938	4778	4628	4421
5300	6302	5810	5391	5204	5029	4866	4713	4503
5400	6416	5915	5489	5298	5120	4955	4799	4585
5500	6529	6020	5586	5393	5211	5043	4885	4667
5600	6643	6125	5683	5487	5302	5132	4970	4749
5700	6756	6229	5781	5582	5393	5220	5055	4831
5800	6869	6334	5878	5676	5484	5308	5141	4912
5900	6982	6438	5975	5770	5575	5396	5226	4994

FPPA_3_03

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October 30, 2016

HJ1-29000-003-001

Page 5-189

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Ht [ft MSL]	Pressure Altitude [feet] Ambient Temperature °C							
	-40	-20	0	10	20	30	40	55
6000	7094	6542	6071	5863	5665	5484	5311	5075
6100	7207	6646	6168	5957	5755	5572	5396	5156
6200	7319	6750	6264	6051	5846	5659	5480	5238
6300	7431	6853	6361	6144	5936	5747	5565	5319
6400	7543	6956	6457	6237	6026	5834	5649	5400
6500	7654	7060	6553	6330	6115	5922	5734	5480
6600	7766	7163	6649	6424	6205	6009	5818	5561
6700	7877	7265	6744	6516	6294	6096	5902	5642
6800	7988	7368	6840	6609	6384	6183	5986	5722
6900	8099	7471	6935	6702	6473	6269	6070	5803
7000	8209	7573	7030	6794	6562	6356	6153	5883
7100	8319	7675	7125	6887	6651	6443	6237	5963
7200	8430	7777	7220	6979	6740	6529	6321	6043
7300	8540	7879	7315	7071	6829	6615	6404	6123
7400	8649	7980	7410	7163	6917	6702	6487	6203
7500	8759	8082	7504	7254	7005	6788	6570	6283
7600	8869	8183	7599	7346	7094	6874	6653	6362
7700	8979	8285	7693	7438	7183	6959	6737	6442
7800	9089	8387	7788	7529	7271	7045	6820	6521
7900	9199	8489	7883	7620	7360	7131	6903	6600
8000	9309	8590	7977	7711	7448	7216	6986	6680
8100	9419	8692	8072	7802	7536	7301	7069	6759
8200	9529	8793	8166	7893	7625	7387	7152	6838
8300	9638	8894	8260	7984	7713	7472	7235	6916
8400	9747	8995	8354	8074	7801	7557	7317	6995
8500	9856	9096	8448	8165	7888	7641	7400	7074
8600	9965	9197	8542	8255	7976	7726	7482	7152
8700	10074	9297	8635	8345	8064	7811	7565	7231
8800	10183	9398	8729	8435	8151	7895	7647	7309
8900	10291	9498	8822	8525	8238	7979	7729	7387

FPPA_6_03

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October 30, 2016

HJ1-29000-003-001

Page 5-190

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Ht [ft MSL]	Pressure Altitude [feet] Ambient Temperature °C							
	-40	-20	0	10	20	30	40	55
9000	10399	9598	8915	8615	8325	8064	7811	7465
9100	10507	9698	9008	8704	8412	8148	7893	7543
9200	10615	9798	9101	8794	8499	8232	7974	7621
9300	10723	9897	9194	8883	8586	8316	8056	7699
9400	10830	9997	9286	8972	8673	8399	8137	7776
9500	10938	10096	9379	9061	8759	8483	8219	7854
9600	11045	10195	9471	9150	8846	8566	8300	7931
9700	11152	10294	9563	9239	8932	8650	8381	8008
9800	11259	10393	9655	9327	9018	8733	8462	8086
9900	11365	10492	9747	9416	9104	8816	8543	8163
10000	11472	10590	9839	9504	9190	8899	8624	8240
10100	11578	10689	9931	9592	9276	8982	8705	8316
10200	11684	10787	10022	9681	9362	9064	8786	8393
10300	11790	10885	10113	9768	9447	9147	8866	8470
10400	11896	10983	10205	9856	9533	9229	8947	8546
10500	12002	11081	10296	9944	9618	9312	9027	8623
10600	12107	11178	10387	10031	9703	9394	9107	8699
10700	12212	11276	10478	10119	9789	9476	9187	8775
10800	12317	11373	10568	10206	9874	9558	9267	8851
10900	12422	11470	10659	10293	9958	9640	9347	8927
11000	12527	11567	10749	10380	10043	9722	9427	9003
11100	12631	11664	10839	10467	10128	9803	9506	9079
11200	12736	11760	10929	10553	10212	9885	9586	9154
11300	12840	11857	11019	10640	10296	9966	9665	9230
11400	12944	11953	11109	10726	10381	10047	9745	9305
11500	13048	12049	11199	10812	10465	10128	9824	9380
11600	13151	12145	11289	10898	10549	10209	9903	9455
11700	13255	12241	11378	10984	10633	10290	9982	9531
11800	13358	12337	11467	11070	10716	10371	10061	9605
11900	13461	12433	11556	11156	10800	10451	10139	9680
12000	13564	12528	11645	11241	10883	10532	10218	9755

FPPA_9_03

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October 30, 2016

HJ1-29000-003-001

Page 5-191

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
-1000	-40	10.4	9.5	8.7	7.9	7.2	6.5	5.8	5.2
	15	10.7	9.8	8.9	8.2	7.4	6.7	6.0	5.3
	20	10.7	9.8	9.0	8.2	7.4	6.7	6.0	5.4
	25	10.7	9.9	9.0	8.2	7.5	6.7	6.0	5.4
	30	10.1	9.2	8.4	7.6	6.9	6.2	5.5	4.9
	35	8.9	8.1	7.3	6.6	5.9	5.2	4.6	4.0
	40	7.7	7.0	6.2	5.5	4.8	4.2	3.6	3.0
	45	6.5	5.8	5.0	4.4	3.7	3.2	2.6	2.1
	50	5.2	4.5	3.9	3.3	2.7	2.2	1.7	1.2
	55	4.0	3.4	2.8	2.2	1.7	1.2	0.8	0.3
Sea Level	-40	10.3	9.4	8.6	7.9	7.1	6.4	5.7	5.1
	15	10.6	9.7	8.9	8.1	7.4	6.6	5.9	5.3
	20	10.6	9.8	8.9	8.1	7.4	6.7	6.0	5.3
	25	10.7	9.8	8.9	8.2	7.4	6.7	6.0	5.3
	30	9.6	8.7	7.9	7.2	6.4	5.7	5.1	4.4
	35	8.4	7.6	6.8	6.1	5.4	4.7	4.1	3.5
	40	7.2	6.3	5.6	4.9	4.3	3.7	3.1	2.6
	45	5.9	5.2	4.5	3.9	3.3	2.7	2.2	1.7
	50	4.6	4.0	3.3	2.8	2.2	1.7	1.2	0.8
	55	3.4	2.8	2.3	1.7	1.2	0.8	0.3	-0.1
1000	-40	9.9	9.1	8.3	7.5	6.8	6.1	5.4	4.8
	10	10.2	9.4	8.5	7.8	7.0	6.3	5.6	5.0
	15	10.2	9.4	8.6	7.8	7.0	6.3	5.6	5.0
	20	10.3	9.4	8.6	7.8	7.1	6.3	5.7	5.0
	25	9.9	9.0	8.2	7.4	6.7	6.0	5.3	4.7
	30	8.8	7.9	7.2	6.4	5.7	5.0	4.4	3.8
	35	7.6	6.8	6.1	5.4	4.7	4.1	3.5	2.9
	40	6.4	5.7	4.9	4.3	3.7	3.1	2.5	2.0
	45	5.2	4.5	3.8	3.2	2.6	2.1	1.6	1.1
	50	3.9	3.3	2.7	2.2	1.6	1.2	0.7	0.3

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October 30, 2016

HJ1-29000-003-001

Page 5-192

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
2000	-40	9.6	8.7	8.0	7.2	6.5	5.8	5.1	4.5
	10	9.9	9.0	8.2	7.4	6.7	6.0	5.3	4.7
	15	9.9	9.1	8.2	7.5	6.7	6.0	5.4	4.7
	20	9.9	9.1	8.3	7.5	6.8	6.0	5.4	4.7
	25	9.1	8.2	7.5	6.7	6.0	5.3	4.7	4.1
	30	8.0	7.2	6.4	5.7	5.0	4.4	3.8	3.2
	35	6.9	6.1	5.4	4.7	4.0	3.4	2.9	2.4
	40	5.7	4.9	4.2	3.6	3.0	2.5	2.0	1.5
	45	4.4	3.8	3.2	2.6	2.1	1.6	1.1	0.6
	50	3.2	2.6	2.1	1.6	1.1	0.6	0.2	-0.2
3000	-40	9.2	8.4	7.6	6.9	6.2	5.5	4.8	4.2
	5	9.5	8.7	7.9	7.1	6.4	5.7	5.0	4.4
	10	9.5	8.7	7.9	7.1	6.4	5.7	5.0	4.4
	15	9.6	8.7	7.9	7.1	6.4	5.7	5.1	4.4
	20	9.4	8.5	7.7	7.0	6.2	5.5	4.9	4.3
	25	8.2	7.4	6.7	5.9	5.2	4.6	4.0	3.4
	30	7.2	6.3	5.6	4.9	4.3	3.7	3.1	2.6
	35	6.1	5.3	4.6	4.0	3.4	2.8	2.3	1.8
	40	4.9	4.2	3.6	3.0	2.4	1.9	1.4	1.0
	45	3.7	3.1	2.5	2.0	1.5	1.0	0.6	0.1
4000	-40	8.9	8.1	7.3	6.5	5.8	5.2	4.5	3.9
	5	9.1	8.3	7.5	6.8	6.0	5.3	4.7	4.1
	10	9.1	8.3	7.5	6.8	6.0	5.4	4.7	4.1
	15	9.2	8.3	7.5	6.8	6.0	5.4	4.7	4.1
	20	8.5	7.7	6.9	6.2	5.5	4.8	4.2	3.6
	25	7.5	6.6	5.9	5.2	4.5	3.9	3.3	2.8
	30	6.3	5.5	4.8	4.2	3.6	3.0	2.5	1.9
	35	5.2	4.5	3.8	3.2	2.7	2.1	1.6	1.2
	40	4.1	3.5	2.9	2.3	1.8	1.3	0.9	0.4
	45	3.1	2.5	1.9	1.4	1.0	0.5	0.1	-0.3

FPCG_TO_0_2_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-193

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
5000	-40	8.5	7.7	7.0	6.2	5.5	4.9	4.2	3.7
	5	8.7	7.9	7.2	6.4	5.7	5.0	4.4	3.8
	10	8.7	7.9	7.2	6.4	5.7	5.0	4.4	3.8
	15	8.7	7.9	7.2	6.4	5.7	5.0	4.4	3.8
	20	7.7	6.9	6.1	5.4	4.7	4.1	3.5	3.0
	25	6.5	5.8	5.1	4.4	3.8	3.2	2.7	2.1
	30	5.5	4.7	4.1	3.4	2.9	2.3	1.8	1.3
	35	4.4	3.7	3.1	2.6	2.0	1.5	1.1	0.6
	40	3.4	2.8	2.2	1.7	1.2	0.7	0.3	-0.1
	45	2.4	1.8	1.3	0.8	0.4	0.0	-0.4	-0.8
6000	-40	8.2	7.4	6.6	5.9	5.2	4.6	4.0	3.4
	0	8.4	7.6	6.8	6.0	5.3	4.7	4.1	3.5
	5	8.3	7.5	6.8	6.0	5.3	4.7	4.1	3.5
	10	8.3	7.5	6.7	6.0	5.3	4.7	4.0	3.5
	15	7.9	7.1	6.3	5.6	4.9	4.3	3.7	3.2
	20	6.8	6.1	5.3	4.6	4.0	3.4	2.9	2.3
	25	5.7	5.0	4.3	3.7	3.1	2.5	2.0	1.5
	30	4.6	4.0	3.3	2.8	2.2	1.7	1.2	0.8
	35	3.6	3.0	2.4	1.9	1.4	0.9	0.5	0.1
	40	2.6	2.0	1.5	1.0	0.6	0.2	-0.3	-0.6
7000	-40	7.9	7.1	6.3	5.6	4.9	4.3	3.7	3.1
	0	8.0	7.2	6.4	5.7	5.0	4.3	3.8	3.2
	5	7.9	7.1	6.3	5.6	5.0	4.3	3.7	3.2
	10	7.9	7.1	6.3	5.6	4.9	4.3	3.7	3.1
	15	7.1	6.3	5.6	4.9	4.2	3.6	3.0	2.5
	20	6.0	5.3	4.6	3.9	3.3	2.8	2.2	1.7
	25	4.9	4.2	3.6	3.0	2.4	1.9	1.4	1.0
	30	3.8	3.2	2.6	2.1	1.6	1.1	0.6	0.2
	35	2.9	2.3	1.8	1.3	0.8	0.4	-0.1	-0.5
	40	1.9	1.4	0.9	0.4	0.0	-0.4	-0.8	-1.2

FPCG_TO_0_5_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-194

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
8000	-40	7.5	6.7	6.0	5.3	4.6	4.0	3.4	2.9
	-5	7.6	6.8	6.0	5.3	4.6	4.0	3.4	2.9
	0	7.5	6.7	6.0	5.3	4.6	4.0	3.4	2.9
	5	7.5	6.7	6.0	5.3	4.6	4.0	3.4	2.8
	10	7.4	6.5	5.8	5.1	4.4	3.8	3.2	2.7
	15	6.2	5.5	4.8	4.1	3.5	2.9	2.4	1.9
	20	5.2	4.5	3.8	3.2	2.6	2.1	1.6	1.1
	25	4.1	3.5	2.9	2.3	1.8	1.3	0.8	0.4
	30	3.1	2.5	2.0	1.5	1.0	0.5	0.1	-0.3
	35	2.1	1.6	1.1	0.7	0.2	-0.2	-0.6	-1.0
9000	-40	7.2	6.4	5.7	5.0	4.3	3.7	3.2	2.6
	-5	7.2	6.4	5.6	4.9	4.3	3.7	3.1	2.6
	0	7.2	6.3	5.6	4.9	4.3	3.7	3.1	2.6
	5	7.1	6.3	5.6	4.9	4.3	3.7	3.1	2.5
	10	6.4	5.7	5.0	4.3	3.7	3.1	2.6	2.1
	15	5.4	4.7	4.0	3.4	2.8	2.3	1.8	1.3
	20	4.4	3.7	3.1	2.5	2.0	1.5	1.0	0.6
	25	3.3	2.7	2.2	1.7	1.2	0.7	0.3	-0.1
	30	2.4	1.8	1.3	0.8	0.4	0.0	-0.4	-0.8
	35	1.4	0.9	0.5	0.0	-0.4	-0.7	-1.1	-1.5
10000	-40	6.9	6.1	5.4	4.7	4.1	3.5	2.9	2.4
	-5	6.8	6.0	5.3	4.6	4.0	3.4	2.8	2.3
	0	6.8	6.0	5.3	4.6	4.0	3.4	2.8	2.3
	5	6.7	5.9	5.2	4.5	3.9	3.3	2.8	2.2
	10	5.5	4.8	4.1	3.5	2.9	2.4	1.9	1.4
	15	4.6	3.9	3.3	2.7	2.1	1.6	1.2	0.7
	20	3.6	3.0	2.4	1.8	1.4	0.9	0.4	0.0
	25	2.6	2.0	1.5	1.0	0.6	0.1	-0.3	-0.7
	30	1.7	1.2	0.7	0.2	-0.2	-0.6	-0.9	-1.3
	35	0.7	0.3	-0.1	-0.5	-0.9	-1.3	-1.6	-1.9

FPCG_TO_0_8_04

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October 30, 2016

HJ1-29000-003-001

Page 5-195

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
11000	-40	6.4	5.7	5.0	4.3	3.7	3.1	2.6	2.1
	-10	6.4	5.6	4.9	4.3	3.6	3.1	2.5	2.0
	-5	6.4	5.6	4.9	4.3	3.6	3.1	2.5	2.0
	0	6.4	5.6	4.9	4.2	3.6	3.1	2.5	2.0
	5	5.9	5.2	4.5	3.8	3.2	2.7	2.2	1.7
	10	4.8	4.1	3.5	2.9	2.3	1.8	1.3	0.9
	15	3.8	3.2	2.6	2.0	1.5	1.1	0.6	0.2
	20	2.8	2.3	1.7	1.2	0.8	0.3	-0.1	-0.5
	25	1.9	1.4	0.9	0.4	0.0	-0.4	-0.8	-1.1
	30	1.0	0.5	0.1	-0.3	-0.7	-1.1	-1.4	-1.8
12000	-40	6.1	5.3	4.6	4.0	3.4	2.8	2.3	1.8
	-10	6.0	5.3	4.6	3.9	3.3	2.8	2.3	1.8
	-5	6.1	5.3	4.6	4.0	3.4	2.8	2.3	1.8
	0	6.0	5.3	4.6	3.9	3.3	2.8	2.2	1.7
	5	5.1	4.4	3.7	3.1	2.6	2.1	1.6	1.1
	10	4.0	3.4	2.8	2.2	1.7	1.2	0.8	0.3
	15	3.0	2.5	1.9	1.4	0.9	0.5	0.1	-0.3
	20	2.1	1.6	1.1	0.6	0.2	-0.2	-0.6	-1.0
	25	1.2	0.7	0.3	-0.1	-0.5	-0.9	-1.3	-1.6
	30	0.4	0.0	-0.5	-0.8	-1.2	-1.5	-1.9	-2.2

FPCG_TO_0_11_04

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
-1000	-40	10.4	9.5	8.7	8.0	7.2	6.5	5.8	5.2
	-30	10.4	9.6	8.7	8.0	7.2	6.5	5.8	5.2
	-25	10.4	9.6	8.8	8.0	7.2	6.5	5.8	5.2
	-20	10.4	9.6	8.8	8.0	7.3	6.5	5.9	5.2
	-15	10.5	9.6	8.8	8.0	7.3	6.6	5.9	5.2
	-10	10.5	9.6	8.8	8.0	7.3	6.6	5.9	5.2
	-5	10.6	9.7	8.9	8.1	7.3	6.6	5.9	5.3
	0	10.6	9.7	8.9	8.1	7.4	6.7	6.0	5.3
	5	10.6	9.8	8.9	8.2	7.4	6.7	6.0	5.3
	10	10.7	9.8	9.0	8.2	7.4	6.7	6.0	5.4
Sea Level	-40	10.3	9.5	8.7	7.9	7.2	6.4	5.8	5.1
	-30	10.3	9.5	8.7	7.9	7.2	6.5	5.8	5.1
	-25	10.4	9.5	8.7	7.9	7.2	6.5	5.8	5.2
	-20	10.4	9.5	8.7	8.0	7.2	6.5	5.8	5.2
	-15	10.4	9.6	8.7	8.0	7.2	6.5	5.8	5.2
	-10	10.5	9.6	8.8	8.0	7.3	6.5	5.9	5.2
	-5	10.5	9.7	8.8	8.0	7.3	6.6	5.9	5.2
	0	10.6	9.7	8.9	8.1	7.3	6.6	5.9	5.3
	5	10.6	9.7	8.9	8.1	7.4	6.6	5.9	5.3
	10	10.6	9.7	8.9	8.1	7.4	6.7	6.0	5.3
1000	-40	9.9	9.1	8.3	7.5	6.8	6.1	5.4	4.8
	-30	10.0	9.1	8.3	7.6	6.8	6.1	5.5	4.8
	-25	10.0	9.2	8.4	7.6	6.9	6.2	5.5	4.9
	-20	10.0	9.2	8.4	7.6	6.9	6.2	5.5	4.9
	-15	10.1	9.2	8.4	7.6	6.9	6.2	5.5	4.9
	-10	10.1	9.3	8.5	7.7	6.9	6.2	5.6	4.9
	-5	10.2	9.3	8.5	7.7	7.0	6.3	5.6	5.0
	0	10.2	9.3	8.5	7.7	7.0	6.3	5.6	5.0
	5	10.2	9.4	8.5	7.8	7.0	6.3	5.6	5.0
	10	9.8	8.9	8.1	7.3	6.6	5.9	5.2	4.6

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October 30, 2016

HJ1-29000-003-001

Page 5-197

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
2000	-40	9.6	8.8	8.0	7.2	6.5	5.8	5.2	4.5
	-30	9.7	8.8	8.0	7.3	6.6	5.8	5.2	4.6
	-25	9.7	8.9	8.1	7.3	6.6	5.9	5.2	4.6
	-20	9.7	8.9	8.1	7.3	6.6	5.9	5.2	4.6
	-15	9.8	8.9	8.1	7.4	6.6	5.9	5.3	4.7
	-10	9.8	9.0	8.2	7.4	6.7	6.0	5.3	4.7
	-5	9.9	9.0	8.2	7.4	6.7	6.0	5.3	4.7
	0	9.9	9.0	8.2	7.5	6.7	6.0	5.3	4.7
	5	9.9	9.1	8.2	7.5	6.7	6.0	5.4	4.7
	10	8.9	8.1	7.3	6.6	5.9	5.2	4.6	4.0
3000	-40	9.3	8.4	7.7	6.9	6.2	5.5	4.9	4.3
	-30	9.3	8.5	7.7	7.0	6.2	5.6	4.9	4.3
	-25	9.4	8.6	7.8	7.0	6.3	5.6	5.0	4.3
	-20	9.4	8.6	7.8	7.1	6.3	5.6	5.0	4.4
	-15	9.5	8.6	7.8	7.1	6.3	5.7	5.0	4.4
	-10	9.5	8.7	7.9	7.1	6.4	5.7	5.0	4.4
	-5	9.5	8.7	7.9	7.1	6.4	5.7	5.0	4.4
	0	9.5	8.7	7.9	7.1	6.4	5.7	5.1	4.4
	5	9.3	8.5	7.7	6.9	6.2	5.5	4.9	4.3
	10	8.1	7.3	6.5	5.8	5.1	4.5	3.9	3.3
4000	-40	8.9	8.1	7.3	6.6	5.9	5.2	4.6	4.0
	-30	9.0	8.2	7.4	6.6	5.9	5.3	4.6	4.0
	-25	9.0	8.2	7.4	6.7	6.0	5.3	4.7	4.1
	-20	9.1	8.2	7.5	6.7	6.0	5.3	4.7	4.1
	-15	9.1	8.3	7.5	6.7	6.0	5.3	4.7	4.1
	-10	9.1	8.3	7.5	6.8	6.0	5.3	4.7	4.1
	-5	9.1	8.3	7.5	6.8	6.0	5.4	4.7	4.1
	0	9.2	8.3	7.5	6.8	6.1	5.4	4.7	4.1
	5	8.5	7.7	6.9	6.1	5.5	4.8	4.2	3.6
	10	7.2	6.4	5.7	5.0	4.3	3.7	3.1	2.6

FPCG_TO_2_2_02

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October 30, 2016

HJ1-29000-003-001

Page 5-198

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
5000	-40	8.6	7.8	7.0	6.3	5.6	4.9	4.3	3.7
	-30	8.6	7.8	7.1	6.3	5.6	5.0	4.3	3.8
	-25	8.7	7.9	7.1	6.3	5.6	5.0	4.4	3.8
	-20	8.7	7.9	7.1	6.4	5.7	5.0	4.4	3.8
	-15	8.7	7.9	7.2	6.4	5.7	5.0	4.4	3.8
	-10	8.8	7.9	7.2	6.4	5.7	5.0	4.4	3.8
	-5	8.8	8.0	7.2	6.4	5.7	5.1	4.4	3.8
	0	8.8	8.0	7.2	6.4	5.7	5.1	4.4	3.8
	5	7.6	6.8	6.1	5.4	4.7	4.1	3.5	2.9
	10	6.2	5.5	4.8	4.1	3.5	2.9	2.4	1.9
6000	-40	8.2	7.5	6.7	5.9	5.3	4.6	4.0	3.4
	-30	8.3	7.5	6.7	6.0	5.3	4.7	4.1	3.5
	-25	8.3	7.5	6.8	6.0	5.3	4.7	4.1	3.5
	-20	8.4	7.6	6.8	6.0	5.4	4.7	4.1	3.5
	-15	8.4	7.6	6.8	6.1	5.4	4.7	4.1	3.5
	-10	8.4	7.6	6.8	6.1	5.4	4.7	4.1	3.5
	-5	8.4	7.6	6.8	6.1	5.4	4.7	4.1	3.5
	0	8.0	7.2	6.4	5.7	5.0	4.3	3.7	3.2
	5	6.6	5.9	5.2	4.5	3.9	3.3	2.7	2.2
	10	5.3	4.6	3.9	3.3	2.7	2.2	1.7	1.2
7000	-40	7.9	7.2	6.4	5.7	5.0	4.3	3.7	3.2
	-30	8.0	7.2	6.4	5.7	5.0	4.4	3.8	3.2
	-25	8.0	7.2	6.4	5.7	5.1	4.4	3.8	3.3
	-20	8.0	7.3	6.5	5.7	5.1	4.4	3.8	3.3
	-15	8.1	7.3	6.5	5.8	5.1	4.4	3.8	3.3
	-10	8.0	7.3	6.5	5.7	5.1	4.4	3.8	3.3
	-5	8.0	7.2	6.4	5.7	5.0	4.4	3.8	3.2
	0	7.1	6.2	5.5	4.8	4.2	3.6	3.0	2.5
	5	5.7	5.0	4.3	3.7	3.1	2.5	2.0	1.5
	10	4.4	3.7	3.1	2.5	2.0	1.5	1.0	0.6

FPCG_TO_2_5_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-199

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
8000	-40	7.6	6.8	6.1	5.4	4.7	4.1	3.5	2.9
	-30	7.7	6.9	6.1	5.4	4.7	4.1	3.5	3.0
	-25	7.7	6.9	6.1	5.4	4.7	4.1	3.5	3.0
	-20	7.7	6.9	6.1	5.4	4.7	4.1	3.5	3.0
	-15	7.7	6.9	6.1	5.4	4.7	4.1	3.5	3.0
	-10	7.6	6.8	6.1	5.4	4.7	4.1	3.5	2.9
	-5	7.4	6.6	5.9	5.2	4.5	3.9	3.3	2.8
	0	6.1	5.4	4.7	4.0	3.4	2.8	2.3	1.8
	5	4.8	4.1	3.5	2.9	2.3	1.8	1.3	0.9
	10	3.5	2.9	2.3	1.8	1.3	0.9	0.4	0.0
9000	-40	7.3	6.5	5.8	5.1	4.4	3.8	3.2	2.7
	-30	7.3	6.5	5.8	5.1	4.4	3.8	3.2	2.7
	-25	7.3	6.5	5.8	5.1	4.4	3.8	3.2	2.7
	-20	7.3	6.5	5.7	5.0	4.4	3.8	3.2	2.7
	-15	7.3	6.4	5.7	5.0	4.4	3.8	3.2	2.6
	-10	7.2	6.4	5.7	5.0	4.3	3.7	3.2	2.6
	-5	6.5	5.7	5.0	4.3	3.7	3.1	2.6	2.1
	0	5.2	4.5	3.8	3.2	2.7	2.1	1.6	1.2
	5	3.9	3.3	2.7	2.2	1.6	1.2	0.7	0.3
	10	2.7	2.2	1.6	1.1	0.7	0.2	-0.2	-0.6
10000	-40	6.9	6.1	5.4	4.7	4.1	3.5	2.9	2.4
	-30	6.9	6.1	5.4	4.7	4.1	3.5	2.9	2.4
	-25	6.9	6.1	5.4	4.7	4.1	3.5	2.9	2.4
	-20	6.8	6.1	5.3	4.7	4.0	3.4	2.9	2.3
	-15	6.8	6.0	5.3	4.6	4.0	3.4	2.8	2.3
	-10	6.8	6.0	5.3	4.6	4.0	3.4	2.8	2.3
	-5	5.6	4.9	4.2	3.6	3.0	2.5	2.0	1.5
	0	4.3	3.7	3.1	2.5	2.0	1.5	1.0	0.6
	5	3.1	2.5	1.9	1.4	1.0	0.5	0.1	-0.3
	10	1.9	1.4	0.9	0.5	0.0	-0.4	-0.7	-1.1

FPCG_TO_2_8_02

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October 30, 2016

HJ1-29000-003-001

Page 5-200

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps TO/APPR, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
11000	-40	6.5	5.7	5.0	4.4	3.7	3.2	2.6	2.1
	-30	6.4	5.7	5.0	4.3	3.7	3.1	2.6	2.1
	-25	6.4	5.7	5.0	4.3	3.7	3.1	2.6	2.0
	-20	6.4	5.6	4.9	4.3	3.6	3.1	2.5	2.0
	-15	6.4	5.6	4.9	4.3	3.6	3.1	2.5	2.0
	-10	6.0	5.2	4.5	3.9	3.3	2.7	2.2	1.7
	-5	4.7	4.1	3.4	2.8	2.3	1.8	1.3	0.8
	0	3.5	2.8	2.3	1.8	1.3	0.8	0.4	0.0
	5	2.3	1.7	1.2	0.8	0.3	-0.1	-0.5	-0.9
	10	1.2	0.7	0.3	-0.2	-0.6	-0.9	-1.3	-1.6
12000	-40	6.1	5.4	4.7	4.0	3.4	2.8	2.3	1.8
	-30	6.1	5.3	4.6	4.0	3.4	2.8	2.3	1.8
	-25	6.0	5.3	4.6	4.0	3.4	2.8	2.3	1.8
	-20	6.0	5.3	4.6	3.9	3.3	2.8	2.3	1.8
	-15	6.0	5.3	4.6	3.9	3.3	2.8	2.3	1.8
	-10	5.1	4.4	3.8	3.2	2.6	2.1	1.6	1.1
	-5	3.8	3.2	2.6	2.1	1.6	1.1	0.6	0.2
	0	2.6	2.1	1.6	1.1	0.6	0.2	-0.2	-0.6
	5	1.5	1.0	0.5	0.1	-0.3	-0.7	-1.1	-1.4
	10	0.5	0.0	-0.4	-0.8	-1.1	-1.5	-1.8	-2.1

FPCG_TO_2_11_02

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
-1000	-40	11.9	11.0	10.2	9.4	8.6	7.9	7.2	6.5
	15	12.2	11.3	10.4	9.6	8.8	8.1	7.4	6.7
	20	12.2	11.3	10.4	9.6	8.8	8.1	7.4	6.7
	25	12.3	11.3	10.5	9.7	8.9	8.1	7.4	6.8
	30	11.6	10.7	9.8	9.1	8.3	7.6	6.9	6.2
	35	10.4	9.6	8.8	8.0	7.3	6.6	6.0	5.4
	40	9.2	8.4	7.7	7.0	6.3	5.6	5.0	4.5
	45	8.0	7.3	6.6	5.9	5.3	4.6	4.1	3.6
	50	6.8	6.1	5.5	4.9	4.3	3.7	3.2	2.7
	55	5.7	5.1	4.4	3.9	3.3	2.8	2.3	1.8
Sea Level	-40	11.8	10.9	10.1	9.3	8.5	7.8	7.1	6.5
	15	12.1	11.2	10.4	9.5	8.8	8.0	7.3	6.7
	20	12.2	11.2	10.4	9.6	8.8	8.0	7.3	6.7
	25	12.2	11.3	10.4	9.6	8.8	8.1	7.4	6.7
	30	11.1	10.2	9.4	8.6	7.8	7.1	6.5	5.8
	35	9.9	9.0	8.3	7.5	6.8	6.1	5.5	4.9
	40	8.6	7.8	7.1	6.4	5.8	5.1	4.6	4.0
	45	7.5	6.7	6.1	5.4	4.8	4.2	3.6	3.1
	50	6.3	5.6	5.0	4.4	3.8	3.2	2.7	2.3
	55	5.2	4.6	4.0	3.4	2.9	2.3	1.9	1.4
1000	-40	11.4	10.6	9.7	8.9	8.2	7.5	6.8	6.2
	10	11.7	10.8	10.0	9.2	8.4	7.7	7.0	6.4
	15	11.8	10.9	10.0	9.2	8.4	7.7	7.0	6.4
	20	11.8	10.9	10.0	9.2	8.5	7.7	7.0	6.4
	25	11.4	10.5	9.6	8.9	8.1	7.4	6.7	6.1
	30	10.2	9.4	8.6	7.9	7.1	6.5	5.8	5.2
	35	9.1	8.3	7.6	6.8	6.2	5.5	4.9	4.4
	40	7.9	7.2	6.5	5.8	5.2	4.6	4.0	3.5
	45	6.8	6.1	5.4	4.8	4.2	3.6	3.1	2.6
	50	5.7	5.0	4.4	3.8	3.2	2.7	2.2	1.8

FPCG_UP_0_1_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-202

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
2000	-40	11.1	10.2	9.4	8.6	7.9	7.2	6.5	5.9
	10	11.4	10.5	9.7	8.9	8.1	7.4	6.7	6.1
	15	11.4	10.5	9.7	8.9	8.1	7.4	6.7	6.1
	20	11.5	10.6	9.7	8.9	8.2	7.4	6.8	6.1
	25	10.6	9.7	8.9	8.1	7.4	6.7	6.1	5.5
	30	9.4	8.6	7.9	7.2	6.5	5.8	5.2	4.6
	35	8.4	7.6	6.9	6.2	5.5	4.9	4.3	3.8
	40	7.2	6.5	5.8	5.2	4.6	4.0	3.4	2.9
	45	6.1	5.4	4.8	4.2	3.6	3.1	2.6	2.1
	50	5.0	4.4	3.8	3.3	2.7	2.2	1.7	1.3
3000	-40	10.7	9.9	9.1	8.3	7.6	6.9	6.2	5.6
	5	11.0	10.1	9.3	8.5	7.8	7.1	6.4	5.8
	10	11.0	10.2	9.3	8.6	7.8	7.1	6.4	5.8
	15	11.1	10.2	9.4	8.6	7.8	7.1	6.5	5.8
	20	10.9	10.0	9.2	8.4	7.7	7.0	6.3	5.7
	25	9.7	8.9	8.1	7.4	6.7	6.0	5.4	4.8
	30	8.6	7.9	7.1	6.4	5.8	5.1	4.6	4.0
	35	7.6	6.9	6.2	5.5	4.9	4.3	3.7	3.2
	40	6.5	5.9	5.2	4.6	4.0	3.4	2.9	2.4
	45	5.5	4.8	4.2	3.7	3.1	2.6	2.1	1.6
4000	-40	10.4	9.5	8.7	8.0	7.3	6.6	5.9	5.3
	5	10.6	9.8	9.0	8.2	7.5	6.8	6.1	5.5
	10	10.6	9.8	9.0	8.2	7.5	6.8	6.1	5.5
	15	10.7	9.8	9.0	8.2	7.5	6.8	6.1	5.5
	20	10.0	9.2	8.4	7.6	6.9	6.3	5.6	5.0
	25	8.9	8.1	7.4	6.7	6.0	5.4	4.8	4.2
	30	7.8	7.1	6.4	5.7	5.1	4.5	3.9	3.4
	35	6.8	6.1	5.5	4.8	4.2	3.7	3.1	2.7
	40	5.8	5.2	4.6	4.0	3.4	2.9	2.4	1.9
	45	4.9	4.2	3.7	3.1	2.6	2.1	1.6	1.2

FPCG_UP_0_2_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-203

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
5000	-40	10.0	9.2	8.4	7.7	6.9	6.3	5.6	5.1
	5	10.2	9.4	8.6	7.9	7.1	6.5	5.8	5.2
	10	10.2	9.4	8.6	7.9	7.1	6.4	5.8	5.2
	15	10.2	9.4	8.6	7.8	7.1	6.4	5.8	5.2
	20	9.2	8.4	7.6	6.9	6.2	5.6	5.0	4.4
	25	8.1	7.3	6.6	6.0	5.3	4.7	4.1	3.6
	30	7.0	6.3	5.7	5.0	4.4	3.8	3.3	2.8
	35	6.1	5.4	4.8	4.2	3.6	3.1	2.6	2.1
	40	5.1	4.5	3.9	3.3	2.8	2.3	1.8	1.4
	45	4.2	3.6	3.1	2.5	2.0	1.5	1.1	0.7
6000	-40	9.7	8.8	8.1	7.4	6.7	6.0	5.4	4.8
	0	9.9	9.0	8.2	7.5	6.8	6.1	5.5	4.9
	5	9.8	9.0	8.2	7.5	6.8	6.1	5.5	4.9
	10	9.8	9.0	8.2	7.5	6.8	6.1	5.5	4.9
	15	9.4	8.6	7.8	7.1	6.4	5.8	5.2	4.6
	20	8.3	7.6	6.9	6.2	5.5	4.9	4.3	3.8
	25	7.3	6.6	5.9	5.3	4.6	4.1	3.5	3.0
	30	6.3	5.6	5.0	4.4	3.8	3.2	2.7	2.3
	35	5.4	4.7	4.1	3.5	3.0	2.5	2.0	1.6
	40	4.4	3.8	3.3	2.7	2.2	1.7	1.3	0.9
7000	-40	9.3	8.5	7.8	7.1	6.4	5.7	5.1	4.6
	0	9.4	8.6	7.9	7.2	6.5	5.8	5.2	4.6
	5	9.4	8.6	7.8	7.1	6.4	5.8	5.2	4.6
	10	9.4	8.6	7.8	7.1	6.4	5.8	5.2	4.6
	15	8.6	7.8	7.1	6.4	5.7	5.1	4.5	4.0
	20	7.6	6.8	6.1	5.5	4.9	4.3	3.7	3.2
	25	6.5	5.8	5.2	4.6	4.0	3.4	2.9	2.4
	30	5.6	4.9	4.3	3.7	3.2	2.6	2.2	1.7
	35	4.7	4.1	3.5	2.9	2.4	1.9	1.5	1.0
	40	3.8	3.2	2.7	2.1	1.7	1.2	0.8	0.4

FPCG_UP_0_5_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-204

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
8000	-40	9.0	8.2	7.5	6.8	6.1	5.5	4.9	4.3
	-5	9.1	8.3	7.5	6.8	6.1	5.5	4.9	4.3
	0	9.0	8.2	7.5	6.8	6.1	5.5	4.9	4.3
	5	9.0	8.2	7.5	6.8	6.1	5.4	4.8	4.3
	10	8.8	8.0	7.3	6.6	5.9	5.3	4.7	4.1
	15	7.8	7.0	6.3	5.7	5.0	4.4	3.9	3.4
	20	6.8	6.1	5.4	4.8	4.2	3.6	3.1	2.6
	25	5.8	5.1	4.5	3.9	3.4	2.8	2.3	1.9
	30	4.9	4.3	3.7	3.1	2.6	2.1	1.6	1.2
	35	4.0	3.4	2.9	2.4	1.9	1.4	1.0	0.6
9000	-40	8.7	7.9	7.2	6.5	5.8	5.2	4.6	4.1
	-5	8.6	7.9	7.1	6.4	5.8	5.1	4.6	4.0
	0	8.6	7.9	7.1	6.4	5.8	5.1	4.6	4.0
	5	8.6	7.8	7.1	6.4	5.7	5.1	4.5	4.0
	10	8.0	7.2	6.5	5.8	5.2	4.6	4.0	3.5
	15	7.0	6.3	5.6	5.0	4.4	3.8	3.3	2.8
	20	6.0	5.4	4.7	4.1	3.6	3.0	2.5	2.1
	25	5.1	4.5	3.9	3.3	2.8	2.3	1.8	1.4
	30	4.2	3.6	3.1	2.6	2.0	1.6	1.1	0.7
	35	3.4	2.8	2.3	1.8	1.3	0.9	0.4	0.1
10000	-40	8.4	7.6	6.9	6.2	5.5	4.9	4.4	3.8
	-5	8.3	7.5	6.8	6.1	5.5	4.9	4.3	3.8
	0	8.3	7.5	6.8	6.1	5.5	4.9	4.3	3.8
	5	8.2	7.4	6.7	6.0	5.4	4.8	4.2	3.7
	10	7.1	6.4	5.7	5.1	4.5	3.9	3.4	2.9
	15	6.2	5.5	4.9	4.3	3.7	3.2	2.7	2.2
	20	5.3	4.7	4.1	3.5	3.0	2.4	2.0	1.5
	25	4.4	3.8	3.2	2.7	2.2	1.7	1.3	0.8
	30	3.6	3.0	2.5	2.0	1.5	1.0	0.6	0.2
	35	2.7	2.2	1.7	1.2	0.8	0.3	-0.1	-0.4

FPCG_UP_0_8_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-205

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection Off, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
11000	-40	8.0	7.2	6.5	5.9	5.2	4.6	4.1	3.5
	-10	7.9	7.2	6.5	5.8	5.1	4.5	4.0	3.5
	-5	7.9	7.2	6.5	5.8	5.2	4.6	4.0	3.5
	0	7.9	7.2	6.4	5.8	5.1	4.5	4.0	3.5
	5	7.5	6.7	6.1	5.4	4.8	4.2	3.6	3.1
	10	6.4	5.7	5.1	4.5	3.9	3.3	2.8	2.4
	15	5.5	4.9	4.3	3.7	3.1	2.6	2.1	1.7
	20	4.6	4.0	3.5	2.9	2.4	1.9	1.4	1.0
	25	3.8	3.2	2.7	2.1	1.7	1.2	0.8	0.4
	30	3.0	2.4	1.9	1.4	1.0	0.5	0.1	-0.2
12000	-40	7.6	6.9	6.2	5.5	4.9	4.3	3.8	3.3
	-10	7.6	6.8	6.1	5.5	4.9	4.3	3.7	3.2
	-5	7.6	6.8	6.2	5.5	4.9	4.3	3.7	3.2
	0	7.6	6.8	6.1	5.5	4.9	4.3	3.7	3.2
	5	6.7	6.0	5.4	4.7	4.1	3.6	3.1	2.6
	10	5.7	5.1	4.4	3.8	3.3	2.8	2.3	1.8
	15	4.8	4.2	3.6	3.1	2.5	2.0	1.6	1.2
	20	4.0	3.4	2.9	2.3	1.8	1.4	0.9	0.5
	25	3.2	2.6	2.1	1.6	1.1	0.7	0.3	-0.1
	30	2.4	1.9	1.4	0.9	0.5	0.1	-0.3	-0.7

FPCG_UP_0_11_04

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-206

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
-1000	-40	11.9	11.0	10.2	9.4	8.6	7.9	7.2	6.6
	-30	12.0	11.0	10.2	9.4	8.6	7.9	7.2	6.6
	-25	12.0	11.1	10.2	9.4	8.6	7.9	7.2	6.6
	-20	12.0	11.1	10.2	9.4	8.7	7.9	7.2	6.6
	-15	12.0	11.1	10.2	9.5	8.7	7.9	7.2	6.6
	-10	12.0	11.1	10.3	9.5	8.7	8.0	7.3	6.6
	-5	12.1	11.2	10.3	9.5	8.7	8.0	7.3	6.7
	0	12.1	11.2	10.4	9.6	8.8	8.0	7.3	6.7
	5	12.2	11.3	10.4	9.6	8.8	8.1	7.4	6.7
	10	12.2	11.3	10.4	9.6	8.8	8.1	7.4	6.7
Sea Level	-40	11.8	11.0	10.1	9.3	8.5	7.8	7.1	6.5
	-30	11.9	11.0	10.1	9.3	8.6	7.8	7.1	6.5
	-25	11.9	11.0	10.2	9.4	8.6	7.8	7.2	6.5
	-20	11.9	11.0	10.2	9.4	8.6	7.9	7.2	6.5
	-15	12.0	11.0	10.2	9.4	8.6	7.9	7.2	6.6
	-10	12.0	11.1	10.2	9.4	8.7	7.9	7.2	6.6
	-5	12.0	11.1	10.3	9.5	8.7	7.9	7.3	6.6
	0	12.1	11.2	10.3	9.5	8.7	8.0	7.3	6.6
	5	12.1	11.2	10.3	9.5	8.8	8.0	7.3	6.7
	10	12.2	11.2	10.4	9.6	8.8	8.0	7.3	6.7
1000	-40	11.5	10.6	9.7	9.0	8.2	7.5	6.8	6.2
	-30	11.5	10.6	9.8	9.0	8.2	7.5	6.8	6.2
	-25	11.5	10.6	9.8	9.0	8.3	7.5	6.9	6.2
	-20	11.6	10.7	9.8	9.1	8.3	7.6	6.9	6.3
	-15	11.6	10.7	9.9	9.1	8.3	7.6	6.9	6.3
	-10	11.6	10.7	9.9	9.1	8.3	7.6	6.9	6.3
	-5	11.7	10.8	9.9	9.2	8.4	7.6	7.0	6.3
	0	11.7	10.8	10.0	9.2	8.4	7.7	7.0	6.4
	5	11.8	10.8	10.0	9.2	8.4	7.7	7.0	6.4
	10	11.3	10.4	9.6	8.8	8.0	7.3	6.6	6.0

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-207

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
2000	-40	11.1	10.2	9.4	8.7	7.9	7.2	6.5	5.9
	-30	11.2	10.3	9.5	8.7	8.0	7.2	6.6	6.0
	-25	11.2	10.3	9.5	8.7	8.0	7.3	6.6	6.0
	-20	11.2	10.4	9.5	8.8	8.0	7.3	6.6	6.0
	-15	11.3	10.4	9.6	8.8	8.0	7.3	6.7	6.0
	-10	11.3	10.4	9.6	8.8	8.1	7.3	6.7	6.1
	-5	11.4	10.5	9.6	8.9	8.1	7.4	6.7	6.1
	0	11.4	10.5	9.7	8.9	8.1	7.4	6.7	6.1
	5	11.4	10.5	9.7	8.9	8.1	7.4	6.7	6.1
	10	10.4	9.6	8.8	8.0	7.3	6.6	6.0	5.4
3000	-40	10.8	9.9	9.1	8.3	7.6	6.9	6.3	5.7
	-30	10.8	10.0	9.2	8.4	7.7	7.0	6.3	5.7
	-25	10.9	10.0	9.2	8.5	7.7	7.0	6.3	5.7
	-20	10.9	10.1	9.2	8.5	7.7	7.0	6.4	5.8
	-15	11.0	10.1	9.3	8.5	7.8	7.1	6.4	5.8
	-10	11.0	10.1	9.3	8.5	7.8	7.1	6.4	5.8
	-5	11.0	10.1	9.3	8.6	7.8	7.1	6.4	5.8
	0	11.0	10.2	9.3	8.6	7.8	7.1	6.5	5.8
	5	10.8	10.0	9.1	8.4	7.6	6.9	6.3	5.7
	10	9.6	8.7	8.0	7.3	6.6	5.9	5.3	4.7
4000	-40	10.4	9.6	8.8	8.0	7.3	6.6	6.0	5.4
	-30	10.5	9.6	8.8	8.1	7.4	6.7	6.0	5.4
	-25	10.5	9.7	8.9	8.1	7.4	6.7	6.1	5.5
	-20	10.6	9.7	8.9	8.2	7.4	6.7	6.1	5.5
	-15	10.6	9.7	8.9	8.2	7.4	6.8	6.1	5.5
	-10	10.6	9.8	9.0	8.2	7.5	6.8	6.1	5.5
	-5	10.6	9.8	9.0	8.2	7.5	6.8	6.1	5.5
	0	10.7	9.8	9.0	8.2	7.5	6.8	6.2	5.6
	5	10.0	9.1	8.4	7.6	6.9	6.2	5.6	5.0
	10	8.7	7.9	7.2	6.5	5.8	5.2	4.6	4.1

FPCG_UP_2_2_02

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-208

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
5000	-40	10.1	9.2	8.4	7.7	7.0	6.3	5.7	5.1
	-30	10.1	9.3	8.5	7.8	7.1	6.4	5.8	5.2
	-25	10.2	9.3	8.5	7.8	7.1	6.4	5.8	5.2
	-20	10.2	9.4	8.6	7.8	7.1	6.4	5.8	5.2
	-15	10.2	9.4	8.6	7.9	7.1	6.4	5.8	5.2
	-10	10.2	9.4	8.6	7.9	7.1	6.5	5.8	5.2
	-5	10.3	9.4	8.6	7.9	7.2	6.5	5.8	5.3
	0	10.3	9.4	8.6	7.9	7.2	6.5	5.8	5.3
	5	9.1	8.3	7.6	6.9	6.2	5.5	4.9	4.4
	10	7.8	7.0	6.3	5.7	5.0	4.4	3.9	3.4
6000	-40	9.7	8.9	8.1	7.4	6.7	6.1	5.4	4.9
	-30	9.8	9.0	8.2	7.5	6.8	6.1	5.5	4.9
	-25	9.8	9.0	8.2	7.5	6.8	6.1	5.5	4.9
	-20	9.9	9.0	8.3	7.5	6.8	6.2	5.5	5.0
	-15	9.9	9.1	8.3	7.5	6.8	6.2	5.5	5.0
	-10	9.9	9.1	8.3	7.6	6.8	6.2	5.6	5.0
	-5	9.9	9.1	8.3	7.6	6.8	6.2	5.6	5.0
	0	9.4	8.6	7.9	7.2	6.5	5.8	5.2	4.6
	5	8.2	7.4	6.7	6.0	5.4	4.8	4.2	3.7
	10	6.9	6.2	5.5	4.9	4.3	3.7	3.2	2.7
7000	-40	9.4	8.6	7.8	7.1	6.4	5.8	5.2	4.6
	-30	9.5	8.7	7.9	7.2	6.5	5.8	5.2	4.7
	-25	9.5	8.7	7.9	7.2	6.5	5.9	5.2	4.7
	-20	9.5	8.7	7.9	7.2	6.5	5.9	5.3	4.7
	-15	9.5	8.7	8.0	7.2	6.5	5.9	5.3	4.7
	-10	9.5	8.7	8.0	7.2	6.5	5.9	5.3	4.7
	-5	9.5	8.7	7.9	7.2	6.5	5.9	5.2	4.7
	0	8.6	7.8	7.0	6.4	5.7	5.1	4.5	4.0
	5	7.3	6.6	5.9	5.2	4.6	4.0	3.5	3.0
	10	6.1	5.4	4.8	4.2	3.6	3.0	2.5	2.1

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October 30, 2016

HJ1-29000-003-001

Page 5-209

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
8000	-40	9.1	8.3	7.5	6.8	6.2	5.5	4.9	4.4
	-30	9.1	8.3	7.6	6.9	6.2	5.5	4.9	4.4
	-25	9.1	8.3	7.6	6.9	6.2	5.6	5.0	4.4
	-20	9.1	8.3	7.6	6.9	6.2	5.6	5.0	4.4
	-15	9.1	8.3	7.6	6.9	6.2	5.6	4.9	4.4
	-10	9.1	8.3	7.6	6.9	6.2	5.5	4.9	4.4
	-5	8.9	8.1	7.4	6.7	6.0	5.3	4.7	4.2
	0	7.7	6.9	6.2	5.6	4.9	4.4	3.8	3.3
	5	6.4	5.8	5.1	4.5	3.9	3.4	2.8	2.4
	10	5.3	4.6	4.0	3.5	2.9	2.4	1.9	1.5
9000	-40	8.7	8.0	7.2	6.5	5.9	5.2	4.7	4.1
	-30	8.8	8.0	7.3	6.6	5.9	5.3	4.7	4.1
	-25	8.8	8.0	7.3	6.6	5.9	5.3	4.7	4.1
	-20	8.8	8.0	7.2	6.5	5.9	5.2	4.7	4.1
	-15	8.7	7.9	7.2	6.5	5.8	5.2	4.6	4.1
	-10	8.7	7.9	7.2	6.5	5.8	5.2	4.6	4.1
	-5	8.0	7.3	6.6	5.9	5.2	4.6	4.1	3.6
	0	6.8	6.1	5.5	4.8	4.2	3.7	3.1	2.7
	5	5.6	5.0	4.4	3.8	3.2	2.7	2.2	1.8
	10	4.5	3.9	3.4	2.8	2.3	1.8	1.4	0.9
10000	-40	8.4	7.6	6.9	6.2	5.6	5.0	4.4	3.9
	-30	8.4	7.6	6.9	6.2	5.6	5.0	4.4	3.9
	-25	8.4	7.6	6.9	6.2	5.6	4.9	4.4	3.8
	-20	8.3	7.6	6.9	6.2	5.5	4.9	4.3	3.8
	-15	8.3	7.5	6.8	6.2	5.5	4.9	4.3	3.8
	-10	8.3	7.5	6.8	6.1	5.5	4.9	4.3	3.8
	-5	7.2	6.5	5.8	5.2	4.6	4.0	3.4	2.9
	0	6.0	5.3	4.7	4.1	3.5	3.0	2.5	2.0
	5	4.9	4.2	3.7	3.1	2.6	2.1	1.6	1.2
	10	3.8	3.2	2.7	2.2	1.7	1.2	0.8	0.4

FPCG_UP_2_8_02

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HJ1-29000-003-001

Page 5-210

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HA-420 AFM

PERFORMANCE

Uncorrected Net Climb Gradient [%] at pressure altitude									
Flaps UP, Gear UP, Ice Protection On, V2									
F. ALT [ft]	TEMP [°C]	Takeoff Weight [lb]							
		7800	8200	8600	9000	9400	9800	10200	10600
11000	-40	8.0	7.3	6.6	5.9	5.2	4.6	4.1	3.6
	-30	8.0	7.2	6.5	5.9	5.2	4.6	4.1	3.5
	-25	7.9	7.2	6.5	5.8	5.2	4.6	4.0	3.5
	-20	7.9	7.2	6.5	5.8	5.2	4.6	4.0	3.5
	-15	7.9	7.2	6.4	5.8	5.1	4.5	4.0	3.5
	-10	7.5	6.8	6.1	5.4	4.8	4.2	3.7	3.2
	-5	6.4	5.7	5.1	4.4	3.9	3.3	2.8	2.3
	0	5.2	4.6	4.0	3.4	2.9	2.4	1.9	1.4
	5	4.1	3.5	3.0	2.5	2.0	1.5	1.0	0.6
	10	3.1	2.6	2.1	1.6	1.1	0.7	0.3	-0.1
12000	-40	7.6	6.9	6.2	5.6	4.9	4.3	3.8	3.3
	-30	7.6	6.9	6.2	5.5	4.9	4.3	3.8	3.2
	-25	7.6	6.8	6.2	5.5	4.9	4.3	3.7	3.2
	-20	7.6	6.8	6.1	5.5	4.9	4.3	3.7	3.2
	-15	7.6	6.8	6.1	5.5	4.9	4.3	3.7	3.2
	-10	6.7	6.0	5.4	4.8	4.2	3.6	3.1	2.6
	-5	5.6	4.9	4.3	3.7	3.2	2.6	2.2	1.7
	0	4.5	3.9	3.3	2.7	2.2	1.7	1.3	0.9
	5	3.4	2.9	2.3	1.8	1.4	0.9	0.5	0.1
	10	2.5	1.9	1.5	1.0	0.5	0.1	-0.3	-0.6

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

ENROUTE CLIMB

Use of enroute climb tables:

1. Determine the airplane gross weight.
2. Obtain ambient information:
 - ambient temperature
 - Determine that the temperature is within the ambient temperature limits found in the limitations section.
 - pressure altitude
3. Using the gross weight determined in step 1, find the 1 Engine and/or 2 Engine enroute climb gradient and/or rate of climb using pages 213 through 231.

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HA-420 AFM

PERFORMANCE

		Single Engine En Route Gross Climb Gradient [%]																		
		FLAPS UP, GEAR UP, Ice Protection Off, MCT, 140 [KIAS]																		
Alt [ft]	Wt [lb]	TEMPERATURE [°C]																		
		-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	15	20	25	30	35	40		
SL	8000	10.6	10.6	10.6	10.6	10.6	10.6	10.6	10.7	10.7	10.8	10.8	10.8	10.9	10.9	9.7	8.7	7.8		
	8500	9.6	9.6	9.6	9.6	9.6	9.7	9.7	9.8	9.8	9.8	9.8	9.9	9.9	9.9	8.8	7.9	7.0		
	9000	8.7	8.7	8.7	8.8	8.8	8.8	8.8	8.9	8.9	8.9	8.9	9.0	9.0	9.0	8.0	7.1	6.2		
	9500	7.9	7.9	7.9	8.0	8.0	8.0	8.0	8.1	8.1	8.1	8.1	8.2	8.2	8.2	7.2	6.4	5.6		
	10000	7.2	7.2	7.2	7.2	7.2	7.2	7.3	7.3	7.3	7.4	7.4	7.4	7.4	7.4	6.5	5.7	5.0		
5000	10500	6.5	6.5	6.5	6.5	6.5	6.6	6.6	6.6	6.6	6.7	6.7	6.7	6.7	6.7	5.9	5.1	4.4		
	8000	8.8	8.8	8.9	8.9	8.9	8.9	9.0	9.0	9.0	9.0	9.0	9.0	7.9	6.8	5.9				
	8500	7.9	8.0	8.0	8.0	8.1	8.1	8.1	8.1	8.1	8.1	8.1	8.1	7.1	6.1	5.2				
	9000	7.2	7.2	7.2	7.3	7.3	7.3	7.3	7.3	7.3	7.3	7.3	7.3	6.3	5.4	4.6				
	9500	6.4	6.5	6.5	6.5	6.5	6.6	6.6	6.6	6.6	6.6	6.6	6.6	5.7	4.8	4.0				
10000	10000	5.8	5.8	5.8	5.9	5.9	5.9	5.9	5.9	5.9	5.9	5.9	5.9	5.0	4.2	3.5				
	10500	5.2	5.2	5.2	5.2	5.3	5.3	5.3	5.3	5.3	5.3	5.3	5.3	4.5	3.7	3.0				
	8000	7.2	7.2	7.2	7.2	7.1	7.1	7.1	7.1	7.1	7.1	6.1	5.2	4.4						
	8500	6.4	6.4	6.4	6.4	6.4	6.3	6.3	6.3	6.3	6.3	5.4	4.6	3.8						
	9000	5.7	5.7	5.7	5.7	5.7	5.6	5.6	5.6	5.6	5.6	4.8	4.0	3.2						
10000	9500	5.1	5.1	5.1	5.1	5.0	5.0	5.0	5.0	5.0	5.0	4.2	3.5	2.7						
	10000	4.5	4.5	4.5	4.5	4.4	4.4	4.4	4.4	4.4	4.4	3.7	2.9	2.3						
	10500	3.9	4.0	3.9	3.9	3.9	3.9	3.9	3.8	3.9	3.9	3.2	2.5	1.8						

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Single Engine En Route Gross Climb Gradient [%]																
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 140 [KIAS]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	
15000	8000				5.4	5.4	5.4	5.4	5.4	5.4	5.4	4.6	3.8	3.0	2.3	
	8500				4.7	4.7	4.7	4.7	4.7	4.8	4.8	4.0	3.2	2.5	1.8	
	9000				4.1	4.1	4.1	4.1	4.1	4.2	4.2	3.4	2.7	2.0	1.4	
	9500				3.6	3.6	3.6	3.6	3.6	3.6	3.6	2.9	2.2	1.6	1.0	
	10000				3.0	3.0	3.1	3.1	3.1	3.1	3.1	2.4	1.8	1.1	0.6	
20000	10500				2.6	2.6	2.6	2.6	2.6	2.6	2.6	2.0	1.3	0.8	0.2	
	8000			4.2	4.2	4.2	4.2	4.2	4.2	3.4	2.7	1.9	1.2			
	8500			3.6	3.6	3.6	3.6	3.7	3.6	2.9	2.2	1.5	0.8			
	9000			3.1	3.1	3.1	3.1	3.1	3.0	2.4	1.7	1.1	0.4			
	9500			2.6	2.6	2.6	2.6	2.6	2.6	2.0	1.3	0.7	0.1			
25000	10000			2.1	2.1	2.1	2.1	2.2	2.1	1.5	0.9	0.3	-0.3			
	10500			1.7	1.7	1.7	1.7	1.7	1.7	1.1	0.5	0.0	-0.6			
	8000	3.4	3.4	3.4	3.5	3.5	3.1	2.4	1.8	1.2	0.6					
	8500	2.9	2.9	2.9	2.9	2.9	2.6	1.9	1.4	0.8	0.2					
	9000	2.4	2.4	2.4	2.4	2.5	2.1	1.5	1.0	0.4	-0.1					
TERGG_UP_0.2_05	9500	1.9	1.9	2.0	2.0	2.0	1.6	1.1	0.6	0.1	-0.5					
	10000	1.5	1.5	1.5	1.5	1.6	1.2	0.7	0.2	-0.3	-0.8					
	10500	1.1	1.1	1.1	1.1	1.1	0.8	0.3	-0.1	-0.6	-1.1					

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

		Single Engine En Route Rate of Climb [FPM]																
		FLAPS UP, GEAR UP, Ice Protection Off, MCT, 140 [KIAS]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]																
		-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	15	20	25	30	35	40
SL	8000	1346	1361	1377	1393	1409	1424	1443	1462	1482	1501	1518	1535	1553	1566	1417	1281	1147
	8500	1224	1238	1252	1267	1281	1295	1312	1330	1348	1366	1381	1397	1413	1425	1285	1156	1029
	9000	1113	1125	1138	1152	1165	1177	1193	1210	1226	1243	1257	1271	1286	1297	1164	1042	922
	9500	1011	1022	1034	1046	1058	1070	1084	1099	1115	1130	1143	1156	1170	1180	1053	938	823
	10000	917	927	938	949	960	971	984	998	1012	1026	1038	1050	1063	1071	951	841	732
5000	10500	830	839	849	859	869	879	891	904	917	930	941	952	963	971	856	751	647
	8000	1227	1246	1264	1282	1298	1314	1329	1344	1358	1371	1381	1390	1231	1076	941		
	8500	1108	1125	1141	1158	1173	1187	1200	1214	1227	1239	1248	1256	1106	960	831		
	9000	999	1015	1030	1045	1058	1072	1084	1096	1107	1118	1127	1133	991	853	731		
	9500	899	913	927	941	953	965	976	987	998	1008	1015	1021	886	754	639		
10000	10000	807	820	832	845	856	867	877	887	897	906	912	917	789	663	553		
	10500	721	733	744	756	766	776	785	794	803	811	816	821	698	578	472		
	8000	1103	1116	1126	1134	1140	1147	1153	1162	1173	1185	1039	895	756				
	8500	986	999	1007	1014	1019	1025	1030	1038	1048	1059	921	785	653				
	9000	880	891	899	904	909	913	918	925	934	944	812	684	559				
10000	9500	781	791	798	803	807	811	815	820	829	837	713	590	471				
	10000	690	699	705	709	712	716	719	724	731	739	620	503	390				
	10500	606	614	619	622	624	627	629	633	640	647	533	421	313				

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Single Engine En Route Rate of Climb [FPM]																
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 140 [KIAS]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	
15000	8000				914	925	937	949	961	973	985	839	695	556	426	
	8500				803	813	823	834	845	856	867	729	592	461	338	
	9000				701	709	719	728	738	748	758	627	497	373	256	
	9500				607	614	622	631	639	648	657	532	408	290	179	
	10000				519	525	532	540	547	555	563	444	326	213	107	
20000	10500				436	442	448	454	461	468	475	361	248	140	38	
	8000			771	783	795	806	817	810	678	530	391	249			
	8500			664	675	685	695	705	698	572	433	301	167			
	9000			566	575	584	593	601	594	475	342	217	90			
	9500			474	482	490	498	505	497	384	258	138	17			
25000	10000			388	395	402	409	415	407	298	178	64	-51			
	10500			308	313	319	325	331	322	218	103	-6	-117			
	8000	681	694	706	719	731	650	517	392	263	125					
	8500	574	585	597	608	618	542	416	297	175	44					
	9000	476	485	495	505	514	441	321	209	93	-31					
TERROC_UP_0_2_05	9500	384	392	401	409	417	347	233	126	15	-103					
	10000	298	305	312	319	326	259	150	47	-59	-172					
	10500	217	223	229	235	241	176	71	-27	-129	-237					

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Single Engine En Route Gross Climb Gradient [%]												
FLAPS UP, GEAR UP, Ice Protection On, MCT, 140 [KIAS]												
Alt [ft]	Wt [lb]	TEMPERATURE [°C]										
		-40	-35	-30	-25	-20	-15	-10	-5	0	5	10
SL	8000	10.6	10.6	10.7	10.7	10.7	10.7	10.8	10.8	10.9	10.9	10.9
	8500	9.7	9.7	9.7	9.7	9.7	9.8	9.8	9.8	9.9	9.9	9.9
	9000	8.8	8.8	8.8	8.8	8.9	8.9	8.9	9.0	9.0	9.0	9.0
	9500	8.0	8.0	8.0	8.0	8.0	8.1	8.1	8.1	8.2	8.2	8.2
	10000	7.2	7.2	7.3	7.3	7.3	7.3	7.4	7.4	7.4	7.4	7.4
5000	10500	6.5	6.5	6.6	6.6	6.6	6.6	6.7	6.7	6.7	6.7	6.7
	8000	8.9	9.0	9.0	9.0	9.0	9.1	9.1	9.1	9.1	7.8	6.5
	8500	8.1	8.1	8.1	8.1	8.2	8.2	8.2	8.2	8.2	7.0	5.8
	9000	7.3	7.3	7.3	7.3	7.4	7.4	7.4	7.4	7.4	6.2	5.1
	9500	6.6	6.6	6.6	6.6	6.6	6.7	6.7	6.7	6.7	5.6	4.5
10000	10000	5.9	5.9	5.9	5.9	6.0	6.0	6.0	6.0	6.0	4.9	3.9
	10500	5.3	5.3	5.3	5.3	5.3	5.3	5.4	5.4	5.4	4.4	3.4
	8000	7.3	7.3	7.3	7.2	7.2	7.2	7.1	6.0	4.9	3.8	2.8
	8500	6.5	6.5	6.5	6.5	6.4	6.4	6.4	5.3	4.3	3.3	2.3
	9000	5.8	5.8	5.8	5.8	5.7	5.7	5.7	4.7	3.7	2.7	1.9
10000	9500	5.2	5.2	5.2	5.1	5.1	5.1	5.0	4.1	3.1	2.3	1.4
	10000	4.6	4.6	4.6	4.5	4.5	4.5	4.4	3.6	2.6	1.8	1.0
	10500	4.0	4.0	4.0	4.0	3.9	3.9	3.9	3.0	2.2	1.4	0.6

1ERCG_UP_2_1_02

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Single Engine En Route Gross Climb Gradient [%]																
FLAPS UP, GEAR UP, Ice Protection On, MCT, 140 [KIAS]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	
15000	8000				5.4	5.4	5.4	5.4	5.5	4.5	3.4	2.4	1.5	0.7	-0.4	
	8500				4.8	4.8	4.8	4.8	4.8	3.9	2.9	1.9	1.1	0.3	-0.7	
	9000				4.1	4.2	4.2	4.2	4.2	3.3	2.4	1.5	0.7	-0.1	-1.0	
	9500				3.6	3.6	3.6	3.6	3.6	2.8	1.9	1.1	0.3	-0.4	-1.3	
	10000				3.1	3.1	3.1	3.1	3.1	2.3	1.5	0.7	-0.1	-0.7	-1.6	
20000	10500				2.6	2.6	2.6	2.6	2.6	1.9	1.0	0.3	-0.4	-1.0	-1.8	
	8000			4.2	4.2	4.2	3.3	2.4	1.4	0.5	-0.4	-1.1	-1.9			
	8500			3.6	3.6	3.6	2.8	1.9	1.0	0.1	-0.7	-1.4	-2.1			
	9000			3.1	3.1	3.1	2.3	1.5	0.6	-0.2	-1.0	-1.6	-2.3			
	9500			2.6	2.6	2.6	1.8	1.1	0.3	-0.5	-1.3	-1.9	-2.5			
25000	10000			2.1	2.1	2.1	1.4	0.7	-0.1	-0.9	-1.5	-2.1	-2.7			
	10500			1.7	1.7	1.6	1.0	0.3	-0.4	-1.2	-1.8	-2.4	-2.9			
	8000	3.4	3.4	3.0	2.3	1.6	0.8	0.0	-0.8	-1.5	-3.1					
	8500	2.9	2.9	2.5	1.8	1.1	0.4	-0.3	-1.1	-1.8	-3.2					
	9000	2.4	2.4	2.0	1.4	0.7	0.0	-0.6	-1.3	-2.0	-3.4					
TERGG_UP_2_2_02	9500	1.9	1.9	1.6	1.0	0.4	-0.3	-0.9	-1.6	-2.3	-3.5					
	10000	1.5	1.5	1.2	0.6	0.0	-0.6	-1.2	-1.9	-2.5	-3.7					
	10500	1.1	1.1	0.8	0.2	-0.3	-0.9	-1.5	-2.1	-2.7	-3.9					

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Single Engine En Route Rate of Climb [FPM]													
FLAPS UP, GEAR UP, Ice Protection On, MCT, 140 [KIAS]													
Alt [ft]	Wt [lb]	TEMPERATURE [°C]											
		-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	
SL	8000	1346	1361	1378	1396	1414	1431	1451	1470	1488	1505	1521	
	8500	1223	1237	1253	1269	1286	1301	1319	1337	1354	1369	1384	
	9000	1112	1124	1138	1153	1169	1183	1199	1216	1231	1246	1259	
	9500	1010	1021	1034	1048	1062	1075	1090	1105	1119	1132	1144	
	10000	915	926	938	950	963	975	989	1003	1016	1028	1039	
5000	10500	828	837	848	860	872	883	895	908	920	931	941	
	8000	1240	1256	1273	1290	1306	1322	1336	1350	1361	1175	991	
	8500	1119	1134	1150	1165	1179	1194	1206	1220	1230	1054	881	
	9000	1009	1023	1037	1051	1064	1077	1089	1101	1110	944	779	
	9500	909	921	934	946	959	970	981	992	1000	842	686	
10000	10000	815	827	838	850	861	872	881	891	898	748	599	
	10500	729	739	750	760	770	780	788	797	803	660	518	
	8000	1110	1122	1131	1138	1145	1150	1156	986	807	637	478	
	8500	992	1003	1012	1018	1023	1028	1033	873	703	543	392	
	9000	885	895	902	907	912	916	920	768	607	455	313	
10000	9500	786	795	801	806	810	813	816	672	519	374	239	
	10000	695	702	708	711	715	717	720	582	436	299	169	
	10500	609	616	621	624	626	628	630	498	359	227	104	

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Single Engine En Route Rate of Climb [FPM]																
FLAPS UP, GEAR UP, Ice Protection On, MCT, 140 [KIAS]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	5	10	
15000	8000				909	919	930	942	954	789	607	437	272	120	-74	
	8500				798	808	817	828	838	683	510	350	195	51	-133	
	9000				697	705	713	723	732	584	421	269	122	-15	-188	
	9500				603	610	617	625	634	493	338	193	53	-77	-242	
	10000				515	521	528	535	542	408	260	122	-12	-135	-293	
20000	10500				433	438	444	450	457	328	187	55	-73	-192	-342	
	8000			774	778	784	630	459	280	97	-71	-218	-372			
	8500			666	669	674	529	367	198	25	-133	-273	-418			
	9000			567	569	573	435	282	121	-42	-193	-325	-463			
	9500			475	476	479	348	202	49	-106	-250	-376	-507			
25000	10000			388	389	391	266	127	-19	-168	-305	-425	-550			
	10500			307	307	309	189	55	-84	-226	-357	-472	-592			
	8000	683	694	616	470	325	165	8	-163	-333	-671					
	8500	575	585	511	373	236	84	-64	-226	-387	-705					
	9000	476	485	414	283	153	9	-133	-286	-438	-740					
TERROC_UP_2_2_02	9500	384	391	323	198	74	-63	-197	-343	-489	-775					
	10000	297	303	238	118	0	-131	-260	-399	-538	-810					
	10500	215	221	158	43	-71	-196	-319	-453	-585	-846					

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Gross Climb Gradient [%]																	
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 210 / 0.57 [KIAS / MACH]																	
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														35	40
		-40	-20	-15	-10	-5	0	5	10	15	20	25	30				
SL	8000	23.6	23.6	23.7	23.7	23.8	23.9	23.9	24.0	24.1	24.1	24.2	22.2	20.2	18.0		
	8500	22.0	22.1	22.1	22.2	22.2	22.3	22.4	22.4	22.5	22.6	22.6	20.7	18.8	16.8		
	9000	20.7	20.7	20.7	20.8	20.8	20.9	21.0	21.1	21.1	21.2	21.2	19.4	17.6	15.7		
	9500	19.4	19.5	19.5	19.5	19.6	19.7	19.7	19.8	19.9	19.9	19.9	18.3	16.6	14.8		
	10000	18.3	18.4	18.4	18.4	18.5	18.5	18.6	18.7	18.7	18.8	18.8	17.2	15.6	13.9		
5000	10500	17.3	17.3	17.4	17.4	17.4	17.5	17.6	17.6	17.7	17.7	17.8	16.2	14.7	13.1		
	8000	20.4	20.7	20.8	20.8	20.8	20.8	20.9	20.9	20.9	18.8	16.9	15.3				
	8500	19.0	19.3	19.4	19.4	19.5	19.5	19.5	19.5	19.5	17.6	15.8	14.2				
	9000	17.8	18.1	18.2	18.2	18.2	18.2	18.3	18.3	18.3	16.5	14.8	13.3				
	9500	16.8	17.0	17.1	17.1	17.1	17.1	17.2	17.2	17.2	15.5	13.9	12.4				
10000	10000	15.8	16.0	16.1	16.1	16.1	16.1	16.2	16.2	16.2	14.5	13.0	11.7				
	10500	14.9	15.1	15.2	15.2	15.2	15.2	15.2	15.3	15.3	13.7	12.2	11.0				
	8000	17.4	17.5	17.4	17.4	17.3	17.3	17.3	15.6	13.8	12.2						
	8500	16.3	16.3	16.3	16.2	16.1	16.1	16.1	14.5	12.9	11.4						
	9000	15.2	15.3	15.2	15.2	15.1	15.1	15.1	13.6	12.0	10.6						
15000	9500	14.3	14.3	14.3	14.2	14.2	14.1	14.2	12.7	11.2	9.9						
	10000	13.4	13.5	13.4	13.4	13.3	13.3	13.3	11.9	10.5	9.3						
	10500	12.6	12.7	12.6	12.6	12.5	12.5	12.5	11.2	9.9	8.7						
	8000	14.0	14.0	14.0	14.0	14.1	12.6	11.2	9.8								
	8500	13.0	13.0	13.0	13.1	13.1	11.8	10.4	9.1								
	9000	12.2	12.1	12.2	12.2	12.2	11.0	9.7	8.5								
	9500	11.4	11.4	11.4	11.4	11.4	10.2	9.1	7.9								
	10000	10.7	10.6	10.7	10.7	10.7	9.6	8.5	7.3								
	10500	10.0	10.0	10.0	10.0	10.1	9.0	7.9	6.9								

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Gross Climb Gradient [%]																
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 210 / 0.57 [KIAS / MACH]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-65	-60	-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	
20000	8000					12.0	12.0	12.0	12.1	12.1	12.0	10.5	9.1	7.8	6.6	
	8500					11.1	11.1	11.2	11.2	11.2	11.1	9.8	8.5	7.2	6.0	
	9000					10.4	10.4	10.4	10.4	10.5	10.3	9.1	7.9	6.7	5.6	
	9500					9.7	9.7	9.7	9.7	9.8	9.7	8.5	7.3	6.2	5.1	
	10000					9.0	9.1	9.1	9.1	9.1	9.0	7.9	6.8	5.7	4.7	
25000	10500					8.5	8.5	8.5	8.5	8.6	8.5	7.4	6.3	5.3	4.4	
	8000			10.9	11.0	11.0	11.1	11.1	10.0	8.9	7.7	6.6	5.5			
	8500			10.2	10.2	10.2	10.3	10.3	9.2	8.2	7.1	6.1	5.0			
	9000			9.5	9.5	9.5	9.5	9.6	8.6	7.6	6.6	5.6	4.6			
	9500			8.8	8.8	8.9	8.9	8.9	8.0	7.1	6.1	5.1	4.2			
30000	10000			8.2	8.3	8.3	8.3	8.4	7.4	6.6	5.7	4.7	3.9			
	10500			7.7	7.7	7.8	7.8	7.8	6.9	6.1	5.2	4.4	3.5			
	8000	8.2	8.2	8.3	8.3	8.3	7.6	6.4	5.4	4.4	3.5					
	8500	7.6	7.6	7.6	7.6	7.7	7.0	5.9	4.9	4.0	3.2					
	9000	7.0	7.0	7.1	7.1	7.1	6.5	5.5	4.5	3.7	2.9					
	9500	6.5	6.5	6.6	6.6	6.6	6.0	5.0	4.1	3.3	2.6					
	10000	6.0	6.1	6.1	6.1	6.1	5.5	4.6	3.8	3.0	2.3					
	10500	5.6	5.6	5.7	5.7	5.7	5.1	4.3	3.5	2.7	2.1					

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Gross Climb Gradient [%]													
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 210 / 0.57 [KIAS / MACH]													
Alt [ft]	Wt [lb]	TEMPERATURE [°C]											
		-75	-70	-65	-60	-55	-50	-45	-40	-35	-30		
35000	8000	8.0	8.0	8.0	8.0	7.8	7.0	6.1	5.1	4.1	3.2		
	8500	7.3	7.3	7.3	7.3	7.1	6.4	5.6	4.6	3.7	2.8		
	9000	6.7	6.7	6.7	6.7	6.5	5.8	5.0	4.1	3.3	2.4		
	9500	6.1	6.1	6.1	6.1	6.0	5.3	4.6	3.7	2.9	2.1		
	10000	5.6	5.6	5.6	5.6	5.4	4.8	4.1	3.3	2.5	1.7		
40000	10500	5.1	5.1	5.1	5.1	5.0	4.4	3.7	2.9	2.2	1.4		
	8000	5.2	5.2	5.2	5.2	4.8	4.1	3.2	2.4	1.5	0.7		
	8500	4.6	4.7	4.7	4.6	4.3	3.6	2.7	2.0	1.2	0.4		
	9000	4.1	4.1	4.2	4.1	3.8	3.1	2.3	1.6	0.9	0.1		
	9500	3.6	3.7	3.7	3.6	3.3	2.7	1.9	1.3	0.6	-0.2		
43000	10000	3.2	3.2	3.2	3.2	2.9	2.3	1.6	0.9	0.3	-0.4		
	10500	2.8	2.8	2.8	2.8	2.5	1.9	1.2	0.6	0.0	-0.6		
	8000	3.6	3.6	3.6	3.5	3.1	2.4	1.7	0.9	0.1	-0.7		
	8500	3.1	3.1	3.1	3.0	2.6	1.9	1.3	0.6	-0.2	-1.0		
	9000	2.6	2.6	2.6	2.5	2.2	1.5	0.9	0.3	-0.5	-1.2		
43000	9500	2.2	2.2	2.2	2.1	1.8	1.2	0.6	-0.1	-0.7	-1.5		
	10000	1.8	1.8	1.8	1.7	1.4	0.8	0.3	-0.4	-1.0	-1.7		
	10500	1.4	1.4	1.4	1.3	1.0	0.5	-0.1	-0.6	-1.3	-1.9		

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

		All Engine En Route Rate of Climb [FPM]																
		FLAPS UP, GEAR UP, Ice Protection Off, MCT, 210 / 0.57 [KIAS / MACH]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]																
		-40	-20	-15	-10	-5	0	5	10	15	20	25	30	35	40			
SL	8000	4510	4713	4762	4818	4877	4941	5003	5062	5120	5179	5230	4837	4434	3998			
	8500	4217	4406	4452	4504	4560	4620	4678	4733	4788	4843	4890	4521	4141	3730			
	9000	3955	4132	4176	4224	4277	4333	4388	4440	4491	4543	4587	4238	3879	3490			
	9500	3719	3886	3926	3972	4022	4074	4126	4175	4223	4272	4314	3983	3643	3274			
	10000	3505	3662	3700	3744	3790	3840	3889	3935	3981	4027	4066	3752	3428	3078			
5000	10500	3310	3458	3495	3536	3580	3627	3673	3717	3760	3804	3841	3541	3233	2899			
	8000	4266	4513	4570	4622	4674	4720	4771	4823	4863	4423	4008	3639					
	8500	3985	4216	4269	4318	4366	4409	4457	4506	4543	4128	3738	3390					
	9000	3733	3950	4000	4046	4091	4131	4176	4222	4257	3865	3496	3167					
	9500	3506	3710	3757	3801	3843	3881	3923	3966	3999	3627	3278	2966					
10000	10000	3300	3493	3537	3578	3618	3654	3694	3734	3765	3411	3079	2782					
	10500	3112	3295	3336	3375	3413	3447	3484	3523	3552	3215	2898	2615					
	8000	4000	4181	4209	4235	4260	4292	4334	3941	3526	3152							
	8500	3732	3900	3926	3950	3974	4003	4043	3673	3282	2929							
	9000	3491	3649	3673	3695	3718	3745	3782	3432	3063	2729							
15000	9500	3274	3422	3445	3466	3486	3512	3546	3215	2865	2548							
	10000	3077	3216	3237	3257	3276	3300	3333	3017	2684	2383							
	10500	2897	3029	3048	3066	3084	3107	3138	2837	2519	2232							
	8000	3531	3675	3717	3763	3810	3454	3099	2739									
	8500	3287	3421	3460	3503	3547	3211	2877	2538									
	9000	3068	3193	3230	3270	3311	2993	2677	2357									
	9500	2871	2987	3022	3059	3098	2797	2497	2193									
	10000	2691	2800	2832	2867	2904	2617	2332	2043									
	10500	2526	2629	2659	2692	2727	2453	2181	1905									

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Rate of Climb [FPM]																
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 210 / 0.57 [KIAS / MACH]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-65	-60	-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	
20000	8000					3292	3338	3382	3426	3470	3465	3087	2705	2342	1977	
	8500					3059	3102	3143	3184	3225	3220	2863	2503	2161	1818	
	9000					2850	2890	2929	2967	3005	3000	2662	2322	1999	1674	
	9500					2661	2698	2734	2770	2806	2800	2480	2158	1851	1543	
	10000					2488	2523	2557	2591	2624	2619	2315	2007	1716	1422	
25000	10500					2330	2363	2395	2426	2458	2453	2162	1869	1591	1311	
	8000			3259	3307	3354	3403	3453	3131	2811	2476	2134	1794			
	8500			3025	3070	3114	3159	3206	2902	2601	2285	1963	1642			
	9000			2815	2857	2898	2940	2984	2697	2411	2112	1808	1504			
	9500			2625	2664	2702	2742	2783	2510	2239	1956	1667	1379			
30000	10000			2451	2488	2524	2561	2599	2340	2082	1812	1538	1264			
	10500			2292	2327	2360	2395	2431	2184	1938	1681	1418	1157			
	8000	2644	2689	2727	2762	2799	2591	2223	1883	1562	1258					
	8500	2443	2485	2520	2553	2587	2391	2044	1723	1421	1134					
	9000	2262	2301	2334	2365	2396	2210	1882	1579	1293	1021					
	9500	2098	2134	2165	2193	2222	2046	1734	1447	1176	918					
	10000	1948	1982	2010	2037	2064	1896	1599	1325	1067	822					
	10500	1810	1842	1868	1893	1918	1757	1474	1213	967	733					

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-225

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Rate of Climb [FPM]												
FLAPS UP, GEAR UP, Ice Protection Off, MCT, 210 / 0.57 [KIAS / MACH]												
Alt [ft]	Wt [lb]	TEMPERATURE [°C]										
		-75	-70	-65	-60	-55	-50	-45	-40	-35	-30	
35000	8000	2526	2560	2593	2625	2589	2364	2088	1759	1435	1108	
	8500	2310	2341	2372	2401	2366	2153	1893	1583	1277	968	
	9000	2115	2143	2171	2198	2164	1962	1716	1422	1132	840	
	9500	1936	1962	1987	2012	1979	1787	1553	1274	998	721	
	10000	1771	1795	1819	1841	1809	1626	1403	1136	874	610	
40000	10500	1619	1641	1663	1683	1652	1477	1263	1009	758	506	
	8000	1647	1674	1701	1706	1606	1361	1079	819	534	244	
	8500	1468	1493	1517	1520	1425	1195	928	682	413	139	
	9000	1304	1327	1349	1351	1260	1041	788	555	300	40	
	9500	1154	1174	1193	1194	1108	899	658	437	194	-53	
43000	10000	1014	1032	1049	1049	966	767	537	326	94	-141	
	10500	883	899	915	914	833	643	423	221	0	-226	
	8000	1137	1153	1167	1156	1035	802	577	322	42	-254	
	8500	975	989	1002	990	875	654	442	200	-64	-344	
	9000	826	838	849	837	726	517	316	86	-164	-430	
	9500	688	698	707	694	589	389	197	-21	-260	-512	
	10000	558	567	574	561	460	269	85	-123	-351	-592	
	10500	437	444	449	436	338	155	-21	-221	-438	-669	

2ERROC_UP_0_3_05

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HA-420 AFM

PERFORMANCE

All Engine En Route Gross Climb Gradient [%]													
FLAPS UP, GEAR UP, Ice Protection On, MCT, 210/0.57 [KIAS/MACH]													
Alt [ft]	Wt [lb]	TEMPERATURE [°C]											
		-45	-40	-35	-30	-25	-20	-15	-10	-5	0	5	10
SL	8000		23.7	23.7	23.7	23.8	23.8	23.8	23.9	24.0	24.1	24.1	24.2
	8500		22.2	22.2	22.2	22.2	22.2	22.3	22.3	22.4	22.5	22.6	22.6
	9000		20.8	20.8	20.8	20.8	20.9	20.9	20.9	21.0	21.1	21.2	21.2
	9500		19.5	19.5	19.6	19.6	19.6	19.6	19.7	19.8	19.8	19.9	19.9
	10000		18.4	18.4	18.4	18.5	18.5	18.5	18.6	18.6	18.7	18.8	18.8
	10500		17.4	17.4	17.4	17.4	17.4	17.5	17.5	17.6	17.7	17.7	17.8
5000	8000		20.6	20.7	20.7	20.8	20.9	20.9	21.0	21.0	21.0	21.1	19.9
	8500		19.2	19.3	19.4	19.5	19.5	19.6	19.6	19.6	19.7	19.7	18.6
	9000		18.0	18.1	18.2	18.2	18.3	18.3	18.4	18.4	18.4	18.4	17.4
	9500		16.9	17.0	17.0	17.1	17.2	17.2	17.2	17.3	17.3	17.3	16.3
	10000		15.9	16.0	16.0	16.1	16.2	16.2	16.2	16.3	16.3	16.3	15.4
	10500		15.0	15.1	15.1	15.2	15.2	15.3	15.3	15.3	15.4	15.4	14.5
10000	8000		17.6	17.6	17.7	17.7	17.6	17.6	17.5	17.4	16.5	14.5	12.7
	8500		16.4	16.5	16.5	16.5	16.4	16.4	16.3	16.3	15.4	13.5	11.8
	9000		15.4	15.4	15.4	15.4	15.4	15.3	15.3	15.2	14.4	12.6	11.0
	9500		14.4	14.4	14.5	14.5	14.4	14.4	14.3	14.3	13.5	11.8	10.3
	10000		13.5	13.6	13.6	13.6	13.5	13.5	13.5	13.4	12.7	11.1	9.6
	10500		12.8	12.8	12.8	12.8	12.8	12.7	12.7	12.6	11.9	10.4	9.0
15000	8000		14.1	14.1	14.1	14.1	14.1	14.1	13.5	11.9	10.3	8.7	7.3
	8500		13.1	13.1	13.1	13.1	13.1	13.1	12.5	11.0	9.5	8.1	6.8
	9000		12.3	12.2	12.2	12.2	12.2	12.3	11.7	10.3	8.9	7.5	6.2
	9500		11.5	11.4	11.4	11.4	11.5	11.5	10.9	9.6	8.2	7.0	5.8
	10000		10.8	10.7	10.7	10.7	10.7	10.8	10.2	8.9	7.7	6.5	5.3
	10500		10.1	10.1	10.1	10.1	10.1	10.1	9.6	8.4	7.2	6.0	4.9
20000	8000	12.1	12.1	12.1	12.2	11.4	9.8	8.3	6.9	5.5	4.3		
	8500	11.2	11.2	11.3	11.3	10.6	9.1	7.7	6.3	5.1	3.9		
	9000	10.5	10.5	10.5	10.5	9.8	8.5	7.1	5.8	4.7	3.6		
	9500	9.8	9.8	9.8	9.8	9.2	7.9	6.6	5.4	4.3	3.2		
	10000	9.1	9.1	9.2	9.2	8.6	7.3	6.1	5.0	3.9	2.9		
	10500	8.5	8.6	8.6	8.6	8.0	6.8	5.7	4.6	3.6	2.6		

2ERCG_UP_2_1_02

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October 30, 2016

HJ1-29000-003-001

Page 5-227

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HA-420 AFM

PERFORMANCE

All Engine En Route Gross Climb Gradient [%]														
FLAPS UP, GEAR UP, Ice Protection On, MCT, 210/0.57 [KIAS/MACH]														
Alt [ft]	Wt [lb]	TEMPERATURE [°C]												
		-75	-65	-60	-55	-50	-45	-40	-35	-30	-25	-20	-15	-10
25000	8000				11.0	11.1	11.1	10.7	9.4	8.3	7.1	5.9	4.6	3.4
	8500				10.2	10.3	10.3	10.0	8.7	7.7	6.5	5.4	4.2	3.1
	9000				9.5	9.6	9.6	9.3	8.1	7.1	6.0	4.9	3.8	2.7
	9500				8.9	8.9	8.9	8.6	7.5	6.6	5.6	4.5	3.4	2.5
	10000				8.3	8.3	8.4	8.1	7.0	6.1	5.1	4.2	3.1	2.2
	10500				7.8	7.8	7.8	7.5	6.6	5.7	4.7	3.8	2.8	1.9
30000	8000		8.2	8.3	8.3	8.0	7.1	5.9	4.8	3.8	2.8	1.8		
	8500		7.6	7.7	7.7	7.4	6.6	5.4	4.4	3.5	2.5	1.6		
	9000		7.1	7.1	7.1	6.9	6.1	5.0	4.0	3.1	2.2	1.3		
	9500		6.5	6.6	6.6	6.4	5.6	4.6	3.7	2.8	2.0	1.1		
	10000		6.1	6.1	6.1	5.9	5.2	4.2	3.3	2.5	1.7	0.9		
	10500		5.6	5.7	5.7	5.5	4.8	3.9	3.0	2.3	1.5	0.7		
35000	8000	8.0	8.0	7.6	6.9	5.8	4.8	3.7	2.7	1.6				
	8500	7.3	7.3	7.0	6.2	5.3	4.3	3.3	2.3	1.3				
	9000	6.7	6.7	6.4	5.7	4.8	3.9	2.9	1.9	1.0				
	9500	6.1	6.1	5.8	5.2	4.3	3.4	2.5	1.6	0.7				
	10000	5.6	5.6	5.3	4.7	3.9	3.1	2.2	1.3	0.5				
	10500	5.1	5.1	4.9	4.3	3.5	2.7	1.9	1.1	0.2				
40000	8000	5.2	5.1	4.6	3.7	2.8	1.9	0.9	-0.5	-1.6				
	8500	4.7	4.6	4.1	3.3	2.4	1.5	0.5	-0.7	-1.8				
	9000	4.1	4.0	3.6	2.8	2.0	1.2	0.3	-0.9	-2.0				
	9500	3.7	3.6	3.1	2.4	1.6	0.8	0.0	-1.1	-2.1				
	10000	3.2	3.1	2.7	2.0	1.3	0.5	-0.3	-1.3	-2.3				
	10500	2.8	2.7	2.3	1.7	1.0	0.3	-0.5	-1.5	-2.4				
43000	8000	3.6	3.4	2.8	2.0	1.1	-0.1	-1.0	-2.1	-2.6				
	8500	3.1	2.9	2.3	1.6	0.7	-0.3	-1.3	-1.6	-2.4				
	9000	2.6	2.5	1.9	1.2	0.4	-0.6	-1.5	-1.8	-2.6				
	9500	2.2	2.0	1.5	0.8	0.1	-0.9	-1.7	-2.0	-2.7				
	10000	1.8	1.6	1.1	0.5	-0.2	-1.1	-1.9	-2.2	-2.9				
	10500	1.4	1.3	0.8	0.2	-0.5	-1.4	-2.1	-2.4	-3.0				

2ERCG_UP_2_2_02

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Rate of Climb [FPM]																
FLAPS UP, GEAR UP, Ice Protection On, MCT, 210 / 0.57 [KIAS / MACH]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-40	-35	-30	-25	-20	-15	-10	-5	0	5	10				
SL	8000	4512	4563	4614	4667	4718	4769	4831	4895	4957	5018	5075				
	8500	4218	4266	4314	4363	4411	4459	4517	4577	4635	4692	4745				
	9000	3956	4001	4046	4092	4136	4182	4236	4293	4347	4401	4451				
	9500	3720	3761	3804	3847	3889	3932	3983	4036	4088	4138	4185				
	10000	3505	3545	3585	3626	3665	3706	3754	3804	3853	3901	3945				
5000	10500	3310	3347	3385	3424	3461	3500	3545	3593	3639	3684	3726				
	8000	4284	4348	4410	4473	4532	4586	4640	4692	4742	4789	4831				
	8500	4001	4061	4120	4178	4233	4284	4335	4383	4430	4474	4517				
	9000	3749	3804	3860	3914	3966	4014	4061	4107	4151	4192	4236				
	9500	3521	3573	3625	3677	3725	3771	3815	3858	3899	3938	3979				
10000	10000	3314	3363	3412	3461	3507	3550	3591	3632	3671	3707	3745				
	10500	3125	3172	3218	3265	3308	3348	3388	3426	3463	3497	3536				
	8000	4021	4073	4117	4159	4189	4217	4243	4268	4288	4308	4326				
	8500	3751	3800	3841	3880	3908	3934	3958	3981	3998	4015	4031				
	9000	3509	3555	3593	3630	3656	3680	3703	3724	3742	3758	3774				
15000	9500	3291	3334	3370	3404	3429	3451	3472	3492	3508	3523	3538				
	10000	3093	3133	3167	3199	3222	3244	3263	3281	3297	3313	3328				
	10500	2912	2950	2982	3012	3034	3054	3072	3089	3105	3121	3137				
	8000	3544	3572	3606	3646	3687	3730	3769	3809	3848	3886	3923				
	8500	3299	3325	3357	3394	3432	3472	3509	3546	3582	3617	3652				
20000	9000	3079	3103	3133	3167	3204	3241	3279	3316	3352	3388	3423				
	9500	2881	2903	2931	2963	2997	3032	3067	3101	3135	3168	3201				
	10000	2700	2721	2747	2777	2809	2842	2874	2905	2936	2966	2996				
	10500	2535	2554	2579	2607	2637	2668	2697	2726	2754	2781	2808				
	10500	2535	2554	2579	2607	2637	2668	2697	2726	2754	2781	2808				

2ERROC_UP_2_1_02

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Rate of Climb [FPM]																
FLAPS UP, GEAR UP, Ice Protection On, MCT, 210 / 0.57 [KIAS / MACH]																
Alt [ft]	Wt [lb]	TEMPERATURE [°C]														
		-65	-60	-55	-50	-45	-40	-35	-30	-25	-20	-15	-10	-5	0	
20000	8000					3308	3352	3396	3440	3255	2838	2433	2029	1648	1292	
	8500					3074	3115	3156	3197	3023	2629	2248	1867	1508	1173	
	9000					2864	2902	2940	2978	2814	2442	2081	1721	1381	1065	
	9500					2674	2709	2745	2781	2624	2271	1929	1588	1266	965	
	10000					2500	2533	2567	2601	2452	2116	1791	1466	1160	874	
25000	10500					2342	2373	2404	2436	2294	1974	1663	1353	1061	789	
	8000			3272	3318	3366	3288	2926	2592	2237	1871	1476	1108			
	8500			3037	3080	3124	3051	2710	2395	2060	1715	1343	996			
	9000			2826	2866	2907	2837	2515	2217	1900	1574	1222	894			
	9500			2635	2672	2711	2644	2338	2056	1755	1446	1112	800			
30000	10000			2461	2495	2532	2468	2177	1908	1622	1328	1010	714			
	10500			2301	2334	2368	2307	2029	1773	1499	1219	916	633			
	8000	2649	2692	2730	2669	2397	2017	1661	1326	986	646					
	8500	2447	2488	2523	2465	2209	1850	1515	1199	878	558					
	9000	2266	2303	2336	2281	2038	1699	1382	1083	780	476					
	9500	2101	2136	2166	2114	1883	1562	1261	977	689	401					
	10000	1951	1983	2011	1961	1741	1435	1149	879	605	331					
	10500	1812	1843	1869	1820	1611	1319	1045	788	526	265					

2ERROC_UP_2_2_02

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

All Engine En Route Rate of Climb [FPM]												
FLAPS UP, GEAR UP, Ice Protection On, MCT, 210 / 0.57 [KIAS / MACH]												
Alt [ft]	Wt [lb]	TEMPERATURE [°C]										
		-75	-70	-65	-60	-55	-50	-45	-40	-35	-30	
35000	8000	2531	2564	2595	2509	2277	1965	1630	1276	923	545	
	8500	2315	2345	2374	2292	2072	1778	1462	1128	795	438	
	9000	2119	2146	2173	2095	1886	1608	1309	992	677	339	
	9500	1939	1965	1989	1914	1716	1452	1167	867	567	247	
	10000	1775	1798	1820	1749	1559	1307	1036	750	464	159	
40000	10500	1623	1644	1664	1595	1414	1173	914	641	368	77	
	8000	1648	1674	1658	1505	1240	943	639	293	-164	-576	
	8500	1469	1493	1477	1331	1081	800	513	187	-244	-632	
	9000	1306	1327	1311	1172	935	669	397	88	-320	-688	
	9500	1155	1174	1158	1025	799	546	288	-6	-394	-743	
43000	10000	1015	1032	1015	889	673	432	185	-95	-464	-797	
	10500	884	899	883	761	555	324	88	-180	-532	-850	
	8000	1129	1146	1111	923	658	365	-17	-347	-730	-902	
	8500	968	982	949	771	521	243	-118	-430	-566	-845	
	9000	820	832	799	630	392	129	-213	-508	-636	-900	
	9500	682	692	660	498	272	22	-304	-585	-705	-955	
	10000	553	561	529	375	159	-80	-390	-658	-772	-1010	
	10500	431	438	407	258	51	-178	-474	-730	-837	-1065	

2ERROC_UP_2_3_02

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

APPROACH AND LANDING

Use of approach and landing information tables:

1. Determine the airplane gross weight and expected airplane configuration based on icing conditions at the time of landing.
2. Obtain airport information:
 - active runway
 - available runway length
 - runway gradient
 - ambient temperature
 - Determine that the temperature is within the ambient temperature limits found in the limitations section.
 - pressure altitude
 - wind
 - Determine the wind component parallel to active runway from the crosswind component chart on page 24.
3. Check the maximum landing weight permitted by climb requirement for the planned airplane configuration on page 234. If the limitations restrict the landing gross weight, the pilot must burn fuel prior to landing.
4. Using the landing gross weight determined in step 3, determine V_{REF} and the uncorrected landing field length on pages 236 – 251.
5. For runway gradients and wind, the landing field length must be corrected using the correction tables on pages 239 – 253.
6. If the available runway length is less than the required landing field length, the airplane weight must be reduced until this requirement can be met.
7. Approach climb and landing climb gradient tables are provided on pages 242 – 259.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Use of approach and landing information tables (continued):

NOTE Landing distances include a 15 % factor to account for operational factors including crosswind and landing technique. If required, use the Operational Factors Required for Landing Distance section to remove or increase the factor based on the operational rules required.

NOTE Landing data for 10,600 lbs are provided for reference and interpolation only in case an emergency landing is required.

NOTE The landing field length provided in the performance section of the flight manual is based on a dry runway. If landing on a wet runway, it is recommended to increase the predicted landing field length by 30 %.

NOTE Ice Protection ON assumes both ENG and WING A/I are ON. If only ENG A/I is on, the Ice Protection ON tables must be used.

NOTE The landing field length wind and runway slope correction tables are valid for Ice Protection Off and On.

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Maximum Landing Weight – Climb and Brake Energy Limited

For reference, data has been provided for weights over the maximum landing weight.

Red shading on landing tables indicates conditions where the airplane does not meet the climb requirements, but can be used for interpolation.

Example:

Ambient Conditions:

Temperature	30 °C
Airport Altitude	926 ft
Wind	10 kts Tailwind
Runway Gradient	-1.6 %

Aircraft Configuration:

Flaps	LDG
Weight	9860 lbs

Bleed Setting:

Ice Protection	Off
----------------	-----

Using the Weight Limit Chart:

Weight Limit	Not Limited
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Using the Tables:

Uncorrected

- V_{REF} 112 KIAS
- Field Length 3716 ft
- Approach Gradient 5.7 %

Wind Correction:

- Field Length 4262 ft

Slope Correction:

- Field Length: 6016 ft

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October 30, 2016

HJ1-29000-003-001

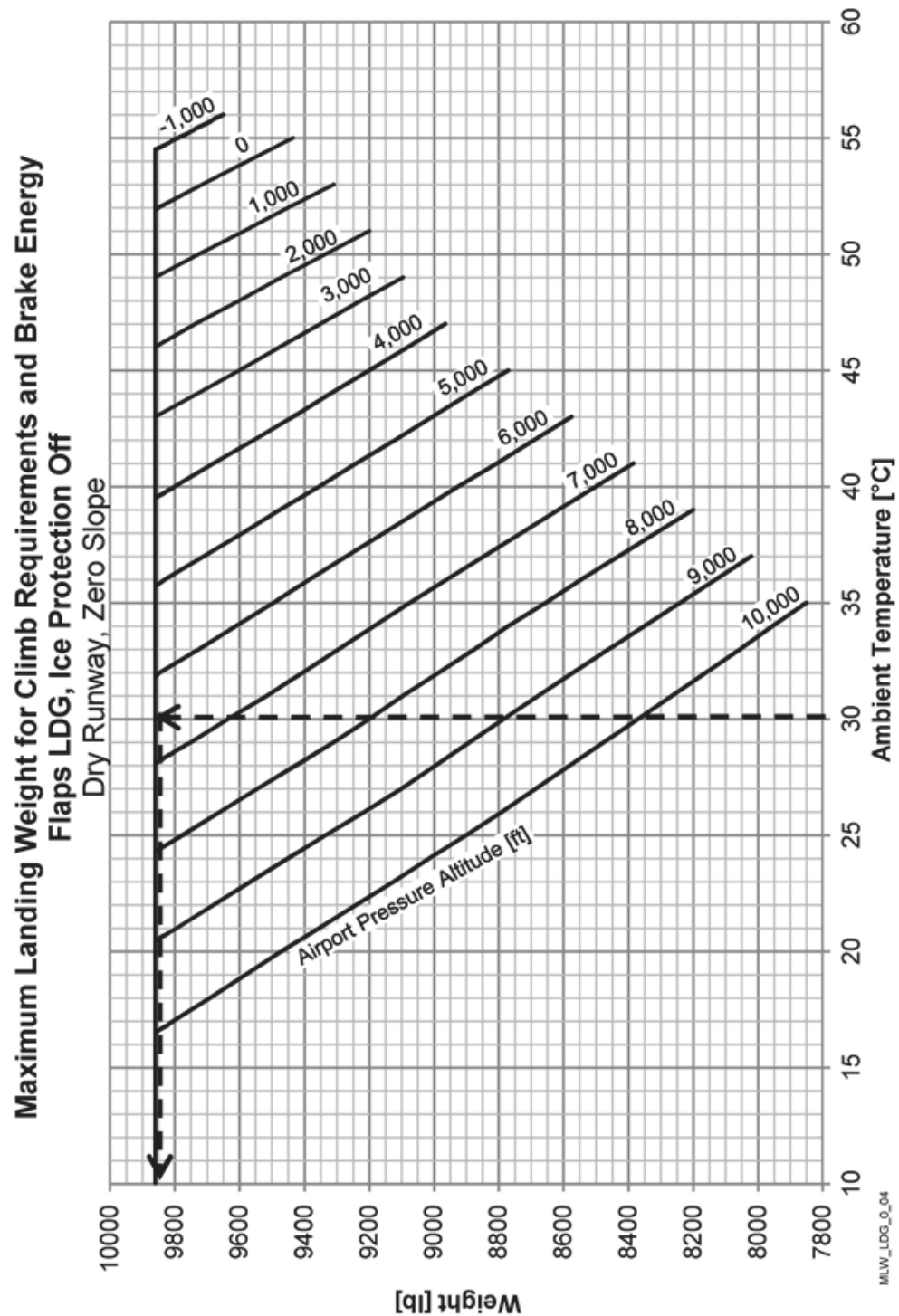
Page 5-234

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-235

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Landing Field Length [feet]								
Dry Runway, Zero Slope, No Wind								
Flaps LDG, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
-1000	-40	2539	2602	2703	2802	2900	2970	3112
	15	2906	2981	3101	3219	3335	3418	3587
	25	2973	3051	3173	3294	3414	3500	3674
	35	3041	3120	3246	3371	3494	3581	3760
	45	3108	3190	3319	3446	3573	3663	3846
	50	3141	3224	3355	3484	3612	3703	3889
	55	3174	3258	3390	3521	3651	3743	3931
Sea Level	-40	2596	2662	2765	2867	2968	3040	3186
	15	2976	3054	3177	3298	3418	3504	3678
	25	3046	3126	3252	3377	3500	3588	3767
	35	3116	3198	3327	3455	3582	3672	3856
	45	3185	3269	3402	3534	3663	3756	3945
	50	3219	3305	3439	3572	3704	3797	3989
	55	3253	3340	3476	3611	3744	3839	4032
1000	-40	2657	2724	2830	2935	3039	3113	3264
	10	3014	3093	3218	3341	3462	3549	3726
	20	3085	3167	3295	3422	3547	3636	3818
	30	3157	3241	3373	3503	3631	3723	3910
	40	3229	3315	3450	3583	3715	3810	4001
	45	3265	3352	3488	3624	3757	3853	4047
	50	3300	3388	3526	3663	3799	3895	4092
2000	-40	2721	2791	2900	3008	3115	3191	3346
	10	3090	3172	3300	3427	3552	3642	3823
	20	3165	3249	3381	3511	3640	3731	3919
	30	3239	3325	3461	3594	3727	3821	4013
	40	3313	3402	3540	3678	3813	3910	4108
	45	3349	3439	3580	3719	3856	3955	4155
	50	3386	3477	3619	3760	3899	3998	4201

LFL_LDG_0_-1_06

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October 30, 2016

HJ1-29000-003-001

Page 5-236

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Landing Field Length [feet]								
Dry Runway, Zero Slope, No Wind								
Flaps LDG, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
3000	-40	2788	2860	2973	3084	3194	3272	3432
	5	3132	3215	3345	3473	3600	3691	3876
	15	3208	3294	3428	3560	3691	3784	3974
	25	3285	3373	3510	3646	3781	3876	4072
	35	3362	3452	3593	3732	3870	3969	4170
	40	3399	3491	3634	3775	3915	4015	4218
	45	3437	3530	3674	3817	3959	4060	4266
4000	-40	2858	2933	3048	3163	3276	3357	3521
	0	3173	3258	3390	3520	3649	3741	3928
	10	3253	3340	3475	3610	3742	3837	4030
	20	3332	3422	3561	3699	3835	3933	4131
	35	3450	3543	3689	3832	3974	4076	4282
	40	3489	3584	3731	3876	4020	4123	4332
	45	3528	3623	3772	3919	4065	4169	4382
5000	-40	2931	3008	3127	3245	3362	3445	3614
	0	3258	3346	3481	3616	3748	3843	4036
	10	3340	3430	3570	3708	3845	3942	4141
	30	3503	3598	3746	3892	4036	4140	4350
	35	3544	3640	3790	3938	4084	4189	4402
	40	3584	3681	3833	3983	4131	4237	4453
	45	3623	3722	3876	4028	4178	4285	4504
6000	-40	3007	3087	3210	3332	3452	3537	3712
	-5	3303	3393	3531	3667	3801	3898	4094
	5	3388	3480	3622	3762	3901	4000	4202
	15	3473	3568	3714	3858	4001	4103	4311
	30	3599	3698	3850	4000	4149	4255	4472
	35	3640	3740	3894	4047	4198	4306	4525
	40	3682	3783	3939	4093	4247	4356	4578

LFL_LDG_0_3_05

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Landing Field Length [feet]								
Dry Runway, Zero Slope, No Wind								
Flaps LDG, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
7000	-40	3087	3169	3296	3421	3545	3633	3813
	-10	3349	3441	3580	3719	3855	3953	4152
	10	3525	3622	3770	3917	4062	4166	4377
	25	3656	3756	3911	4064	4216	4324	4544
	30	3699	3801	3958	4113	4266	4376	4600
	35	3742	3845	4004	4161	4317	4428	4655
	40	3784	3889	4050	4209	4367	4480	4709
8000	-40	3169	3254	3385	3514	3642	3733	3919
	-15	3396	3489	3631	3771	3910	4009	4211
	0	3532	3630	3778	3925	4071	4175	4386
	20	3713	3816	3973	4129	4283	4393	4617
	25	3758	3862	4021	4179	4336	4447	4675
	30	3802	3908	4070	4230	4388	4501	4732
	35	3846	3953	4117	4280	4440	4555	4789
9000	-40	3254	3343	3478	3611	3743	3837	4028
	-15	3490	3586	3733	3877	4021	4123	4331
	15	3771	3876	4036	4194	4351	4463	4691
	20	3817	3924	4086	4247	4406	4519	4751
	25	3863	3971	4136	4299	4460	4576	4810
	30	3909	4019	4186	4351	4514	4631	4869
	35	3955	4066	4235	4402	4568	4687	4928
10000	-40	3343	3435	3574	3711	3847	3944	4142
	-20	3539	3637	3785	3932	4078	4182	4393
	15	3878	3986	4152	4315	4477	4593	4828
	20	3925	4036	4203	4369	4533	4651	4890
	25	3973	4085	4255	4423	4590	4709	4951
	30	4021	4134	4306	4477	4645	4766	5012
	35	4068	4183	4357	4530	4701	4823	5073

LFL_LDG_0_7_05

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-238

FOR TRAINING PURPOSES ONLY

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HA-420 AFM

PERFORMANCE

Wind Corrected Landing Field Length [feet]								
Flaps LDG, Ice Protection Off								
Tailwind			Headwind					
-10	-5	◀ REF [0] ▶	5	10	15	20	25	30
2450	2230	2000	1950	1900	1840	1790	1730	1670
2556	2333	2100	2050	1998	1938	1886	1826	1766
2661	2436	2200	2149	2097	2035	1983	1922	1861
2767	2538	2300	2249	2195	2133	2079	2018	1957
2872	2641	2400	2348	2294	2230	2176	2114	2052
2978	2744	2500	2448	2392	2328	2272	2210	2148
3084	2847	2600	2548	2490	2426	2368	2306	2244
3189	2950	2700	2647	2589	2523	2465	2402	2339
3295	3052	2800	2747	2687	2621	2561	2498	2435
3400	3155	2900	2846	2786	2718	2658	2594	2530
3506	3258	3000	2946	2884	2816	2754	2690	2626
3612	3361	3100	3046	2982	2914	2850	2786	2722
3717	3464	3200	3145	3081	3011	2947	2882	2817
3823	3566	3300	3245	3179	3109	3043	2978	2913
3928	3669	3400	3344	3278	3206	3140	3074	3008
4034	3772	3500	3444	3376	3304	3236	3170	3104
4140	3875	3600	3544	3474	3402	3332	3266	3200
4245	3978	3700	3643	3573	3499	3429	3362	3295
4351	4080	3800	3743	3671	3597	3525	3458	3391
4456	4183	3900	3842	3770	3694	3622	3554	3486
4562	4286	4000	3942	3868	3792	3718	3650	3582
4668	4389	4100	4042	3966	3890	3814	3746	3678
4773	4492	4200	4141	4065	3987	3911	3842	3773
4879	4594	4300	4241	4163	4085	4007	3938	3869
4984	4697	4400	4340	4262	4182	4104	4034	3964
5090	4800	4500	4440	4360	4280	4200	4130	4060
5196	4903	4600	4540	4458	4378	4296	4226	4156
5301	5006	4700	4639	4557	4475	4393	4322	4251
5407	5108	4800	4739	4655	4573	4489	4418	4347
5512	5211	4900	4838	4754	4670	4586	4514	4442
5618	5314	5000	4938	4852	4768	4682	4610	4538

LWC_LDG_0_06

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October 30, 2016

HJ1-29000-003-001

Page 5-239

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Landing Field Length [feet]								
Flaps LDG, Ice Protection Off								
Runway Gradient [%]								
-2.0	-1.5	-1.0	-0.5	◀ REF [0] ▶	0.5	1.0	1.5	2.0
2010	1957	1834	1726	1600	1526	1432	1379	1300
2197	2100	1955	1835	1700	1619	1521	1464	1383
2385	2244	2077	1943	1800	1713	1611	1550	1465
2572	2387	2198	2052	1900	1806	1700	1635	1548
2760	2530	2320	2160	2000	1900	1790	1720	1630
2948	2673	2442	2268	2100	1994	1880	1805	1712
3135	2816	2563	2377	2200	2087	1969	1890	1795
3323	2960	2685	2485	2300	2181	2059	1976	1877
3510	3103	2806	2594	2400	2274	2148	2061	1960
3698	3246	2928	2702	2500	2368	2238	2146	2042
3886	3389	3050	2810	2600	2462	2328	2231	2124
4073	3532	3171	2919	2700	2555	2417	2316	2207
4261	3676	3293	3027	2800	2649	2507	2402	2289
4448	3819	3414	3136	2900	2742	2596	2487	2372
4636	3962	3536	3244	3000	2836	2686	2572	2454
4824	4105	3658	3352	3100	2930	2776	2657	2536
5011	4248	3779	3461	3200	3023	2865	2742	2619
5199	4392	3901	3569	3300	3117	2955	2828	2701
5386	4535	4022	3678	3400	3210	3044	2913	2784
5574	4678	4144	3786	3500	3304	3134	2998	2866
5762	4821	4266	3894	3600	3398	3224	3083	2948
5949	4964	4387	4003	3700	3491	3313	3168	3031
6137	5108	4509	4111	3800	3585	3403	3254	3113
6324	5251	4630	4220	3900	3678	3492	3339	3196
6512	5394	4752	4328	4000	3772	3582	3424	3278
6700	5537	4874	4436	4100	3866	3672	3509	3360
6887	5680	4995	4545	4200	3959	3761	3594	3443
7075	5824	5117	4653	4300	4053	3851	3680	3525
7262	5967	5238	4762	4400	4146	3940	3765	3608
7450	6110	5360	4870	4500	4240	4030	3850	3690
7638	6253	5482	4978	4600	4334	4120	3935	3772

LSC1_LDG_0_06

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-240

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Slope Corrected Landing Field Length [feet]								
Flaps LDG, Ice Protection Off								
Runway Gradient [%]								
-2.0	-1.5	-1.0	-0.5	◀ REF [0] ▶	0.5	1.0	1.5	2.0
7638	6253	5482	4978	4600	4334	4120	3935	3772
7825	6396	5603	5087	4700	4427	4209	4020	3855
8013	6540	5725	5195	4800	4521	4299	4106	3937
8200	6683	5846	5304	4900	4614	4388	4191	4020
8388	6826	5968	5412	5000	4708	4478	4276	4102
8576	6969	6090	5520	5100	4802	4568	4361	4184
8763	7112	6211	5629	5200	4895	4657	4446	4267
8951	7256	6333	5737	5300	4989	4747	4532	4349
9138	7399	6454	5846	5400	5082	4836	4617	4432
9326	7542	6576	5954	5500	5176	4926	4702	4514
9514	7685	6698	6062	5600	5270	5016	4787	4596
9701	7828	6819	6171	5700	5363	5105	4872	4679

LSC2_LDg_0_06

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October 30, 2016

HJ1-29000-003-001

Page 5-241

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Discontinued Approach Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
-1000	-40	12.6	11.4	10.1	8.9	7.9	7.2	5.9
	15	12.9	11.7	10.3	9.1	8.1	7.4	6.1
	25	12.9	11.7	10.4	9.2	8.1	7.4	6.1
	35	10.9	9.8	8.6	7.5	6.5	5.9	4.7
	45	8.4	7.4	6.3	5.3	4.5	3.9	2.9
	50	7.1	6.2	5.2	4.3	3.5	2.9	2.0
	55	5.9	5.0	4.1	3.2	2.5	2.0	1.1
Sea Level	-40	12.3	11.2	9.9	8.7	7.7	7.0	5.7
	15	12.6	11.5	10.1	9.0	7.9	7.2	5.9
	25	12.7	11.5	10.2	9.0	8.0	7.3	6.0
	35	10.2	9.2	8.0	6.9	6.0	5.4	4.2
	45	7.7	6.8	5.7	4.8	4.0	3.4	2.4
	50	6.4	5.6	4.6	3.7	3.0	2.5	1.5
	55	5.2	4.4	3.5	2.7	2.0	1.5	0.7
1000	-40	11.9	10.8	9.5	8.3	7.3	6.7	5.4
	10	12.2	11.1	9.8	8.6	7.6	6.9	5.6
	20	12.3	11.1	9.8	8.6	7.6	6.9	5.7
	30	10.6	9.5	8.3	7.2	6.3	5.6	4.5
	40	8.2	7.2	6.1	5.2	4.3	3.8	2.8
	45	6.9	6.0	5.0	4.2	3.4	2.8	1.9
	50	5.7	4.9	4.0	3.1	2.4	1.9	1.0
2000	-40	11.5	10.4	9.1	8.0	7.0	6.4	5.2
	10	11.8	10.7	9.4	8.3	7.3	6.6	5.4
	20	11.9	10.7	9.4	8.3	7.3	6.6	5.4
	30	9.7	8.7	7.5	6.5	5.6	5.0	3.9
	40	7.4	6.5	5.5	4.5	3.7	3.2	2.2
	45	6.2	5.4	4.4	3.6	2.8	2.3	1.4
	50	5.1	4.3	3.4	2.6	1.9	1.4	0.6

MAC_TO_0_-1_07

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October 30, 2016

HJ1-29000-003-001

Page 5-242

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Discontinued Approach Gradient [%]								
Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
3000	-40	11.1	10.0	8.8	7.7	6.7	6.0	4.9
	5	11.4	10.3	9.0	7.9	6.9	6.3	5.1
	15	11.4	10.3	9.0	7.9	6.9	6.3	5.1
	25	10.0	8.9	7.8	6.7	5.8	5.2	4.1
	35	7.7	6.8	5.7	4.8	4.0	3.4	2.5
	40	6.6	5.8	4.8	3.9	3.1	2.6	1.7
	45	5.5	4.7	3.8	3.0	2.2	1.8	0.9
4000	-40	10.7	9.6	8.4	7.3	6.4	5.7	4.6
	0	10.9	9.8	8.6	7.5	6.6	5.9	4.8
	10	11.0	9.9	8.6	7.6	6.6	5.9	4.8
	20	10.3	9.2	8.0	7.0	6.0	5.4	4.3
	35	6.9	6.0	5.0	4.1	3.3	2.8	1.9
	40	5.8	5.0	4.1	3.2	2.5	2.0	1.1
	45	4.8	4.0	3.2	2.4	1.7	1.2	0.4
5000	-40	10.3	9.2	8.0	7.0	6.1	5.4	4.3
	0	10.5	9.4	8.2	7.2	6.2	5.6	4.5
	10	10.5	9.4	8.2	7.2	6.2	5.6	4.5
	30	7.1	6.2	5.2	4.3	3.5	3.0	2.0
	35	6.1	5.2	4.3	3.5	2.7	2.2	1.3
	40	5.1	4.3	3.4	2.6	1.9	1.4	0.6
	45	4.1	3.3	2.5	1.8	1.1	0.7	-0.1
6000	-40	9.9	8.9	7.7	6.7	5.8	5.2	4.0
	-5	10.1	9.0	7.9	6.8	5.9	5.3	4.2
	5	10.0	9.0	7.8	6.8	5.9	5.3	4.2
	15	9.6	8.6	7.4	6.4	5.5	4.9	3.8
	30	6.3	5.5	4.5	3.7	2.9	2.4	1.5
	35	5.3	4.5	3.6	2.8	2.1	1.6	0.8
	40	4.3	3.6	2.7	2.0	1.3	0.9	0.1

MAC_TO_0_3_05

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Discontinued Approach Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
7000	-40	9.6	8.5	7.4	6.4	5.5	4.9	3.8
	-10	9.7	8.7	7.5	6.5	5.6	5.0	3.9
	10	9.6	8.6	7.4	6.4	5.5	4.9	3.8
	25	6.6	5.7	4.7	3.9	3.1	2.6	1.7
	30	5.6	4.7	3.8	3.0	2.3	1.8	0.9
	35	4.6	3.8	3.0	2.2	1.5	1.1	0.3
	40	3.6	2.9	2.1	1.4	0.8	0.3	-0.4
8000	-40	9.2	8.2	7.1	6.1	5.2	4.6	3.6
	-15	9.3	8.3	7.1	6.1	5.3	4.7	3.6
	0	9.2	8.2	7.1	6.1	5.2	4.6	3.5
	20	6.8	5.9	4.9	4.1	3.3	2.8	1.8
	25	5.8	5.0	4.0	3.2	2.5	2.0	1.1
	30	4.8	4.0	3.2	2.4	1.7	1.3	0.4
	35	3.9	3.2	2.3	1.6	1.0	0.5	-0.2
9000	-40	8.9	7.9	6.8	5.8	4.9	4.4	3.3
	-15	8.8	7.9	6.8	5.8	4.9	4.3	3.3
	15	7.0	6.1	5.1	4.2	3.4	2.9	2.0
	20	6.0	5.2	4.2	3.4	2.7	2.2	1.3
	25	5.0	4.2	3.4	2.6	1.9	1.4	0.6
	30	4.1	3.4	2.5	1.8	1.1	0.7	-0.1
	35	3.2	2.5	1.7	1.0	0.4	0.0	-0.7
10000	-40	8.5	7.5	6.4	5.5	4.6	4.1	3.0
	-20	8.4	7.5	6.4	5.4	4.6	4.0	3.0
	15	6.2	5.4	4.4	3.6	2.8	2.3	1.4
	20	5.3	4.5	3.6	2.8	2.1	1.6	0.7
	25	4.3	3.5	2.7	2.0	1.3	0.9	0.1
	30	3.4	2.7	1.9	1.2	0.6	0.2	-0.6
	35	2.5	1.9	1.1	0.5	-0.1	-0.5	-1.2

MAC_TO_0_7_06

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October 30, 2016

HJ1-29000-003-001

Page 5-244

FOR TRAINING PURPOSES ONLY

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Balked Landing Gradient [%]								
Dry Runway, Zero Slope, No Wind								
Flaps LDG, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
-1000	-40	32.2	29.8	27.1	24.8	22.6	21.2	18.7
	15	32.6	30.3	27.6	25.2	23.0	21.6	19.0
	25	32.8	30.4	27.7	25.3	23.1	21.7	19.1
	35	28.7	26.6	24.1	21.9	19.9	18.6	16.3
	45	23.6	21.7	19.5	17.6	15.8	14.7	12.6
	50	21.0	19.3	17.2	15.4	13.8	12.7	10.8
	55	18.6	16.9	15.0	13.3	11.8	10.8	9.0
Sea Level	-40	32.0	29.7	27.0	24.6	22.5	21.1	18.6
	15	32.5	30.1	27.4	25.0	22.9	21.5	18.9
	25	32.6	30.2	27.5	25.1	23.0	21.6	19.0
	35	27.6	25.5	23.1	20.9	19.0	17.8	15.4
	45	22.4	20.6	18.5	16.6	14.9	13.8	11.7
	50	19.9	18.2	16.2	14.4	12.9	11.8	9.9
	55	17.4	15.9	14.0	12.4	10.9	10.0	8.2
1000	-40	31.1	28.8	26.2	23.9	21.8	20.4	17.9
	10	31.6	29.3	26.6	24.3	22.2	20.8	18.3
	20	31.7	29.4	26.7	24.4	22.3	20.9	18.4
	30	28.3	26.2	23.7	21.5	19.6	18.3	16.0
	40	23.4	21.5	19.3	17.4	15.6	14.5	12.4
	45	20.9	19.1	17.1	15.3	13.7	12.6	10.7
	50	18.4	16.8	14.9	13.2	11.7	10.7	8.9
2000	-40	30.2	28.0	25.4	23.2	21.1	19.8	17.4
	10	30.8	28.5	25.9	23.6	21.6	20.2	17.7
	20	30.9	28.6	26.0	23.7	21.6	20.3	17.8
	30	26.5	24.5	22.1	20.0	18.2	17.0	14.7
	40	21.8	20.0	17.9	16.1	14.4	13.3	11.3
	45	19.4	17.8	15.8	14.1	12.5	11.5	9.6
	50	17.1	15.5	13.7	12.1	10.6	9.7	8.0

BLC_LDg_0_-1_06

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HA-420 AFM

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Uncorrected Balked Landing Gradient [%]								
Dry Runway, Zero Slope, No Wind								
Flaps LDG, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
3000	-40	29.4	27.2	24.7	22.5	20.5	19.2	16.8
	5	29.9	27.7	25.2	22.9	20.9	19.6	17.2
	15	30.0	27.8	25.2	23.0	21.0	19.6	17.2
	25	27.1	25.0	22.6	20.5	18.7	17.4	15.1
	35	22.5	20.6	18.5	16.7	15.0	13.9	11.9
	40	20.3	18.6	16.6	14.8	13.2	12.2	10.3
	45	18.0	16.4	14.6	12.9	11.4	10.4	8.7
4000	-40	28.6	26.5	24.0	21.8	19.8	18.6	16.2
	0	29.0	26.9	24.4	22.2	20.2	18.9	16.5
	10	29.1	26.9	24.4	22.2	20.2	18.9	16.6
	20	27.7	25.6	23.2	21.0	19.1	17.9	15.6
	35	20.8	19.1	17.1	15.3	13.7	12.6	10.7
	40	18.7	17.1	15.2	13.5	12.0	11.0	9.2
	45	16.7	15.1	13.3	11.8	10.3	9.4	7.7
5000	-40	27.8	25.7	23.3	21.1	19.2	17.9	15.6
	0	28.2	26.1	23.6	21.4	19.5	18.2	15.9
	10	28.2	26.1	23.6	21.4	19.5	18.2	15.9
	30	21.3	19.5	17.5	15.7	14.0	13.0	11.0
	35	19.2	17.6	15.6	13.9	12.4	11.4	9.6
	40	17.2	15.6	13.8	12.2	10.8	9.8	8.1
	45	15.2	13.7	12.0	10.5	9.2	8.3	6.7
6000	-40	27.0	24.9	22.6	20.5	18.6	17.3	15.1
	-5	27.3	25.2	22.9	20.7	18.8	17.6	15.3
	5	27.2	25.2	22.8	20.7	18.8	17.6	15.3
	15	26.3	24.3	22.0	19.9	18.0	16.8	14.6
	30	19.7	18.0	16.1	14.3	12.8	11.7	9.9
	35	17.7	16.1	14.2	12.6	11.2	10.2	8.4
	40	15.7	14.2	12.5	10.9	9.6	8.7	7.1

BLC_LDG_0_3_05

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October 30, 2016

HJ1-29000-003-001

Page 5-246

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HA-420 AFM

PERFORMANCE

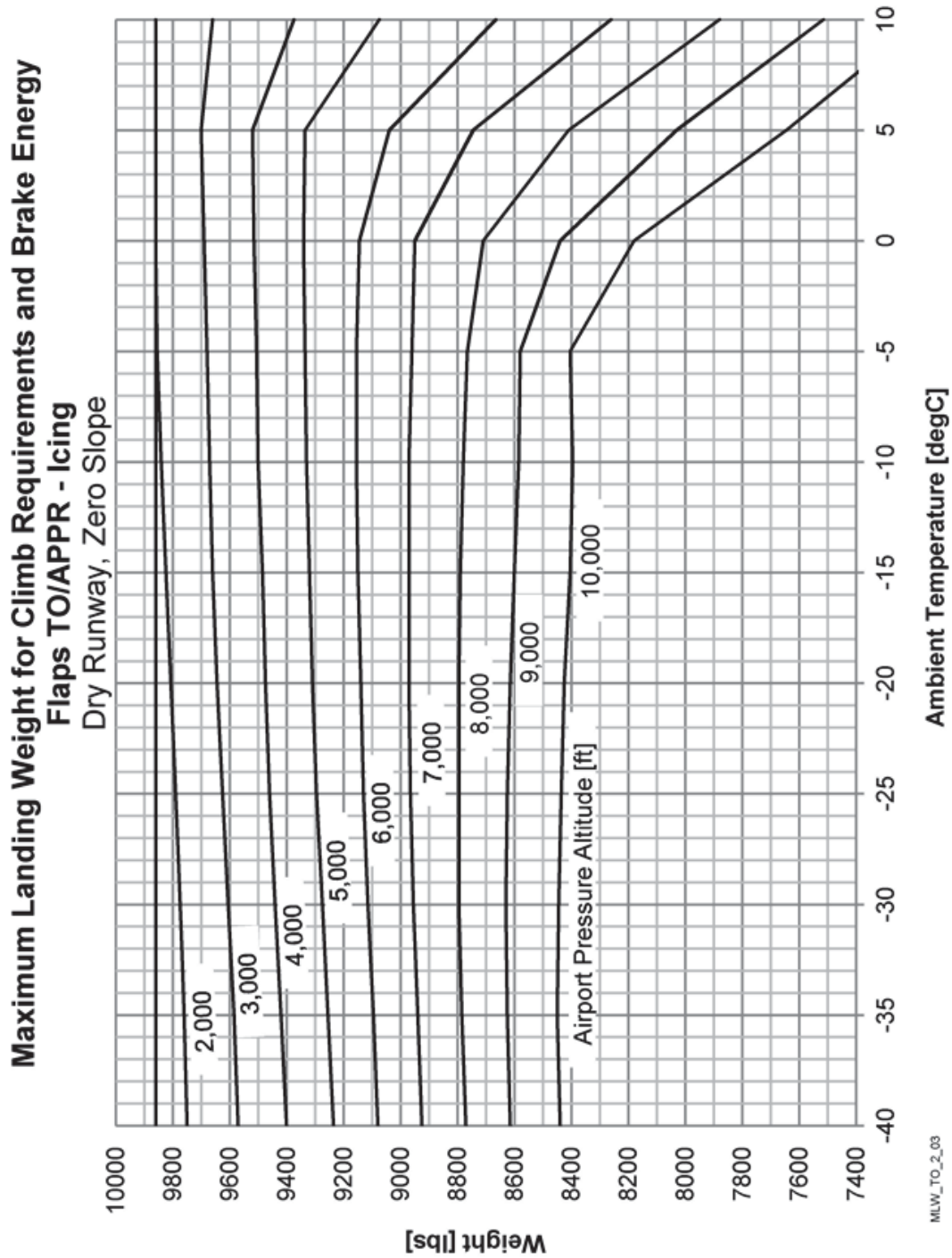
Uncorrected Balked Landing Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps LDG, Ice Protection Off								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		100	102	105	108	111	113	117
7000	-40	26.3	24.2	21.9	19.8	18.0	16.8	14.6
	-10	26.5	24.4	22.1	20.0	18.2	17.0	14.7
	10	26.3	24.3	21.9	19.9	18.0	16.8	14.6
	25	20.2	18.5	16.5	14.7	13.2	12.1	10.2
	30	18.1	16.5	14.7	13.0	11.5	10.6	8.8
	35	16.2	14.7	12.9	11.4	10.0	9.1	7.4
	40	14.2	12.8	11.2	9.7	8.4	7.6	6.0
8000	-40	25.5	23.5	21.3	19.2	17.4	16.2	14.1
	-15	25.7	23.7	21.4	19.3	17.5	16.3	14.2
	0	25.5	23.5	21.2	19.2	17.4	16.2	14.0
	20	20.7	19.0	17.0	15.2	13.6	12.5	10.6
	25	18.6	17.0	15.1	13.4	11.9	10.9	9.1
	30	16.6	15.1	13.3	11.7	10.3	9.4	7.7
	35	14.8	13.3	11.7	10.2	8.9	8.0	6.4
9000	-40	24.8	22.9	20.6	18.6	16.9	15.7	13.6
	-15	24.8	22.9	20.6	18.6	16.9	15.7	13.6
	15	21.1	19.3	17.3	15.5	13.9	12.8	10.9
	20	19.1	17.4	15.5	13.8	12.3	11.3	9.5
	25	17.1	15.5	13.7	12.1	10.7	9.8	8.0
	30	15.2	13.7	12.0	10.5	9.2	8.3	6.7
	35	13.3	12.0	10.4	9.0	7.7	6.9	5.4
10000	-40	24.1	22.1	20.0	18.0	16.3	15.1	13.0
	-20	24.0	22.1	19.9	17.9	16.2	15.1	13.0
	15	19.5	17.8	15.9	14.1	12.6	11.6	9.7
	20	17.5	16.0	14.1	12.5	11.1	10.1	8.4
	25	15.6	14.1	12.4	10.9	9.5	8.6	7.0
	30	13.8	12.4	10.8	9.3	8.1	7.2	5.7
	35	12.0	10.7	9.2	7.8	6.6	5.9	4.4

BLC_LDg_0_7_05

Honda Aircraft Company

HA-420 AFM

PERFORMANCE



FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 5-248

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HA-420 AFM

PERFORMANCE

Uncorrected Landing Field Length [feet] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
-1000	-40	2818	2909	3021	3131	3241	3319	3478
	-20	2964	3062	3182	3301	3419	3503	3674
	-10	3056	3156	3280	3402	3523	3610	3786
	-5	3090	3192	3317	3442	3565	3653	3831
	0	3123	3227	3354	3481	3606	3695	3877
	5	3156	3261	3391	3519	3646	3737	3921
	10	3187	3294	3425	3556	3684	3776	3964
Sea Level	-40	2878	2971	3086	3200	3312	3393	3557
	-20	3030	3131	3255	3377	3498	3585	3761
	-10	3128	3231	3359	3484	3609	3698	3879
	-5	3163	3268	3397	3525	3652	3742	3927
	0	3198	3304	3435	3565	3694	3786	3973
	5	3231	3339	3473	3605	3736	3829	4019
	10	3262	3371	3507	3641	3773	3868	4061
1000	-40	2940	3036	3155	3273	3389	3472	3640
	-20	3104	3207	3335	3461	3585	3675	3856
	-10	3203	3309	3440	3570	3698	3790	3977
	-5	3240	3347	3481	3612	3743	3836	4026
	0	3274	3383	3519	3653	3785	3880	4073
	5	3308	3419	3556	3692	3827	3923	4118
	10	3342	3455	3594	3732	3869	3966	4164
2000	-40	3006	3105	3228	3349	3469	3554	3728
	-20	3183	3289	3420	3549	3678	3769	3956
	-10	3281	3391	3526	3659	3792	3886	4079
	-5	3319	3430	3567	3703	3838	3934	4129
	0	3354	3466	3606	3744	3880	3978	4177
	5	3389	3504	3645	3785	3924	4023	4225
	10	3425	3542	3685	3827	3968	4068	4272

LFL_TO_2-1_03

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Landing Field Length [feet] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
3000	-40	3080	3182	3308	3433	3556	3644	3824
	-20	3265	3374	3509	3643	3775	3869	4062
	-10	3365	3477	3616	3754	3890	3987	4186
	-5	3401	3516	3657	3797	3935	4034	4236
	0	3437	3553	3697	3839	3980	4080	4285
	5	3475	3593	3738	3883	4026	4128	4336
	10	3512	3632	3780	3926	4071	4174	4385
4000	-40	3162	3267	3397	3525	3652	3743	3928
	-20	3345	3458	3597	3735	3871	3968	4167
	-10	3448	3564	3707	3849	3989	4090	4294
	-5	3486	3604	3750	3894	4037	4139	4346
	0	3524	3644	3792	3938	4083	4187	4398
	5	3562	3684	3834	3983	4130	4235	4449
	10	3603	3726	3878	4029	4178	4284	4501
5000	-40	3246	3354	3487	3619	3750	3844	4034
	-20	3431	3547	3690	3832	3973	4073	4277
	-10	3536	3656	3804	3950	4095	4198	4409
	-5	3574	3696	3846	3995	4142	4247	4461
	0	3614	3738	3890	4041	4191	4298	4516
	5	3655	3780	3935	4089	4241	4349	4570
	10	3696	3823	3980	4135	4289	4398	4622
6000	-40	3335	3446	3584	3720	3855	3951	4147
	-20	3523	3643	3791	3937	4082	4185	4396
	-10	3627	3750	3902	4053	4202	4309	4526
	-5	3667	3792	3947	4100	4252	4360	4581
	0	3709	3836	3993	4149	4303	4413	4638
	5	3751	3880	4040	4198	4355	4467	4695
	10	3794	3925	4087	4247	4406	4520	4751

LFL_TO_2_3_03

Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Landing Field Length [feet] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
7000	-40	3427	3541	3683	3823	3962	4061	4264
	-20	3611	3734	3887	4037	4187	4294	4511
	-10	3720	3847	4004	4159	4313	4423	4648
	-5	3762	3891	4051	4209	4365	4477	4705
	0	3806	3937	4099	4260	4418	4532	4763
	5	3851	3984	4148	4311	4473	4588	4823
	10	3895	4029	4196	4361	4525	4642	4880
8000	-40	3519	3637	3783	3927	4070	4173	4381
	-20	3708	3835	3992	4148	4302	4412	4636
	-10	3817	3948	4110	4270	4429	4542	4774
	-5	3861	3994	4158	4321	4482	4597	4832
	0	3907	4042	4208	4374	4538	4655	4893
	5	3954	4091	4260	4428	4594	4713	4955
	10	4001	4140	4312	4482	4651	4771	5017
9000	-40	3611	3732	3883	4032	4180	4285	4500
	-20	3805	3936	4098	4259	4418	4531	4763
	-10	3917	4052	4219	4384	4548	4665	4903
	-5	3964	4101	4271	4439	4605	4724	4966
	0	4012	4152	4324	4494	4663	4784	5030
	5	4061	4202	4377	4550	4722	4844	5094
	10	4113	4256	4434	4609	4783	4908	5161
10000	-40	3706	3831	3986	4140	4293	4402	4624
	-20	3908	4043	4210	4376	4540	4657	4896
	-10	4023	4162	4334	4505	4674	4795	5041
	-5	4072	4213	4388	4562	4734	4856	5107
	0	4123	4266	4444	4620	4795	4920	5174
	5	4173	4319	4499	4678	4855	4982	5240
	10	4230	4378	4561	4743	4922	5051	5312

LFL_TO_2_7_03

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October 30, 2016

HJ1-29000-003-001

Page 5-251

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HA-420 AFM

PERFORMANCE

Wind Corrected Landing Field Length [feet]								
Flaps TO/APPR - Icing								
Tailwind		◀ REF [0] ▶	Headwind					
-10	-5		5	10	15	20	25	30
2440	2220	2000	1945	1890	1830	1775	1720	1660
2546	2323	2100	2044	1989	1928	1872	1816	1756
2651	2426	2200	2144	2087	2026	1969	1912	1851
2757	2529	2300	2243	2186	2123	2065	2008	1947
2862	2632	2400	2342	2284	2221	2162	2104	2042
2968	2735	2500	2441	2383	2319	2259	2200	2138
3073	2838	2600	2541	2481	2417	2356	2296	2233
3179	2941	2700	2640	2580	2514	2452	2392	2329
3284	3044	2800	2739	2678	2612	2549	2488	2424
3390	3147	2900	2838	2777	2710	2646	2584	2520
3495	3250	3000	2938	2875	2808	2743	2680	2615
3601	3353	3100	3037	2974	2905	2839	2776	2711
3706	3456	3200	3136	3072	3003	2936	2872	2806
3812	3559	3300	3235	3171	3101	3033	2968	2902
3917	3662	3400	3335	3269	3199	3130	3064	2997
4023	3765	3500	3434	3368	3296	3226	3160	3093
4128	3868	3600	3533	3466	3394	3323	3256	3188
4234	3971	3700	3632	3565	3492	3420	3352	3284
4339	4074	3800	3732	3663	3590	3517	3448	3379
4445	4177	3900	3831	3762	3687	3613	3544	3475
4550	4280	4000	3930	3860	3785	3710	3640	3570
4656	4383	4100	4029	3959	3883	3807	3736	3666
4761	4486	4200	4129	4057	3981	3904	3832	3761
4867	4589	4300	4228	4156	4078	4000	3928	3857
4972	4692	4400	4327	4254	4176	4097	4024	3952
5078	4795	4500	4426	4353	4274	4194	4120	4048
5183	4898	4600	4526	4451	4372	4291	4216	4143
5289	5001	4700	4625	4550	4469	4387	4312	4239
5394	5104	4800	4724	4648	4567	4484	4408	4334
5500	5207	4900	4823	4747	4665	4581	4504	4430
5605	5310	5000	4923	4845	4763	4678	4600	4525

LWC_TO_2_02

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October 30, 2016

HJ1-29000-003-001

Page 5-252

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HA-420 AFM

PERFORMANCE

Slope Corrected Landing Field Length [feet]								
Flaps TO/APPR - Icing								
Runway Gradient [%]								
-2.0	-1.5	-1.0	-0.5	◀ REF [0] ▶	0.5	1.0	1.5	2.0
3134	2806	2558	2361	2200	2076	1966	1880	1793
3320	2950	2680	2470	2300	2170	2055	1965	1875
3506	3094	2802	2579	2400	2264	2144	2050	1957
3692	3238	2923	2688	2500	2358	2233	2135	2039
3877	3382	3045	2798	2600	2452	2323	2220	2121
4063	3525	3166	2907	2700	2545	2412	2305	2203
4249	3669	3288	3016	2800	2639	2501	2390	2285
4435	3813	3409	3125	2900	2733	2590	2475	2367
4620	3957	3531	3235	3000	2827	2680	2560	2448
4806	4101	3652	3344	3100	2921	2769	2645	2530
4992	4245	3774	3453	3200	3015	2858	2730	2612
5178	4388	3895	3562	3300	3108	2947	2815	2694
5363	4532	4017	3672	3400	3202	3037	2900	2776
5549	4676	4138	3781	3500	3296	3126	2985	2858
5735	4820	4260	3890	3600	3390	3215	3070	2940
5921	4964	4382	3999	3700	3484	3304	3155	3022
6107	5108	4503	4108	3800	3578	3393	3240	3104
6292	5252	4625	4218	3900	3672	3483	3325	3186
6478	5395	4746	4327	4000	3765	3572	3410	3268
6664	5539	4868	4436	4100	3859	3661	3495	3350
6850	5683	4989	4545	4200	3953	3750	3580	3432
7035	5827	5111	4655	4300	4047	3840	3665	3513
7221	5971	5232	4764	4400	4141	3929	3750	3595
7407	6115	5354	4873	4500	4235	4018	3835	3677
7593	6258	5475	4982	4600	4328	4107	3920	3759
7778	6402	5597	5092	4700	4422	4197	4005	3841
7964	6546	5718	5201	4800	4516	4286	4090	3923
8150	6690	5840	5310	4900	4610	4375	4175	4005
8336	6834	5962	5419	5000	4704	4464	4260	4087
8522	6978	6083	5528	5100	4798	4553	4345	4169
8707	7122	6205	5638	5200	4892	4643	4430	4251

LSC_TO_2_02

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October 30, 2016

HJ1-29000-003-001

Page 5-253

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Honda Aircraft Company

HA-420 AFM

PERFORMANCE

Uncorrected Discontinued Approach Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
-1000	-40	8.0	6.9	5.6	4.4	3.4	2.7	1.5
	-20	8.1	6.9	5.6	4.4	3.4	2.7	1.5
	-10	8.1	6.9	5.6	4.5	3.4	2.8	1.5
	-5	8.1	7.0	5.7	4.5	3.5	2.8	1.6
	0	8.2	7.0	5.7	4.5	3.5	2.8	1.6
	5	8.2	7.1	5.7	4.6	3.5	2.9	1.6
	10	8.3	7.1	5.8	4.6	3.6	2.9	1.6
Sea Level	-40	7.8	6.6	5.3	4.2	3.2	2.5	1.3
	-20	7.8	6.7	5.4	4.2	3.2	2.6	1.3
	-10	7.9	6.7	5.4	4.3	3.3	2.6	1.4
	-5	7.9	6.8	5.5	4.3	3.3	2.6	1.4
	0	8.0	6.8	5.5	4.4	3.4	2.7	1.5
	5	8.0	6.9	5.6	4.4	3.4	2.7	1.5
	10	8.1	6.9	5.6	4.4	3.4	2.7	1.5
1000	-40	7.4	6.2	5.0	3.9	2.9	2.2	1.0
	-20	7.4	6.3	5.0	3.9	2.9	2.3	1.1
	-10	7.5	6.4	5.1	4.0	3.0	2.3	1.1
	-5	7.6	6.4	5.2	4.0	3.0	2.4	1.2
	0	7.6	6.5	5.2	4.1	3.1	2.4	1.2
	5	7.6	6.5	5.2	4.1	3.1	2.4	1.2
	10	7.7	6.5	5.2	4.1	3.1	2.4	1.2
2000	-40	7.0	5.8	4.6	3.5	2.6	1.9	0.7
	-20	7.1	6.0	4.7	3.6	2.7	2.0	0.8
	-10	7.2	6.0	4.8	3.7	2.7	2.1	0.9
	-5	7.2	6.1	4.8	3.7	2.7	2.1	0.9
	0	7.2	6.1	4.9	3.8	2.8	2.1	0.9
	5	7.2	6.1	4.9	3.8	2.8	2.1	0.9
	10	7.3	6.1	4.9	3.8	2.8	2.2	1.0

MAC_TO_2_1_03

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PERFORMANCE

Uncorrected Discontinued Approach Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
3000	-40	6.5	5.5	4.3	3.2	2.2	1.6	0.5
	-20	6.7	5.6	4.4	3.3	2.4	1.7	0.6
	-10	6.8	5.7	4.5	3.4	2.4	1.8	0.6
	-5	6.8	5.7	4.5	3.4	2.4	1.8	0.6
	0	6.8	5.7	4.5	3.4	2.4	1.8	0.7
	5	6.8	5.7	4.5	3.4	2.5	1.8	0.7
	10	6.7	5.7	4.4	3.4	2.4	1.8	0.6
4000	-40	6.2	5.1	3.9	2.9	1.9	1.3	0.2
	-20	6.3	5.3	4.1	3.0	2.1	1.4	0.3
	-10	6.4	5.3	4.1	3.0	2.1	1.5	0.3
	-5	6.4	5.3	4.1	3.1	2.1	1.5	0.4
	0	6.4	5.3	4.1	3.1	2.1	1.5	0.4
	5	6.4	5.4	4.2	3.1	2.1	1.5	0.4
	10	6.1	5.0	3.9	2.8	1.9	1.3	0.2
5000	-40	5.8	4.7	3.6	2.6	1.6	1.0	-0.1
	-20	5.9	4.9	3.7	2.7	1.8	1.2	0.0
	-10	6.0	4.9	3.8	2.7	1.8	1.2	0.1
	-5	6.0	4.9	3.8	2.7	1.8	1.2	0.1
	0	6.0	5.0	3.8	2.7	1.8	1.2	0.1
	5	6.0	4.9	3.8	2.7	1.8	1.2	0.1
	10	5.4	4.4	3.3	2.2	1.3	0.8	-0.3
6000	-40	5.4	4.4	3.3	2.3	1.4	0.8	-0.3
	-20	5.6	4.5	3.4	2.4	1.5	0.9	-0.2
	-10	5.6	4.6	3.4	2.4	1.5	0.9	-0.2
	-5	5.6	4.6	3.4	2.4	1.5	0.9	-0.2
	0	5.6	4.5	3.4	2.4	1.5	0.9	-0.2
	5	5.3	4.3	3.2	2.2	1.3	0.7	-0.4
	10	4.5	3.5	2.4	1.5	0.6	0.1	-1.0

MAC_TO_2_3_03

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PERFORMANCE

Uncorrected Discontinued Approach Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
7000	-40	5.1	4.1	3.0	2.0	1.1	0.5	-0.6
	-20	5.2	4.2	3.0	2.0	1.2	0.6	-0.5
	-10	5.2	4.2	3.0	2.1	1.2	0.6	-0.5
	-5	5.1	4.1	3.0	2.0	1.1	0.6	-0.5
	0	5.1	4.1	3.0	2.0	1.1	0.5	-0.5
	5	4.7	3.7	2.6	1.6	0.8	0.2	-0.8
	10	3.6	2.7	1.6	0.7	-0.1	-0.6	-1.6
8000	-40	4.7	3.7	2.6	1.7	0.8	0.2	-0.8
	-20	4.8	3.8	2.7	1.7	0.8	0.3	-0.8
	-10	4.7	3.8	2.7	1.7	0.8	0.3	-0.8
	-5	4.7	3.7	2.6	1.7	0.8	0.2	-0.8
	0	4.6	3.6	2.5	1.6	0.7	0.1	-0.9
	5	3.9	3.0	1.9	1.0	0.2	-0.4	-1.4
	10	2.7	1.8	0.9	0.0	-0.8	-1.3	-2.2
9000	-40	4.4	3.4	2.3	1.4	0.5	0.0	-1.0
	-20	4.4	3.4	2.3	1.4	0.5	0.0	-1.0
	-10	4.3	3.3	2.3	1.3	0.5	-0.1	-1.1
	-5	4.3	3.3	2.3	1.3	0.5	-0.1	-1.1
	0	4.0	3.0	2.0	1.1	0.2	-0.3	-1.3
	5	3.1	2.2	1.2	0.3	-0.5	-1.0	-1.9
	10	1.9	1.1	0.2	-0.7	-1.4	-1.9	-2.8
10000	-40	4.0	3.0	2.0	1.1	0.2	-0.3	-1.3
	-20	3.9	3.0	2.0	1.0	0.2	-0.3	-1.3
	-10	3.9	2.9	1.9	1.0	0.1	-0.4	-1.4
	-5	3.9	3.0	1.9	1.0	0.2	-0.4	-1.4
	0	3.4	2.5	1.5	0.6	-0.2	-0.8	-1.7
	5	2.2	1.4	0.4	-0.4	-1.2	-1.6	-2.5
	10	1.1	0.4	-0.5	-1.3	-2.0	-2.5	-3.3

MAC_TO_2_7_03

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HA-420 AFM

PERFORMANCE

Uncorrected Balked Landing Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
-1000	-40	29.5	27.1	24.4	22.1	20.0	18.6	16.0
	-20	29.5	27.1	24.5	22.1	20.0	18.6	16.1
	-10	29.6	27.2	24.5	22.2	20.1	18.7	16.1
	-5	29.7	27.3	24.6	22.3	20.1	18.8	16.2
	0	29.8	27.4	24.7	22.3	20.2	18.8	16.3
	5	29.9	27.5	24.8	22.4	20.3	18.9	16.3
	10	29.9	27.5	24.8	22.5	20.3	18.9	16.4
Sea Level	-40	29.3	26.9	24.3	21.9	19.8	18.4	15.9
	-20	29.4	27.0	24.3	22.0	19.9	18.5	16.0
	-10	29.5	27.1	24.4	22.1	20.0	18.6	16.0
	-5	29.6	27.2	24.5	22.1	20.0	18.7	16.1
	0	29.7	27.3	24.6	22.2	20.1	18.7	16.2
	5	29.7	27.3	24.6	22.3	20.2	18.8	16.2
	10	29.8	27.4	24.7	22.3	20.2	18.8	16.3
1000	-40	28.4	26.1	23.5	21.2	19.1	17.8	15.3
	-20	28.6	26.2	23.6	21.3	19.2	17.9	15.4
	-10	28.7	26.4	23.7	21.4	19.3	18.0	15.5
	-5	28.8	26.4	23.8	21.5	19.4	18.0	15.6
	0	28.9	26.5	23.9	21.5	19.5	18.1	15.6
	5	28.9	26.6	23.9	21.6	19.5	18.1	15.6
	10	29.0	26.6	24.0	21.6	19.5	18.2	15.7
2000	-40	27.6	25.3	22.8	20.5	18.5	17.2	14.7
	-20	27.8	25.5	23.0	20.7	18.7	17.3	14.9
	-10	28.0	25.7	23.1	20.8	18.8	17.5	15.0
	-5	28.0	25.7	23.2	20.9	18.8	17.5	15.0
	0	28.1	25.8	23.2	20.9	18.9	17.6	15.1
	5	28.1	25.8	23.2	21.0	18.9	17.6	15.1
	10	28.2	25.9	23.3	21.0	19.0	17.6	15.2

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October 30, 2016

HJ1-29000-003-001

Page 5-257

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HA-420 AFM

PERFORMANCE

Uncorrected Balked Landing Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
3000	-40	26.8	24.5	22.0	19.8	17.9	16.6	14.2
	-20	27.1	24.8	22.3	20.1	18.1	16.8	14.4
	-10	27.2	24.9	22.4	20.2	18.2	16.9	14.5
	-5	27.2	25.0	22.4	20.2	18.2	16.9	14.5
	0	27.3	25.0	22.5	20.2	18.2	16.9	14.5
	5	27.3	25.0	22.5	20.3	18.3	17.0	14.5
	10	27.1	24.9	22.3	20.1	18.1	16.8	14.4
4000	-40	26.0	23.8	21.3	19.2	17.2	15.9	13.6
	-20	26.3	24.1	21.6	19.4	17.5	16.2	13.8
	-10	26.4	24.2	21.7	19.5	17.5	16.2	13.9
	-5	26.4	24.2	21.7	19.5	17.6	16.3	13.9
	0	26.4	24.2	21.7	19.5	17.6	16.3	13.9
	5	26.5	24.2	21.8	19.6	17.6	16.3	14.0
	10	25.8	23.6	21.2	19.0	17.1	15.8	13.5
5000	-40	25.2	23.0	20.6	18.5	16.6	15.3	13.1
	-20	25.5	23.3	20.9	18.8	16.8	15.6	13.3
	-10	25.6	23.4	21.0	18.8	16.9	15.6	13.3
	-5	25.6	23.4	21.0	18.8	16.9	15.7	13.3
	0	25.6	23.4	21.0	18.9	16.9	15.7	13.4
	5	25.6	23.4	21.0	18.9	17.0	15.7	13.4
	10	24.4	22.3	20.0	17.9	16.0	14.8	12.5
6000	-40	24.4	22.3	20.0	17.9	16.0	14.8	12.5
	-20	24.7	22.6	20.2	18.1	16.2	15.0	12.7
	-10	24.8	22.6	20.3	18.2	16.3	15.0	12.8
	-5	24.8	22.6	20.3	18.2	16.3	15.1	12.8
	0	24.7	22.6	20.2	18.1	16.3	15.0	12.8
	5	24.2	22.1	19.8	17.7	15.9	14.7	12.4
	10	22.5	20.5	18.3	16.3	14.5	13.3	11.2

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HA-420 AFM

PERFORMANCE

Uncorrected Balked Landing Gradient [%] Dry Runway, Zero Slope, No Wind								
Flaps TO/APPR - Icing								
ALT [ft]	TEMP [°C]	Landing Weight [lb]						
		7600	8000	8500	9000	9500	9860	10600
		VREF [KIAS]						
		109	111	115	118	121	124	128
7000	-40	23.7	21.6	19.3	17.3	15.4	14.2	12.0
	-20	23.9	21.8	19.5	17.4	15.6	14.4	12.2
	-10	23.9	21.8	19.5	17.5	15.6	14.4	12.2
	-5	23.9	21.8	19.5	17.4	15.6	14.4	12.2
	0	23.8	21.7	19.4	17.4	15.5	14.3	12.1
	5	22.9	20.8	18.6	16.6	14.8	13.6	11.5
	10	20.7	18.8	16.6	14.7	13.1	11.9	9.9
8000	-40	23.0	20.9	18.7	16.7	14.9	13.7	11.5
	-20	23.1	21.0	18.8	16.8	15.0	13.8	11.6
	-10	23.0	21.0	18.7	16.7	14.9	13.7	11.6
	-5	22.9	20.9	18.7	16.7	14.9	13.7	11.5
	0	22.7	20.7	18.4	16.4	14.7	13.5	11.3
	5	21.3	19.4	17.2	15.3	13.6	12.5	10.4
	10	18.9	17.1	15.1	13.3	11.7	10.6	8.7
9000	-40	22.3	20.3	18.0	16.1	14.3	13.2	11.0
	-20	22.3	20.3	18.0	16.1	14.3	13.2	11.0
	-10	22.1	20.1	17.9	16.0	14.2	13.1	10.9
	-5	22.1	20.1	17.9	15.9	14.2	13.0	10.9
	0	21.4	19.5	17.3	15.4	13.7	12.5	10.5
	5	19.6	17.7	15.7	13.8	12.2	11.1	9.2
	10	17.3	15.5	13.6	11.9	10.4	9.4	7.5
10000	-40	21.5	19.6	17.4	15.5	13.7	12.6	10.5
	-20	21.4	19.5	17.3	15.4	13.6	12.5	10.4
	-10	21.3	19.3	17.2	15.2	13.5	12.4	10.3
	-5	21.3	19.3	17.2	15.3	13.5	12.4	10.3
	0	20.3	18.4	16.3	14.4	12.7	11.6	9.6
	5	17.9	16.1	14.1	12.4	10.8	9.8	8.0
	10	15.7	14.0	12.2	10.6	9.1	8.2	6.4

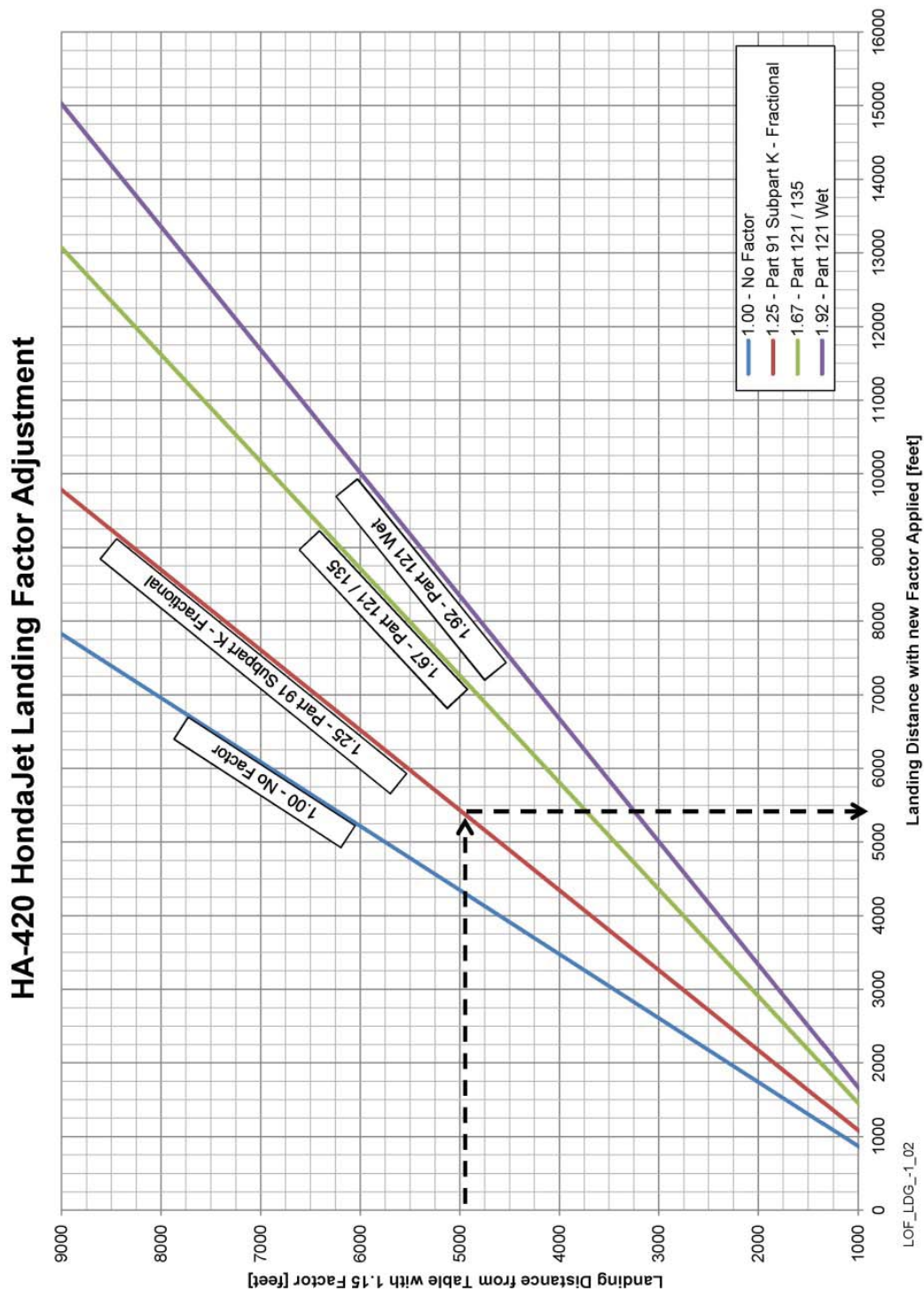
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PERFORMANCE

Operational Factors for Landing Distance



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HJ1-29000-003-001

Page 5-260

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PERFORMANCE

TURNAROUND

Landing Brake Energy

A minimum of 30 minutes between landing and takeoff in the event of the following sequence:

1. Maximum landing weight full stop.
2. Takeoff in 15 minutes or less.
3. Less than 1 hour flight.

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PERFORMANCE

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HJ1-29000-003-001

Page 5-262

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HA-420 AFM

WEIGHT AND BALANCE

SECTION 6 WEIGHT AND BALANCE

TABLE OF CONTENTS

INTRODUCTION.....	6-3
WEIGHT AND BALANCE DEFINITIONS	6-4
WEIGHING INSTRUCTIONS	6-7
LOADING INSTRUCTIONS	6-18
CABIN FLOORPLAN / PAYLOAD DATA.....	6-21
CONSUMABLES AND BAGGAGE.....	6-23
CENTER OF GRAVITY LIMITS	6-24
USABLE FUEL	6-26
ENGINE OIL / UNUSABLE FUEL	6-27
APPENDIX A – BLANK FORMS.....	6-28

Honda Aircraft Company

HA-420 AFM

WEIGHT AND BALANCE

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 6-2

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HA-420 AFM

WEIGHT AND BALANCE

WEIGHT AND BALANCE

INTRODUCTION

This section of the Airplane Flight Manual contains the HA-420 model (HondaJet) Weight and Balance instructions, forms and data required by 14 CFR Part 23.1583 Operating limitations (Airplane Flight Manual).

Specifically this section complies with the following 14 CFR regulations:

1. Weight limits
2. Empty weight and corresponding center of gravity
3. Weight and center of gravity
4. Loading information

Each airplane will have a specific Basic Empty Weight (BEW) and Balance Form in the “as delivered” from manufacturer condition, along with an example loading in the “as delivered” condition as shown in WEIGHING INSTRUCTIONS.

Also included are the forms and data necessary to compute BEW and perform loading calculations along with the airplane Equipment List.

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HA-420 AFM

WEIGHT AND BALANCE

WEIGHT AND BALANCE DEFINITIONS

Term	Definitions
Approved Loading Envelope	Those combinations of airplane weight and center of gravity which define the limits beyond which loading is not approved
Basic Empty Weight	The completed airplane (all structure and systems) including enclosed fluids such as unusable fuel, engine oil, hydraulic oil, ECS refrigerants and oxygen. BEW corresponds to the 14 CFR Part 23.29 description
Center of Gravity (CG)	The point at which the weight of an item can be considered concentrated for weight and balance purposes
CG Limits	The CG locations within which the airplane must be operated at a given weight
Empty Weight	The airplane weight prior to fuel or engine oil being added. Empty weight includes all permanently installed equipment, full system operating fluids and fixed ballast. Essentially manufacturers “dry” weight of the completed airplane
Engine Oil (Drainable)	The amount of oil that can be drained from the engine
Fuselage Station (F-Sta)	The longitudinal distance (in inches) from the zero fuselage reference data plane to a specific point

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October 30, 2016

HJ1-29000-003-001

Page 6-4

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HA-420 AFM

WEIGHT AND BALANCE

Jack Point(s)	Appropriate position(s) specified by the airplane manufacturer as correct point for jacking the airplane during weighing or maintenance operation
Leveling Point(s)	Those points which are used to level the airplane in pitch and roll during the weighing operation
Maximum Landing Weight (MLW)	The maximum weight of the airplane allowed by structural design that should normally not be exceeded at touchdown
Maximum Ramp Weight (MRW)	The highest weight of the airplane allowed by structural design that should not be exceeded
Maximum Takeoff Weight (MTOW)	The highest permissible weight of the airplane at break release for takeoff (ramp weight minus taxi and run up fuel = MTOW)
Maximum Zero Fuel Weight (MZFW)	The Maximum permissible weight of the loaded airplane before usable fuel is added
Moment (mom)	A measure of rotational tendency of an item about a specified line mathematically equal to the product of the weight and fuse station (weight * F-Sta = moment)
Operating Weight Empty (OWE)	BEW plus crew and mission specific equipment (does not include fuel)

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October 30, 2016

HJ1-29000-003-001

Page 6-5

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HA-420 AFM

WEIGHT AND BALANCE

Payload	Payload is any combination of items (passengers, baggage or consumables) added to the airplane for mission purposes. Fuel is not considered payload
Ramp Weight	The airplane weight at engine start after loading is complete
Tare	The weight indicated by a scale prior to applied load, or, the weight indicated by a scale when the load is removed. (Indicated weight minus tare = actual weight)
Unusable Fuel	<p>The quantity of fuel remaining in the tanks that cannot be used by the aircraft engines when in flight. Unusable fuel is further divided into two areas:</p> <ul style="list-style-type: none">• Unusable/Drainable The amount of fuel that can be drained from the fuel tanks when starting with the unusable quantity• Unusable/Undrainable The amount of fuel remaining in the tanks after draining and “sumping” also known as “Trapped Fuel”
Zero Fuel Weight	Airplane operating weight plus payload without usable fuel

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HA-420 AFM

WEIGHT AND BALANCE

WEIGHING INSTRUCTIONS

Weighing Procedure

1. Airplane must be weighed in an enclosed area.
2. Ventilation system (heating or cooling) must not impinge on the airplane.
3. Floor must be level (slope not to exceed $\frac{1}{4}$ " per foot).
4. Airplane must be defueled and drained per normal maintenance procedure to obtain BEW.
5. Each engine must have full Engine Oil.
6. Flaps to be in the retracted position.
7. Note the airplane configuration prior to weighing and any overages or shortages.

Additional or missing Equipment List Items, Life Rafts, changes to Interior configuration, Potable Fluids, Consumables, etc. should be recorded on page 2 of the Basic Empty Weight and Balance Form. These overage and shortage items will reconcile the "as weighed" value to BEW.

8. Check for current weighing device calibration and allow sufficient "warm up" time per manufacturer's instructions and ensure all readings are stable prior to applying load.
9. If weighing on platform scales (weight on wheels) measurements will be taken from a known Fuselage Reference Point (FRP). If weighing on Jacking points, refer to Basic Empty Weight and Balance Form. The FRP is the intersection of the wing leading edge to the fuselage fairing and corresponds to F-Sta 222.454".

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WEIGHT AND BALANCE

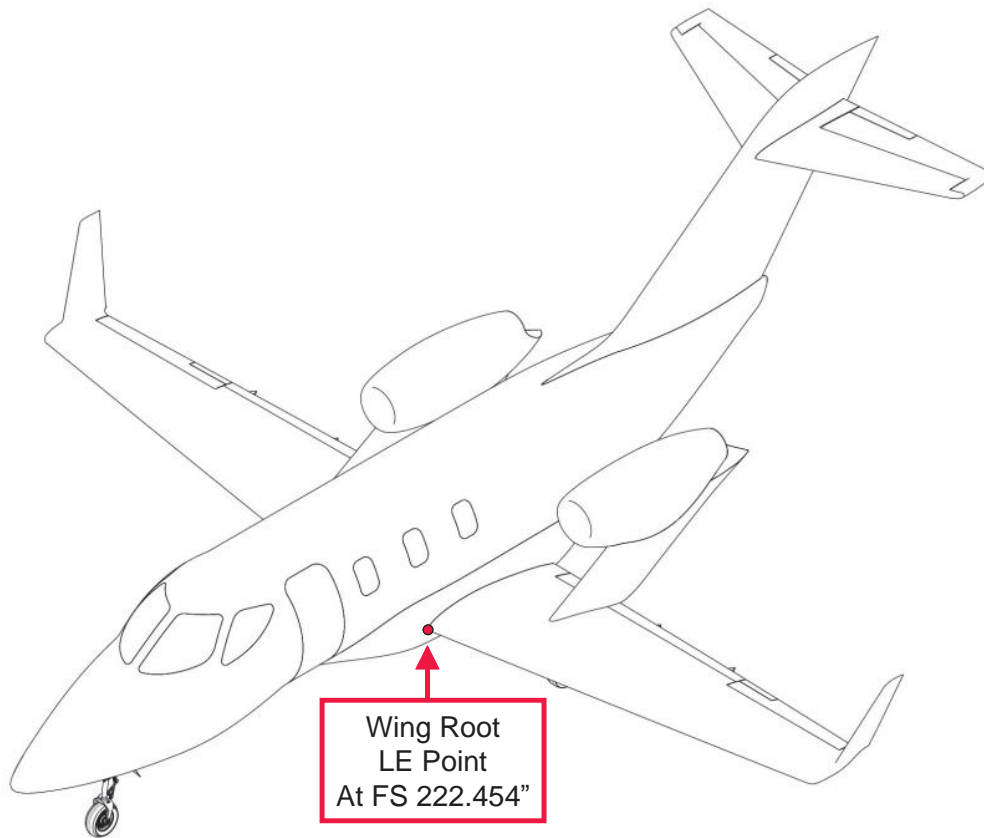


Figure 14. Weighing Procedure – Wing Root LE Point

10. Airplane must be leveled in Pitch and Roll by the use of an inclinometer (or other level) placed on the marked Cabin Floor Panel

↑ LEVEL AIRCRAFT HERE ↑

11. Remove Inclinometer/Leveling device and close passenger door.
12. Note reactions (scale or load cell readings) at each location and enter on "Basic Empty Weight and Balance Form" (see example on following pages).
13. If weighing on jack points with load cells no distance measurements are required.

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October 30, 2016

HJ1-29000-003-001

Page 6-8

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HA-420 AFM

WEIGHT AND BALANCE

14. If weighing on Landing gear, obtain by measuring the landing gear contact (reaction) points and enter the values in the F-Sta column of the “as weighed” table on the BEW and Balance Form (see example on following pages).

Static Landing Gear contact points are shown on the BEW and Balance form, however due to trailing link configuration and payload variation it is better to verify reaction points by measurement. For detailed measuring instructions; see **Measuring Landing Gear Tire Contact Points**.

15. The airplane “as weighed” condition is computed on Basic Empty Weight and Balance Form from reactions (noted in step 12) and measurements (taken in step 14). Add and subtract any overage or shortage items to “as weighed” to obtain BEW.

If there are many overages and shortages; utilize BEW and Balance Form sheet 2.

16. Measuring Landing Gear Tire Contact Points. One method is shown below in steps a through h.

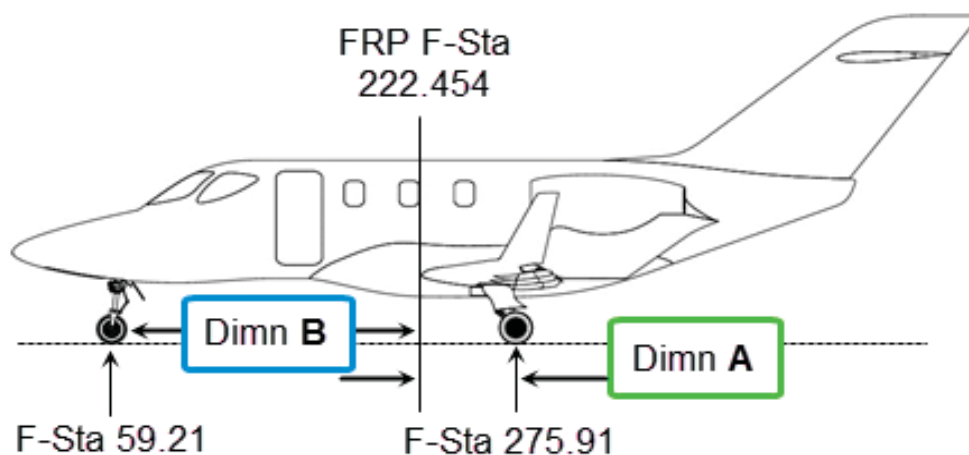


Figure 15. Method to Measure Landing Gear Tire Contact Points

- a) Mark the floor at the center of each main gear tire, a plumb line dropped from the axle center will give the correct reaction (contact) point.

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HA-420 AFM

WEIGHT AND BALANCE

- b) Mark the floor at the center of each side of the NLG tire, employing the same method as the main gear.
- c) Mark a point on the floor below the FRP on the left and right side of the fuselage using a similar method as the gear.
- d) Measure the perpendicular distance between the left and right FRP points and the MLG points, add the two numbers, then divide by two. This is dimension (Dimn) A.

Example measurements of 53.33” and 53.35” give Dimn A = 53.34”.

- e) Add Dimn A to the FRP value and enter the value in the “Main Total” row F-Sta column intersect cell of the BEW and Balance Form.

For example, $222.454 + 53.34 = 275.79$ (see example table below).

- f) Measure the perpendicular distance between the left and right NLG points and the FRP points, add the two numbers, then divide by two. This is Dimension (Dimn) B.

Example measurements of 163.82” and 163.86” give Dimn B = 163.84”.

- g) Minus Dimn B from the FRP value and enter the value in the “Nose” F-Sta cell of the BEW and Balance Form. See entry example below.

For example, $222.454 - 163.84 = 58.61$

- h) Landing Gear Reaction Point Table Example

Reaction Points		Scale Reading	Tare	Net Weight	F-Sta	Mom/100	% MAC
Quality Stamp	Left Main	3,231.0	6.2	3,224.8			
	Right Main	3,218.0	5.9	3,212.1			
	Main Total			6,436.9	275.79	17,752	
	Nose	719.0	6.9	712.1	58.61	417	
As Weighed Total (lb)				7,149.0	254.15	18,169	36.76%

Honda Aircraft Company

HA-420 AFM

WEIGHT AND BALANCE

17. After inputting the individual weight values and reaction points take note of any Tare weight at each reaction point and enter the value on the form.

NOTE Tare weight is typically “chock weight” or a reading other than zero prior to applying load. If the Tare weight is negative, add the value to the scale reading to obtain net weight. If the Tare weight is positive, subtract the value from the scale reading to obtain net weight.

18. Add the Left and Right Main net weights together and input the result in Net Weight Main Total cell.
19. Using the F-Sta values (obtained either from Jack Points, Static reference or actual measurement), multiply this value by the net weight and divide by 100 and add this value in the Mom/100 column.
20. Add the Nose and Main Net weights together to obtain the “As weighed Total”.
21. Add the Main Total Mom/100 and Nose Mom/100 values together and input value into the “As weighed Total” Mom/100
22. Divide the As weighed Total mom/100 by the As Weighed Total Net and multiply by 100 to obtain the F-Sta CG
23. To obtain the as weighed % MAC value use the following formula

Equation 1: As Weighed % MAC Formula

$$\text{As Weighed \% MAC} = \frac{(\text{As Weighed } F_{sta}CG - 232.20)}{59.72} * 100$$

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HA-420 AFM

WEIGHT AND BALANCE

24. The Basic Empty Weight can now be computed by adding and subtracting the overages and shortages (noted step 15) to the as weighed total. Overages should be subtracted from the as weighed to obtain BEW. Shortages should be added to the as weighed to obtain BEW.
25. Basic Empty Weight and Balance Form (example) – A blank form is available in Appendix A.

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HA-420 AFM

WEIGHT AND BALANCE

Sheet 1 of 2

HA-420 - Basic Empty Weight & Balance Form

Wing: 269.34 Nose: 69.0 Wing 269.34
Production Jacking Positions

FRP F-Sta 222.454
Dimn B Dimn A
F-Sta 59.21 F-Sta 275.91

Location: HACI R&D Hangar Model: HA-420 Serial #: P11

Date: 4/15/2015 Weighed by: _____

Reaction Points		Scale Reading	Tare	Net Weight	F-Sta	Mom/100	% MAC
Quality Stamp	Left Main	3,231.0	6.2	3,224.8			
	Right Main	3,218.0	5.9	3,212.1			
	Main Total			6,436.9	275.79	17,752	
	Nose	719.0	6.9	712.1	58.61	417	
As Weighed Total (lb)				7,149.0	254.15	18,169	36.76%

Measurements:

	Input	static ref.
Dimn A *FRP to Center of Main Gear	53.34 inches	53.55
Dimn B *FRP to Center of Nose Gear	163.84 inches	163.21

Additions or Subtractions to as weighed condition.	lb	F-Sta	Mom/100	% MAC
Subtract:				
Nose Jack Pad	0.7	69.00	0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
Add:				
Unusable Fuel (Left Wing)	18.0	258.00	46	
Unusable Fuel (Right Wing)	18.0	258.00	46	
Unusable Fuel (Ctr)	8.9	247.06	22	
Crew Headset	1.5	125.07	2	
			0	
			0	
			0	
			0	
			0	
			0	
Basic Empty Weight	7,194.7	254.15	18,286	36.76%

420AFM06_2514A

Figure 16. Basic Empty Weight and Balance Form Sheet 1 (Example)

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HJ1-29000-003-001

October 30, 2016

Page 6-13

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HA-420 AFM

WEIGHT AND BALANCE

26. Basic Empty Weight and Balance Form Sheet 2 (example) – A blank form is available in Appendix A.

This form is used to record and reconcile specific configurations, and to detail items too numerous for sheet 1.

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HA-420 AFM

WEIGHT AND BALANCE

Sheet 2 of 2				
HA-420 - Basic Empty Weight & Balance Form				
Extra worksheet to document and capture items weighed but not part of basic empty weight (Overages) and basic empty weight items not in airplane at time of weighing (Shortages).				
Additions or Subtractions to as weighed condition.	lb	F-Sta	Moment/100	% MAC
<u>Subtract:</u> (Overages)			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
Total Subtractions	0.0		0	
<u>Add:</u> (Shortages)			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
			0	
Total Additions	0.0		0	
<u>Measuring Equipment:</u>				
Scale Type: <u>In-Ground Floor Scales</u>		Increments/Accuracy: <u>better than 0.1%</u>		
Model Number: <u>IND780 5607882-5FK</u>		Calibration Date: <u>May 26, 2015</u>		
<u>Airplane Configuration:</u> *Circle one				
Fuel Qty: <u>0</u> <u>lb</u> /gallon*		Flaps Retracted: <u>Yes</u> /No*		
Defueled: <u>Yes</u> /No*		Drained: <u>Yes</u> /No*		
Level Info: <u>HA-2485 07/31/15</u> <u>HA-3356 4/30/16</u>		Lateral: <u>Yes</u> /No* 0.1° Longitudinal: <u>Yes</u> /No* 0.02°		
<u>Notes/Remarks:</u>				
P11 BEW determination. Plumbob CG method.				

420AFM06_2517A

Figure 17. Basic Empty Weight and Balance Form Sheet 2 (Example)

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HJ1-29000-003-001

October 30, 2016

Page 6-15

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HA-420 AFM

WEIGHT AND BALANCE

27. Basic Empty Weight and Balance Record – A blank form is available in Appendix A.

A running total of BEW should be kept to record changes in equipment and its effect on weight and balance; see the following example.

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HA-420 AFM

WEIGHT AND BALANCE

[illegible]

Figure 18. Basic Empty Weight and Balance Record

FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 6-17

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HA-420 AFM

WEIGHT AND BALANCE

LOADING INSTRUCTIONS

GENERAL

It is the airplane owner/operator's responsibility to ensure the airplane is correctly loaded. Honda Aircraft Company will furnish the BEW and all necessary data to compute specific airplane loadings at the time of delivery. All subsequent changes to the airplane BEW and loading configurations are the responsibility of the owner/operator.

LIMITS

The airplane weight and CG (for all flight and ground operations) must be maintained within the approved weight and CG limits described in **CENTER OF GRAVITY LIMITS**.

UNITS

All airplane and individual weight information is given in pounds, Stations and measurements are in inches and moments are in inch-pounds divided by one hundred (in-lbs/100).

FORMULA

When performing the computing procedure the following formulae display the relationship between the respective CG terms.

F-Sta Distance from Fuselage reference plane in inches

F-Sta to % MAC $(F-Sta - 232.20) / 59.72 * 100$

F-Sta to Moment $(F-Sta * weight) / 100$

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HA-420 AFM

WEIGHT AND BALANCE

LOADING FORM COMPUTING PROCEDURE

Enter the latest BEW data from either the Basic Empty Weight and Balance Form or the Basic Empty Weight and Balance Record onto Line 1 of the Weight and Balance Loading Form.

1. Enter the Crew, Passenger and Payload weights onto lines 2 through 15. A diagram of the interior floorplan/payload location (based on the delivered configuration) can be found in the next section (**Cabin Floorplan**). Payload weight and moment data is included in **Payload Data**.
2. Total the weight of lines 1-15 to determine Zero Fuel Weight (ZFW). ZFW must not exceed 8,800 lbs.
3. Calculate ZFW CG by totaling the moments of lines 1-15, dividing by the weight obtained in step 2 and multiplying by 100. Ensure CG is within ZFW limits; payload may have to be relocated or removed to respect limits (forward and aft ZFW limits are shown graphically in **Weight and CG Envelope** and numerically in **Loading Envelope Limits**).
4. Enter the Fuel weight and moment data on line 17. Corresponding moments for fuel loads can be found in **USABLE FUEL**.
5. Total lines 16 and 17 (ZFW plus Fuel) to obtain Ramp Weight. Ramp weight must not exceed 10,680 lbs. Check the Ramp Weight and CG is within the Loading Envelope Limits.
6. Enter Taxi/Run up fuel on line 19 and subtract from line 18 to obtain Takeoff Weight. Takeoff weight must not exceed 10,600 lbs.
7. Complete Loading Form by entering estimated post flight fuel remaining on line 22 (usable fuel moments can be found in **USABLE FUEL**). Add line 22 values to line 21 (same as line 16 – Zero Fuel weight) to determine Landing Weight. Landing weight must not exceed 9,860 lbs.

Honda Aircraft Company

HA-420 AFM

WEIGHT AND BALANCE

8. Weight and Balance Loading Form (see Loading example below). Loading example includes: 2 Crew, 4 Cabin Passengers, Aft Baggage and a fuel load of 2144 lb. Blank forms are available in Appendix A.

Weight & Balance Loading Form					
				HA-420 Serial#	P12
				Date:	2/16/2015
Line	Item Description	weight (lbs)	arm (in)	moment (lbs.in)/100	%MAC
1	Basic Empty Weight	7,000.0	253.00	17,710.0	34.83%
2	Pilot	200.0	121.40	242.8	
3	Copilot	200.0	121.40	242.8	
4	Crew Equipment			0.0	
5	Provisions-Cabinet	15.0	142.50	21.4	
6	Operating Weight	7,415.0	245.68	18,217.0	22.57%
7	Side Facing Passenger		158.71	0.0	
8	Fwd LH Passenger	170.0	188.37	320.2	
9	Fwd RH Passenger	170.0	188.37	320.2	
10	Aft LH Passenger	170.0	239.43	407.0	
11	Aft RH Passenger	170.0	239.43	407.0	
12	Miscellaneous			0.0	
13	Cabin Baggage			0.0	
14	Aft Baggage (Max 400)	150.0	311.65	467.5	
15	Fwd Baggage (Max 100)		54.30	0.0	
16	Zero Fuel Weight (Limit 8,800)	8,245.0	244.26	20,139.0	20.19%
17	Fuel Weight	2,144.0		5,516.0	
18	Ramp Weight (Limit 10,680)	10,389.0	246.94	25,655.0	24.69%
19	Minus Taxi/Run-up Fuel	100.0	288.59	288.6	
20	Take-off Weight (Limit 10,600)	10,289.0	246.54	25,366.4	24.01%
21	Zero Fuel Weight (s/as line 16)	8245.0	244.26	20139.0	
22	Estimated Fuel Remaining	1,005.0		2,595.0	
23	Landing Weight (Limit 9,860)	9,250.0	245.77	22,734.0	22.73%

Figure 19. Weight and Balance Loading Form (Example)

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HA-420 AFM

WEIGHT AND BALANCE

CABIN FLOORPLAN / PAYLOAD DATA

CABIN FLOORPLAN

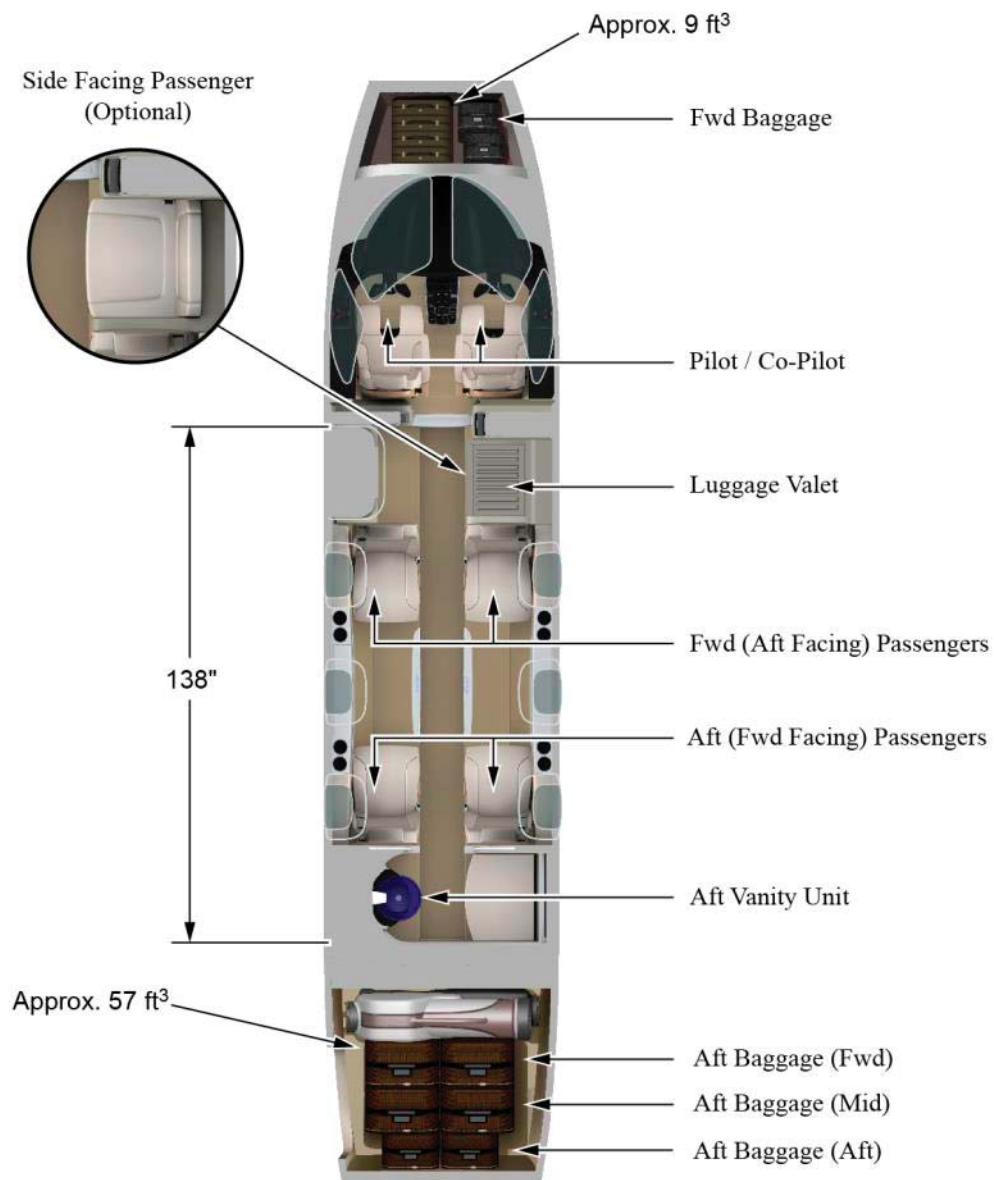


Figure 20. Cabin Floorplan

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HA-420 AFM

WEIGHT AND BALANCE

PAYLOAD DATA

Occupants

Weight (lb)	Pilot / Copilot		Side Face Passenger		Fwd Passengers		Rear Passengers	
	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb)	F-Sta (in)	mom/100 (lb.in)
80	121.40	97.1	158.71	127.0	188.37	150.7	239.43	191.5
90	121.40	109.3	158.71	142.8	188.37	169.5	239.43	215.5
100	121.40	121.4	158.71	158.7	188.37	188.4	239.43	239.4
110	121.40	133.5	158.71	174.6	188.37	207.2	239.43	263.4
120	121.40	145.7	158.71	190.5	188.37	226.0	239.43	287.3
130	121.40	157.8	158.71	206.3	188.37	244.9	239.43	311.3
140	121.40	170.0	158.71	222.2	188.37	263.7	239.43	335.2
150	121.40	182.1	158.71	238.1	188.37	282.6	239.43	359.1
160	121.40	194.2	158.71	253.9	188.37	301.4	239.43	383.1
170	121.40	206.4	158.71	269.8	188.37	320.2	239.43	407.0
180	121.40	218.5	158.71	285.7	188.37	339.1	239.43	431.0
190	121.40	230.7	158.71	301.5	188.37	357.9	239.43	454.9
200	121.40	242.8	158.71	317.4	188.37	376.7	239.43	478.9
210	121.40	254.9	158.71	333.3	188.37	395.6	239.43	502.8
220	121.40	267.1	158.71	349.2	188.37	414.4	239.43	526.7
230	121.40	279.2	158.71	365.0	188.37	433.3	239.43	550.7
240	121.40	291.4	158.71	380.9	188.37	452.1	239.43	574.6
250	121.40	303.5	158.71	396.8	188.37	470.9	239.43	598.6

Figure 21. Occupants Weights

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HA-420 AFM

WEIGHT AND BALANCE

CONSUMABLES AND BAGGAGE

Forward Baggage Compartment is limited to 100 lbs capacity.

Aft Baggage Compartment is limited to 400 lb capacity with the load evenly spread between the three baggage areas (forward 160 lb, Mid 120 lb, Aft 120 lb); items should be secured with the straps provided.

Weight (lb)	Fwd Baggage		Aft Baggage (Fwd)		Aft Baggage (Mid)		Aft Baggage (Aft)	
	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb)	F-Sta (in)	mom/100 (lb.in)
10	54.3	5.4	311.65	31.2	331.5	33.2	347.65	34.8
20	54.3	10.9	311.65	62.3	331.5	66.3	347.65	69.5
30	54.3	16.3	311.65	93.5	331.5	99.5	347.65	104.3
40	54.3	21.7	311.65	124.7	331.5	132.6	347.65	139.1
50	54.3	27.2	311.65	155.8	331.5	165.8	347.65	173.8
60	54.3	32.6	311.65	187.0	331.5	198.9	347.65	208.6
100	54.3	54.3	311.65	311.7	331.5	331.5	347.65	347.7
120	Max Compartment Allowable = 100.0 lb		311.65	374.0	331.5	397.8	347.65	417.2
160			311.65	498.6				

420AFM06_2521A

Figure 22. Baggage

Weight (lb)	LH Cabinet		RH Cabinet		Luggage Valet	
	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb)	F-Sta (in)	mom/100 (lb.in)
10	141.77	14.2	143.86	14.4	162.63	16.3
20	141.77	28.4	143.86	28.8	162.63	32.5
22			143.86	31.6	162.63	35.8
30					162.63	48.8
40					162.63	65.1
50					162.63	81.3
	Max Refreshment Cabinet Load = 20.0 lb		Max Refreshment Cabinet Load = 22.0 lb		Max Compartment Allowable = 50.0 lb	

Weight (lb)	Vanity Carry-on		Vanity Water		Lav Water	
	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb)	F-Sta (in)	mom/100 (lb.in)
1.25	270.84	3.4				
5.0			271.42	13.6		
10.6					273.16	28.9

Figure 23. Consumables and Storage

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HA-420 AFM

WEIGHT AND BALANCE

CENTER OF GRAVITY LIMITS

WEIGHT AND CG ENVELOPE

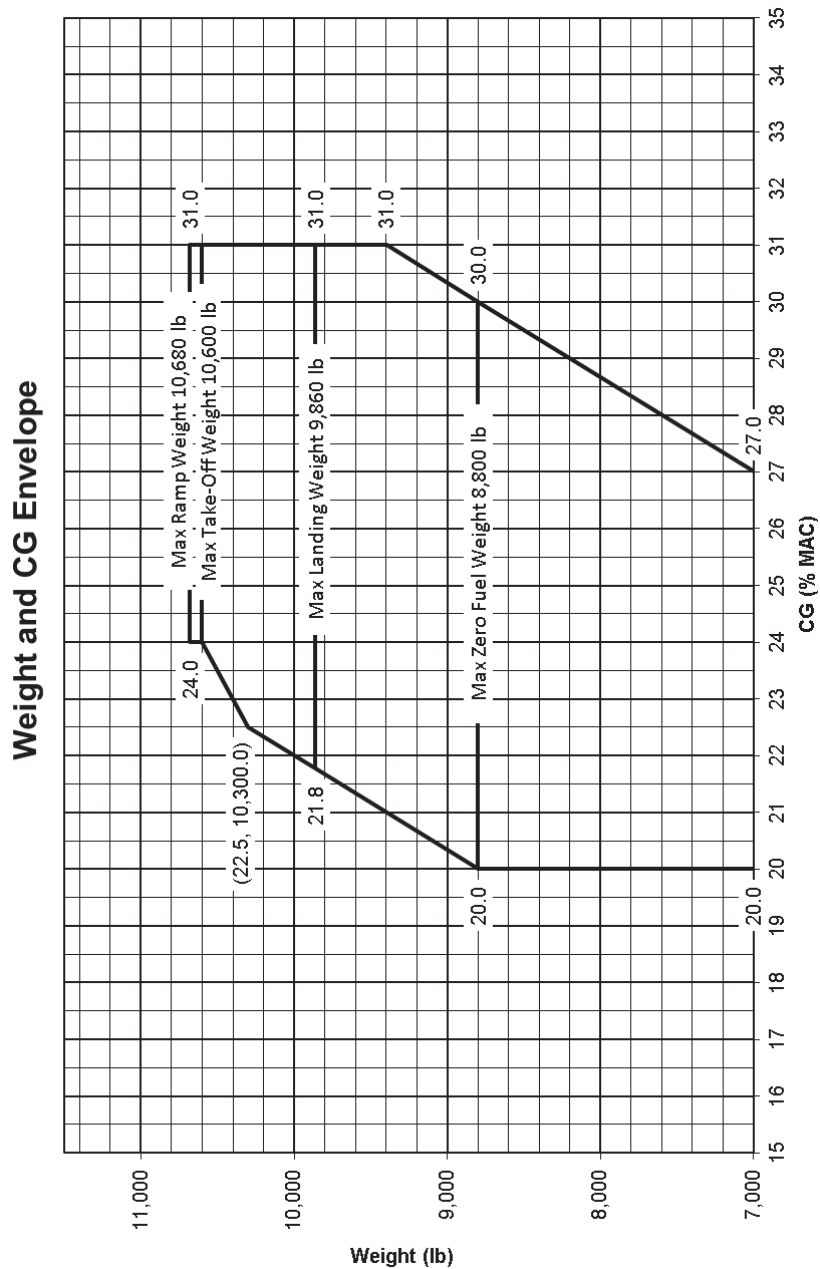


Figure 24. Weight and CG Envelope

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HA-420 AFM

WEIGHT AND BALANCE

LOADING ENVELOPE LIMITS

Below are the forward and aft limits of the weight and CG envelope.

	Weight lb	Fwd Limit			Aft Limit		
		F-Sta	Moment/100	%MAC	F-Sta	Moment/100	% MAC
	7,000	244.14	17,090.1	20.00	248.32	17,382.7	27.00
	7,100	244.14	17,334.2	20.00	248.42	17,638.1	27.17
	7,200	244.14	17,578.4	20.00	248.52	17,893.7	27.33
	7,300	244.14	17,822.5	20.00	248.62	18,149.5	27.50
	7,400	244.14	18,066.7	20.00	248.72	18,405.5	27.67
	7,500	244.14	18,310.8	20.00	248.82	18,661.7	27.83
	7,600	244.14	18,554.9	20.00	248.92	18,918.0	28.00
	7,700	244.14	18,799.1	20.00	249.02	19,174.6	28.17
	7,800	244.14	19,043.2	20.00	249.12	19,431.4	28.33
	7,900	244.14	19,287.4	20.00	249.22	19,688.4	28.50
	8,000	244.14	19,531.5	20.00	249.32	19,945.6	28.67
	8,100	244.14	19,775.7	20.00	249.42	20,203.0	28.83
	8,200	244.14	20,019.8	20.00	249.52	20,460.5	29.00
	8,300	244.14	20,264.0	20.00	249.62	20,718.3	29.17
	8,400	244.14	20,508.1	20.00	249.72	20,976.3	29.33
	8,500	244.14	20,752.2	20.00	249.82	21,234.5	29.50
	8,600	244.14	20,996.4	20.00	249.92	21,492.9	29.67
	8,700	244.14	21,240.5	20.00	250.02	21,751.4	29.83
MZFW	8,800	244.14	21,484.7	20.00	250.12	22,010.2	30.00
	8,900	244.24	21,737.7	20.17	250.22	22,269.2	30.17
	9,000	244.34	21,990.9	20.33	250.32	22,528.4	30.33
	9,100	244.44	22,244.3	20.50	250.41	22,787.7	30.50
	9,200	244.54	22,497.9	20.67	250.51	23,047.3	30.67
	9,300	244.64	22,751.7	20.83	250.61	23,307.1	30.83
	9,400	244.74	23,005.7	21.00	250.71	23,567.0	31.00
	9,500	244.84	23,259.9	21.17	250.71	23,817.8	31.00
	9,600	244.94	23,514.3	21.33	250.71	24,068.5	31.00
	9,700	245.04	23,768.9	21.50	250.71	24,319.2	31.00
	9,800	245.14	24,023.7	21.67	250.71	24,569.9	31.00
MLW	9,860	245.20	24,176.6	21.77	250.71	24,720.3	31.00
	9,900	245.24	24,278.6	21.83	250.71	24,820.6	31.00
	10,000	245.34	24,533.8	22.00	250.71	25,071.3	31.00
	10,100	245.44	24,789.2	22.17	250.71	25,322.0	31.00
	10,200	245.54	25,044.8	22.33	250.71	25,572.7	31.00
	10,300	245.64	25,300.6	22.50	250.71	25,823.5	31.00
	10,400	245.94	25,577.3	23.00	250.71	26,074.2	31.00
	10,500	246.23	25,854.6	23.50	250.71	26,324.9	31.00
MTOW	10,600	246.53	26,132.5	24.00	250.71	26,575.6	31.00
MRW	10,680	246.53	26,329.7	24.00	250.71	26,776.2	31.00

Figure 25. Loading Envelope Limits

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HA-420 AFM

WEIGHT AND BALANCE

USABLE FUEL

The figure below shows the values for fuel quantity in gallons along with the corresponding weight and fuselage station for fuel of various densities.

Fuel Quantity (gallons)	6.3 lb/gallon		6.5 lb/gallon		6.7 lb/gallon		6.9 lb/gallon		7.1 lb/gallon	
	weight (lb)	mom/100 (lb.in)	weight (lb)	mom/100 (lb.in)	weight (lb)	mom/100 lb.in	weight (lb)	mom/100 (lb.in)	weight (lb)	mom/100 (lb.in)
10	63	165	65	170	67	175	69	181	71	186
20	126	329	130	340	134	350	138	360	142	371
30	189	493	195	509	201	524	207	540	213	555
40	252	656	260	677	268	698	276	719	284	739
50	315	819	325	845	335	871	345	897	355	923
60	378	981	390	1,013	402	1,044	414	1,075	426	1,106
70	441	1,144	455	1,180	469	1,216	483	1,253	497	1,289
80	504	1,306	520	1,347	536	1,388	552	1,430	568	1,471
90	567	1,467	585	1,514	603	1,560	621	1,607	639	1,653
100	630	1,629	650	1,680	670	1,732	690	1,784	710	1,836
110	693	1,790	715	1,847	737	1,904	759	1,961	781	2,017
120	756	1,954	780	2,016	804	2,078	828	2,140	852	2,202
130	819	2,117	845	2,184	871	2,252	897	2,319	923	2,386
140	882	2,279	910	2,352	938	2,424	966	2,496	994	2,569
150	945	2,440	975	2,517	1,005	2,595	1,035	2,672	1,065	2,749
160	1,008	2,598	1,040	2,680	1,072	2,763	1,104	2,845	1,136	2,928
170	1,071	2,756	1,105	2,844	1,139	2,931	1,173	3,019	1,207	3,106
180	1,134	2,916	1,170	3,009	1,206	3,101	1,242	3,194	1,278	3,286
190	1,197	3,075	1,235	3,173	1,273	3,271	1,311	3,368	1,349	3,466
200	1,260	3,235	1,300	3,338	1,340	3,440	1,380	3,543	1,420	3,646
210	1,323	3,395	1,365	3,503	1,407	3,610	1,449	3,718	1,491	3,826
220	1,386	3,554	1,430	3,667	1,474	3,780	1,518	3,893	1,562	4,005
230	1,449	3,713	1,495	3,831	1,541	3,949	1,587	4,066	1,633	4,184
240	1,512	3,870	1,560	3,992	1,608	4,115	1,656	4,238	1,704	4,361
250	1,575	4,026	1,625	4,153	1,675	4,281	1,725	4,409	1,775	4,537
260	1,638	4,181	1,690	4,314	1,742	4,447	1,794	4,580	1,846	4,712
270	1,701	4,337	1,755	4,474	1,809	4,612	1,863	4,750	1,917	4,887
280	1,764	4,492	1,820	4,635	1,876	4,777	1,932	4,920	1,988	5,062
290	1,827	4,654	1,885	4,802	1,943	4,950	2,001	5,098	2,059	5,245
300	1,890	4,825	1,950	4,979	2,010	5,132	2,070	5,285	2,130	5,438
310	1,953	5,003	2,015	5,162	2,077	5,320	2,139	5,479	2,201	5,638
320	2,016	5,187	2,080	5,351	2,144	5,516	2,208	5,681	2,272	5,845
330	2,079	5,371	2,145	5,542	2,211	5,712	2,277	5,883	2,343	6,053
340	2,142	5,554	2,210	5,731	2,278	5,907	2,346	6,083	2,414	6,260
350	2,205	5,739	2,275	5,921	2,345	6,104	2,415	6,286	2,485	6,468
360	2,268	5,925	2,340	6,113	2,412	6,302	2,484	6,490	2,556	6,678
370	2,331	6,112	2,405	6,306	2,479	6,500	2,553	6,694	2,627	6,888
380	2,394	6,298	2,470	6,497	2,546	6,697	2,622	6,897	2,698	7,097
390	2,457	6,479	2,535	6,684	2,613	6,890	2,691	7,096	2,769	7,302
400	2,520	6,661	2,600	6,873	2,680	7,084	2,760	7,296	2,840	7,507
410	2,583	6,823	2,665	7,040	2,747	7,256	2,829	7,473	2,911	7,689
420	2,646	6,977	2,730	7,198	2,814	7,420	2,898	7,641	2,982	7,863
424	2,671	7,039	2,756	7,262	2,841	7,485	2,926	7,709	3,010	7,932

420AFM06_2524A

Figure 26. Usable Fuel

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October 30, 2016

HJ1-29000-003-001

Page 6-26

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HA-420 AFM

WEIGHT AND BALANCE

ENGINE OIL / UNUSABLE FUEL

Weight (lb)	Engine Oil		Trapped Fuel (Undrainable)		Drainable Unusable Fuel		Total Unusable Fuel	
	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb.in)	F-Sta (in)	mom/100 (lb)	F-Sta (in)	mom/100 (lb.in)
12.3	320.00	39.2						
8.4			254.87	21.4				
37.0					257.70	95.4		
45.4							257.18	116.8

420AFM06_2525A

NOTE Unusable fuel (both Trapped and Drainable) is considered part of the Basic Empty Weight.

Figure 27. Engine Oil / Unusable Fuel

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WEIGHT AND BALANCE

APPENDIX A – BLANK FORMS

The following blank forms are available in this Appendix A:

Basic Empty Weight and Balance Form: Sheet 1

Basic Empty Weight and Balance Form: Sheet 2

Basic Empty Weight and Balance Record

Weight and Balance Loading Form

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WEIGHT AND BALANCE

[illegible]

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October 30, 2016

HJ1-29000-003-001

Page 6-29

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WEIGHT AND BALANCE

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October 30, 2016

HJ1-29000-003-001

Page 6-30

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WEIGHT AND BALANCE

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FAA APPROVED
October 30, 2016

HJ1-29000-003-001

Page 6-31

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HA-420 AFM

WEIGHT AND BALANCE

Weight & Balance Loading Form					
					HA-420 Serial#
					Date:
Line	Item Description	weight (lbs)	arm (in)	moment (lbs.in)/100	%MAC
1	Basic Empty Weight				
2	Pilot				
3	Copilot				
4	Crew Equipment				
5	Provisions-Cabinet				
6	Operating Weight				
7	Side Facing Passenger				
8	Fwd LH Passenger				
9	Fwd RH Passenger				
10	Aft LH Passenger				
11	Aft RH Passenger				
12	Miscellaneous				
13	Cabin Baggage				
14	Aft Baggage (Max 400)				
15	Fwd Baggage (Max 100)				
16	Zero Fuel Weight (Limit 8,800)				
17	Fuel Weight				
18	Ramp Weight (Limit 10,680)				
19	Minus Taxi/Run-up Fuel				
20	Take-off Weight (Limit 10,600)				
21	Zero Fuel Weight (s/as line 16)				
22	Estimated Fuel Remaining				
23	Landing Weight (Limit 9,860)				

420AFM06_3265A

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October 30, 2016

HJ1-29000-003-001

Page 6-32

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